

Article **On the Statistical Size Effect of Cast Aluminium**

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Abstract: Manufacturing process based imperfections can reduce the theoretical fatigue strength since they can be considered as pre-existent microcracks. The statistical distribution of fatigue fracture initiating defect sizes also varies with the highly-stressed volume, since the probability of a larger highly-stressed volume to inherit a potentially critical defect is elevated. This fact is widely known by the scientific community as the statistical size effect. The assessment of this effect within this paper is based on the statistical distribution of defect sizes in a reference volume V_0 compared to an arbitrary enlarged volume *Vα*. By implementation of the crack resistance curve in the Kitagawa–Takahashi diagram, a fatigue assessment model, based on the volume-dependent probability of occurrence of inhomogeneities, is set up, leading to a multidimensional fatigue assessment map. It is shown that state-of-the-art methodologies for the evaluation of the statistical size effect can lead to noticeable over-sizing in fatigue design of approximately 10%. On the other hand, the presented approach, which links the statistically based distribution of defect sizes in an arbitrary highly-stressed volume to a crack-resistant dependent Kitagawa–Takahashi diagram leads to a more accurate fatigue design with a maximal conservative deviation of 5% to the experimental validation data. Therefore, the introduced fatigue assessment map improves fatigue design considering the statistical size effect of lightweight aluminium cast alloys.

Keywords: cast aluminium; fatigue assessment; shrinkage pores; statistical distribution; extreme value statistics; highly-stressed volume

1. Introduction

In order to ensure conservative fatigue design of heterogeneous material, size effects have to be assessed properly, since they significantly influence the fatigue strength of engineering components. A study in [\[1\]](#page-15-0) separates the size effects into statistical [\[2–](#page-15-1)[4\]](#page-15-2), geometrical [\[1](#page-15-0)[,5](#page-15-3)[,6\]](#page-15-4), technological [\[7](#page-15-5)[,8\]](#page-15-6) and surface technology [\[9\]](#page-15-7) contributions. The statistical size effect, whose improved assessment is the aim of this work, leads to a decrease of fatigue strength with elevated size of the structure or specimen due to increased probabilities of critical defect sizes. Griffith's investigations in [\[10\]](#page-15-8) revealed such an effect, due to the presence of microcracks. One of the most used statistical models is the weakest link concept of Weibull [\[11\]](#page-15-9). Accordingly, the probability of a structure or specimen to inherit a critical defect size elevates in line with the volume of the component—see [\[12\]](#page-15-10). This concept was later confirmed to be valid for the high cycle fatigue fracture by Heckel in [\[13\]](#page-15-11). A more recent study [\[14\]](#page-15-12) revealed similar effects in the very high cycle fatigue (VHCF) region, pointing out the need for using large specimens in VHCF testing, in order to consider critical defect sizes. Another method to characterize the statistical size effect was proposed by Carpinteri in [\[15–](#page-15-13)[17\]](#page-15-14). Hereby, a fractal geometry approach is utilized for the characterization of the fracture surfaces, assuming them to possess non-integer dimensions. By modelling the self-similar fracture surfaces, mechanical properties depending on the fractal dimension can be deduced—see [\[15](#page-15-13)[–17\]](#page-15-14). Hence, the Basquin equation [\[18\]](#page-16-0) is modified invoking the fractal approach. Another methodology to assess the statistical size effect is given in [\[19\]](#page-16-1). Hereby, the highly-stressed volume, defined as the volume which is stressed by over 95% of the peak stress value, is considered in an exponential approach—see Equation [\(1\)](#page-1-0):

$$
\sigma_{\text{LLF}} = B \cdot V_{95}^{-a},\tag{1}
$$

with *B* and *a* being material dependent coefficients and V_{95} the 95% highly-stressed volume, with respect to the maximum stress. In order to evaluate the highly-stressed volume, a numerical procedure to evaluate stress integrals is provided in [\[20\]](#page-16-2). Sonsino [\[21\]](#page-16-3) proposed a similar volumetric model, invoking the 90% highly-stressed volume, *V*⁹⁰ and the Weibull exponent *κ*—see Equation [\(2\)](#page-1-1):

$$
\frac{\sigma_{\text{LLF},1}}{\sigma_{\text{LLF},2}} = \left(\frac{V_{90,2}}{V_{90,1}}\right)^{\frac{1}{\kappa}}.
$$
\n(2)

The Weibull exponent is a material dependent coefficient, which characterizes the slope in the double logarithmic $\sigma_{a,V_i} - V_i$ plot and therefore the decrease in fatigue strength at an increased highly-stressed volume. Its value is listed for various material classes in a common fatigue design guideline—see [\[22\]](#page-16-4). In case of cast aluminium, *κ* is proposed to possess a value of ten [\[1](#page-15-0)[,22\]](#page-16-4). It has to be mentioned that, with increasing highly-stressed volume, a saturation effect was observed, such that no noticeable further decrease in fatigue strength can be observed at volumes $V \geq V_{\kappa\infty}$. The threshold volume $V_{\kappa\infty}$ is proposed to be 8000 mm³ for cast material in [\[23\]](#page-16-5). Investigations in [\[24\]](#page-16-6) modified this approach, invoking the highly-stressed surface instead of an highly-stressed volume, since surface near defects are revealed to be more critical in terms of fatigue life, as also stated in [\[25](#page-16-7)[–28\]](#page-16-8).

The validation of the different presented models for the size effect, respectively the corresponding fatigue strength assessment, reveals that the weakest link model and the volumetric model match the experimental fatigue data best—see [\[29\]](#page-16-9). Furthermore, it is stated within the same study that both the weakest link model as well as the volumetric model lead to similar results. Preliminary studies in [\[30,](#page-16-10)[31\]](#page-16-11), which investigate the statistical size effect of an aluminium alloy containing artificial surface defects, reveal the volume approach to be more suitable than the surface approach. Thus, the investigation of the statistical size effect in this study is conducted by means of the volumetric model proposed by Sonsino as given in [\[23\]](#page-16-5). Kitagawa and Takahashi proposed in [\[32\]](#page-16-12) a fatigue assessment method for material inheriting small cracks respectively heterogeneities. Due to its sound applicability, the methodology proposed by Kitagawa and Takahashi has been utilized for the fatigue strength assessment of imperfect materials in several preliminary studies [\[33](#page-16-13)[–37\]](#page-16-14).

Hereby, the long-life fatigue strength can be assessed invoking two major boundaries. On the one hand, there is the fatigue strength of the defect free material $\Delta \sigma_0$ with crack sizes tending towards zero and on the other hand, when defect respectively crack sizes *a* become sufficiently large, there is the long crack threshold ∆*Kth*,*lc*. The long life fatigue strength *σ^a* of a material inheriting flaws can therefore be estimated fracture mechanically assessed by Equation [\(3\)](#page-1-2):

$$
\Delta \sigma_{\text{LLF}} = \frac{\Delta K_{th,lc}}{Y \sqrt{\pi a}},\tag{3}
$$

with *Y* being the geometry factor, taking the defect size, shape and location into account—see [\[38\]](#page-16-15). A qualitative study on the influence of the defect orientation and location on the stress concentration factor is given in [\[39\]](#page-16-16). The original Kitagawa–Takahashi diagram KTD was later modified by El Haddad in [\[40,](#page-16-17)[41\]](#page-16-18), such that the fatigue strength steadily decreases with elevating defect sizes *a*.

A further modification of the KTD is given by the implementation of the crack resistance curve (R-Curve), as introduced by Chapetti in [\[42\]](#page-17-0). The cyclic R-curve characterizes the resistance of the material to crack initiation, respectively propagation. Therefore, the crack resistance rises from the intrinsic threshold ∆*Kth*,*e f f* to the long crack threshold ∆*Kth*,*lc* with elevating crack length ∆*a* due to crack closure effects [\[43\]](#page-17-1). Crack closure effects lead to premature physical contact of the crack shoulders, such that the original load range ∆*K* = *Kmax* − *Kmin* is reduced to an effective one ΔK_{eff} = K_{max} − K_{op} —see Figure [1.](#page-2-0)

Figure 1. Effective load stress intensity factor due to crack closure effects.

The most distinguished crack closure effects are the roughness- [\[44,](#page-17-2)[45\]](#page-17-3), plasticity- [\[46](#page-17-4)[,47\]](#page-17-5), and oxide-induced [\[48,](#page-17-6)[49\]](#page-17-7) effects. Hence, the composition of aforementioned contributions leads to a crack resistance curve. The R-curve can be characterized by an exponential law as proposed in [\[50\]](#page-17-8). A more elaborated procedure to describe the course of the R-curve was proposed by Maierhofer et al. in [\[51\]](#page-17-9). This methodology takes the contributions v_i of the very crack closure effects into account. Each effect with index *i* is assumed to be completely built up after reaching the crack extension length l_i —see Equation [\(4\)](#page-2-1):

$$
\Delta K_{th,\Delta a} = \Delta K_{th,eff} + \left(\Delta K_{th,lc} - \Delta K_{th,eff}\right) \left[1 - \sum_{i=0}^{n} v_i \cdot \exp\left(-\frac{\Delta a}{l_i}\right)\right],\tag{4}
$$

with

$$
\sum_{i=0}^{n} \nu_i \equiv 1.
$$

Since the contributions of the very crack closure effects cannot be defined a priori due to influences of the specimen geometry [\[52\]](#page-17-10), this model represents a potent tool to qualitatively assess the elevating crack resistance with increasing crack length. Thus, the cyclic R-curve can be implemented in the KTD by inserting Equation [\(4\)](#page-2-1) in the Kitagawa–Takahashi relationship—see Equation [\(3\)](#page-1-2). A preliminary study in [\[53\]](#page-17-11) reveals this approach to be suitable to assess the mean stress sensitivity of both imperfect and near defect free material. Hereby, a three-dimensional KTD is set up by implementing a mean stress sensitivity approach, provided in a common guideline [\[22\]](#page-16-4) and a generalized R-curve considering varying load stress ratios, as proposed in [\[54\]](#page-17-12).

Hence, the extension of the Kitagawa–Takahashi diagram by the crack resistance R-curve is suitable to assess both physically short and long cracks in terms of fatigue strength, where the crack length *^a* is substituted by an equivalent flaw size [√] *area* if sharp defects such as micro shrinkage pores are considered. The [√] *area* parameter was introduced by Murakami in [\[55\]](#page-17-13) and is given by the square root of the projected area of the fatigue fracture initiating heterogeneity, measured perpendicular to the maximum principal load direction. Since a preliminary study in [\[56\]](#page-17-14) revealed that the [√] *area* parameter correlates well with the stress field around the inhomogeneity, this parameter is further utilized in this work for the estimation of defect sizes. The defect size measurement methodology invoked

in this work is in detail described in [\[57](#page-17-15)[–59\]](#page-17-16). According to investigations in [\[60,](#page-17-17)[61\]](#page-17-18), the statistical distribution of defect sizes can be characterized by the limiting extreme value distribution of the Lognormal (LN) distribution, namely the Gumbel or Extreme Value type one distribution (EV) [\[62](#page-17-19)[,63\]](#page-17-20). Since the Generalized Extreme Value (GEV) distribution [\[64\]](#page-17-21) inherits the Extreme Value distributions of type one, two and three (Gumbel, Frechet and Weibull), the GEV distribution is also applicable for the characterization of fatigue fracture initiating defect sizes—see [\[65\]](#page-17-22). The cumulative distribution function of the GEV is given in Equation [\(5\)](#page-3-0):

$$
P(\sqrt{area}) = \exp\left\{-\left[1 + \zeta\left(\frac{\sqrt{area} - \mu}{\delta}\right)\right]^{-\frac{1}{\zeta}}\right\} \tag{5}
$$

and

$$
P_{Occ} = 1 - P(\sqrt{area}).\tag{6}
$$

Its course is defined by the location μ , scale δ and shape ζ parameter which are evaluated by means of a maximum likelihood method, as proposed in [\[66](#page-17-23)[,67\]](#page-17-24). Another methodology for calculating the maximum inclusion, inherited in enlarged geometries, is given in [\[68](#page-18-0)[,69\]](#page-18-1). By evaluating the most extremal defect in a given reference volume *V*0, the obtained defect sizes are to be Gumbel distributed, with the cumulative distribution function $P(\sqrt{area})$ of Equation [\(7\)](#page-3-1), characterized by its location- μ and scale parameter *δ*:

$$
P(\sqrt{area}) = exp\left\{-exp\left[-\frac{\sqrt{area} - \mu}{\delta}\right]\right\}.
$$
 (7)

This methodology, also referred to as extreme value inclusion rating *EVIR*, enables the prediction of the size of a potentially critical inclusion with a spatial extent [√] *area*(*T*) in an enlarged control Volume *Vα*, which is defined by the return period *T*—see Equation [\(8\)](#page-3-2):

$$
\sqrt{\text{area}}(T) = \mu - \delta \cdot \ln\left[-\ln\left(1 - \frac{1}{T}\right)\right],\tag{8}
$$

with

$$
T = \frac{V_{\alpha}}{V_0}.\tag{9}
$$

Hereby, the very fatigue fracture initiating defect sizes of every specimen are sorted and indexed with $j = 1, \ldots, n$. Subsequently, the cumulative probability F_j can be derived for each sample *j* invoking Equation [\(10\)](#page-3-3). Accordingly, a reduced variate y_j can be estimated, as depicted in Equation [\(11\)](#page-3-4):

$$
F_j = \frac{j}{n+1} \tag{10}
$$

and

$$
y_j = -\ln\left[-\ln\left(F_j\right)\right].\tag{11}
$$

Furthermore, Murakami [\[68\]](#page-18-0) proposes an empirical formula for the defect based fatigue strength of various material types—see Equation [\(12\)](#page-4-0). Since the fatigue strength of metallic materials is revealed to correlate with the Vickers hardness HV, Murakami invokes this material constant, in line with two fit parameters, C_1 and C_2 . Hereby, C_1 acts as a geometry factor, depending on the location of the defect, being 1.43 for a surface-near and 1.56 in case of a subsurface defect. The coefficient C_2 is regularly a priori defined to possess a value of 120. Nevertheless, preliminary studies [\[57](#page-17-15)[,70\]](#page-18-2) propose a factor *C*² of zero for similar materials, induced by the deviation in defect size measurement methodologies:

$$
\sigma_{\text{LLF}} = C_1 \frac{C_2 + \text{HV}}{\sqrt{\text{area}^{\frac{1}{6}}}}.
$$
\n(12)

Although the statistical size effect due to natural defects is a well researched field, the probabilistic fatigue assessment method presented in this work is a potent tool for conservative fatigue design. Hence, the focus of this work lies on the evaluation of the statistical distribution of fatigue fracture initiating defect sizes in an arbitrary highly-stressed volume and its impact on the fatigue strength. Finally, the presented fatigue assessment map facilitates fatigue design of parts inheriting manufacturing process based imperfections with improved accuracy in terms of probabilistic assessment.

2. Investigated Material

The specimens are taken out from two positions, *A* and *B* from an engineering component made of EN AC-46200 in T6 heat treatment. The nominal chemical composition in line with the standard given in [\[71\]](#page-18-3) of this aluminium alloy is listed in Table [1.](#page-4-1) In order to build up a KTD, the fatigue strength of the defect free material has to be evaluated. Therefore, additional specimens are taken out from the same positions with additional preliminary hot isostatic pressing (HIP) treatment.

Table 1. Nominal chemical composition of the investigated cast alloy EN AC-46200 in weight percent.

| Alloy | | Si $\binom{9}{0}$ Cu $\binom{9}{0}$ Fe $\binom{9}{0}$ Mn $\binom{9}{0}$ Mg $\binom{9}{0}$ Ti $\binom{9}{0}$ Al $\binom{1}{1}$ | | |
|------------------------------------------------------------------|--|-------------------------------------------------------------------------------------------------------------------------------|--|--|
| EN AC-46200 7.5–8.5 2.0–3.5 0.8 0.15–0.65 0.05–0.55 0.25 balance | | | | |

The microstructural properties in terms of the dendrite arm spacing (DAS) are evaluated by a procedure proposed in [\[72\]](#page-18-4), since the DAS correlates well with both quasi-static [\[73–](#page-18-5)[75\]](#page-18-6) and fatigue properties [\[76,](#page-18-7)[77\]](#page-18-8) due to its dependency with inter-dendritic shrinkage pore sizes [\[57\]](#page-17-15). The investigations of the DAS reveal a minor deviation of just five percent between sampling position A and B, such that technological size effects are negligible at these two positions. Thus, differently sized specimens can be taken from these two positions to evaluate the statistical size effects. Furthermore, the specimens have been designed by numerical shape optimization of the fillet region, such that a minor stress concentration factor of just 1.02 is not exceeded. This enables a variation of the highly-stressed volume by scaling the length *lconst* of the constant specimen diameter, where the general shape of the specimen is kept similar. The high cycle fatigue (HCF) specimen geometry type A and B are displayed in Figures [2](#page-5-0) and [3,](#page-5-1) respectively. As depicted in Figure [3,](#page-5-1) the length of the constant testing diameter *l*_{const} of specimen B is significantly enhanced by a factor of 18 with respect to specimen geometry A, to force a significant statistical size effect. Subsequently, the two specimen types are only labelled in relation to their different geometry as A, small-sized reference sample, and B, enhanced highly-stressed volume size.

The highly-stressed volume is furthermore numerically evaluated. In line with the subsequent testing procedure, the linear-elastic finite element analysis is conducted with an unified uni-axial tension load. Hereby, the numerical model is set up invoking eight node axisymmetric CAX8R elements. In Figure [4,](#page-5-2) the results of the numerical analysis are depicted, showing the normalized maximum principle stress.

The evaluation of the 95% highly-stressed volumes hence leads to a ratio between geometry B and A of $V_{95,B}/V_{95,A} \approx 10$, as described in [\[57\]](#page-17-15). Hence, the highly-stressed volume of position A is *V*95,*^A* considered as reference volume *V*0, where the increased volume of position B is in the following referred to as highly-stressed volume *V*1.

Figure 2. HCF specimen geometry A.

Figure 3. HCF specimen geometry B.

Figure 4. FEM analysis of specimen types A and B.

3. Results

3.1. Fatigue Strength

The fatigue strength of both specimens is evaluated at the same resonant testing machine, utilizing a testing frequency of just beneath 120 Hz and fully reversed/tension compression loading. The run out number was preliminarily defined to be ten million cycles in order to ensure a statistically assessed long life fatigue strength. As proposed in prior studies [\[78\]](#page-18-9) and applied in [\[4,](#page-15-2)[70\]](#page-18-2), the slope of the S/N-curve in the long life region k_2 is considered to scale with the slope in the finite life k_1 , such that $k_2 = 5 \cdot k_1$ —see [\[78\]](#page-18-9). The evaluation of failures in the finite life region is conducted by the statistical procedure given in the standard [\[79\]](#page-18-10). The long life fatigue strength is assessed by means of the arcsin \sqrt{p} methodology—see [\[80\]](#page-18-11).

In order to evaluate the fatigue strength of the defect free material, specimens of geometry A are taken from an HIP treated component. Again, it was ensured by DAS-evaluation that no technical size effect occurred at this position. The fatigue data as well as the statistically estimated slopes in the finite and long life region as well as the scatter bands are depicted in Figure [5.](#page-6-0) The fatigue strength of the near defect free HIP material at ten million load cycles is considered as a reference value in the further work.

The statistically evaluated S/N-curve of position A, inheriting a highly-stressed volume V_0 , is depicted in Figure [6,](#page-6-1) along with the 90 and 10 percent scatter band. All fatigue data is normalized by

the fatigue strength of the near defect free material of position A. Thus, the long life fatigue strength σ_{LIF} decrease by about 20% in the presence of shrinkage pores. Since the highly-stressed volume *V*₁ of position B is majorly enhanced, its fatigue strength reveals a significant statistical size effect in terms of an additional decrease in fatigue strength by approximately fifteen percent—see Figure [7.](#page-7-0)

Figure 5. S/N curve of near defect free position A.

Figure 6. S/N curve of position A.

The evaluated slopes k_1 of the imperfective specimens in the finite life correlate well with a value of about 4.5. Thus, the derived slopes *k*² in the long life region match with the proposed value of 22 in the standard [\[81\]](#page-18-12). It has to be mentioned that the scatter band of the specimens of position B is revealed to be minor with respect to the specimens of position A, where the number of cycles N_T of the transition knee-point is somewhat enhanced in position B—see Table [2.](#page-7-1)

The long life fatigue strength *σ*LLF is additionally listed in Table [2,](#page-7-1) where all fatigue data are normalized by the fatigue strength of near defect free position A at ten million load cycles, considering a probability of survival *P^S* of 50%. Additionally, the slope of the finite life region *k*1, the number of load cycles for the transition point *NT*, and the statistically evaluated fatigue scatter band *T^S* is given in Table [2.](#page-7-1)

Figure 7. S/N curve of position B.

Table 2. Evaluated results of the fatigue tests.

| Position | HT | Volume k_1 [-] | | σ_{LLF} [-] | N_T [-] | T_S [-] |
|----------|------------|------------------|------|---------------------------|------------|-----------|
| А | HIP | V٥ | 7.10 | 1.00 | 438.627 | 1:1.27 |
| А | T6 | V_0 | 4.70 | 0.79 | 780.732 | 1:1.23 |
| B | T6 | V1 | 4.54 | 0.68 | 11,184.783 | 1:1.16 |

Thus, the statistical size effect can be assessed by means of the volumetric model, revealing a deviation of 8% with respect to the experimentally observed fatigue strength. Furthermore, a local experimental Weibull factor *κexp* can be derived, revealing a significantly enhanced value of 16.6, rather than the originally proposed value of 10. The volumetric model, along with the experimental fatigue test results at 50% probability of survival *PS*, is depicted in Figure [8.](#page-7-2) This significant deviation points out the need for more precise methodologies for the fatigue assessment of defective materials.

Figure 8. Volumetric model with the theoretical and experimentally deduced Weibull factor *κ*.

3.2. Fractography

In addition, the fatigue fracture initiating defect sizes of the HCF-specimens are investigated subsequent to the testing by means of digital optical microscopy respectively scanning electron microscopy.

In contrast to the method proposed by Murakami in [\[68\]](#page-18-0), where "effective" heterogeneity sizes are measured, the defect sizes are evaluated rather form fitting in this work. Thus, the investigation of the spatial extent of a flaw will lead to comparably smaller sizes. Nevertheless, this methodology is of utmost importance in order to minimize any distortion regarding the subsequent statistical analysis.

Figure [9](#page-8-0) displays the fatigue fracture surface of a representative specimen of position *A*, revealing an equivalent crack size [√] *area* of just over 85 µm. Accordingly, Figure [10](#page-8-1) depicts a fracture surface and the critical defect as the very point of crack initiation at geometry *B*. The evaluated critical defect size displayed in Figure [10](#page-8-1) is significantly enhanced with respect to average inhomogeneity sizes in position A, in line with the general presumption of the statistical size effect. Hence, the evaluated decrease in terms of fatigue strength of position B is caused by the elevated spatial extent of pre-existent microcracks respectively micro shrinkage pores.

Figure 9. Representative fracture initiating defect at specimen geometry A.

Figure 10. Representative fracture initiating defect at specimen geometry B.

In order to statistically evaluate the most critical defect sizes, the generalized extreme value distribution with the cumulative distribution function (CDF) given in Equation [\(5\)](#page-3-0) is invoked, as proposed in [\[65\]](#page-17-22). Additionally, a Kolmogorov–Smirnov test for the goodness of fit is utilized for the statistical assessment of the distributions—see [\[82\]](#page-18-13). Considering a reference volume V_0 , the course of the probability of occurrence *POcc* of a heterogeneity with a critical defect is characterized by the location, shape and scale parameters of the reference distribution *P*. Furthermore, the cumulative distribution function of a *α*-times enlarged volume V_α can be derived as the following [\[62\]](#page-17-19):

$$
V_0 \sim P(\sqrt{\text{area}}; \mu, \delta, \xi) = \exp\left\{-\left[1 + \xi \left(\frac{\sqrt{\text{area}} - \mu}{\delta}\right)\right]^{-\frac{1}{\xi}}\right\},\tag{13}
$$

$$
V_{\alpha} \sim P^{\alpha}, \qquad (14)
$$

which leads to

$$
P^{\alpha} = \exp\left\{-\left[1 + \xi \left(\frac{\sqrt{\text{area}} - (\mu + \frac{\delta}{\xi}(\alpha^{\xi} - 1))}{\delta \alpha^{\xi}}\right)\right]^{-\frac{1}{\xi}}\right\},\tag{15}
$$

$$
V_{\alpha} \sim P\left(\sqrt{\text{area}}; \mu + \frac{\delta}{\xi}\left(\alpha^{\xi} - 1\right), \delta \alpha^{\xi}, \xi\right). \tag{16}
$$

Thus, the parameters for the critical defect size distribution in an α -times greater volume V_α can be easily derived, based on the distribution of the reference volume *V*₀—see Equation [\(13\)](#page-9-0). Figure [11](#page-9-1) depicts the probability of occurrence *POcc* of a flaw with a given size in both specimen geometries *A* and *B*. Furthermore, the parameters of each distribution are statistically estimated by means of a maximum likelihood function, as proposed in [\[66\]](#page-17-23).

Figure 11. Probability of occurrence of critical defect sizes in specimen geometry A and B.

Finally, the derived model for the generalized extreme value distribution in a volume *V^α* is displayed in Figure [11.](#page-9-1) As proposed by the numerical investigation of the highly-stressed volumes in each specimen geometry, the factor *α* is defined to possess a value of ten. As depicted, the critical defect sizes in geometry *B* have minor deviation with respect to the theoretical distribution of *Vα*, which is caused by variations of fatigue fracture initiating micropore sizes due to the invoked measurement methodology.

Nevertheless, the critical pore size with a probability of occurrence of 50% matches the experimental data, as listed in Table [3.](#page-10-0)

| Position Volume | | $\boldsymbol{\mathcal{u}}$ | δ | $\sqrt{areaP_{Occ=0.5}}$ (µm) |
|-----------------|--------------|----------------------------|-----------------|-------------------------------|
| А | V_0 | | 78.6 21.6 0.11 | 86 |
| B | V1 | | 134.6 38.6 0.01 | 148 |
| | V_{α} | | 135.0 27.7 0.11 | 145 |

Table 3. Statistically evaluated and derived distribution parameters for the GEV.

3.3. Fatigue Assessment

In order to assess the decrease in fatigue strength with elevated probabilities of critical defects, the *EVIR* methodology of Murakami [\[68\]](#page-18-0) is invoked, as utilized in various preliminary studies [\[83–](#page-18-14)[85\]](#page-18-15). Hereby, a reduced variate *y^j* is calculated and plotted in a Gumbel probability paper—see Figure [12.](#page-10-1) Afterwards, the defect in an enhanced volume is computed by invoking the return period T. Thus, the defect sizes in a given control volume and therefore the corresponding fatigue strength are subsequently estimated by the [√] *area* approach of Murakami [\[68](#page-18-0)[,69\]](#page-18-1)—see Table [4.](#page-12-0)

Figure 12. *EVIR* methodology as proposed in [\[68\]](#page-18-0).

Another possibility to assess the fatigue strength in the presence of microcracks or defects is the Kitagawa–Takahashi diagram. Hence, crack propagation tests were conducted to obtain the materials fracture mechanical threshold value. The single edge notched specimens are taken from identical positions as the HCF specimens in order to minimize microstructural deviations. The testing procedure and extensive specimen preparation, as well as the crack propagation results, are given in detail in a preliminary study—see [\[37\]](#page-16-14). As proposed by the author in [\[37\]](#page-16-14), the crack resistance curve is implemented in the model, such that the KTD is valid both for physically short as well as long cracks. Accordingly, the fatigue strength of specimens with a subsequent HIP treatment characterizes the near defect free material, which defines the left-hand side limit of the KTD, where defect size tends to be zero. Figure [13](#page-11-0) displays the schematic set up of the fatigue assessment model. The starting point is given by the defect size in a reference volume V_0 represented by the size of an inhomogeneity μ_0 with a probability of occurrence of $P_{Occ} = 50\%$. Subsequently, the fatigue strength is estimated by the R-curve extension of the KTD [\[42\]](#page-17-0), with respect to the reference volume *V*0. Afterwards, the critical defect size with a similar *POcc* in an arbitrary enhanced volume *V^α* is evaluated, invoking Equation [\(13\)](#page-9-0). The corresponding fatigue strength ∆*σ*LLF,*V^α* is computed in line with the procedure for the fatigue strength of the reference volume. Hence, a local factor *κ* depending on the return period *α* of the

highly-stressed volume and the reference defect distribution represented by μ_0 can be derived by Equation [\(17\)](#page-11-1):

Figure 13. Schematic set up of the fatigue assessment model.

In Figure [14,](#page-11-2) the set up of the fatigue assessment model based on the KTD is depicted. In line with the S/N-curves, the fatigue range is again normalized by the fatigue strength at ten million load cycles of the near defect free HIP material. Furthermore, Figure [14](#page-11-2) displays the experimental results of the fatigue strength at ten million load cycles with a probability of survival of 50%. In addition, the probability density functions of the fatigue fracture initiation shrinkage pore sizes is plotted.

Figure 14. Validation of the fatigue assessment model.

In order to estimate the fatigue strength of a given defect distribution, the size of the heterogeneity in an arbitrary control volume V_α with a probability of occurrence $P_{Occ} = 0.5$ is considered. Subsequently, the corresponding fatigue strength is evaluated, invoking the Chapetti approach as in [\[42](#page-17-0)[,51\]](#page-17-9) and the model of El-Haddad—see [\[40](#page-16-17)[,41\]](#page-16-18). In addition, a local Weibull factor *κ* can be derived in order to enhance state-of-the-art fatigue design guidelines [\[22\]](#page-16-4), depending on a reference inhomogeneity population represented by its location parameter μ_0 , as depicted in Figure [15.](#page-12-1) Hence,

 (17)

the local *κ* factor steadily increases with elevated return period *α* of an highly-stressed volume, which will eventually lead to a saturation of the statistical size effect, in line with conclusions made in preliminary studies—see [\[23\]](#page-16-5).

Figure 15. Weibull factor *κ* depending on the return period *α* and the local defect population *µ*0.

Table [4](#page-12-0) lists the comparison of commonly invoked fatigue assessment models, accounting for a statistical size effect. Furthermore, the different models are validated by the experimental data of the two specimen geometries with varying highly-stressed volumes. Thus, the volumetric model turns out to be significantly conservative with a decrease in fatigue strength of 8%, leading to over-sizing of critical components in the design process. It has to be mentioned that the volumetric model is based on the statistically estimated fatigue strength of specimen position A, such that position A itself cannot be validated by this approach.

| Position V | Experiment Volumetric Δ ElHaddad Δ Chapetti Δ Murakami | | | | | | | | Δ |
|------------|-----------------------------------------------------------------------------|-------|-------|------|--------|-------------|--------|---------------------|----------|
| | 0.79 | Basis | Basis | 0.80 | $+2\%$ | $0.78 -2\%$ | | 0.68 ^a | -14% |
| | 0.68 | 0.63 | $-8%$ | 0.72 | $+5\%$ | 0.65 | -5% | 0.64 ^a | -7% |

^a Modified Murakami approach invoking a coefficient *C*² of zero.

Furthermore, the original [√] *area* model of Murakami was modified with a coefficient *C*² of zero rather than 120, since preliminary studies revealed a proper accordance in fatigue strength assessment with *C*² equal to zero considering similar cast alloys—see [\[58](#page-17-25)[,70\]](#page-18-2). The modified *EVIR* approach of Murakami [\[68\]](#page-18-0) hence leads to a similar conservative assessment of the fatigue strength as the volumetric approach [\[21\]](#page-16-3), with 14% for the small-sized and respectively 7% for the large-sized specimen with respect to the experimental results.

On the other hand, the presented approach in this work, considering the theoretical statistical distribution of flaw sizes in an arbitrary enhanced volume *Vα*, reveals a sound correlation with the fatigue data of both geometries A and B with a slightly conservative estimation of the fatigue strength—see Table [4.](#page-12-0) Finally, a graphical illustration is given in Figure [16.](#page-13-0) Thus, the fatigue strength of larger components, inheriting an elevated highly-stressed volume, can be properly calculated by the presented model in this work, especially when crack closure effects are considered by means of the implemented R-curve approach.

Figure 16. Validation of the fatigue assessment model considering the statistical size effect.

4. Conclusions

This work evaluates the fatigue strength of two differently sized specimens made of EN AC-46200 in T6 condition. Technological size effects are suppressed as the microstructural properties reveal no significant deviation between the investigated sample positions A and B. The presented approach considers both physically short and long crack growth invoking crack closure effects evaluated as the crack resistance R-curve. A probabilistic model of the fatigue fracture initiating flaw sizes in an arbitrary highly-stressed volume is set up, in order to act as input parameter for the subsequent fatigue assessment based on the Kitagawa–Takahashi diagram. Based on the results presented within this paper, the following conclusions can be drawn:

- The return period *α* possesses a value of approximately ten utilizing the numerically determined highly-stressed volumes of the two specimen geometries A and B.
- Extensive high cycle fatigue tests are statistically evaluated in both the finite as well as the long life region. The fatigue assessment of the empirical HCF data reveals a significant statistical size effect with a decrease in fatigue strength of about 14% in terms of specimen geometry B compared to the reference geometry A.
- The theoretical probability of occurrence of fatigue fracture initiating defect sizes for a given highly-stressed volume is evaluated based on the distribution of critical heterogeneities in a reference volume. The validation of the theoretical distribution with the estimated spatial extents of crack-initiating flaws reveals a minor deviation of just two percent, evaluated for a probability of occurrence of 50%.
- The defect size at a $P_{Occ} = 0.5$ subsequently acts as equivalent crack size for the fatigue assessment invoking the Kitagawa–Takahashi diagram with respect to its extensions for short crack growth.
- The validation of the defect based probabilistic fatigue assessment model with the empirical fatigue data of both specimen types with varying highly-stressed volumes reveals that the introduced R-curve concept to assess the fatigue strength best has a conservative deviation of just five percent.

Future work will deal with the implementation of the statistically distributed fatigue strength of the defect free material as well as the long crack threshold, as originally presented in [\[37\]](#page-16-14), along with a mean stress sensitivity model, as provided in [\[53\]](#page-17-11), in order to set up a holistic fatigue assessment map.

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Abbreviations

The following abbreviations and symbols are used in this manuscript:

References

- 1. Hertel, O.; Vormwald, M. Statistical and geometrical size effects in notched members based on weakest-link and short-crack modelling. *Eng. Fract. Mech.* **2012**, *95*, 72–83. [\[CrossRef\]](http://dx.doi.org/10.1016/j.engfracmech.2011.10.017)
- 2. Makkonen, M. Statistical size effect in the fatigue limit of steel. *Int. J. Fatigue* **2001**, *23*, 395–402. [\[CrossRef\]](http://dx.doi.org/10.1016/S0142-1123(01)00003-2)
- 3. Bazios, I.; Gudladt, H.J. The fatigue lifetime estimation in consideration of the statistical size effect as example for the AlMgSi0,7 alloy. *Materialwiss. Werkstofftech.* **2004**, *35*, 21–28. [\[CrossRef\]](http://dx.doi.org/10.1002/mawe.200300691)
- 4. Leitner, M.; Garb, C.; Remes, H.; Stoschka, M. Microporosity and statistical size effect on the fatigue strength of cast aluminium alloys EN AC-45500 and 46200. *Mater. Sci. Eng. A* **2017**, *707*, 567–575. [\[CrossRef\]](http://dx.doi.org/10.1016/j.msea.2017.09.023)
- 5. Kloos, K.H.; Buch, A.; Zankov, D. Pure Geometrical Size Effect in Fatigue Tests with Constant Stress Amplitude and in Programme Tests. *Materialwiss. Werkstofftech.* **1981**, *12*, 40–50. [\[CrossRef\]](http://dx.doi.org/10.1002/mawe.19810120205)
- 6. Taylor, D. Geometrical effects in fatigue: A unifying theoretical model. *Int. J. Fatigue* **1999**, *21*, 413–420. [\[CrossRef\]](http://dx.doi.org/10.1016/S0142-1123(99)00007-9)
- 7. Savaidis, G.; Savaidis, A.; Tsamasphyros, G.; Zhang, C. On size and technological effects in fatigue analysis and prediction of engineering materials and components. *Inter. J. Mech. Sci.* **2002**, *44*, 521–543. [\[CrossRef\]](http://dx.doi.org/10.1016/S0020-7403(01)00105-9)
- 8. Shirani, M.; Härkegård, G. Fatigue life distribution and size effect in ductile cast iron for wind turbine components. *Eng. Failure Anal.* **2011**, *18*, 12–24. [\[CrossRef\]](http://dx.doi.org/10.1016/j.engfailanal.2010.07.001)
- 9. Pavlov, V.F.; Kirpichev, V.A.; Vakulyuk, V.S.; Sazanov, V.P. Surface hardening influence on the fatigue limit for cylindrical parts of different diameter. *Russ. Aeronaut.* **2014**, *57*, 324–326. [\[CrossRef\]](http://dx.doi.org/10.3103/S1068799814030180)
- 10. Griffith, A.A. The Phenomena of Rupture and Flow in Solids. *Philos. Trans. R. Soc. Math. Phys. Eng. Sci.* **1921**, *221*, 163–198. [\[CrossRef\]](http://dx.doi.org/10.1098/rsta.1921.0006)
- 11. Weibull, W. Zur Abhängigkeit der Festigkeit von der Probengröße. *Arch. Appl. Mech.* **1959**, *28*, 360–362.
- 12. Carpinteri, A. Decrease of apparent tensile and bending strength with specimen size: Two different explanations based on fracture mechanics. *Inter. J. Solids Struct.* **1989**, *25*, 407–429. [\[CrossRef\]](http://dx.doi.org/10.1016/0020-7683(89)90056-5)
- 13. Heckel, K.; Köhler, J.; München, T. Experimental Investigation of the Statistical Size-Effect in Fatigue Tests with Unnotched Specimens. *Materialwiss. Werkstofftech.* **1975**, *6*, 52–54. [\[CrossRef\]](http://dx.doi.org/10.1002/mawe.19750060204)
- 14. Furuya, Y. Notable size effects on very high cycle fatigue properties of high-strength steel. *Mater. Sci. Eng. A* **2011**, *528*, 5234–5240. [\[CrossRef\]](http://dx.doi.org/10.1016/j.msea.2011.03.082)
- 15. Carpinteri, A.; Spagnoli, A.; Vantadori, S. An approach to size effect in fatigue of metals using fractal theories. *Fract. Eng. Mater. Struct.* **2002**, *25*, 619–627. [\[CrossRef\]](http://dx.doi.org/10.1046/j.1460-2695.2002.00506.x)
- 16. Carpinteri, A.; Puzzi, S. A fractal approach to indentation size effect. *Eng. Fract. Mech.* **2006**, *73*, 2110–2122. [\[CrossRef\]](http://dx.doi.org/10.1016/j.engfracmech.2006.04.020)
- 17. Carpinteri, A.; Puzzi, S. The fractal-statistical approach to the size-scale effects on material strength and toughness. *Probab. Eng. Mech.* **2009**, *24*, 75–83. [\[CrossRef\]](http://dx.doi.org/10.1016/j.probengmech.2008.01.003)
- 18. Basquin, O.H. The exponential law of endurance tests. *Am Soc. Test. Mater. Proc.* **1910**, *10*, 625–630.
- 19. Kuguel, R. A relation between theoretical stress concentration factor and fatigue notch factor deduced from the concept of highly stressed volume. *Proc. ASTM* **1961**, *61*, 732–748.
- 20. Diemar, A.; Thumser, R.; Bergmann, J.W. Statistischer Größeneinfluss und Bauteilfestigkeit. *Mater. Test.* **2004**, *46*, 16–21. [\[CrossRef\]](http://dx.doi.org/10.3139/120.100559)
- 21. Sonsino, C.M. Zur Bewertung des Schwingfestigkeitsverhaltens von Bauteilen mit Hilfe örtlicher Beanspruchungen. *Konstruktion* **1993**, *45*, 25–33.
- 22. Rennert, R.; Maschinenbau F.F. *Rechnerischer Festigkeitsnachweis für Maschinenbauteile aus Stahl, Eisenguss- und Aluminiumwerkstoffen*, 6th ed.; FKM-Richtlinie, VDMA-Verl.: Frankfurt am Main, Germany, 2012.
- 23. Sonsino, C.M.; Ziese, J. Fatigue strength and applications of cast aluminium alloys with different degrees of porosity. *Int. J. Fatigue* **1993**, *15*, 75–84. [\[CrossRef\]](http://dx.doi.org/10.1016/0142-1123(93)90001-7)
- 24. Böhm, J.; Heckel, K. The Prediction of the Fatigue Strength with Regard to the Statistical Size Effect. *Materialwiss. Werkstofftech.* **1982**, *13*, 120–128. [\[CrossRef\]](http://dx.doi.org/10.1002/mawe.19820130408)
- 25. Kuwazuru, O.; Murata, Y.; Hangai, Y.; Utsunomiya, T.; Kithara, S.; Yoshikawa, N. X-Ray CT Inspection for Porosities and Its Effect on Fatigue of Die Cast Aluminium Alloy. *J. Solid Mech. Mater. Eng.* **2008**, *2*, 1220–1231. [\[CrossRef\]](http://dx.doi.org/10.1299/jmmp.2.1220)
- 26. Rotella, A.; Nadot, Y.; Piellard, M.; Augustin, R.; Fleuriot, M. Fatigue limit of a cast Al-Si-Mg alloy (A357-T6) with natural casting shrinkages using ASTM standard X-ray inspection. *Int. J. Fatigue* **2018**, *114*, 177–188. [\[CrossRef\]](http://dx.doi.org/10.1016/j.ijfatigue.2018.05.026)
- 27. González, R.; González, A.; Talamantes-Silva, J.; Valtierra, S.; Mercado-Solís, R.D.; Garza-Montes-de Oca, N.F.; Colás, R. Fatigue of an aluminium cast alloy used in the manufacture of automotive engine blocks. *Int. J. Fatigue* **2013**, *54*, 118–126. [\[CrossRef\]](http://dx.doi.org/10.1016/j.ijfatigue.2013.03.018)
- 28. González, R.; Martínez, D.I.; González, J.A.; Talamantes, J.; Valtierra, S.; Colás, R. Experimental investigation for fatigue strength of a cast aluminium alloy. *Int. J. Fatigue* **2011**, *33*, 273–278. [\[CrossRef\]](http://dx.doi.org/10.1016/j.ijfatigue.2010.09.002)
- 29. Gänser, H.P. Some notes on gradient, volumetric and weakest link concepts in fatigue. *Comput. Mater. Sci.* **2008**, *44*, 230–239. [\[CrossRef\]](http://dx.doi.org/10.1016/j.commatsci.2008.03.021)
- 30. Abroug, F.; Pessard, E.; Germain, G.; Morel, F.; Hénaff, G. Fatigue size effect due to defects in an AA7050 alloy. *MATEC Web Conf.* **2018**, *165*, 14015. [\[CrossRef\]](http://dx.doi.org/10.1051/matecconf/201816514015)
- 31. Abroug, F.; Pessard, E.; Germain, G.; Morel, F. A probabilistic approach to study the effect of machined surface states on HCF behavior of a AA7050 alloy. *Int. J. Fatigue* **2018**, *116*, 473–489. [\[CrossRef\]](http://dx.doi.org/10.1016/j.ijfatigue.2018.06.048)
- 32. Kitagawa, H.; Takahashi, S. Applicability of fracture mechanics to very small cracks or the cracks in the early stage. In Proceedings of the 2nd International Conference on Mechanical Behaviour of Materials, Boston, MA, USA, 16–20 August 1976; pp. 627–631.
- 33. Garb, C.; Leitner, M.; Stauder, B.; Schnubel, D.; Grün, F. Application of modified Kitagawa-Takahashi diagram for fatigue strength assessment of cast Al-Si-Cu alloys. *Int. J. Fatigue* **2018**, *111*, 256–268. [\[CrossRef\]](http://dx.doi.org/10.1016/j.ijfatigue.2018.01.030)
- 34. Iben Houria, M.; Nadot, Y.; Fathallah, R.; Roy, M.; Maijer, D.M. Influence of casting defect and SDAS on the multiaxial fatigue behaviour of A356-T6 alloy including mean stress effect. *Int. J. Fatigue* **2015**, *80*, 90–102. [\[CrossRef\]](http://dx.doi.org/10.1016/j.ijfatigue.2015.05.012)
- 35. Tenkamp, J.; Koch, A.; Knorre, S.; Krupp, U.; Michels, W.; Walther, F. Defect-correlated fatigue assessment of A356-T6 aluminum cast alloy using computed tomography based Kitagawa-Takahashi diagrams. *Int. J. Fatigue* **2018**, *108*, 25–34. [\[CrossRef\]](http://dx.doi.org/10.1016/j.ijfatigue.2017.11.003)
- 36. Roy, M.J.; Nadot, Y.; Nadot-Martin, C.; Bardin, P.G.; Maijer, D.M. Multiaxial Kitagawa analysis of A356-T6. *Int. J. Fatigue* **2011**, *33*, 823–832. [\[CrossRef\]](http://dx.doi.org/10.1016/j.ijfatigue.2010.12.011)
- 37. Aigner, R.; Pusterhofer, S.; Pomberger, S.; Leitner, M.; Stoschka, M. A probabilistic Kitagawa-Takahashi diagram for fatigue strength assessment of cast aluminium alloys. *Mater. Sci. Eng. A* **2019**, *745*, 326–334. [\[CrossRef\]](http://dx.doi.org/10.1016/j.msea.2018.12.108)
- 38. Radaj, D. Geometry correction for stress intesity at elliptical cracks. *Weld. Cut.* **1977**, *29*, 198–402.
- 39. Groza, M.; Nadot, Y.; Varadi, K. Defect size map for nodular cast iron components with ellipsoidal surface defects based on the defect stress gradient approach. *Int. J. Fatigue* **2018**, *112*, 206–215. [\[CrossRef\]](http://dx.doi.org/10.1016/j.ijfatigue.2018.03.025)
- 40. El Haddad, M.H.; Smith, K.N.; Topper, T.H. Fatigue Crack Propagation of Short Cracks. *J. Eng. Mater. Technol.* **1979**, *101*, 42. [\[CrossRef\]](http://dx.doi.org/10.1115/1.3443647)
- 41. El Haddad, M.H.; Topper, T.H.; Smith, K.N. Prediction of non propagating cracks. *Eng. Fract. Mech.* **1979**, *11*, 573–584. [\[CrossRef\]](http://dx.doi.org/10.1016/0013-7944(79)90081-X)
- 42. Chapetti, M.D. Fatigue propagation threshold of short cracks under constant amplitude loading. *Int. J. Fatigue* **2003**, *25*, 1319–1326. [\[CrossRef\]](http://dx.doi.org/10.1016/S0142-1123(03)00065-3)
- 43. Zerbst, U.; Vormwald, M.; Pippan, R.; Gänser, H.P.; Sarrazin-Baudoux, C.; Madia, M. About the fatigue crack propagation threshold of metals as a design criterion—A review. *Eng. Fract. Mech.* **2016**, *153*, 190–243. [\[CrossRef\]](http://dx.doi.org/10.1016/j.engfracmech.2015.12.002)
- 44. Suresh, S.; Ritchie, R.O. A geometric model for fatigue crack closure induced by fracture surface roughness. *Metall. Trans. A* **1982**, *13*, 1627–1631. [\[CrossRef\]](http://dx.doi.org/10.1007/BF02644803)
- 45. Wasén, J.; Heier, E. Fatigue crack growth thresholds—the influence of Young's modulus and fracture surface roughness. *Int. J. Fatigue* **1998**, *20*, 737–742. [\[CrossRef\]](http://dx.doi.org/10.1016/S0142-1123(98)00034-6)
- 46. Elber, W. Fatigue crack closure under cyclic tension. *Eng. Fract. Mech.* **1970**, *2*, 37–45. [\[CrossRef\]](http://dx.doi.org/10.1016/0013-7944(70)90028-7)
- 47. Kim, J.H.; Lee, S.B. Behavior of plasticity-induced crack closure and roughness-induced crack closure in aluminum alloy. *Int. J. Fatigue* **2001**, *23*, 247–251. [\[CrossRef\]](http://dx.doi.org/10.1016/S0142-1123(01)00155-4)
- 48. Suresh, S.; Zamiski, G.F.; Ritchie, D.R.O. Oxide-Induced Crack Closure: An Explanation for Near-Threshold Corrosion Fatigue Crack Growth Behavior. *Metall. Mater. Trans. A* **1981**, *12*, 1435–1443. [\[CrossRef\]](http://dx.doi.org/10.1007/BF02643688)
- 49. Newman, J.A.; Piascik, R.S. Interactions of plasticity and oxide crack closure mechanisms near the fatigue crack growth threshold. *Int. J. Fatigue* **2004**, *26*, 923–927. [\[CrossRef\]](http://dx.doi.org/10.1016/j.ijfatigue.2004.02.001)
- 50. McEvily, A.J.; Endo, M.; Murakami, Y. On the relationship and the short fatigue crack threshold. *Fatigue Fract. Eng. Mater. Struct.* **2003**, *26*, 269–278. [\[CrossRef\]](http://dx.doi.org/10.1046/j.1460-2695.2003.00636.x)
- 51. Maierhofer, J.; Pippan, R.; Gänser, H.P. Modified NASGRO equation for physically short cracks. *Int. J. Fatigue* **2014**, *59*, 200–207. [\[CrossRef\]](http://dx.doi.org/10.1016/j.ijfatigue.2013.08.019)
- 52. Ishihara, S.; Sugai, Y.; McEvily, A.J. On the Distinction Between Plasticity- and Roughness-Induced Fatigue Crack Closure. *Metall. Mater. Trans. A* **2012**, *43*, 3086–3096. [\[CrossRef\]](http://dx.doi.org/10.1007/s11661-012-1121-9)
- 53. Aigner, R.; Leitner, M.; Stoschka, M. On the mean stress sensitivity of cast aluminium considering imperfections. *Mater. Sci. Eng. A* **2019**. [\[CrossRef\]](http://dx.doi.org/10.1016/j.msea.2019.04.119)
- 54. Kolitsch, S.; Gänser, H.P.; Maierhofer, J.; Pippan, R. Fatigue crack growth threshold as a design criterion statistical scatter and load ratio in the Kitagawa-Takahashi diagram. *IOP Conf. Ser. Mater. Sci. Eng.* **2016**, *119*, 012015. [\[CrossRef\]](http://dx.doi.org/10.1088/1757-899X/119/1/012015)
- 55. Murakami, Y.; Endo, M. Effects of defects, inclusions and inhomogeneities on fatigue strength. *Int. J. Fatigue* **1994**, *16*, 163–182. [\[CrossRef\]](http://dx.doi.org/10.1016/0142-1123(94)90001-9)
- 56. Li, P.; Lee, P.D.; Maijer, D.M.; Lindley, T.C. Quantification of the interaction within defect populations on fatigue behavior in an aluminum alloy. *Acta Mater.* **2009**, *57*, 3539–3548. [\[CrossRef\]](http://dx.doi.org/10.1016/j.actamat.2009.04.008)
- 57. Aigner, R.; Leitner, M.; Stoschka, M. Fatigue strength characterization of Al-Si cast material incorporating statistical size effect. *MATEC Web Conf.* **2018**, *165*, 14002. [\[CrossRef\]](http://dx.doi.org/10.1051/matecconf/201816514002)
- 58. Aigner, R.; Garb, C.; Leitner, M.; Stoschka, M.; Grün, F. Application of a $\sqrt{ }$ area -Approach for Fatigue Assessment of Cast Aluminum Alloys at Elevated Temperature. *Metals* **2018**, *8*, 1033. [\[CrossRef\]](http://dx.doi.org/10.3390/met8121033)
- 59. Aigner, R.; Leitner, M.; Stoschka, M.; Hannesschläger, C.; Wabro, T.; Ehart, R. Modification of a Defect-Based Fatigue Assessment Model for Al-Si-Cu Cast Alloys. *Materials* **2018**, *11*. [\[CrossRef\]](http://dx.doi.org/10.3390/ma11122546) [\[PubMed\]](http://www.ncbi.nlm.nih.gov/pubmed/30558138)
- 60. Tiryakioğlu, M. On the size distribution of fracture-initiating defects in Al- and Mg-alloy castings. Mater. *Sci. Eng. A* **2008**, *476*, 174–177. [\[CrossRef\]](http://dx.doi.org/10.1016/j.msea.2007.04.088)
- 61. Anderson, K.V.; Daniewicz, S.R. Statistical analysis of the influence of defects on fatigue life using a Gumbel distribution. *Int. J. Fatigue* **2018**, *112*, 78–83. [\[CrossRef\]](http://dx.doi.org/10.1016/j.ijfatigue.2018.03.008)
- 62. Gumbel, E.J. *Statistics of Extremes*; Columbia University Press: New York, NY, USA, 1958.
- 63. Gnedenko, B. Sur la distribution limite du terme maximum d'une serie aleatoire *Ann. Math.* **1943**. *44*. [\[CrossRef\]](http://dx.doi.org/10.2307/1968974)
- 64. Jenkinson, A.F. The frequency distribution of the annual maximum (or minimum) values of meteorological elements. *Q. J. R. Meteorolog. Soc.* **1955**, *87*, 145–158. [\[CrossRef\]](http://dx.doi.org/10.1002/qj.49708134804)
- 65. Tiryakioğlu, M. Statistical distributions for the size of fatigue-initiating defects in Al–7%Si–0.3%Mg alloy castings: A comparative study. *Mater. Sci. Eng. A* **2008**, *497*, 119–125. [\[CrossRef\]](http://dx.doi.org/10.1016/j.msea.2008.06.023)
- 66. Beretta, S.; Murakami, Y. Statistical analysis of defects for fatigue strength prediction and quality control of materials. *Fatigue Fract. Eng. Mater. Struct.* **1998**, *21*, 1049–1065. [\[CrossRef\]](http://dx.doi.org/10.1046/j.1460-2695.1998.00104.x)
- 67. Raynal-Villasenor, J.A.; Raynal-Gutierrez, M.E. Estimation procedures for the GEV distribution for the minima. *J. Hydrol.* **2014**, *519*, 512–522. [\[CrossRef\]](http://dx.doi.org/10.1016/j.jhydrol.2014.07.045)
- 68. Murakami, Y. *Metal Fatigue: Effects of Small Defects and Nonmetallic Inclusions*; Elsevier: Amsterdam, The Netherlands, 2002.
- 69. Murakami, Y. Material defects as the basis of fatigue design. *Int. J. Fatigue* **2012**, *41*, 2–10. [\[CrossRef\]](http://dx.doi.org/10.1016/j.ijfatigue.2011.12.001)
- 70. Garb, C.; Leitner, M.; Grün, F. Application of $\sqrt{\text{area-concept}}$ to assess fatigue strength of AlSi7Cu0.5Mg casted components. *Eng. Fract. Mech.* **2017**, *185*, 61–71. [\[CrossRef\]](http://dx.doi.org/10.1016/j.engfracmech.2017.03.018)
- 71. DIN EN 1706. *Aluminium and Aluminium Alloys-Castings-Chemical Composition and Mechanical Propertie*; EN 1706:2010; German version; NSAI: Dublin, Ireland, 2010.
- 72. Vandersluis, E.; Ravindran, C. Comparison of Measurement Methods for Secondary Dendrite Arm Spacing. *Metall. Microstruct. Anal.* **2017**, *6*, 89–94. [\[CrossRef\]](http://dx.doi.org/10.1007/s13632-016-0331-8)
- 73. Boileau, J.M.; Zindel, J.W.; Allison, J.E. The Effect of Solidification Time on the Mechanical Properties in a Cast A356-T6 Aluminum Alloy. *SAE Trans.* **1997**, *106*, 63–74.
- 74. Zhang, L.Y.; Jiang, Y.H.; Ma, Z.; Shan, S.F.; Jia, Y.Z.; Fan, C.Z.; Wang, W.K. Effect of cooling rate on solidified microstructure and mechanical properties of aluminium-A356 alloy. *J. Mater. Process. Technol.* **2008**, *207*, 107–111. [\[CrossRef\]](http://dx.doi.org/10.1016/j.jmatprotec.2007.12.059)
- 75. Ceschini, L.; Boromei, I.; Morri, A.; Seifeddine, S.; Svensson, I.L. Microstructure, tensile and fatigue properties of the Al–10%Si–2%Cu alloy with different Fe and Mn content cast under controlled conditions. *J. Mater. Process. Technol.* **2009**, *209*, 5669–5679. [\[CrossRef\]](http://dx.doi.org/10.1016/j.jmatprotec.2009.05.030)
- 76. Zhang.; Chen.; Poirier. Effect of solidification cooling rate on the fatigue life of A356.2-T6 cast aluminium alloy. *Fatigue Fract. Eng. Mater. Struct.* **2000**, *23*, 417–423. [\[CrossRef\]](http://dx.doi.org/10.1046/j.1460-2695.2000.00299.x)
- 77. Boileau, J.M.; Allison, J.E. The effect of solidification time and heat treatment on the fatigue properties of a cast 319 aluminum alloy. *Metall. Trans.* **2003**, *34*, 1807–1820. [\[CrossRef\]](http://dx.doi.org/10.1007/s11661-003-0147-4)
- 78. Leitner, H. Simulation des Ermüdungsverhaltens von Aluminiumgusslegierungen. Ph.D. Thesis, Montanuniversität Leoben, Leoben, Austria, 2001.
- 79. ASTM International E 739. *Standard Practice for Statistical Analysis of Linear or Linearized Stress-Life (S-N) and Strain Life (E-N) Fatigue Data*; ASTM International: West Conshohocken, PA, USA, 1998.
- 80. Dengel, D.; Harig, H. Estimation of the fatigue limit by progressively-increasing load tests. *Fatigue Fract. Eng. Mater. Struct.* **1980**, *3*, 113–128. [\[CrossRef\]](http://dx.doi.org/10.1111/j.1460-2695.1980.tb01108.x)
- 81. DIN 50100. *Load Controlled Fatigue Testing-Execution and Evaluation of Cyclic Tests at Constant Load Amplitudes on Metallic Specimens and Components*; DIN: Berlin, Germany, 2016.
- 82. Massey, F.J., Jr. The Kolmogorov-Smirnov test for goodness of fit. *J. Am. Stat. Assoc.* **1951**, *46*, 68–78. [\[CrossRef\]](http://dx.doi.org/10.1080/01621459.1951.10500769)
- 83. Le, V.D.; Morel, F.; Bellett, D.; Saintier, N.; Osmond, P. Simulation of the Kitagawa-Takahashi diagram using a probabilistic approach for cast Al-Si alloys under different multiaxial loads. *Int. J. Fatigue* **2016**, *93*, 109–121. [\[CrossRef\]](http://dx.doi.org/10.1016/j.ijfatigue.2016.08.014)
- 84. Linder, J.; Axelsson, M.; Nilsson, H. The influence of porosity on the fatigue life for sand and permanent mould cast aluminium. *Int. J. Fatigue* **2006**, *28*, 1752–1758. [\[CrossRef\]](http://dx.doi.org/10.1016/j.ijfatigue.2006.01.001)
- 85. Makkonen, L.; Rabb, R.; Tikanmäki, M. Size effect in fatigue based on the extreme value distribution of defects. *Mater. Sci. Eng. A* **2014**, *594*, 68–71. [\[CrossRef\]](http://dx.doi.org/10.1016/j.msea.2013.11.045)

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