

Supplementary material

Strategies Regarding High-Temperature Applications w.r.t Strength, Toughness, and Fatigue Life for SA508 Alloy

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1. Simulation Conditions for Optimization

We use Jmat-Pro (version 7) and Thermo-Calc software (2019a) for simulations, so we some assumptions for these calculations. Thermocalc is working on Gibbs's free energy models [1] for equilibrium conditions. For mechanical calculations related to SA508 analysis, it was assumed at γ -FCC, grain size is 10–15 μ m, and calculations were performed at 650 K temperature. For stress-strain calculations, the strain rate was assumed to be 0.1 at 650 K temperature.

Creep is reflected to be a temperature-dependent process and satisfies the power-law equation [2].

$$
\varepsilon = A\left(\frac{\sigma}{E}\right) \exp\left(\frac{-Q}{RT}\right) \tag{S1}
$$

T is the temperature, σ is significant stress, E is the Modulus, and A is a material-dependent constant; however, R, Q, and n are gas constant, activation energy, and stress exponent. R, T, and A are constants or known, so only Q, n, E, and σ are responsible parameters that change the material's steady-state creep rate.

Figure S2. Acs(ferrite to Austenite transition temperature) and Ac4(Austinite to Ferrite transition temperature) temperature calculations (**a**) FCC and BCC for MS1 samples; (**b**) FCC and BCC for MS2 samples; (**c**) FCC and BCC for MS3 samples.

Figure S3. Martensite temperature start (Ms), Ac³ and Ac⁴ temperatures w.r.t (**a**) MS1 (changing Carbon contents); (**b**) MS2 (changing Mangenese contents); (**c**) MS3 (changing Silicon contents).

Figure S4. Composition of phases w.r.t elements (**a**) MC_ETA; (**b**) M3C2;(**c**) M7C3; (**d**)Cementite; (**e**) M23C6; (**f**) G-phase.

Figure S5. Site fractions of phases w.r.t elements (**a**) MC_ETA, (**b**) M3C2, (**c**) M7C3, (**d**)Cementite, (**e**) M23C6, (**f**) Gphase.

Figure S6. (**a**) Volume fraction Austenite and pearlite regarding Carbon; (**b**) Volume fraction Austenite and pearlite regarding Mn; (**c**) Volume fraction Austenite and pearlite regarding Silicon.

Figure S7. Stress-strain calculation for SA508 (**a**); Silicon, (**b**); Manganese; (**c**) Carbon.

2. Fatigue Equations

Fatigue is an effect of permanent microplastic deformation when a material is exposed to cyclic loading. This often occurs together with thermal and/or corrosive attack owing to the presence of oxygen and hydrogen, causing a phenomenon known as stress corrosion cracking [3].

$$
\frac{\Delta\epsilon}{2} = \frac{\Delta\epsilon_e}{2} + \frac{\Delta\epsilon_p}{2} = \frac{\sigma_f'}{E} (2N)^b + \epsilon_f' (2N)^c \tag{S2}
$$

In this equation $\Delta \epsilon$, σ'_f , ϵ'_f , E, and 2N are strain amplitude, fracture stress, fracture strain, Young`s Modulus and number of cycles respectively, where b(-0.1) and c(-0.9) are user defined constants [4].

Figure S8. Scheil solidification diagram for the SUS304 composition and their effect on solidifications.

References

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