

*Article*



# **Influence of Clearance Flow on Dynamic Hydraulic Forces of Pump-Turbine during Runaway Transient Process**

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**Abstract:** The clearance flow around the pump-turbine runner has significant influences on unit vibrations, which may cause accidents in transient processes. The dynamic hydraulic forces and flow patterns in the clearance flow channel (CFC) of a low specific-speed pump-turbine were analyzed based on 3D CFD simulations during the runaway oscillating process. It is shown that the axial force of the runner periodically fluctuates with large amplitudes, and its components in CFC and the main flow channel (MFC) demonstrate a similar significance level. The CFC component was formulated as a function of the clearance inlet pressure and rotational speed, while the MFC component as a function of the momentum changing rate and the runner outlet pressure force. The fluctuation of runner radial force is mainly caused by the flow evolution in MFC, however, the flow in CFC aggravates it. The pressure in CFC shows a few pulsating signals from MFC, and the radial pressure drop in CFC is proportional to the square of both radius and rotational speed. In CFC, strong rotating shear flow containing a velocity core region in the circumferential direction is formed, and rotational speed is the dominant factor.

**Keywords:** pump-turbine; runaway transient process; runner forces; clearance leakage; clearance flow patterns

# **1. Introduction**

As the main way to store energy on large scale, the pumped-storage power generally undertakes functions such as peak load regulation, valley load regulation, frequency modulation, phase modulation and spinning reserve in the power grid, to match with thermal, nuclear, wind and solar powers. Fast and frequent transitions of operating modes, along with various off-design operations are needed for pump-turbines [\[1,](#page-18-0)[2\]](#page-18-1). During violent transient processes, many dynamic instability problems, such as unit vibration, powerhouse noise and excessive water hammer were reported frequently [\[3–](#page-18-2)[5\]](#page-18-3), even if the extreme accidents of rotor lifting and rotor-stator crashing happened occasionally [\[6](#page-18-4)[,7\]](#page-18-5). These accidents were actually caused by the fluctuations of runner forces during transients. The runner forces can be classified into two parts: the force in the main flow channel (MFC) and the force in the clearance flow channel (CFC), which is the small gaps between the rotating and stationary parts of the pump-turbine. It was recognized that the pressure, distribution and evolution in CFC have significant influences on the total runner forces. However, most of the existing research on runner forces only focused on the flow in MFC, because of the difficulties in measuring and simulating the flow in CFC.

Regarding the force in MFC, Dai et al. [\[8\]](#page-18-6) investigated the axial forces of the pumpturbine in Tianhuangping hydropower station and found that the axial force measured by the model test was greatly different from the prototype unit, due to the difficulty in



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ensuring similarity. Therefore, numerical simulation is the main means to study runner forces. Xia et al. [\[9,](#page-18-7)[10\]](#page-18-8) numerically studied pressure pulsations and runner forces of a model pump-turbine during the runaway transient process and found that the reverse flow around the runner inlet could cause high-frequency axial force fluctuations. Yang et al. [\[11\]](#page-18-9) pointed out that the unstable vortex structure in MFC of the pump-turbine is another reason for the force fluctuations during the runaway transient process after pump-trip. Widmer et al. [\[12\]](#page-18-10) found that the rotating stall in the flow channel of the pump-turbine was the main reason resulting in the instability of the runner during the turbine braking process. Furthermore, during the load rejection transient process, Fu et al. [\[13\]](#page-18-11) concluded that the fluctuation frequency of axial force was caused by the reverse flow and helical vortex flow in the draft tube. All above studies on force fluctuations neglected the hydraulic force in CFC, which should be considered in analyzing the possibilities of rotor lifting and

Up to now, the force in CFC was rarely studied. Most studies on CFC mainly paid attention to the sealing ring, which is the most important component of CFC. The sealing ring was made into various zigzag shapes to increase the local kinetic energy loss, so as to reduce the fluid leakage [\[14\]](#page-18-12). One of the most common research directions was to compare the sealing effect of different sealing types, such as straight-through, stepped and labyrinth type [\[15](#page-18-13)[–17\]](#page-18-14). Another was to explore the influence of sealing size on the clearance leakage and unit efficiency [\[18,](#page-18-15)[19\]](#page-18-16). Koirala et al. [\[20\]](#page-18-17) found that the leakage increased with the increase of sealing size, but the energy conversion efficiency of the turbine decreased. Dong et al. [\[21\]](#page-19-0) simulated the clearance flow of the pump-turbine and found that the leakage flow rate could not only cause volume loss, but also change the mainstream flow patterns. Moreover, Jürgen et al. [\[22\]](#page-19-1) and Čelič et al. [[23\]](#page-19-2) found that the unit efficiency agreed with the reality better when CFC was considered. In recent years, some researchers began to study the force in CFC. Wu et al. [\[24,](#page-19-3)[25\]](#page-19-4) studied the axial hydraulic force of pump-turbine with high head and found that the clearance leakage had a sensitive influence on the axial hydraulic force, which was the main hydraulic factor leading to rotor lifting. Fu et al. [\[26\]](#page-19-5) and Zhang et al. [\[27\]](#page-19-6) compared the axial force with and without the clearance flow during the load rejection process and confirmed that there was a huge difference in value and direction. However, the correlated mechanism was not analyzed thoroughly. In summary, the research on the clearance flow patterns and clearance hydraulic force of pump-turbine are still very scarce, and the influence mechanism of clearance flow is not clear. How clearance flow and forces change during the transient process, and how they affect the total runner force, are needed to be clarified.

Due to the small size and complex shape of CFC, model tests and prototype observations are largely limited, and computational fluid dynamics (CFD) has become an important method to study the clearance flow. In this paper, the dynamic hydraulic forces and flow patterns in CFC of a low specific-speed pump-turbine were analyzed based on three-dimensional (3D) CFD simulations of the runaway oscillating process. The analysis focused on the influence degree of clearance hydraulic force on the runner forces, the influence factors and change mechanism of hydraulic forces, the change laws of clearance flow patterns and pressure distribution in CFC.

#### **2. CFD Simulation Methods for the Pump-Turbine**

#### *2.1. Pump-Turbine Model*

rotor-stator crashing.

In this study, a prototype pump-turbine with specific-speed  $n_q = 30.5$  (calculated by Equation (1)) was selected. The computational domain contains MFC and CFC. As shown in Figure [1,](#page-2-0) the MFC is composed of spiral-casing, stay and guide vanes, runner and draft tube; the CFC contains the clearance region between the hub upper surface (rotating surface) and the cover lower surface (stationary surface), denoted as CHC, and the clearance region between the shroud lower surface (rotating surface) and the bottom ring upper surface (stationary surface), denoted as CSB; the CHC domain contains clearance inlet, pressure external cavity, sealing ring, pressure internal cavity, clearance outlet and

<span id="page-2-0"></span>

pressure equalizing pipes (PEP). The sealing sizes of CHC and CSB are 1.4 mm and 1.6 mm, respectively. The relevant parameters of the runner are shown in Table [1.](#page-2-1)<br> *<u>rage</u>* 

$$
n_{\mathbf{q}} = \frac{n_{\mathbf{r}}\sqrt{Q_{\mathbf{r}}}}{H_{\mathbf{r}}^{3/4}}
$$
 (1)

Figure 1. 3D model of the pump-turbine. (a) Computational domains. (b) Flow channels. (c) Detail drawing of CHC.

<span id="page-2-1"></span>



Runner outlet diameter *D*2 (m) 1.955 Level of the sealing ring 7

#### *2.2. Multi-Scale Grid Coupling*

Compared with MFC, CFC is very thin and needs to be discretized with extremely small grids. Therefore, appropriately connecting the macro-scale grids in MFC with the micro-scale grids in CFC is the key to effectively transfer the numerical values. The multi-scale grid coupling method was used (Figure [2a](#page-3-0)), and a transient region was set between the guide vanes and the runner. The macro-scale grids were used in the main flow direction, while the micro-scale grids were used in the clearance channel direction, which were suitable for the micro-scale grids in the clearance channel. The grids of the hub upper surface (rotating surface) are shown in Figure 2b.

<span id="page-3-0"></span>

Figure 2. Simulation grid discretization. (a) Grid coupling of different scales. (b) Grids on the hub upper surface. upper surface.

## *2.3. Grid Independence Verification 2.3. Grid Independence Verification*

Hybrid grids were generated by commercial software ANSYS ICEM. Structural hex-Hybrid grids were generated by commercial software ANSYS ICEM. Structural hexahedral grids were used in all domains except for the guide vane domain, in which unstructured wedge-shaped grids were used. Since the unstructured grid in guide vane main is better to adapt to the dynamic grid, which will be applied in the load rejection domain is better to adapt to the dynamic grid, which will be applied in the load rejection transition process in the future. Grid independence verification adopted five stepwise levels, from 3.73 to 4.10, 4.61, 5.17, 5.55 and 6.00 million, respectively. In the rated operating condition, when the number of cells was more than 4.61 million, the variation ranges of torque and discharge were 0.7% and 0.2%, respectively. At the off-design operating condition (the discharge was half of the rated discharge), the moment variation range was small enough to 1.3%. Therefore, the 5.17 million grid level was used, and the grid allocation in each computational domain was shown in Table [2.](#page-3-1) In order to verify the grid sensitivity of the clearance regions, the influence of the CHC grid number on the leakage was further analyzed. The leakage remained relatively constant (within 0.32%) when the cell number of CHC is more than 0.51 million, therefore, the grids of 0.74 million and 0.6 million were used in CHC and CSB, respectively.

<span id="page-3-1"></span>**Table 2.** Number of cells (million). **Table 2.** Number of cells (million).



# *2.4. Numerical Methods 2.4. Numerical Methods*

The finite volume method of commercial software Fluent was used for the simulation. Combining the advantage of RANS and LES model, the SAS turbulence model was generally applied to the transient processes of the turbine, and good results were obtained [\[9](#page-18-7)[,10](#page-18-8)[,28\]](#page-19-7). Therefore, the SAS-SST turbulence model and the SIMPLEC pressure velocity coupling algorithm were selected. For the prototype pump-turbine with a large size, it is almost impossible to meet  $y+=1$  because of the high Reynolds number; to ensure the high Reynolds number; to ensure  $\frac{1}{2}$  accuracy of the turbulence model of SAS, the y+ value was more than 30 in the whole domain, so that the wall function approach can be used in the simulation. The second order<br>domain, so that the wall function approach can be used in the simulation. The second order 0.001 c scheme was used in space and time discretization. The time step size was set as 0.001 s. the accuracy of the turbulence model of SAS, the y+ value was more than 30 in the whole

#### $(1)$ Boundary conditions

The total pressure of 7,393,218.21 Pa was set at the inlet of the spiral casing, and the static pressure of 979,234.2 Pa was set at the outlet of the draft tube. All the walls were set as the non-slip wall.

## (2) Rotational speed

During the runaway transient process, the rotational speed of each time step is determined by the water torque, which can be calculated as follows:

$$
\omega_{t} = \frac{T}{J} \Delta t + \omega_{t-\Delta t} \tag{2}
$$

where *ω*t−∆<sup>t</sup> and *ω*<sup>t</sup> are the rotational speed at previous and current timesteps; ∆*t* is the timestep size; *T* is the hydraulic torque acting on the runner, which contains the torque of MFC and CFC; *J* is the rotational inertia of the rotating part,  $1.125\times10^6$  kg·m<sup>2</sup>.

In addition, the hub upper surface and the shroud lower surface were set as rotational walls, rotating with the same speed as the runner.

## (3) Definition of dimensionless parameters

In order to generally describe the characteristics of the pump-turbine, the pressure coefficient *φ* and velocity coefficient *ϕ* were defined by Equations (3) and (4), respectively. The radial radius coefficient of the external cavity in CHC  $r'$  was defined by Equation (5). The larger the  $r'$ , the closer to the outer edge of CHC.

$$
\phi = \frac{p}{\rho u_1^2 / 2} = \frac{p}{\rho \omega_0^2 R_1^2 / 2} \tag{3}
$$

$$
\varphi = \frac{v}{u_1} = \frac{v}{\omega_0 R_1} \tag{4}
$$

$$
y' = \frac{r - R_m}{R_1 - R_m} \tag{5}
$$

where *p* is the pressure; *ρ* is the density of water;  $\omega_0$  is the rated rotational speed;  $u_1$  is the circumferential velocity at  $R_1$  in the rated operating condition;  $v$  is the flow velocity;  $R_1$  is the radius of the runner inlet, equivalent to the maximum radius of CFC; *r* is the radius in the clearance area;  $R_m$  is the radius at the inlet of the sealing ring of CHC (Figure [1c](#page-2-0)).

*r*

Due to the great differences of force values in different operating conditions, in order to intuitively and concisely describe forces, the axial forces and radial forces were dimensionless processed as follows:

$$
F_z = \frac{N_z}{A_z} = \frac{4N_z}{\rho g H \pi R_2^2}
$$
 (6)

$$
F_{\rm r} = \frac{N_{\rm r}}{A_{\rm r}} = \frac{N_{\rm r}}{\rho g H b_0} \tag{7}
$$

where  $F_z$  and  $F_r$  are the dimensionless axial force and radial force, respectively;  $N_z$  and  $N_r$ are the original axial force and radial force, respectively;  $b_0$  is the guide vane height;  $R_2$  is the radius of runner outlet (in turbine mode); the axial reference force  $A<sub>z</sub>$  and the radial reference force  $A_r$  are defined by multiplying rated head  $H_r$  with the runner outlet and inlet areas at the rated working point, respectively, with  $A_z = 19,244$  kN and  $A_r = 1924$  kN.

#### **3. Clearance Force and Flow Results**

#### *3.1. Macro Parameters during the Runaway Transient Process*

The initial point of the runaway transient process is the rated operating condition. In the case of load rejection transients of pump-turbines, if the guide vanes fail to close, the runner rotational speed, discharge and torque will begin to oscillate periodically with constant and unattenuated amplitudes (Figure [3a](#page-5-0)). The process starts from the turbine mode  $(t = 0$  s to 3.03 s) to the turbine braking mode  $(t = 3.03$  s to 3.41 s) and the reverse pump mode  $(t = 3.41 \text{ s to } 5.15 \text{ s})$ ; then returns from the reverse pump mode to the turbine braking mode (*t* = 5.15 s to 5.77 s) and the turbine mode (*t* = 5.77 s to 8.14 s). Once the load is rejected at *t* = 0 s, the generator resistance torque sharply drops to zero, and the

runner rotational speed increases gradually due to the imbalance of unit torque. With the increase of rotational speed, the discharge gradually decreases due to the influence of runner centrifugal force. After reaching the maximum runaway speed (*t* = 3.03 s), the In the turbine braking mode, in which the hydraulic torque becomes negative, and the discharge braking mode, in which the hydraulic torque becomes negative, and the discharge rapidly decreases to zero (*t* = 3.41 s). Subsequently, the flow reverses and are discharge repress decreases to zero ( $\frac{1}{2}$  in the reverse pump mode, in which the reverse discharge sharply increases to its maximum. Since the rotational speed continues to decrease, the effect of reverse to its maximum. Since the rotational speed continues to decrease, the effect of reverse hydraulic torque gradually weakens, the flow returns to the turbine direction and the unit gradually weakens, the flow returns to the turbine direction and the unit produced to the turbine braking mode. In the turbine braking mode, the reverse hydraulic goes back to the turbine braking mode. In the turbine braking mode, the reverse hydraulic creases rapidly because of the increased positive discharge, then the unit returns to the turbine mode and the hydraulic torque changes to the positive direction. The above to the turbine mode and the hydraulic torque changes to the positive direction. The above description is for the first cycle, and the following process includes many similar oscillating cycles. As shown in Figure [3b](#page-5-0), the operating trajectory shows periodic annular trajectory cycles. As shown in Figure 3b, the operating trajectory shows periodic annular trajectory in the S-shaped region, and the oscillation amplitude cannot be attenuated. The runaway in the S-shaped region, and the oscillation amplitude cannot be attenuated. The runaway oscillation phenomenon is a typical feature of low specific-speed pump-turbines [\[29\]](#page-19-8), in oscillation phenomenon is a typical feature of low specific-speed pump-turbines [29], in which the operating trajectory enters the reverse pump mode deeply, and is easier to produce runaway oscillations, the mechanism of the S-shaped characteristics and the runaway instability could be found in reference [\[9\]](#page-18-7). rance rotational speed increases gradually due to the influence of the torque. While

mode (*t* = 5.15 s to 5.77 s) and the turbine mode (*t* = 5.77 s to 8.14 s). Once the load is

<span id="page-5-0"></span>

Figure 3. Oscillating characteristics during the runaway transient process. (a) Macro parameters. Q: Discharge; n: Rational speed; M: Moment of the runner. (b) Operating trajectory in the  $n_{11}$ - $Q_{11}$  plane.

# *3.2. Runner Force Oscillating Characteristics*

# *3.2. Runner Force Oscillating Characteristics*  3.2.1. Axial Forces

Assuming the shaft of the pump-turbine is vertical, the schematic of the runner axial forces can be described with Figure 4, where  $F_1$  is the axial hydraulic force in MFC, in the upward or downward direction;  $F_2$  is that in CHC, in the downward direction;  $F_3$  is that in CSB, in the upward direction;  $F_4$  is the self-weight of the runner, in the downward direction; *F*<sub>5</sub> is the buoyancy force, in the upward direction; *F*<sub>6</sub> is the axial force acting on the shaft end, in the downward direction. The total axial force on the runner is the summation of these six forces. Compared with other axial forces, the summation of  $F_4$  and  $F_5$  is smaller enough to be neglected. Since the value of  $F_6$  is fixed, it will not be discussed in this paper. Only the axial hydraulic force  $F_1$  in MFC and that  $F_c$  in CFC are analyzed because they are the main components of the total axial hydraulic force. To make clear<br>discussions MFC and they are the distribution the mid-tone *F*<sub>c</sub> extincts which he total discussions, MFC axial force  $F_1$  can be divided into the axial force  $F_b$  acting on the blades,  $m_0$  acting on the nub lower surface and  $T_d$  acting on the should upper surface, hallely  $F_1 = F_b + F_u + F_d$ ; CFC axial force  $F_c$  can be divided as the clearance force  $F_2$  in CHC *F<sub>u</sub>* acting on the hub lower surface and *F*<sub>d</sub> acting on the shroud upper surface, namely, and the clearance force  $F_3$  in CSB, namely,  $F_c = F_2 + F_3$ . The variation law of total axial hydraulic force  $F_w$  ( $F_w = F_1 + F_2 + F_3$ ) will represent the change of the total runner force. The forces were directly monitored in the software.

<span id="page-6-0"></span>

**Figure 4.** Schematic of axial forces.

in Figure [5.](#page-6-1) In MFC,  $F_b = -0.19$  is the smallest;  $F_u = 2.4$  and  $F_d = -2.01$  are larger in values but opposite in directions. These result in a small upward  $F_1 = 0.20$ , which is only 10% of  $Y_0$ . In CrC,  $Y_2 = 2.92$  in CrC and  $Y_3 = 2.04$  in CDD possess hager values than those in MFC. Similarly, the two larger clearance forces counteract due to their opposite directions,  $\mu$ . Similarly, the two larger clearance forces counterate due to their opposite directions,<br>resulting in a small downward  $F_c = -0.13$ , which is only 4.4% of  $F_2$ . Since the CFC axial force  $F_c$  and the MFC axial force  $F_1$  have similar value levels but opposite directions, their counteract result is very small, namely the total axial hydraulic force  $F_w = 0.07$ , which counterties to a professional runner design. The above analysis proves that the force in CFC is oxtromoly important to the total runner force which should not be ignored CFC is extremely important to the total runner force, which should not be ignored. In the rated operating condition, the values of the axial force components are shown Fu. In CFC,  $F_2 = -2.92$  in CHC and  $F_3 = 2.84$  in CSB possess larger values than those in

<span id="page-6-1"></span>

**-2.1 -1.4** *F***b** *F***<sup>u</sup>** *F***d** *F***<sup>2</sup> Figure 5.** Dimensionless axial forces in the rated operating condition. **Figure 5.** Dimensionless axial forces in the rated operating condition.

downward maximum around turbine braking mode (Figure [6a](#page-7-0)). The severe fluctuation is of  $F_1$  in MFC and  $F_c$  in CFC are different, with  $F_1$  always upward and  $F_c$  most of the time  $t_0$  to the value of  $t_c$  decreases rapidly when the working point transits from turblic down to turbine braking mode, and changes to upward when approaching the runaway. boint. In this  $r_c$  upward period,  $r_w$  reaches at its maximum with a value about 0.2 to 0.3, of *F1 in MFC and F* in CFC and *F*<sub>c</sub> in MFC and *F*<sub>1</sub> and *F*<sup>1</sup> and *F*<sub>1</sub> and *F*<sub>1</sub> and *F*<sub>*n*</sub> and *F<sub>1</sub> and F<sub><i>n*</sub> and *F*<sub>*n*</sub><sup>*n*</sup> and *F*<sub>*n*</sub><sup>*n*</sup> and *F*<sub>*n*</sub><sup>*n*</sup> and *F*<sub>*n*</sub><sup>*n*</sup> and *F*<sub>*n*</sub><sup>*n*</sup> these two leads to the largest downward  $F_w$  around the turbine braking mode, which is diese two reads to the hirgest downward  $T_W$  around the tarbine braking mode, which is<br>easy to cause bearing damage.  $F_c$  and  $F_1$  demonstrate a similar significance level, and the  $\frac{1}{2}$  cause beaming damage.  $r_c$  and  $r_1$  demonstrate a similar significance it ver, and the maximum value of  $F_c$  is 1.3 times of  $F_1$ . This means again that the forces in CFC must be considered in transient analysis. The following will further explain why the total axial force has a large difference between considering and without considering clearance hydraulic sition of these two deterministic to the largest download to the turbine turbine turbine turbine turbine braki<br>Force, and how can the avial forces in both MFC and CFC be evaluated force, and how can the axial forces in both MFC and CFC be evaluated. **-2.97** *F***c** *F***<sup>w</sup>** *F***d -3.5** *F***2** tudes, getting the upward maximum before approaching the runaway point and the plitudes, getting the upward maximum before approaching the runaway point and the During the runaway oscillating process, *F<sub>w</sub>* periodically fluctuates with large am-<br>dee osciling the unward maximum hefere annoceshing the winaway point and the downward maximum around turbine braking mode (Figure 6a). The severe fluctuation is corresponding to the S-shaped region in the characteristic plane (Figure 3b). The histories corresponding to the S-shaped region in the characteristic plane (Figure [3b](#page-5-0)). The histories of *F*1 in MFC and *F*c in CFC are different, with *F*<sup>1</sup> always upward and *F*c most of the downward. The value of *F*<sup>c</sup> decreases rapidly when the working point transits from turbine time downward. The value of *F*c decreases rapidly when the working point transits from mode to turbine braking mode, and changes to upward when approaching the runaway turbine mode to turbine braking mode, and changes to upward when approaching the point. In this *F*<sup>c</sup> upward period, *F*<sup>w</sup> reaches at its maximum with a value about 0.2 to 0.3, period in this *F*<sub>C</sub> upward period, *F<sub>W</sub>* reaches at its maximum with a value about on to one, which may induce rotor lifting. The upward *F*<sub>1</sub> gradually decreases in turbine braking mode and near the same time  $F_c$  reaches its downward maximum. The superposition of

<span id="page-7-0"></span>

**Figure 6.** Histories of axial forces during runaway oscillating process. (a) Histories of  $F_w$ . (b) Histories of  $F_1$  and  $F_c$ .

# (1) Axial hydraulic force in MFC—*F*<sup>1</sup> (1) Axial hydraulic force in MFC—*F*<sup>1</sup>

For Francis pump-turbines, assuming the water flows in or out of the runner inlet For Francis pump-turbines, assuming the water flows in or out of the runner inlet (in turbine mode) in the radial direction, while flows out or in the runner outlet (in turbine mode) in the axial direction. Taking the water inside the runner as the control body, the mode) in the axial direction. Taking the water inside the runner as the control body, the resultant force  $\sum F$  acting on the control body is equal to the momentum change of the system according to the momentum theorem, as follows: system according to the momentum theorem, as follows:

total axial force has a large difference between considering and without considering clear-

$$
\sum F = \frac{\partial}{\partial t} \iiint\limits_V \rho v \, dV + \iint\limits_S \rho v^2 \, ds \tag{8}
$$

where *v* is water flow velocity, *V* is the volume of the control body, *s* is the closed surface where *v* is water flow velocity, *V* is the volume of the control body, *s* is the closed surface of the control body. The first term on the right-hand of Equation  $(8)$  is the momentum change with time of control body, and is mainly internal consumption in the control body, which is volume force essentially. However, the surface force plays a predominant factor for pump-turbines. Therefore, the value of the first term is very small, which will be validated by the momentum changing rate of water by following simulation. The second term is the momentum changing rate of water flowing by following in the control body considered, the momentum changing rate of water flowing out the control body. With the special shape of the control body considered, the resultant force *F* acting on the vector  $\frac{1}{2}$  can also be expressed as the vector sum of the axial force  $r_s$  acting on body and the axial pressure force  $F_p$  at the runner outlet, as follows: force  $\sum F$  can also be expressed as the vector sum of the axial force  $F_s$  acting on the water

$$
\sum F = F_{\rm s} - F_{\rm p} \tag{9}
$$

Since *F*<sub>1</sub> is the reaction force of *F*<sub>s</sub>, *F*<sub>1</sub> =  $-F$ <sub>s</sub> can be submitted to Equation (9):

$$
F_1 = -F_s = -(F_p \pm \rho \frac{Q^2}{A_2})
$$
\n(10)

where  $Q$  is the discharge, and  $A_2$  is the cross-sectional area of the runner outlet. The positive sign of the second term in Equation (10) means that the water flows out the runner outlet*,* while the negative sign means that the water flows into the runner outlet.

According to Equation (10), the axial force in MFC  $F_1$  is determined by two terms, the axial pressure force of the runner outlet  $F_p$  and the momentum changing rate  $\rho Q^2/A_2$ . During the runaway transient process, the variation law of  $F_p$ ,  $\rho Q^2/A_2$ , and  $F_1$  are shown in Figure [7a](#page-8-0). The value of  $\rho Q^2/A_2$  is related to the discharge, while *F*<sub>p</sub> is determined by the pressure of the runner outlet. In order to verify the correctness of Equation (10), the simulated *F*<sub>1</sub> is compared with the saludated are by Equation (10) (Figure 7b), and the simulated *F*<sub>1</sub> is compared with the calculated one by Equation (10) (Figure [7b](#page-8-0)), and the  $\frac{1}{2}$  reaches the largest after re-entering the turbine mode, two are in good agreement.  $\rho Q^2/A_2$  reaches the largest after re-entering the turbine mode, which results in the maximum of  $F_1$ . After entering the turbine braking mode,  $F_1$  decreases to the minimum because of the smaller  $F_p$  and  $\rho Q^2/A_2$ .

<span id="page-8-0"></span>

Figure 7. Histories of axial forces in MFC during the runaway transient process. (a) Axial force components. (b) Comparison of *F*<sub>1</sub> between values by simulation and by Equation (9).

Equation (10) is valuable, because we can use it to estimate the MFC axial force history if borrowing the basic results of 1D transient simulation.

(2) Axial hydraulic forces in CFC— $F_2$  and  $F_3$ 

The change laws of  $F_2$  and  $F_3$  a[re](#page-8-1) different from  $F_1$  (Figure 8a). However,  $F_2$  and  $F_3$ possess the same changing trend of magnitude, but in opposite direction. The pressure in the clearance cavity, transmitted from the main flow channel (Section  $3.3.1$ ), is the source of the axial forces in CFC. The variation law of  $|F_2|$  matches well with that of CHC clearance inlet pressure  $P_{\text{in}}$  (dimensionless pressure), indicating that the dominant factor of the clearance axial force is the clearance inlet pressure (Fi[gu](#page-8-1)re 8b). However, the pressure in the clearance region is also influenced by the runner rotational speed. Therefore, the clearance axial force can be represented as a function of the clearance inlet pressure and rotational speed. (Se[ction 3](#page-13-0).3.3).

<span id="page-8-1"></span>

**Figure 8.** Histories of axial forces in CFC during the runaway transient process. (a)  $F_2$ ,  $F_3$  and  $F_1$ . (b)  $|F_2|$  and pressure at the CHC inlet  $P_{\text{in}}$ .

# 3.2.2. Radial Forces 3.2.2. Radial Forces 3.2.2. Radial Forces

The total radial force  $F_r$  can be divided into the component  $F_x$  in the *x*-direction and the component  $F_y$  in the *y*-direction, namely,  $F_r = (F_x^2 + F_y^2)^{0.5}$ . In the rated operating condition, when the runner rotates for a cycle (the period is 0.12 s), the radial forces in MFC and CFC together with the total radial force, show periodically change, not only in value, but also in direction (Figure 9a,b). The [nu](#page-9-0)mbers of the peaks for MFC radial force and the total force in a cycle are equal to the blade number. According to the variation law of pressure in the middle span of the vaneless space (Figure [9c](#page-9-0)), it can be concluded that

the change of radial force in the rated operating condition is mainly affected by stator-rotor interaction. The direction of  $F_x$  and  $F_y$  are shown in Figure 9a, the  $F_x$  is perpendicular to the spiral casing inlet flow. Even though the flow patterns are smooth in the rated operating conditions, the  $F_x$  is greatly asymmetric and random due to the vortexes and backflow around the casing tongue and is dominated by the rotor-stator interference. The periodical change of  $F_x$  dominates the change of the total radial force, and the radial force in CFC is smaller than that in MFC (Figure [9d](#page-9-0)).

<span id="page-9-0"></span>

Figure 9. Variation laws of radial forces in a cycle at the rated working point. (a) The direction of radial forces. (b) Value of radial forces. (c) Histories of pressure in the vaneless space. (d) Oscillating laws of *F*x, *F*y. laws of *F*x, *F*y.

During the runaway transient process, the total radial force increases more than 100 times and reaches the maximum when the working point approaching the runaway point. Its MFC component accounts for more than 70% and dominates the changing tread of the total radial force (Figure 10[a,b\)](#page-10-0). In MFC, when the working point closes to the runaway point, the radial velocity at the hub side rapidly decreases because of the extremely serious point, the radial velocity at the hub side rapidly decreases because of the extremely serious rotating stall, then the reverse flow occurs, which leads to a rapid increase of radial force in MFC (Figure [11\)](#page-10-1). Until the working point enters the turbine braking mode, the reverse in MFC (Figure 11). Until the working point enters the turbine braking mode, the reverse flow gradually transits from the hub side to the middle span, and becomes smooth, then flow gradually transits from the hub side to the middle span, and becomes smooth, then the radial force decreases rapidly. When the working point re-enters the turbine mode, the radial force decreases rapidly. When the working point re-enters the turbine mode, the intensity of the reverse flow at the middle span gradually decreases, and the rotating stalls in blade channels increases, causing the increases of the radial forces. Furthermore, the radial force in MFC is also influenced by the vortex rope, the number of guide vanes, and so on [\[30](#page-19-9)[,31\]](#page-19-10), which would not be discussed in detail. In CFC, the radial forces in CHC and CSB show the same change trend (Figure  $10c,d$  $10c,d$ ). Due to the larger radial projected area, the radial force in CSB is larger than that in CHC. According to Section [3.4.2,](#page-14-0) the strong rotating shear flow cannot generate the radial forces, while the secondary flow vortices in

the radial direction can influence the radial forces in CFC. The strength of secondary flow vortices depends on the rotational speed of the rotating walls. The higher the rotational speed, the larger the centrifugal force and the radial velocity; the stronger the secondary flow vortices and the larger the radial force in CFC. Numerically, the fluctuation of runner tion of reflect that the imperial forces in equality caused by the flow evolution in MFC, however, the flow in CFC aggravates it.

are a, the radial force in CSB is larger than that in  $C\in\mathcal{C}$  is larger than that in  $C\in\mathcal{C}$ 

<span id="page-10-0"></span>

Figure 10. Variation laws of radial forces during the runaway transient process. (a) Total radial force. (b) Radial force in MFC. (c) Radial force in CHC. (d) Radial force in CSB.

(**c**) (**d**)

(**c**) (**d**)

<span id="page-10-1"></span>

**Figure 11.** Radial velocity, circumferential velocity and pressure at the inlet of the runner during the runaway transient process. (**a**) Origin data. (**b**) Filtered data.

# *3.3. Pressure Distribution in CHC and Its Formulation 3.3. Pressure Distribution in CHC and Its Formulation*  3.3.1. Pressure Distribution in CHC

the runaway transient process. (**a**) Origin data. (**b**) Filtered data.

# <span id="page-11-0"></span>3.3.1. Pressure Distribution in CHC

At the rated working point, the pressure gradually decreases inward the MFC and CHC, and the decrease degree in CHC is less than that in MFC (Figure 12a). At the sealing ring position of CHC, the water flows into the cavity in a jet form; the strong jets from the small gaps lead to a huge energy dissipation due to the small size of the sealing ring. Therefore, the pressure drops sharply every time after passing through each labyrinth sealing ring. The pressure distribution is basically uniform along the circumferential direction, namely, there is no circumferential pressure gradient (Figure [12b](#page-11-1)).

<span id="page-11-1"></span>

Figure 12. Pressure distribution at the rated working point. (a) Meridian section of whole flow channels. (**b**) Horizontal cross-section of CHC. channels. (**b**) Horizontal cross-section of CHC.

In order to investigate pressure fluctuations in CHC, several pressure monitoring In order to investigate pressure fluctuations in CHC, several pressure monitoring points on the plane of  $x = 0$  were set in CHC (Figure 13[\), w](#page-11-2)here  $P_1$  is the monitoring point at the vaneless space in MFC, which is located in front of CHC inlet;  $P_2$  to  $P_8$  are in CHC;  $P_9$  is in the draft tube, which is located at the outlet of the pressure equalizing pipe. During the runaway transient process, the pressure changing trends of the measuring points in front of the sealing ring are consistent with that of  $P_1$ , while that behind the sealing ring are consistent with that of  $P_9$  (Figure [14\)](#page-12-0). Therefore, it is the pressure change in MFC that dominates the pressure change in CHC. In addition, the pressure in CFC shows a few pulsating signals from MFC, which indicates that the severe pressure fluctuation in the parally eigened from the cynthetic materials and the severe presente including amplitude decreases at a large scale, more than 75%. decreases at a large scale, more than 75%. decreases at a large scale, more than 75%.  $I = I \cup I \cup I$  order to investigate pressure monitoring in  $C$ product to investigate pressure including in CHC, several pressure infinitoring

<span id="page-11-2"></span>

**Figure 13.** Arrangement of pressure monitoring points. **Figure 13.** Arrangement of pressure monitoring points. **Figure 13.** Arrangement of pressure monitoring points.

#### 3.3.2. Pressure Calculation in CHC

According to the pressure distribution in CHC, the huge pressure is the main contributor to the large clearance forces. Therefore, accurate calculation of the clearance pressure can provide the basis for predicting the clearance forces. For the low specificspeed pump-turbine, the position of the sealing ring is relatively close to the shaft due to the relatively larger  $D_1/D_2$ , leading to a larger external cavity area and smaller internal cavity area. Moreover, the pressure in the external cavity is much higher than that in the internal cavity. Both of that result in a huge axial force in the external cavity, which is the dominant component of CHC forces.



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Figure 14. Pressure variation trend in CHC during the runaway transient process. (a) Original data. (b) Filtered data.

In CHC, the strong rotating shear flow containing a velocity core region in the circumferential direction, and some secondary flow vortices in the meridian section, are formed (Section 3.4.2). Compared with the circumferential velocity, the secondary flow velocity is smaller than a tenth of the circumferential velocity and can be ignored for general analysis. Therefore, the surface forces are equal to the mass forces on any fluid mass in the radial  $\sum_{i=1}^{n}$ direction (Figure 15), and Equation (11) can be listed based on the liquid equilibrium<br>differential equation Figure (11) can be simplified as Equation (12) as follows: differential equation. Equation (11) can be simplified as Equation (12) as follows:

$$
\left(p - \frac{\partial p}{\partial x}\right) dy dz - \left(p + \frac{\partial p}{\partial x}\right) dy dz + \rho f_x dx dy dz = 0
$$
\n(11)

$$
\frac{\partial p}{\partial x} = \rho f_x \tag{12}
$$

where  $p$  is the pressure;  $f_x$  is the mass force. For the water body in CHC, the mass force is only the centrifugal force when ignoring the gravity. Assuming the core region rotational velocity is  $k\omega$ , we can modify Equation (12) to: where  $p$  is the pressure,  $f_X$  is the mass force. For the water body in CFIC, the mass force is

$$
dp = \rho(k\omega)^2 r \tag{13}
$$

<span id="page-12-1"></span>

**Figure 15.** Forces diagram in CHC. **Figure 15.** Forces diagram in CHC.

By integrating Equation (13) from the CHC inlet to the sealing ring inlet, we obtain the radial pressure drop  $\Delta P$  in the external cavity can be obtained as follows:

$$
\int_{p}^{p_{\rm in}} dp = \int_{r}^{R_1} \rho (k\omega)^2 r dr \tag{14}
$$

$$
\Delta P = \frac{\rho(k\omega)^2}{2} (R_1^2 - r^2)
$$
\n(15)

where  $\omega$  is the runner rotational speed;  $k$  is an unknown coefficient,  $r$  is the radius of the clearance area. From Equation (15), it can be seen that the pressure drop in CHC is the clearance area. From Equation (15), it can be seen that the pressure drop in CHC is proportional to the square of both the radius and the rotational speed. proportional to the square of both the radius and the rotational speed.

2

In order to verify the correctness of Equation (15), the results from simulations are com-In order to verify the correctness of Equation (15), the results from simulations are pared with those from Equation (15). Assuming a linear distribution for the circumferential velocity along the cavity height, *k* is equal to 0.5. The pressure drop in the external cavity ∆*P* calculated by Equation (15) and simulation do not match well (Figure [16a](#page-13-1)). However, when the two are in good agreement when let  $k = 0.475$ , the two will agree well. Further study is needed to explore *k* value and the core region velocity.

During the runaway transient process, the oscillating law of ∆*P* calculated by Equation During the runaway transient process, the oscillating law of ∆*P* calculated by Equa- $(15)$  is basically consistent with the simulation result (Fi[gure](#page-13-1) 16b). The rotating speed is the dominant factor for the ∆*P*; the larger the rotating speed, the larger the centrifugal force, fthe larger the pressure drop ∆*P* in the clearance cavity. However, the simulated ∆*P* shows some pulsating signals from the CHC inlet.

<span id="page-13-1"></span>

Figure 16. Pressure in the external cavity of CHC. (a) Pressure distribution along the radial direction at the rated working point. (b) Changing laws of the pressure drop in the external cavity during the runaway transient process.

## <span id="page-13-0"></span>3.3.3. Axial Force Formula 3.3.3. Axial Force Formula

According to the pressure drop formula, the pressure at any position in the external According to the pressure drop formula, the pressure at any position in the external cavity can be calculated according to Equation (16). By integrating, the pressure, we can cavity can be calculated according to Equation (16). By integrating, the pressure, we can obtain the axial force in the external cavity  $F<sub>o</sub>$  as follows:

$$
P = P_{\rm in} - \frac{\rho(k\omega)^2}{2} (R_1^2 - r^2)
$$
 (16)

$$
F_{\rm o} = \int_{R_{\rm m}}^{R_{\rm 1}} [P_{\rm in} \frac{\rho(k\omega)^2}{2} (R_{\rm 1}^2 - r^2)] 2\pi r dr = \pi (R_{\rm 1}^2 - R_{\rm m}^2) [P_{\rm in} - \frac{\rho(k\omega)^2 (R_{\rm 1}^2 - R_{\rm m}^2)}{4}] \tag{17}
$$

where *F*<sub>o</sub> is the axial force of the external cavity; *P*<sub>in</sub> is the inlet pressure of CHC. Therefore, the clearance inlet pressure  $P_{\text{in}}$  and the rotational speed *n* are the influence factors of the clearance axial force.

Similarly, the axial force in the internal cavity can be calculated by integrating the internal cavity pressure. Thus, the axial force in CHC can be expressed as a function of the clearance inlet pressure and the rotational speed. With the runner structure parameters known, we can estimate the axial force in CFC by Equation (17).

#### *3.4. Flow Patterns in the Clearance Region*

# 3.4.1. Streamline Distribution in Clearance Region

In order to qualitatively understand the velocity distribution in CHC, the streamlines in different height sections in CHC are shown in Figure [17.](#page-14-1) In the section  $z/h = 0$ , namely, the upper surface of the hub, the outward spiral flow is formed due to the larger centrifugal force, while in the section  $z/h = 1$ , the lower surface of the cover, the inward spiral flow is formed. In the middle section ( $z/h = 0.5$ ), the radial velocity is close to zero, forming stable strong rotating shear flow. In the meridian section, the secondary flow vortices are shown in Figure [18.](#page-14-2) The pressure difference force is greater than the centrifugal force near the cover and the bottom ring, forming a radial inward flow boundary layer, while exactly opposite near the hub and shroud, where a radial outward flow boundary layer is formed. At the sealing ring position, the flow velocity greatly increases when passing through the sealing ring because of its small size, and the outflow flows into the cavity in a form of jet, which also forms secondary flow vortices.

<span id="page-14-1"></span>

3.4.1. Streamline Distribution in Clearance Region

<span id="page-14-2"></span>Figure 17. Streamline diagrams in different height sections in CHC at the rated working point. (a)  $z/h = 0$ . (b)  $z/h = 0.5$ . *z*/*h* = 1. (**c**) *z*/*h* = 1. *z*/*h* = 1.



Figure 18. Streamline diagrams at the meridian section in CHC and CBS at the rated working point.

<span id="page-14-0"></span>3.4.2. Velocity Distribution in the Clearance Region

In order to quantitatively analyze the velocity evolution in CHC during the runaway process, the velocity change histories at different radius and heights are shown below.

(1) Circumferential velocity distribution

At the rated working point, a velocity core region is formed in the circumferential direction. The circumferential velocity reaches its maximum on the rotating wall, while reduced to zero on the stationary wall, and is between those two for the core region (Figure [19a](#page-15-0)). The velocity core region rotates with a fixed circumferential velocity at the fixed radius similar to a rigid body. The larger the radius, the greater the core region circumferential velocity. From the previous studies, the clearance cavity of a pump-turbine is similar to a thin-layer cavity composed of two disks, in which one disk is stationary and the other is rotating. Batchelor [32] first analyzed the flow between a rotating wall and a stationary wall with infinite boundary and found that the fluid in the central region rotated with a fixed angular velocity. Mukherjee [\[33\]](#page-19-12) tested fluid in finite rotating and stationary disks, and a velocity core region was also observed when the cavity height was large. Then disks, and a velocity core region was also observed when the cavity height was large. Then<br>Singh [34] found that the fluid in the core region rotated with a rotational speed lower than that of the rotating disk. The above existing research on the flow between a rotating wall and a stationary wall proves the correctness of this numerical calculation of clearance flow.

radius similar to a rigid body. The larger the radius, the greater the core region circumfer-

<span id="page-15-0"></span>

Figure 19. Circumferential velocity distribution along the cavity height in CHC. (a) Rated operating condition. (b) Runaway transient process.

During the runaway transient process, the velocity core region is still maintained, its velocity changes with the rotating speed (Figure [19b](#page-15-0)). Similarly, the circumferential velocity near the rotating wall is always the largest. With the increase of rotational speed, the velocity in the core region and the viscous force on the rotating wall become larger.

(2) Radial velocity distribution than a tenth of the circums-

Radial velocity along the cavity height direction at different radii is shown in Figure [20a](#page-16-0). The flow is outward around the rotating wall, while inward around the stationary wall, and the radial velocity is about zero at the middle height of the cavity. Compared with the larger the radial velocity. circumferential velocity, the radial velocity is smaller than a tenth of the circumferential velocity. During the runaway transient process, the radial velocity changes with the rotating speed (Figure [20b](#page-16-0)). The larger the rotating speed of the rotating wall, the greater the centrifugal force, the stronger intensity of the secondary flow vortices and the larger the radial velocity.

<span id="page-16-0"></span>

Figure 20. Radial velocity distribution along cavity height in CHC. (a) Rated operating condition. (b) Runaway transient process. process.

# **4. Conclusions 4. Conclusions**

In this study, the runaway oscillating process of a low specific-speed pump-turbine In this study, the runaway oscillating process of a low specific-speed pump-turbine was simulated by using 3D CFD software. By analyzing the flow patterns, pressure distribution and dynamic hydraulic forces in CFC, we obtained the formulas for estimating the axial hydraulic forces in both MFC and CFC, as well as their influence factors and change mechanism, which benefits to a professional runner design. We found that the change mechanism, which benefits to a professional runner design. We found that the CFC component of the total runner axial hydraulic force shows a similar significance level to the MFC component during transient processes, and this component should be considered when studying the runner forces. The detailed conclusions are as follows:

(1) During the runaway oscillating process, the total runner axial hydraulic force  $F_w$ periodically fluctuates with large amplitudes, and reaches the upward maximum before periodically fluctuates with large amplitudes, and reaches the upward maximum before the runaway point, while comes up to the downward maximum in turbine braking mode. This axial hydraulic force can be quantified by the components in MFC and CFC. The component in MFC can be formulated as a function of the momentum changing rate and the runner outlet axial pressure, while the CFC component as a function of the clearance inlet pressure and rotational speed.<br> $\therefore$ 

(2) The fluctuation of runner radial force is mainly caused by the flow evolution in  $\sum_{n=1}^{\infty}$ MFC, however, the flow in CFC aggravates it. During the runaway process, the extremely services and the extremely serious rotating stating the radio to the rapid increase of radial force in MFC when the rapid increase of  $\sim$ working point closes to the runaway point. For the radial force in CFC, the rotational speed is the main influence factor. serious rotating stall in MFC leads to the rapid increase of radial force in MFC when the is the main influence factor.

speed is the main influence factor. (3) It is the pressure change in MFC that dominates the pressure change in CHC. The (3) It is the pressure change in MFC that dominates the pressure change in CHC. The radial pressure drop ∆*P* in the external cavity of CHC can be formulated as a function radial pressure drop ∆*P* in the external cavity of CHC can be formulated as a function of of the square of both the radius and rotational speed. The pressure in CFC shows a few pulsating signals from MFC, indicating that the severe pressure fluctuation in the vaneless pace can be transmitted into the clearance region, but the amplitude greatly decreases.

Even though we have found the velocity core region in the clearance region, its magnitude and influence factors are still unclear. In order to accurately calculate the clearance pressure and axial hydraulic force, the velocity value of the core region needs to be further studied.

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# **Abbreviations**





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