

# Solar Disc Concentrator: Material Selection for the Receiver

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**Abstract:** Solar concentration is the ability to harness solar radiation in order to increase the temperature of a receiver. The receiver is a component into which a heat transfer fluid can flow in an ORC system, which produces electricity, or it can be used for high-temperature thermal storage or even to implement thermochemical cycles. The choice of material is critical to ensure optimal performance and long-lasting operation. It is also essential that such material can operate at high temperatures and high thermal gradients. In short, material identification involves high thermal stresses that result in structural deformation. Different metal alloys were used to verify that the yield strength limit was not exceeded due to thermal stress induced by concentrated solar radiation. Starting with the general heat equation, the problem was implemented in Matlab. The purpose was to test whether thermal stress exceeds the yield strength, which is the condition in which elastic bonds in the material are changed, causing deformation. This condition, if exceeded, is sufficient to discard the material; otherwise, it is a necessary but not sufficient condition to resist over time. The best material identified was Inconel 740H, which had a high yield strength value and the lowest temperature difference. Under extreme working conditions, it withstood induced thermal shocks.

**Keywords:** solar concentrator; thermal stress; material selection; receiver



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## 1. Introduction

Development of renewable resources is a topic of continuous and growing interest in the current energy and environmental context, within which the solar source is a crucial, rapidly developing technology related to the exploitation of solar concentration for energy purposes [1]. The development of concentrating solar power (CSP) systems is continuous and growing [1,2]. Islam et al. [1] showed the growth of CSP systems, with a steady increase of patents up to 1600 patents/year in 2015. Exploitation of solar radiation, however, is higher when considering photovoltaic (PV) technology because both direct and diffuse radiation are exploited. This is not the case with CSP systems. Another limiting factor of CSP technology lies in its inherent difficulty to be distributed, which is favourable to PV systems. The worldwide installation of PV systems stands at around 1185 GW by the end of 2022 [3], while power installation for CSPs is around 7 GW [4]. There are, however, arguments in favour of the concentration technique, i.e., for the same area occupied, CSP systems generate more electricity than photovoltaic systems. This shows that the economic return of CSP is greater [5]. Several reviews in the industry highlight the latest developments and steps forward in research to lower the cost per installed power and compete more with other power generation systems [1,5,6]. Alami et al. highlight the main critical aspects of such systems [6].

Concentration systems can be used directly or indirectly for the production of thermal energy [7–10], fresh water [11,12] and synthesis gas [13–15]. A further use is for the production of electrical energy using Stirling engines inserted in the focal zone of the concentrator [16–19]. Considerable research studies point out interesting aspects for improvement. For example, the efficiency of thermochemical cycles must be improved [20,21].

Considerable improvements, also in terms of cyclability and repeatability, must be achieved not only in solar-driven thermochemical cycles but also across other applications. In the study of Borghero et al., a critical aspect was shown: the difficulty of working under real conditions by the receiver placed in the focal point of a CSP disc [22].

This issue is common among all possible uses of solar concentration. The choice of material to be placed at the focal point and to be used as a receiver is important for the good functionality of the system. They must be resistant to high temperatures and able to withstand extreme temperature gradients. These aspects are highly stressed for thermochemical cycles where the operating temperatures can exceed 1000 °C [23] and thermal gradients can be as high as 300 °C/min. The strong thermal gradient shows the fragility of ceramic materials, while high-temperature conditions limit the functionality of metal alloys. In Borghero et al., it was shown that the use of sintered alumina is almost impossible to achieve operating conditions for the reduction reaction for Fe<sub>2</sub>O<sub>3</sub> with the use of sintered alumina [22]. Li et al., Liu et al. and Erasmus et al. showed similar issues [24–26]. Research in the field of synthesis gas production from thermochemical cycles is driven by electrical systems (industrial furnaces), and the focus is on the production and realisation of the best catalyst capable of cycling and withstanding such reactions. The research gap is, therefore, related to the identification and choice of material that can withstand the extreme operating conditions of thermochemical cycles. Possible solutions identified in the literature show the possibility of realising systems capable of working at high temperatures and high pressures using composite materials, even if the costs and processing techniques are prohibitive. Solar tubes under a non-uniform solar flux were employed by Du et al., who used two layers of protective metal material placed on a Nickel-based receiver with high thermal conductivity [27].

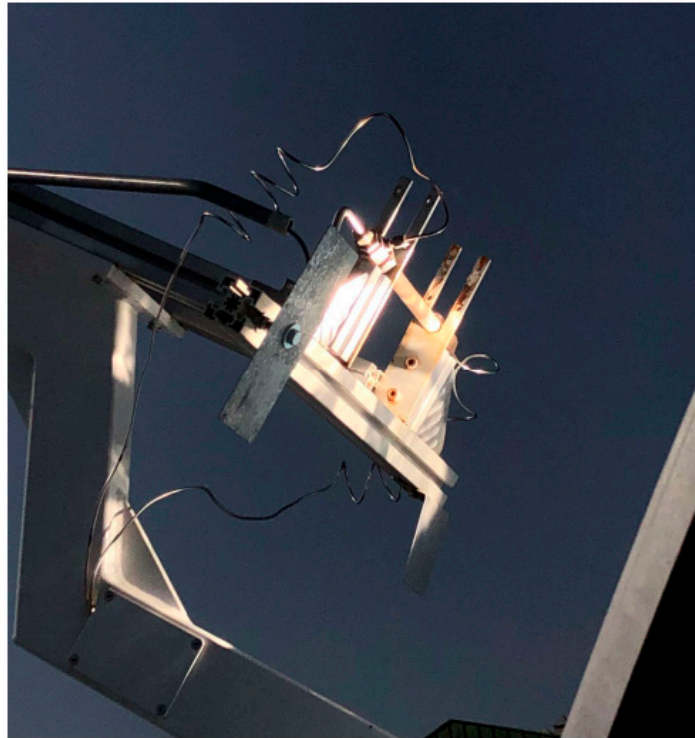
Encouraging results were shown at low thermal gradients and peak temperatures not exceeding 600–700 °C. This work was focused on the identification and behavioural study of thermal stresses induced by certain metal alloys that can be used as solar receivers, with emphasis on low-cost solutions. The solar concentrator at the Energy Center was used, in which a temperature higher than 1000 °C can be achieved with temperature gradients of about 300 °C/min.

The main objective of this work was to identify materials within selected metal alloys to highlight their ability to withstand the thermal stresses associated with strong thermal gradients. Strong thermal gradients can affect receivers under realistic conditions, whereas in most cases published in the literature, there was a focus on stationary thermal conditions.

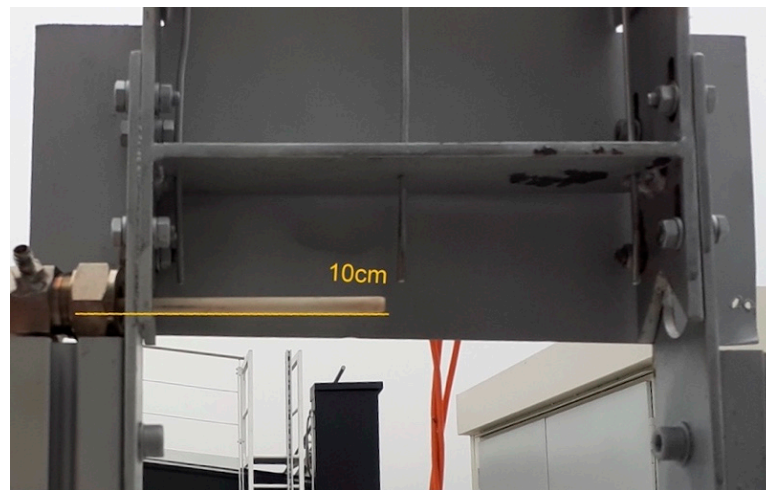
## 2. Materials and Methods

The receiver is the most critical component of the entire CSP structure because it must withstand high temperatures that sometimes exceed 1000 °C and large thermal gradients. The purpose of this project was to investigate different materials in order to select the most suitable one to withstand high-temperature conditions. This is the reason why special attention was paid to researching the most suitable materials to accomplish this task. Metal alloys were selected due to lower costs than composite materials. The receiver must first and foremost satisfy the optical properties, as can be seen from the discussion so far, but it is essential that it also has good thermal and mechanical characteristics. It must have high conductivity values for the heat to reach the heat transfer fluid (which is precisely the purpose of the whole apparatus); it must also possess great resistance to mechanical and thermally induced stresses and resist corrosion. However, it is appropriate to explain what causes of material breakdown or weakening are to be avoided before analysing them individually.

The solar concentrator was described elsewhere [19,28], and the following Figures 1–3 focus on the main receiver elements. Figure 1 shows the receiver illuminated by concentrated solar radiation; each receiver was placed in the indicated housing. The CSP was located in Turin, at the Energy Center rooftop (45.07 and 7.66).



**Figure 1.** Solar concentrator, focus of the receiver housing.



**Figure 2.** Thermocouple T0 (type B) inside the receiver.

Figure 2 highlights the experimental setup without the receiver. The B thermocouple “T0” (Tersid Srl, Sesto San Giovanni, Italy) was placed inside the receiver and located in the middle point.

The geometrical parameters of the solar concentrator (Elma, Net srl, Pregnana Milanese, Italy) are described in Figure 3. They were used in the model implemented in Matlab R2023b (Mathworks, Natick, MA, USA).

Table 1 shows geometrical data relevant to the model of the solar concentrator and solar receiver. The solar concentrator was modelled using the previous model implemented in Comsol Multiphysics 6.0 (Stockholm, Sweden) and presented by Marra et al. [29].

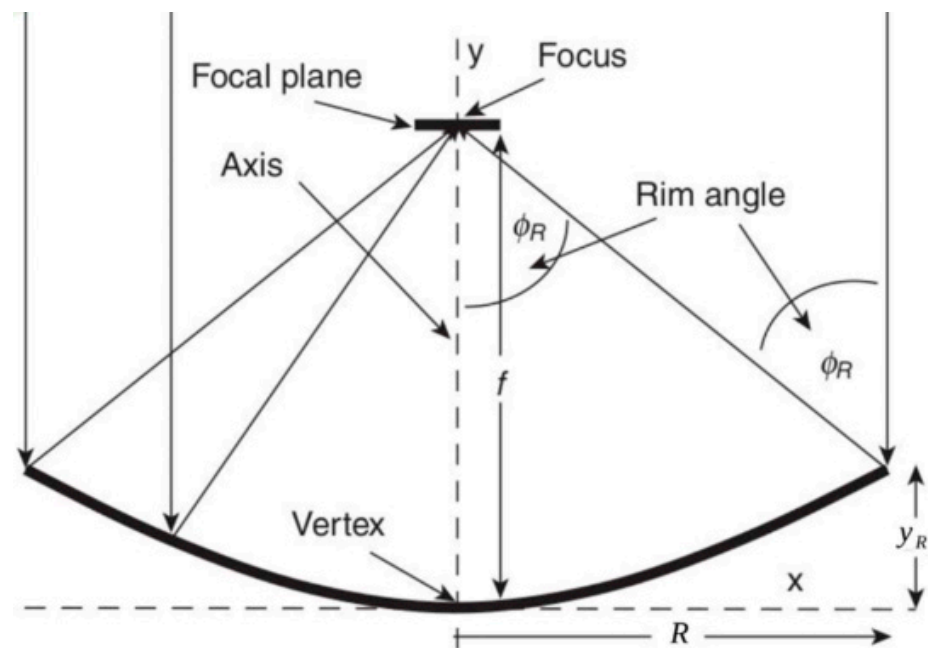


Figure 3. Main geometrical parameters of the solar concentrator.

Table 1. Geometrical data for the solar concentrator.

Name	Expression	Value	Description
$f$	$f = \frac{D}{4 \cdot \tan\left(\frac{\Phi_{rim}}{2}\right)}$	0.92 m	Focal length
$\Phi_{rim}$	$45^\circ$	0.7854 rad	Rim Angle
$D$	-	2.37 m	Diameter of the concentrator
$A_C$	$\pi \cdot d^2 / 4$	2.54 m <sup>2</sup>	Capturing Area of the concentrator
$d$	-	$18.06 \times 10^{-3}$ m	Receiver diameter (outer)
$L$	-	0.2 m	Receiver length
$CR_0$	-	8013	Optical concentration ratio

The metal alloy properties used and implemented in the present study are described in Table 2.

Table 2. Properties of the metal alloys used in this study [30–34].

	Inconel 740H	Alloy 625	Alloy 800H	Haynes 230
<b>T melting</b>	1288–1362 °C	1290–1350 °C	1357–1385 °C	1301–1371 °C
<b>Elastic modulus (E)</b>	186 GPa (@T = 600 °C) 178 GPa (@T = 700 °C) 169 GPa (@T = 800 °C)	170 GPa (@T = 650 °C) 160 GPa (@T = 760 °C) 148 GPa (@T = 870 °C)	157.7 GPa (@T = 600 °C) 150.1 GPa (@T = 700 °C) 141.3 GPa (@T = 800 °C)	175 GPa (@T = 600 °C) 168 GPa (@T = 700 °C) 159 GPa (@T = 800 °C)
<b>Yield strength (<math>\sigma_{serv}</math>)</b>	742 MPa (@Tamb) 608 Mpa (@T = 700 °C) 547 MPa (@T = 800 °C)	414–517 MPa (annealed, @Tamb) 357.2 MPa (@T = 800 °C)	150 MPa (@Tamb) 109 MPa (@T = 700 °C) 90 MPa (@T = 760 °C)	415 MPa (@Tamb) 265 MPa (@871 °C) 294 Mpa (@T = 1000 °C)
<b>Coefficient of linear expansion (<math>\alpha_L</math>)</b>	$15.7 \frac{\mu\text{m}}{\text{m}^\circ\text{C}}$	$15.5 \frac{\mu\text{m}}{\text{m}^\circ\text{C}}$	$18 \frac{\mu\text{m}}{\text{m}^\circ\text{C}}$	$15.3 \frac{\mu\text{m}}{\text{m}^\circ\text{C}}$
<b>Density (<math>\rho</math>)</b>	8050 kg/m <sup>3</sup>	8422 kg/m <sup>3</sup>	7940 kg/m <sup>3</sup>	8968 kg/m <sup>3</sup>
<b>Specific heat (c)</b>	573 J/kgK	600 J/kgK	460 J/kgK	465 J/kgK
<b>Thermal conductivity (k)</b>	22.1 W/mK	15.7 W/mK	11.5 W/mK	16.4 W/mK

The main simplifying hypotheses were:

- Gas flowing inside the receiver has been approximated to air, with minimum airflow fixed at  $\dot{m} = 1.66 \times 10^{-5}$  m<sup>3</sup>/s (calm air condition);
- Heat transport in the fluid (air) due to only advection, while thermal diffusion phenomena in this region are considered negligible;

- Fully developed airflow;
- Spatially and temporally constant properties assessed at a temperature of 800 °C;
- One-way heat transfer along the axial direction;
- Average radiation equal to  $I_0 = 800 \text{ W/m}^2$ .

The useful available energy of the concentrator obtained through energy balance can be calculated considering the irradiation from the sun:

$$q = A_a \cdot I_r \cdot \eta_0 - U_c \cdot (T_m - T_a) \cdot A_r \quad (1)$$

where the first term is associated with the optical losses ( $\eta_0$  optical performance), and the second term is associated with the thermal losses. The overall performance of the concentrator thus results in Equation (2):

$$\eta_C = \frac{q}{I_r \cdot A_a} = \eta_0 - \frac{U_c \cdot (T_m - T_a)}{I_r} \cdot \frac{1}{C_r} \quad (2)$$

The fluid flow regime was evaluated using the Reynolds number:

$$\text{Re} = \frac{\rho * u * D}{\mu} \text{ to evaluate the Laminar or Turbulent Flow regime} \quad (3)$$

This is less than 100, confirming that it is a laminar regime problem. Consequently, given the calm air, we can set Nusselt's number equal to 3.66.

$$\text{Nusselt number constant to 3.66} - h = \frac{Nu \cdot k}{D} \quad (4)$$

From the thermofluidodynamic point of view, the problem can be described by the following relations, see Equations (5) and (6). This was a case of coupled conduction and advection since the material composing the receiver receives heat from the solar radiation, which was assumed to be constant throughout the year and equal to an average value, and gives it up to the air, which, entering the receiver at a lower temperature, also cools the material by licking the inner walls. The goal was to analyse how the problem behaves until a steady-state is reached in which the only thermal gradients are those induced by the geometry, i.e., spatial gradients. They were discretised using an explicit Euler time derivative discretization method and spatially discretised using the centred finite difference method for the conduction and an upwind scheme for the advective term.

The receiver was described using the following equation:

Receiver

$$\rho_m c p_m \frac{\partial T_m}{\partial t} = k_m \cdot \frac{\partial^2 T_m}{\partial x^2} + q - \frac{h \cdot A_r \cdot (T_m - T_a)}{V_m} \quad (5)$$

Internal of the receiver

$$\rho_a c p_a \left( \frac{\partial T_a}{\partial t} + u \frac{\partial T_a}{\partial x} \right) = \frac{h \cdot A_{ri} \cdot (T_m - T_a)}{V_a} \quad (6)$$

The initial conditions are fixed using the Dirichlet boundary condition:

$$T_m(x, t = 0) = 298.5 \text{ K} \quad (7)$$

$$T_a(x, t = 0) = 298.5 \text{ K}$$

While the material boundary conditions (Neumann condition) are fixed as adiabatic extremities

$$-k \frac{\partial T_m}{\partial t} = 0 \quad (8)$$

The spatial course of the temperature at the end of the transient and the temporal course of the most stressed section were the model output.

The thermal stress induced by the temperature gradient was evaluated at the end of the transient (most extreme working condition) using the following formula (Equation (9)).

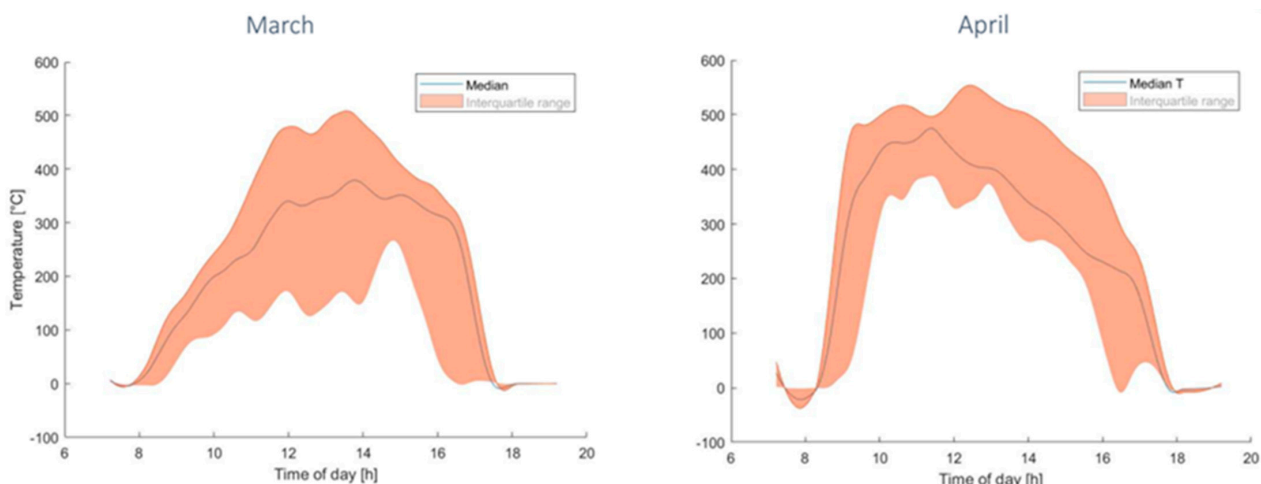
$$\sigma_{th} = E \cdot \alpha_L \cdot \Delta T \quad (9)$$

If this stress ( $\sigma_{th}$ ) exceeds the yield stress ( $\sigma_{serv}$ ), i.e., the condition in which the elastic bonds of the material change, the material deformation and material failure occur (fatigue behaviour is omitted).

### 3. Results and Discussion

#### 3.1. Preliminary Temperature Receiver Recording

An experimental campaign was conducted in March and April 2023. The median value, blue line and interquartile range are shown in Figure 4. The latter describes how far the values deviated from a central value. This fairly high variability is a function of non-fixed solar radiation. Solar radiation in March was more variable than in April, and consequently, so was the receiver temperature.



**Figure 4.** Example of the temperature recorded in the central receiver point (T0) in March and April 2023.

The measured curve is normally a Gaussian curve with the maximum values recorded during the hottest hours of the day, corresponding with the highest value of solar radiation, as shown in previous research [22,35]. In March, the variability was greater, close to 500 °C with a low just above 100 °C in the hottest hours of the day. In April, the minimum value at similar hours was just above 300 °C, while the maximum recorded rises to around 550–600 °C.

As previously reported in our research, the registered temperature increased in the spring and summer seasons [29]. The maximum recorded temperature reached as high as 1100 to 1200 °C. The variability of the data shown in Figure 4 showcases the important value of the thermal gradient. These two aspects affect the receiver structure and its deformation up to the point of structural failure.

#### 3.2. Thermal Stress Induced by the Temperature on the Receiver

Many thermodynamic problems can be solved simply if it is assumed that the internal temperature difference of the body is negligible compared with that between the body and the outside. This assumption, which translates numerically into a  $Bi < 0.1$ , allows the temperature distribution in the solid to be approximated as uniform. This condition is

called the concentrated parameter assumption. For all materials used for the receiver, the Biot number was lower than 0.1, see Table 3.

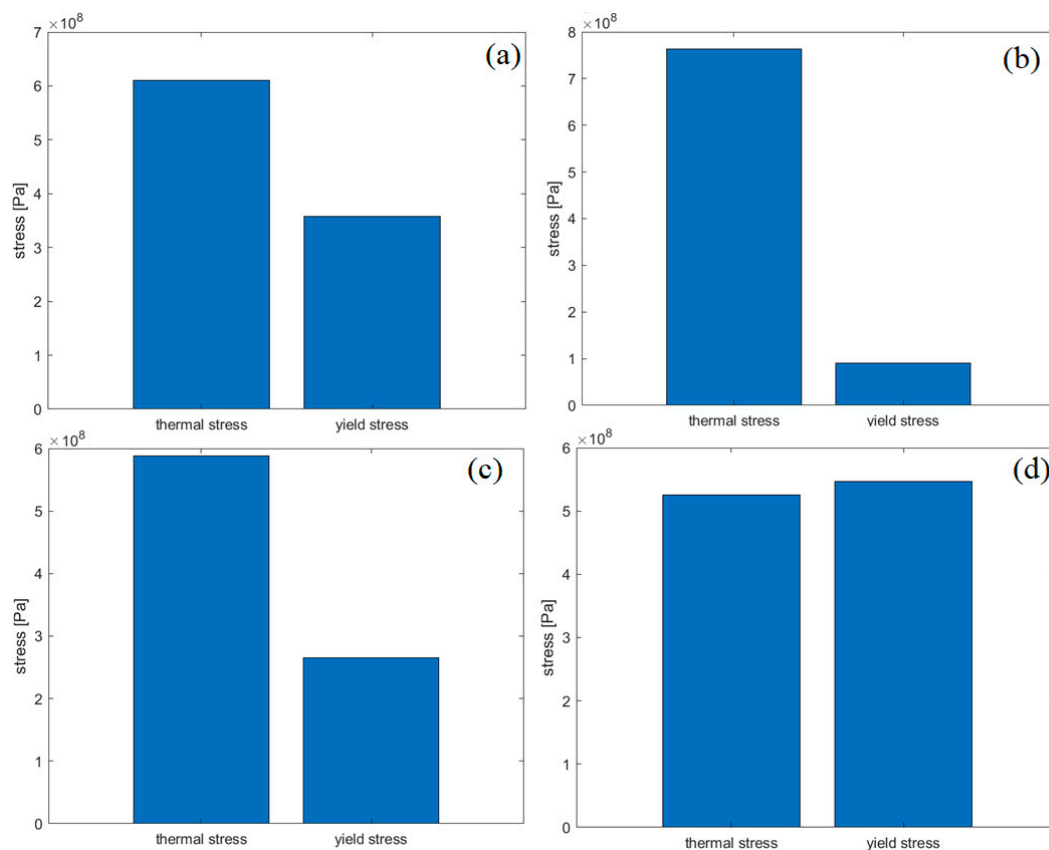
**Table 3.** Biot number evaluated for all the materials.

Material	Biot
Alloy 625	0.0019
Alloy 800H	0.0026
Haynes 230	0.0018
Inconel 740H	0.0013

A similar approach was followed by other researchers to simplify the calculation [36,37].

Using Equation (9), the thermal stress induced by the temperature level variation in the receiver was evaluated. Thermal stress is a function of the material related to the coefficient of linear expansion.

Figure 5a compares the induced efforts in alloy 625. The thermal stress reached ca.  $6 \times 10^8$  Pa, while the yield stress was close to  $4 \times 10^8$  Pa. Figure 6b confirms this result, showing the collapse of the structure. The maximum temperature imposed, 800 °C, was excessively high, as shown by the experimental studies of Suave et al. [38]. They found that the alloy 625, without any prior ageing, is unstable when increasing the temperature above 700–750 °C. Thermal ageing at 650 °C for 500 h induced a stress of around 1000 MPa with a strain of 52%.



**Figure 5.** Evaluation of the induced effort for the metallic alloys—thermal stress versus yield stress: (a) alloy 625, (b) alloy 800H, (c) Haynes 230, (d) Inconel 740H.



**Figure 6.** Experimental activity for the metal alloys used (a) Inconel 740H; (b) alloy 625; (c) Haynes 230.

Figure 5b compares the induced efforts in the alloy 800H. The thermal stress reached ca.  $7.7 \times 10^8$  Pa, while the yield stress was close to  $1 \times 10^8$  Pa. The maximum temperature imposed, 800 °C, was excessively high, as shown by the experimental studies of Cao et al. [39]. Cao et al. studied the hot deformation behaviour of alloy 800H in the temperature range of 825–975 °C [40]. The microstructure was shown to be deformed at 875 °C with elongated parent grains. When the deformation temperature was increased to 975 °C, the degree of recrystallization was higher and the deformed grains affected the structure of the material.

Figure 5c compares the induced efforts in Haynes 230. The thermal stress reached about  $6 \times 10^8$  Pa, while the yield stress was close to  $3 \times 10^8$  Pa. Figure 6c confirms this result and shows the collapse of the structure. The maximum temperature imposed, 800 °C, was excessively high, as shown by the experimental studies of Pataky et al. [41]. They studied the creep deformation and mechanisms in Haynes 230 at 800 and 900 °C. Intergranular failure was observed in all samples, and secondary and tertiary creep were investigated. To improve the operating lifetime, the grain boundary serrations were used to restrict grain boundary sliding working at high temperatures. Unfortunately, this process is quite expensive.

Figure 5d compares the induced efforts in Inconel 740H. Thermal stress was slightly below the yield stress. Figure 6a confirms this result, showing the non-collapse of the structure. This result can be compared to previous studies, such as Kim et al. and de Barbadillo [42,43]. Kim et al. illustrated the deformation behaviour of carbides during the creep investigation at 750 °C for 5000 h. They found that Inconel 740 K was able to maintain the mechanical structure, avoiding the collapse, even when the  $\gamma$  particles increased gradually.

It is superfluous to test the receiver with alloy 800H since the thermal stress induced is much greater than the yield strength compared to the situation achieved with the “similar” alloy 625. For that alloy, which is better than alloy 800H, we reached the failure experimentally, as shown in Figure 6b. Figure 6a shows, on the other hand, the non-breakage of



alloy Inconel 740H, the ability of this alloy to resist extreme temperature conditions and a stronger thermal gradient than the experimentally verified.

#### 4. Conclusions

Solar concentration can be exploited in depth in the near future and the material selection for the receiver element is crucial. High temperatures and high thermal gradients are important issues to be analysed. Different metal alloys were used to verify that the yield strength limit was not exceeded due to thermal stress induced by concentrated solar radiation. The conclusions were as follows:

- Alloy 625 was considered, and the implemented model showed the collapse of the structure. This result was experimentally verified, as reported in Figure 6b.
- Alloy 800H was considered, and the implemented model showed the collapse of the structure. The thermal stress was higher than the yield stress.
- Haynes 230 was considered, and the implemented model showed the collapse of the structure. This result was experimentally verified, as reported in Figure 6c.
- Inconel 740H was considered, and the implemented model showed the non-collapse of the structure. This result was experimentally verified, as reported in Figure 6a. This could be due to its microstructure, even in high temperatures at a long duration, but in stable temperature conditions, as shown by Zielinski et al. [44].

Future works will focus on composite materials, considering metal alloys and ceramic materials used as receiver coatings.

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#### Nomenclature

$\sigma_{snerv}$	yield stress (Pa)
$\sigma_{th}$	thermally induced stress (Pa)
$\alpha_L$	coefficient of linear expansion (m/mK)
$A_a$	concentrator area (m <sup>2</sup> )
$A_r$	receiver area (m <sup>2</sup> )
$A_{ri}$	internal receiver area (m <sup>2</sup> )
$c_p$	specific heat at constant pressure (J/kg K)
$CR_0$	optical concentration ratio
$d$	receiver diameter (m)
$D$	solar concentrator diameter (m)
$E$	elastic modulus of material (Pa)
$f$	focal length (m)
$I_0$	constant solar radiation (W/m <sup>2</sup> )
$I_r$	receiver flow from the solar concentration factor (W/m <sup>2</sup> )
$k$	thermal conductivity (W/mK)
$L$	receiver length (m)
$m$	subscript for the material side
$\dot{m}$	heat transfer fluid mass flow rate (kg/s)
$T_a$	temperature for the air fluid (°C)
$T_{in}$	heat transfer fluid inlet temperature (°C)
$T_m$	material temperature of the receiver (°C)
$T_{out}$	heat transfer fluid outlet temperature (°C)
$U_c$	receiver heat transfer coefficient (W/m <sup>2</sup> K)

$V_m$	receiver volume ( $m^3$ )
$V_a$	internal receiver volume ( $m^3$ )
$\alpha$	solar altitude (rad)
$\gamma$	intercept factor
$\mu$	dynamic viscosity (Pa s)
$\rho$	density ( $kg/m^3$ )
$\tau$	receiver coverage transmittance
$\phi_{rim}$	angle between reflected radiation and vertex-focus junction
$\psi_1$	slope error
$\psi_2$	solar radius error
$\psi_3$	error in solar tracking
$\alpha_r$	absorbance receiver
$\eta_0$	optimal concentrator performance
$\eta_{receiver}$	receiver performance
$\tau_b$	direct optical length
$\tau_d$	diffused optical length

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