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Impact of a Synthetic Component on the Emission of Volatile Organic Compounds during the Combustion Process in a Miniature Turbine Engine

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Abstract: This paper refers to the study of biofuel as an alternative power source for turbine aviation engines. Blends of Jet A-1 fuel and synthesized hydrocarbons from Hydrotreated Esters and Fatty Acids (HEFA) technology at different proportions, such as 25%, 50% and 75%, were used for tests. All the test results were compared with the neat Jet A-1 fuel. A miniature GTM series turbojet engine was used in the test rig studies. During the tests conducted at a specific rotational speed, selected engine operating parameters as well as the emission of volatile organic compounds were measured. In terms of engine performance, no significant differences were found between the test fuels. The results of volatile organic compound emissions indicate that among the most toxic compounds the highest concentrations were obtained for benzene. The addition of the HEFA synthetic component and increasing its proportion in the blend resulted in the obtained concentration values for benzene showing a decreasing trend. The plotted utility profile indicates that the most optimal blend, i.e., the least toxic, is the blend with the share (v/v) of 62.5% of Jet A-1 fuel and 37.5% of HEFA component.

Keywords: synthetic jet fuels; turbine engine; combustion process; exhaust emissions; volatile organic compounds

1. Introduction

In the last 10 years, the need to reduce carbon dioxide (CO_2) emissions has emerged in the aviation industry. One of the main directions that allow for the reduction of these emissions is the introduction of alternative power sources for the turbine engines that power modern aircraft. New fuels, especially the so-called 'drop-in' fuel, play an important role in this area. The 'drop-in' fuel is a fuel consisting of a blend of petroleum jet fuel with synthetic blending component, without the need to modify the aircraft power systems.

Currently, the standard [1] allows for the use in turbine aviation engines of seven technologies for the production of synthetic hydrocarbons as components for blending with fossil jet fuel. The maximum percentage of the synthetic component in the blend, depending on the technology, does not exceed 50%.

The European Commission Directive states that a volatile organic compound is any organic compound having a boiling point lower or equal to 250 °C, measured at a standard pressure of 1013 hPa [2]. Volatile organic compounds (VOCs) are involved in photochemical reactions that produce harmful and toxic products. VOCs may cause serious health effects such as: headaches and dizziness, drowsiness, irritation of mucous membranes, permanent liver or nervous system damage. Many VOCs exhibit toxic, carcinogenic, mutagenic or neurotoxic properties. In addition, many of them produce odours and thus contribute to



Citation: Gawron, B.; Górniak, A.; Białecki, T.; Janicka, A.; Włostowski, R.; Włóka, A.; Molska, J.; Zawiślak, M. Impact of a Synthetic Component on the Emission of Volatile Organic Compounds during the Combustion Process in a Miniature Turbine Engine. *Energies* **2021**, *14*, 8462. https://doi.org/ 10.3390/en14248462

Academic Editors: Jesper Schramm and David Chiaramonti

Received: 28 October 2021 Accepted: 10 December 2021 Published: 15 December 2021

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Copyright: © 2021 by the authors. Licensee MDPI, Basel, Switzerland. This article is an open access article distributed under the terms and conditions of the Creative Commons Attribution (CC BY) license (https:// creativecommons.org/licenses/by/ 4.0/). the deterioration of human living conditions. VOCs enter the body through the respiratory system, the digestive system and through the skin. They are listed as the second most harmful air pollutant after nitrogen oxide (NO_x) [3].

VOCs include many groups of compounds, the most numerous of which are:

- aliphatic hydrocarbons (e.g., ethane, ethylene);
- ring hydrocarbons (cycloalkanes);
- aromatic hydrocarbons (e.g., benzene, toluene, xylenes);
- halogenated hydrocarbons (e.g., chloromethane);
- nitrated hydrocarbons (e.g., nitrobenzene);
- alcohols and phenols (e.g., methanol, ethanol);
- carbonyl derivatives (e.g., acrolein, acetone);
- carboxylic acids and esters (e.g., formic acid, acetic acid);
- heterocyclic organic compounds containing inter alia nitrogen, oxygen, sulphur; most of them are the so-called odours,
- aliphatic sulphur compounds (odours);
- aliphatic amines (odours);
- aromatic amines (e.g., aniline).

Within the group of VOCs, compounds from the BTX group deserve special attention. These are aromatic hydrocarbons such as benzene, toluene and xylenes. Their highest concentrations can be observed along traffic routes. The most toxic one is benzene—the simplest aromatic hydrocarbon, volatile, fragrant, flammable, lighter than water and insoluble in it. Benzene enters the body mainly through the respiratory tract and the skin. It has strong carcinogenic, mutagenic, teratogenic and embryotoxic properties. After ingestion, it causes nausea and vomiting. Acute poisoning may lead to death, preceded by convulsions, arrhythmia or respiratory distress [4]. Benzene occurs naturally in crude oil and coal tar. Benzene in atmospheric air comes partly from evaporation of fuels at petrol stations and fuel depots, as well as from technological processes that use organic solvents and in coking and petrochemical processes. Internal combustion engines are the main source of BTX. In addition, it is estimated that emissions associated with fuel evaporation account for 30–50% of total BTX emissions from vehicles [5].

In the literature, there are articles concerning the research on the emission of volatile organic compounds carried out on aviation turbine engines [6–9]. However, it is worth mentioning that these papers discuss the tests conducted on turbine engines in full scale. An interesting solution may be to carry out tests of alternative fuels for aviation on turbine engines in small scale, which may constitute a preliminary verification of a new fuel before starting tests on larger engines. In the literature, there are numerous papers of different nature on small-scale turbine engines. A few selected items published only in 2020–2021 are included in [10–13].

It is worth mentioning that generally articles describing tests of alternative fuels on small scale turbine engines concern the assessment of fuel influence on the performance and exhaust emissions, mainly regarding gaseous components of exhaust gases and particulates. There is a lack of studies using a small-scale turbine engine for assessment of exhaust gas toxicity, including emissions of volatile organic compounds.

The cooperation between the authors of this paper has already resulted in several publications [14–17], precisely on the assessment of toxicity of exhaust gases emitted during operation of a miniature jet engine powered by various fuels. Papers [14–16] concern toxicity assessment using the BAT-CELL method, which consists of determination of virtual toxic impact of the gas mixture on the living cells. Paper [17] presents the results of studies on the emission of volatile organic compounds (including benzene, toluene) of petroleum fuel with 25% of biobutanol.

The aim of this study is the emission characteristic of volatile organic compounds of a miniature turbojet engine fed with blends of Jet A-1 and HEFA synthetic component. The results obtained for blends were compared with results for neat Jet A-1.

A schematic diagram of the test rig, in which the dynamometer tests were carried out, is presented in Figure 1. The essential element of the rig was a miniature GTM-140 turbojet engine, whose purpose was to generate gases as a result of the process of combustion of test fuels. The gas turbine specification are presented in Table 1.



Figure 1. Test rig (A)—miniature test engine, (B)—exhaust pipe, (C)—exhaust gas collection and storage system.

Parameters	Descriptions		
type of engine	Turbojet—single spool		
compressor	single stage radial compressor		
combustion chamber	annular combustion chamber		
turbine	single stage axial flow turbine		
rpm	33,000–120,000		
max. thrust	140 N		
fuel consumption at max. rpm	420 g/min		
mass flow at max. rpm	350 g/s		
max. exhaust gas temperature	750 °C		

Table 1. Technical parameters of GTM-140 engine.

The presented GTM-140 engine is an main element of the Miniature Jet Engine Test Rig (MiniJETRig), built in Air Force Institute of Technology in Poland. The construction and research capabilities of the engine rig are described by Gawron and Białecki in [18]. Information on the applied measuring sensors regarding basic operating parameters together with their accuracy is presented in [19].

MiniJETRig besides turbojet engine is equipped with two analyser: portable exhaust analyser and particulate matter analyser. The test rig has a control system with data acquisition based on the measurement cards. The operating speed of a miniature GTM-140 turbojet engine is 33,000–120,000 rpm and its maximum thrust is 140 N. The engine the engine is built based on single-stage radial compressor, a single-stage axial turbine and an annular combustion chamber (with vaporizer tubes). Engine is started via an automatic electric starter. It is also possible to start the engine pneumatically which enables it additionally air mass flow rate measurement. The engine's fuel supply system was modified by splitting subsystem of supplying fuel to the combustion chamber and bearings lubrication subsystem. This solution enables the combustion of tested fuel in engine without oil component.

MiniJETRig is used for the research on jet fuel combustion process. The results of previous work carried out on this rig were presented in papers [20–23]. The work included research on blends of petroleum jet fuel with synthetic components from two synthetic hydrocarbon production technologies: HEFA and ATJ. The purpose of the research was

to analyse the effect of the addition of a component to Jet A-1 fuel on the engine performance characteristics and the characteristics of the emissions of exhaust gaseous exhaust components.

The tests were carried out for all test fuels according to the same sample profile (Figure 2). Two independent tests were performed for each fuel. After starting the engine on a given fuel, the rotational speed was gradually increased. It was assumed beforehand that measurements would take place at one selected engine performance condition, i.e., 62,000 rpm. The choice was dictated by the fact that this speed corresponds to approx. 30% of the value of the maximum thrust force for a miniature turbojet engine. The 30% thrust value occurs during the certification procedure in the field of gaseous emissions from aircraft turbine engines [24].



Figure 2. Engine test profile during dynamometer tests.

After reaching a predetermined rotational speed, the period of 30 s was waited in each case for the thermodynamic state of the engine, and thus the combustion process, to stabilise. After this time, the actual measurement began, consisting of the collection of the exhaust gases into a properly prepared system containing Tedlar bags (Figure 3). The test material (exhaust gases) collected in this way was then used to determine the volatile organic compounds. During the tests the basic parameters of engine performance were also recorded: engine thrust (strain gauge bar), rotational speed (optical sensor) and temperature behind the turbine (k-type thermocouple).

The concentration of volatile organic compounds in the samples was evaluated with the aid of gas chromatography according to the Emission Research Laboratory Test Procedure No. 1/2010, using a Varian 450-GC gas chromatograph. The chromatograph was equipped with a flame ionization detector (FID), and a column Varian VF-WAXms 30 m \times 0.25 mm ID, DF 0.25 um. The temperature at the column was 373 K, 523 K at the dispenser and 423 K and the detectors. The samples were collected with YEARS aspirator whose flow rate was adjusted to 30 dm³/h. Each time the 10 dm³ of exhaust gases was aspirated. The gas was adsorbed on active carbon Anasorb[®] SKC CSC with volume of 5 cm³. Extraction proceeded for 20 min. During the test an adequat mixing of the material was ensured. Then 5 µL of the solution was drawn from above the carbon layer. The collected sample was injected into the chromatograph. The total relative error of the method was estimated at 20% [25].





Figure 3. Exhaust gas collection and storage system: (**a**) technical diagram; (**b**) illustrative photograph.

The identification of VOCs is usually performed in three steps: collection of samples, desorption and chemical analysis. In order to identify VOCs and their concentrations, a special probe was placed directly in the exhaust gases, samples of the latter were aspirated and subjected to chromatographic tests.

Engine tests were conducted on selected fuels:

- pure Jet A-1 fuel—Jet designation;
- blends of Jet A-1 with a HEFA technology component (used cooking oil feedstock) in concentrations of: 25%, 50%, 75% share (v/v) of the biocomponent—designation JetX/HEFAX, where X stands for the specific concentration of the given component.

Selected physicochemical properties of test fuels are presented in Table 2. As the biocomponent share in the blend with jet fuel increases, the density of such a blend decreases with simultaneous increase of kinematic viscosity. On the other hand, the parameter—net heat of combustion, which is directly related to the process of fuel combustion, increases with increasing biocomponent share. This may translate into higher combustion heat generated during operation of a turbine engine and higher temperature obtained in the combustion chamber. The neat biocomponent does not contain aromatics; the aromatic content in the blend comes only from the fossil jet fuel. Moreover, the content of aromatics in fuel is undesirable due to its negative impact on the clean of the combustion process. On the other hand, the content of aromatics is important for ensuring an appropriate sealing function in aircraft fuel systems and installations [26].

Lp.	Property Name	Unit of Measurement	Testing Method	Jet	Jet75/ HEFA25	Jet50/ HEFA50	Jet25/ HEFA75	HEFA
1.	Density at 15 $^\circ \mathrm{C}$	kg/m ³	ASTM D 4052	789.3	779.8	770.7	761.4	751.9
2.	Viscosity at $-20\ ^\circ C$	mm ² /s	ASTM D 445	2.998	3.227	3.481	3.754	4.077
3.	Net heat of combustion	MJ/kg	ASTM D 3338	43.312	43.542	43.741	43.948	44.172
4.	Aromatics	% (v/v)	ASTM D 1319	15.1	11.0	7.2	4.0	0

Table 2. Selected physicochemical properties of Jet A-1, HEFA component and their blends.

3. Results and Discussion

As regards engine operating parameters, the results are presented in Table 3. Engine tests for each fuel were performed twice. The table shows the average values of a given parameter for a given engine test.

Fuel	Rotational Speed [rpm]	Engine Thrust [N]	Temperature behind the Turbine [$^{\circ}$ C]
Jet	62,148	40.6	510
Jet75/HEFA25	62,075	40.6	509
Jet50/HEFA50	62,658	42.7	508
Jet25/HEFA75	62,730	41.3	509

 Table 3. Engine operating parameters during tests with different fuels.

On the basis of the presented results, it can be seen that for all tested fuels the engine operated in a comparable range. Similar values of rotational speed, engine thrust and temperature behind the turbine were obtained. During the tests, no fuel consumption was measured that could show differences between the tested fuels related to the influence of both density and neat heat of combustion of a given fuel.

The measured concentration values of volatile organic compounds in the exhaust gases emitted by the GTM-140 miniature engine are shown in Figure 4. The main focus here is on the compounds from the BTX group—i.e., benzene, toluene and xylenes, due to their highest harmfulness among the identified compounds. The most harmful of the BTXs is benzene. In this case, it turns out that the more HEFA component a fuel blend contains, the lower benzene concentrations are obtained. On the other hand, it can be observed that the concentration of toluene and xylene tends to increase with increasing share of the component in the mixture. Nevertheless, the decreasing trend of benzene is dominant here. The aggregate comparison of BTX (orange colour in the graph) also shows a decreasing trend in the harmfulness of the emitted exhaust gases from the miniature engine with an increasing share of the synthetic component in the tested fuel blends.

The effect of fuel blend composition on VOC emissions is presented as the utility profile. This makes it possible to determine the optimum blend composition, in which exhaust toxicity is minimised by minimising the concentration of BTX, while maintaining comparable engine performance. Additionally, it is possible to observe the general effect of fuel composition change on each of the individual parameters measured [27].

Benzene has the highest concentration (Figure 4) compared to other compounds, therefore it was concluded that the concentration of this parameter would be the most representative in the BTX group. Xylenes consist of m-xylene, o-xylene and p-xylene. The concentrations of these compounds were summed and presented as xylenes.

The utility profile is shown in Figure 5. Utility has been coded from 0 to 1, with 0 being considered undesirable and 1 being high, desirable utility. Therefore, in the graph, a high BTX concentration is assigned the utility value of 0 and a low concentration is assigned the high utility (i.e., utility of 1). By analysing the profiles, it can be concluded that the optimal blend of Jet A-1 fuel with HEFA synthetic component consists of a share (v/v) of 62.5% fossil fuel and 37.5% of HEFA component. In this case, the overall utility was 0.70767.



Figure 4. Emission of volatile organic compounds depending on the share of the synthetic component in the tested fuel blend.



Figure 5. Desirability function determining the effect of BTX compounds on the fuel blend contents.

4. Conclusions

In this study, emissions of volatile organic compounds from the miniature turbojet engine were analysed. Research was conducted using synthesized hydrocarbons from HEFA technology. Blends of Jet A-1 fuel with the synthetic component HEFA at different concentration ratios were analysed. All the test results were compared with the neat Jet A-1 fuel.

The main conclusions can be summarized as follows:

- The use of a miniature turbine jet engine for testing the exhaust gas composition, including the assessment of volatile organic compound emissions, gives great benefits related to the possibility of planning much wider tests than in the case of full-scale turbine engine tests.
- In the performance analysis of the miniature jet engine, no significant differences were observed for all test fuels. After reaching the predetermined speed, the engine powered by different fuels generated similar thrust and temperature behind the turbine.
- As regards the emissions of volatile organic compounds in exhaust gases, the focus was on the most harmful compounds—the BTX group (benzene, toluene and xy-lene). Among this group of compounds, the highest concentrations were recorded for benzene.
- As the proportion of HEFA component in the blend increases, the concentration of benzene shows a decreasing trend, while the concentration of toluene and xylenes does not show an unequivocal tendency.
- The developed utility profile indicates that the most optimal, i.e., the least toxic in terms of exhaust gases (minimising the BTX concentration) is a blend with a share (v/v) of 62.5% fossil fuel and 37.5% of HEFA component.
- The next stage of research, planned for the near future, involves testing alternative fuels with regards to VOC emissions using a miniature jet engine at several engine operating modes, i.e., several rotational speed ranges.

Author Contributions: Conceptualization, B.G. and A.G.; methodology, B.G. and A.G.; investigation, B.G., A.G. and R.W.; data analysis, B.G., A.G., A.W. and J.M.; writing—original draft preparation, B.G. and A.G.; writing—review and editing, A.J. and T.B.; visualization, B.G. and A.G.; supervision, A.J., T.B., M.Z.; resources, B.G. and A.G. All authors have read and agreed to the published version of the manuscript.

Funding: This research received no external funding.

Institutional Review Board Statement: Not applicable.

Informed Consent Statement: Not applicable.

Data Availability Statement: Data sharing not applicable.

Acknowledgments: The research was carried out using the R&D infrastructure of the project: Complex GEO-3EM ENERGY ECOLOGY EDUCATION.

Conflicts of Interest: The authors declare no conflict of interest.

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