

Review **Post-Processing Techniques to Enhance the Quality of Metallic Parts Produced by Additive Manufacturing**

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Abstract: Additive manufacturing (AM) processes can produce three-dimensional (3D) near-netshape parts based on computer-aided design (CAD) models. Compared to traditional manufacturing processes, AM processes can generate parts with intricate geometries, operational flexibility and reduced manufacturing time, thus saving time and money. On the other hand, AM processes face complex issues, including poor surface finish, unwanted microstructure phases, defects, wear tracks, reduced corrosion resistance and reduced fatigue life. These problems prevent AM parts from real-time operational applications. Post-processing techniques, including laser shock peening, laser polishing, conventional machining methods and thermal processes, are usually applied to resolve these issues. These processes have proved their capability to enhance the surface characteristics and physical and mechanical properties. In this study, various post-processing techniques and their implementations have been compiled. The effect of post-processing techniques on additively manufactured parts has been discussed. It was found that laser shock peening (LSP) can cause severe strain rate generation, especially in thinner components. LSP can control the surface regularities and local grain refinement, thus elevating the hardness value. Laser polishing (LP) can reduce surface roughness up to 95% and increase hardness, collectively, compared to the as-built parts. Conventional machining processes enhance surface quality; however, their influence on hardness has not been proved yet. Thermal post-processing techniques are applied to eliminate porosity up to 99.99%, increase corrosion resistance, and finally, the mechanical properties' elevation. For future perspectives, to prescribe a particular post-processing technique for specific defects, standardization is necessary. This study provides a detailed overview of the post-processing techniques applied to enhance the mechanical and physical properties of AM-ed parts. A particular method can be chosen based on one's requirements.

Keywords: laser additive manufacturing; 3D printing; post processing techniques; surface characteristics; mechanical properties; hardness; grain refinement

1. Introduction

Additive manufacturing (AM), commonly designated as three-dimensional (3D) printing or rapid prototyping, generates 3D objects in a layer-wise manner based on a computeraided design model [\[1\]](#page-26-0). AM has experienced significant changes in production principle, feedstock, and part performance [\[2\]](#page-26-1). AM can rapidly generate 3D complicated structural elements based on the process characteristics of point-by-point melting and layer-by-layer manufacturing [\[3\]](#page-26-2). The non-equilibrium solidification process may be tweaked to produce

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parts having particular properties for special applications [\[4\]](#page-26-3). AM has various advantages, including a short manufacturing cycle and reasonable manufacturing cost in minibatches [\[5\]](#page-26-4). This technology produces parts using merely raw materials and equipment, thus eliminating the prerequisite of sophisticated tooling or molds, resulting in significant processing and assembly time savings. Near-net-part generation, tiny machining allowance and excellent material utilization are also advantages of AM processes [\[6\]](#page-26-5). In AM, the laser energy density is sufficient to process a wide range of materials [\[7\]](#page-26-6). Because of the high energy density of the laser beam, it can effectively elevate the localized thermal distribution to thousands of degrees, which is enough to melt the majority of metal materials. The AM metallic parts contain substantial residual stresses compared to conventional casting methods [\[8\]](#page-27-0). The layer-by-layer printing method releases the "forming" stress when the deposited layer solidifies. On the other hand, AM processes have numerous advantages, including the capacity to produce a variety of multi-material composites [\[9\]](#page-27-1), high processing efficiency and manufacturing of various complicated structures [\[10\]](#page-27-2).

However, due to layer-by-layer deposition, the surface quality of AM parts is typically inferior compared with the conventional manufacturing processes, which is a significant concern in AM processes [\[11\]](#page-27-3). Surface roughness usually varies depending on the AM technique. As a result, AM alone cannot simultaneously produce parts that meet both mechanical and surface roughness requirements [\[12\]](#page-27-4). In most cases, the most influencing aspect is the lack of process dynamics comprehension. Due to the sophisticated metallurgical and thermo-physical phenomena, the interaction mechanisms between the powder bed and the melt-pool, the powder bed and laser beam, and the melting processes are challenging to explain in the case of laser additive manufacturing (LAM) procedures. For instance, in the case of selective laser melting (SLM), the intense bonding force in processing zones and the quick solidification phenomenon under an ultra-high temperature gradient must be evaluated. Research is needed to evolve the parts' internal structure and thermal stress changes under cyclic heating and cooling. During AM, internal faults such as balling effect, pores formation, cracks generation, powder aggregation, and thermal stresses are usually generated. These flaws significantly impact the manufactured product microstructure and structural mechanics [\[13\]](#page-27-5). As a result, post-processing techniques are frequently needed after the parts are printed to improve mechanical characteristics and surface quality, thus allowing them to be used as intended [\[14\]](#page-27-6). Many post-processing techniques are available, including thermal post-processing to reduce thermally-induced residual stresses. Laser peening is usually applied to lessen micro-defects and to improve surface quality.

This review article discusses the effect of laser peening and polishing, machining, thermal post-processing, and abrasive finishing techniques on AM parts. Section [2](#page-1-0) collects the classification of various AM available as per ISO-ASTM 59200 (2015) standard. Section [3](#page-5-0) discusses the primary defects, including porosity formation, cracks formation, anisotropy and surface roughness problems in the AM-ed parts. Based on the literature review, the cause of these issues has also been highlighted in Section [3.](#page-5-0) For Section [4,](#page-6-0) laser shock peening (LSP), Laser polishing (LP), conventional machining process (CMP), including milling, rolling and chemical and abrasive machining, and heating processes (HP) containing solution heat treatment (SHT), hot isostatic pressing (HIP) and T6-heat treatment (T6-HT). Furthermore, their effects on the AM-ed parts have been highlighted. Moreover, the post-processing techniques' future outlook and conclusion have been provided in Section [5.](#page-24-0)

2. Classification of Additive Manufacturing (AM) Processes

According to ISO-ASTM 52900 (2015), AM is a procedure to join materials, layer-bylayer, to generate three-dimensional parts [\[15\]](#page-27-7). In recent years, AM applications have been expanded into several industrial sectors due to the technology providing opportunities for improved functionality, productivity, and competitiveness. Metal AM has unlimited potential and has recently been explored in the medical, aerospace and automotive industries [\[16\]](#page-27-8). On the one hand, products with complex geometry, operational flexibility, and reduced

manufacturing time can be produced using AM methods. On the other hand, they face several difficulties: poor surface finish, undesirable microstructure phases, porosity and flaws, delamination, wear tracks, a lack of hardness and corrosion resistance, and decreased fatigue life. Parts made using AM suffer due to the issues mentioned above, reducing rangue me. Tarts made using ANT surfer due to the issues membried above, reducing
their mechanical and physical characteristics. Post-processing methods such as laser shock peening, laser polishing, conventional machining processes and heat treatments are often used to fix these problems. These methods can improve the surface characteristics and physical and mechanical properties. The following Sections [2.1](#page-2-0)[–2.7,](#page-5-1) discuss the schematics of AM processes. \mathbf{H}

2.1. Powder Bed Fusion Process. Figure 1 depicts a schematic representation of the powder Bed Fusion Process.

High-energy power sources are used to selectively melt or sinter a metallic powder bed in the powder bed fusion (PBF) process. Figure [1](#page-2-1) depicts a schematic representation of the PBF setup [\[17\]](#page-27-9). The laser beam goes through a series of lenses and is reflected onto the platform surface by a mirror. Mirrors direct the laser beam spot movement along with the pre-determined routes. The platform travels downward after a layer of powder is selectively melted. Following on, a recoating blade spreads another layer of powder from the powder dispenser at the top of the previously deposited layer, and the laser scanning the powder dispenser at the top of the previously deposited layer, and the laser scanning process is repeated. The chamber is often filled with an inert gas such as argon [18]. process is repeated. The chamber is often filled with an inert g[as s](#page-27-10)uch as argon [18].

Figure 1. Powder bed fusion process schematic [17]; published under MDPI open-access license. **Figure 1.** Powder bed fusion process schematic [\[17\]](#page-27-9); published under MDPI open-access license.

2.2. Direct Energy Deposition

The direct energy deposition (DED) schematic is presented in Figure [2](#page-3-0) [\[19\]](#page-27-11). A laser, electron beam, or plasma arc can generate heat. Metallic powder or wire is the raw material ϵ efficiency because only a portion of the entire powder is melted and deposited into the substrate. Typical powder DED machines incorporate an inert gas blasted out of the nozzle along with the powder, shielding the melted area from oxidation [\[20\]](#page-27-12). to produce final products. Compared to metallic wires, powders have a poorer deposition

Figure 2. Direct energy deposition process (**a**) experimental setup and (**b**) schematic [19]; with per-**Figure 2.** Direct energy deposition process (a) experimental setup and (b) schematic [\[19\]](#page-27-11); with permission from Elsevier. mission from Elsevier.

2.3. Binder Jetting Process 2.3. Binder Jetting Process 2.3. Binder Jetting Process

Figur[e 3](#page-3-1) shows the schematic of the Binder jetting process [21]. A layer of powder is Figure 3 shows the schematic of the Binder jetting process [\[21\]](#page-27-13). A layer of powder is spread via a counterrotating roller for each layer. An inkjet print-head then pours/flows the liquid binding agent onto the powder bed to form a layer. Heaters may be used in some binder/powder systems to manage moisture and curing, but heat is not a necessary process requirement. After each layer, the build platform is lowered to print the next layer. The printed items are often brittle and require post-processing to improve the mechanical characteristics [\[21](#page-27-13)]. characteristics [21]. $\frac{1}{1}$ $\frac{1}{1}$ $\frac{1}{2}$ $\frac{1}{2}$

Figure 3. Schematic of Binder Jetting process [21]; with permission from Elsevier. **Figure 3.** Schematic of Binder Jetting process [\[21\]](#page-27-13); with permission from Elsevier.

T_{H} and T_{H} and T_{H} is extracted in the layer form in this process. *2.4. Material Extrusion Process 2.4. Material Extrusion Process*

The material (generally polymers) is extruded in a layer form in this process. This process continues until a 3D part is achieved using a CAD file. Fig[ur](#page-4-0)e 4 shows the schematic of this process [\[22\]](#page-27-14). It allows for a wide range of implant design options, resulting in a wide range of patient-tailored implant products [\[22\]](#page-27-14).

Figure 4. Material extrusion process schematic [\[22](#page-27-14)]; published under MDPI open-access license.

2.5. Material Jetting Process 2.5. Material Jetting Process 2.5. Material Jetting Process

Figure 5 shows the schematic of the material jetting process [23]. It can be divided Figure 5 shows the schematic of the material jetting process [\[23\]](#page-27-15). It can be divided Figur[e 5](#page-4-1) shows the schematic of the material jetting process [23]. It can be divided
into continuous inkjet (CIJ) printing and drop-on-demand (DOD) printing. The timing of droplet production is a critical distinction between CIJ and DOD. CIJ uses an ejection nozzle to break up an ongoing stream of droplets, whereas DOD generates droplets as $\overline{}$ nozzie to break up an ongoing stream or dropiets, whereas DOD generates dropiets as
needed. CIJ and DOD employ ABS, polyamide, PLA, and composites materials [\[24](#page-27-16)].

Figure 5. Material jetting process schematic [23]; published under MDPI open-access license. **Figure 5.** Material jetting process schematic [\[23](#page-27-15)]; published under MDPI open-access license.

2.6. Sheet Lamination Process 2.6. Sheet Lamination Process 2.6. Sheet Lamination Process

The sheet lamination process is a manufacturing process that uses cutting, sequential laminating and bonding to produce items and prototypes. With the help of ultrasonic $\frac{1}{2}$ werding and a laser cutter, tiny adhesive-coated including sheets or layers or plastic can be connected. An example of sheet lamination is illustrated in Figure [6](#page-5-2) [\[25\]](#page-27-17). As the process connected. An example of sheet lamination is illustrated in Figure 6 [25]. As the process ω solid-state bonding and extra admitstration state bonding and extra added to reach its medicines ω temperature for binding. The sheet lamination process can produce objects from various
materials most commonly commis tiles to motals materials, most commonly ceramic tiles to metals. welding and bonding to produce tients and prototypes. While the responsibility of the set of plastic can be welding and a laser cutter, tiny adhesive-coated metallic sheets or layers of plastic can be uses solid-state bonding and extra adhesives, the material does not need to reach its melting
Uses solid-state bonding and extra adhesives, the material does not need to reach its melting Figure 5. Material jetting process schematic [23]; published under MDPI open-access license.
2.6. Sheet Lamination Process
The sheet lamination process is a manufacturing process that uses cutting, sequelaminating and bond

Figure 6. Schematic of sheet lamination process [\[25\]](#page-27-17); with permission from Springer Nature.

2.7. Vat Polymerization Process schematic *profession* process schematic *also schematic [26]*. The photopolymerization process schematic *also schematic also* schematic *also schematic also schematic also schematic also s*

Figur[e 7](#page-5-3) depicts the vat polymerization process schematic [\[26\]](#page-27-18). The photopolymerization of liquid monomers is achieved by UV-assisted photopolymerization. An ultraviolet (UV) laser is used to cure a liquid monomer layer. After the first coating curing, the second layer of resin is sprayed at the top of the previously deposited layer. There are many cycles of re-coating and curing to achieve a 3D part [\[26\]](#page-27-18). $\,$

Figure 7. Schematic of sheet lamination process [26]; with permission from Elsevier. **Figure 7.** Schematic of sheet lamination process [\[26\]](#page-27-18); with permission from Elsevier.

Figure 7. Schematic of sheet lamination process [26]; with permission from Elsevier. **3. Defects in Additively Manufactured Parts 3. Defects in Additively Manufactured Parts**

3. Defects in Additively Manufactured Parts Aluminum, titanium, stainless steel and nickel alloys are the most commonly used feedstocks for AM. High reflectivity and thermal conductivity in aluminum alloys are the primary cause of defects [27]. The AM process's defects are pores, cracks, anisotropy, residual stresses, thermal stresses, laser spattering, poor surface roughness, and shape distortion [\[28,](#page-27-20)[29\]](#page-27-21). The residual stresses are the stresses that persist in a part even in the absence of external loading. In contrast, thermal stresses are usually induced by any change in a material's temperature [\[30\]](#page-27-22). There are various root causes for defects formation in the AM process. Here, we have compiled a few defects and their reasons. \mathbf{H} Aluminum, titanium, stainless steel and nickel alloys are the most commonly used

\mathbf{S} in a material's temperature \mathbf{S} . There are various root causes for defects f AM process. Here, we have compiled a few defects and their reasons. *3.1. Porosity Formation*

cates its material strength, hence the projected end performance [\[32\]](#page-27-24). One may classify the porosities in the AM component into two types: (a) metallurgical pores, which are caused by the absorption of the surrounding gas or by the evaporation of particular alloying elements, and (b) parameter-based holes, which are caused mainly by successive
 dilation/constriction cycles that lead to local failures [\[33\]](#page-27-25). As a result, "macro-porosities" Porosity is a significant flaw in the AM parts [\[31\]](#page-27-23). The density of a component indiare more dangerous than "micro-porosities." Indeed, if macro-porosities are not precisely

spherical, they can act as a cause of fractures. Micro-porosities may become critical, particularly after heat treatment [\[34\]](#page-27-26). Micro-porosities tend to consolidate during such operations, generating macro-porosities that are acicular in shape and hence more prone to cause the previously stated difficulties [\[29\]](#page-27-21). Porosity formation can be attributed to a variety of factors, including process parameters, the presence of impurities in the base material, material's low absorption of laser energy, problems with the solid material's wettability, the combustion chamber's atmospheric conditions, or evaporation phenomena of the alloy's constituent elements [\[35\]](#page-27-27).

3.2. Cracks Formation

Another defect encountered during the AM is crack formation [\[36\]](#page-28-0). The lack of liquid supply to the inter-dendritic spaces causes voids during solidification. It can be attributed to the extensive solidification range. One method to avoid these voids is to modify the composition of the alloy by increasing a particular amount of a specific element, identified by the trial and error method based on the given alloy [\[37\]](#page-28-1). For instance, adding silicon to the aluminum alloy will increase the eutectic phase and reduce the melting temperature and the solidification range, thus decreasing the hot cracking level [\[38\]](#page-28-2).

3.3. Anisotropy in the AM-ed Parts

In AM, another type of defect is the anisotropy of mechanical characteristics. The degree of anisotropy in the generated component depends on the part's building orientation [\[39\]](#page-28-3) and the complicated thermal history that the part has endured during AM [\[40\]](#page-28-4). Thijs et al. [\[41\]](#page-28-5) identified that the part's anisotropy could be controlled by choosing an optimum laser scanning strategy. Prashanth et al. [\[42\]](#page-28-6) showed that some parameter combinations might result in anisotropic characteristics within the produced components. Thus, the building orientation and the supports used during AM play an essential role in part's anisotropy [\[35\]](#page-27-27).

3.4. Surface Roughness Problems in AM-ed Parts

Most of the parts produced via the AM process present poor surface roughness. On the other hand, the best way to take advantage of the AM process is to build a part without post-processing. This goal has not yet been attained. This problem can be resolved by defining a deterministic relationship between operating conditions and surface quality. However, by improving the surface characteristics, one can cause other flaws in the produced components, and thus, the available solutions are limited. The laser parameters utilized directly affect the stability of the melt pool and, as a result, the homogeneity of the final bead [\[43](#page-28-7)[,44\]](#page-28-8). AM also exhibits the balling phenomena, which results in a coarse melt pool. This phenomenon is affected by the surface quality and is dependent on process parameters. Olakanmi [\[45\]](#page-28-9) demonstrated how laser power and scanning rate affect surface morphology. According to Louvis et al. [\[44\]](#page-28-8), low laser scan speeds increase the surface roughness. Furthermore, surface roughness can be addressed by using a contour scan to reduce surface inhomogeneity and skywriting to allow more uniform energy density scanning [\[46\]](#page-28-10).

There are still numerous drawbacks of AM processes that affect their application in demanding branches of industry. However, various techniques are available to improve the structural and mechanical properties of the AM parts. These techniques will be discussed in detail in Section [4](#page-6-0) of this paper.

4. Various Post-Processing Techniques for AM-ed Parts

The following sections discuss the application of post-processing techniques in the AM process.

4.1. Laser Shock Peening (LSP) 4.1. Laser Shock Peening (LSP)

LSP is a lateral expansion procedure that involves the material's plastic compression LSP is a lateral expansion procedure that involves the material's plastic compression perpendicular to the surface. The ability to withstand transverse strain leads to the accu-perpendicular to the surface. The ability to withstand transverse strain leads to the accumulation of local compressive stresses when laser peening is done on thick or restricted mulation of local compressive stresses when laser peening is done on thick or restricted objects [\[47,](#page-28-11)[48\]](#page-28-12). The strain rate is much higher in the thinner parts than the thicker ones objects [47,48]. The strain rate is much higher in the thinner parts than the thicker ones since LSP generates compressive residual stresses in the material [\[49\]](#page-28-13). Figure 8 depicts a since LSP generates compressive residual stresses in the material [49]. Figur[e 8](#page-7-0) depicts a schematic of an LSP process on a metal plate. The heated zone, using a focused laser beam schematic of an LSP process on a metal plate. The heated zone, using a focused laser beam on the metallic surface for 30 ns, reaches 10,000 °C, resulting in plasma formation. The generated plasma absorbs laser energy until the laser-material interaction time is attained. generated plasma absorbs laser energy until the laser-material interaction time is attained. Shock waves transmit the pressure generated by the plasma to the material. Direct ablation means generation of plasma plume by evaporation of laser irradiated material that expands with supersonic velocity, creating a shockwave in the opposite direction with respect to the plume expansion that exerts pressures equivalent to a few tenths of GPa [\[50\]](#page-28-14). To achieve a high amplitude of shock pressure, the LSP process typically employs a "confined mode". The metallic surface is coated with an opaque material such as black paint or aluminum foil, insulated against direct laser radiation by a transparent material. According to recent research, when adopting the confined mode, plasma pressures up to 10 GPa on the metallic surface. With a high magnitude of compressive residual stress, a more powerful pressure pulse may improve the outcome of LSP to a deeper depth [\[51\]](#page-28-15).

Figure 8. Laser shock peening schematic based on the data provided in Reference [51]. **Figure 8.** Laser shock peening schematic based on the data provided in Reference [\[51\]](#page-28-15).

LSP is usually applied to extend the fatigue life of any component, and it has recently LSP is usually applied to extend the fatigue life of any component, and it has recently been used to improve the fatigue life of aircraft componen[ts \[](#page-28-12)48]. LSP has also been used been used to improve the fatigue life of aircraft components [48]. LSP has also been used to enhance the properties of maraging steels [\[52](#page-28-16)[–54\]](#page-28-17). Furthermore, it has also been used to to bend and stretch aircraft fenders to produce more feasible aerodynamic models [48]. bend and stretch aircraft fenders to produce more feasible aerodynamic models [\[48\]](#page-28-12). Short but intense laser pulses form a plasma within the constrained geometry and cause pressure pulses, thus inducing local plastic deformations. The generated pressure can be increased and assist in efficient operation [48].

Fairand et al. [\[55\]](#page-28-18) used a large pulsed laser for producing the stress waves to alter microstructural and mechanical characteristics of 7075 aluminum (Al). The 0.2% offset the microstructural and mechanical characteristics of 7075 aluminum (Al). The 0.2% offset yield strengths of 7075-Al and unaged 7075-Al were increased by as much as 30% over yield strengths of 7075-Al and unaged 7075-Al were increased by as much as 30% over unshocked values. Here, material ageing indicates changes in its original state, but it does not always imply deterioration or degradation [\[56\]](#page-28-19). Ageing can also result in the new substances formation and the stability alteration of existing ones. This impact is desirable in some circumstances. Low pressure-induced residual stresses and their effects on fatigue life and stress corrosion behavior of several metallic alloys have been studied, including titanium [\[57\]](#page-28-20), Al [\[58\]](#page-28-21), steel [\[59\]](#page-28-22), and nickel-base [\[60\]](#page-28-23) alloys. Various scientists also determined the process factors that influence 6061T6-Al mechanical characteristics, fatigue life, and residual stresses [\[61](#page-28-24)[–63\]](#page-28-25).

Salimianrizi et al. [\[64\]](#page-29-0) analyzed the effects of LSP on Al 6061-T6. An Nd:YAG laser beam with 1200 mJ of energy per pulse and an 8 ns pulse duration was used to apply the beam with 1200 mJ of energy per pulse and an 0 ns pulse duration was used to apply the confined LSP. The findings revealed that compressive residual stress could be efficiently confined EST. The findings revealed that compressive residual stress could be efficiently produced on the surface of the treated material. Work hardening and grain refining produced on the surface of the treated material. Work hardening the grain refining were also effective for elevating the material's hardened depth to a maximum of 1875 μ m. Furthermore, surface roughness measurements revealed that the LSP could degrade surface Furthermore, surface roughness measurements revealed that the LSP could degrade surface quality depending on the operating conditions. It can be interpreted as a result of local thermore, surface roughness measurements revealed that the LSP could degrade surface plastic deformation caused by plasma-induced shock waves, the primary source of the quality depending on the operating conditions. It can be interpreted as a result of local plastic deformation caused by plastic matter shock waves, the plastic specific of the set of the sample is a clean straight surface's compressive residual stress. The upper surface of the sample is a clean straight line, as illustrated in Figure [9a](#page-8-0), exhibiting the considerable effect of polishing before LSP. A sample image of a single LSP with 50% overlap is shown in Figure [9b](#page-8-0). Despite using sacrificial confinement layers, the micrograph shows an uneven surface, which could be related to plastic deformations during LSP. \overline{S} Salimianrizi et al. [64] analyzed the effects of LSP on Al 6061-T6. An Nd:YAG laser related to plastic deformations during LSP. onfined LSP. The findings revealed that compressive residual stress could be efficiently roduced on the surface of the treated material. Work hardening and grain refining vere also effective for elevating the material's hardened depth to a maximum of $18/5$ μ m. urthermore, surface roughness measurements revealed that the LSP could degrade surface uality depending on the operating conditions. It can be interpreted as a result of local lastic deformation caused by plasma-induced shock waves, the primary source of the urface's compressive residual stress. The upper surface of the sample is a clean straight ϵ , as illustrated in Figure 9a, exhibiting the considerable effect of polishing before LSP. is ample image or a single LSP with 50% overlap is shown in Figure 9b. Despite using ϵ a sample image of a single sample in the single shows an uneven surface, which could be stated to plastic deformations during LSP.

Figure 9. Optical images of (a) LSP-untreated and (b) LSP-treated surfaces [\[64\]](#page-29-0); with permission from Elsevier. from Elsevier. from Elsevier.

Figure [10a](#page-8-1) shows the schematic of successive laser shots with laser scan overlap. Here, only one laser beam scan was performed along the Z-axis, while a laser scan overlap $\binom{6}{6}$ was applied to perform multiple scans along X-and Y-Axes. Roughness tests were performed on the specimens treated with single-laser-shot along the *X*- and *Y*-directions, performed on the specimens treated with single-laser-shot along the *X*- and *Y*-directions, as illustrated in Figure [10b](#page-8-1) [\[64\]](#page-29-0). Compared to the unprocessed surface, the LSP significantly enhanced in Figure 100 [01]. Computed to the unprocessed surface, the EST significantly
enhances the surface roughness. This increase can be explained by local plastic deformation caused by plasma-induced shock waves, the primary source of compressive residual stress on the surface. The roughness values are also different in the X- and Y-directions due to the scanning pattern and overlaps. It can be seen that increasing the overlap from 20 to 50% improves surface roughness. The 70% overlap, on the other hand, reveals a significant increase in the roughness. $\overline{\text{O}}$ from 20 to 50% improves surface roughness. The 70% overlap, on the other hand, or called a segmentary σ

Figure 10. (**a**) Schematic of successive laser-shots with laser scan overlap and (**b**) overlap (%) effect on the surface roughness [\[64\]](#page-29-0); with permission from Elsevier.

LSP for AM-ed Inconel 718 was reported by Jinoop et al. [\[65\]](#page-29-1), and the parametric LSP for AM-ed Inconel 718 was reported by Jinoop et al. [65], and the parametric analysis was carried out by adjusting peak laser power and the number of shots. For the number of shots. For the hardness and depth of the sample, the laser power was 170 mW, and the number of shots and the number of shots and the sample in the sample of the sample of shots and the sample of shots and the same of shots and the sh shots was 7, respectively. It was found that LSP altered the produced structure's surface shots was 7, respectively. It was found that LSP altered the produced structure's surface morphology and mechanical properties. The surface investigation revealed a maximum profile depth of 10 µm and a hardness of 360 HV measured via an optical profilometer and profile depth of 10 µm and a narraness of 360 HV measured via an optical promoneter and Vickers micro-hardness, respectively. After LSP, the compressive residual stress on the AM vickers micro-hardness, respectively. Their Est, the compressive residual stress on the Third sample surface was 214.9–307.9 MPa. The wear rate of LSP-treated AM samples improved by 1.70 times compared to as-built samples. The wear behavior of untreated and treated by 1.70 times compared to as-built samples. The wear behavior of untreated and treated ϵ and times compared to as-built samples. The wear behavior of untreated and detachment from samples is shown in Figure [11](#page-9-0) [\[65\]](#page-29-1). The SEM images clearly show particle detachment from the sample's surface due to delamination. The plate-like debris particles, in Figure [11a](#page-9-0),b represent indications of delamination that are caused by adhesion and metal-to-metal contact [\[18\]](#page-27-10). It was also discovered that LSP reduces the number of debris particles rising from the surface, linked to the elevated residual compressive stress and hardness. It can be ascribed to the reduced pores quantity in the LSP treated samples compared to untreated samples. As explained above, a specimen under LSP treatment experiences localized melting and re-solidification that reduces the pores percentage [\[50\]](#page-28-14). Figure [11e](#page-9-0) shows the variation in wear rate as a function of different LSP process parameters. It was discovered that when the laser power and number of shots rise, the wear rate reduces significantly. The number of shots seems to be more prominent than laser power variation. The specific wear rate varies according to the micro-hardness data, with the lowest specific wear rate occurring at 200 mW laser power and 7 shots. phology and mechanical properties. The surface investigation revealed and mechanical pro-

Figure 10. (**a**) Schematic of successive laser-shots with laser scan overlap and (**b**) overlap (%) effect

Figure 11. (a,b) Untreated LSP samples wear rates at lower and higher magnifications, respectively, and (c-e) treated LSP samples wear rates at lower and higher magnifications and an evaluation with the untreated samples, respectively [\[65\]](#page-29-1); with permission from Springer. the untreated samples, respectively [65]; with permission from Springer.

Compared to other Al-alloys, AlSi10Mg is an age-hardened cast aluminum alloy with improved mechanical properties and exceptional cast-and-weldability [\[66\]](#page-29-2). LSP's considerable effects on stress corrosion and fatigue characteristics have been widely studied and understood. Damon et al. [\[67\]](#page-29-3) used micro-tomography analysis to compare the shape and porosity distribution of AM AlSi10Mg parts before and after LSP. The LSP process resulted in a significant reduction of porosity (15–30%). Using density measurement, roughness characterization, and hardness measurements, Sagbas [\[68\]](#page-29-4) studied the effects of LSP, abrasive blasting, and laser polishing on textural parameters of direct metal laser sintered AlSi10Mg components.

In comparison to shot blasting, LSP can improve the surface's hardness and strength while reducing the surface roughness. The kurtosis of the LSP surface was <3, indicating that the shot peening surface's height distribution is flattened. As the skewness of the same surface is negative, the surface deviation height is greater than the average, indicating that the kurtosis and skewness profile parameters play an essential role in characterizing surface texture properties. They are indicators of quality in ISO-4287 [\[67\]](#page-29-3). The surface that has been shot-peened has the finest wear resistance.

Different laser intensities were used by Maamoun et al. [\[69\]](#page-29-5) to improve the surface properties of as-processed AlSi10Mg components. Under varying LSP intensities and sample surface textures, the impact of LSP on the microstructure of as-built AlSi10Mg samples was examined. As shown in Figure [5,](#page-4-1) SEM studies revealed a considerable alteration in the as-built microstructure. In the as-built + LSP sample, the distorted layers near the surface due to plastic deformation are depicted in Figure [12a](#page-11-0). In Figure [12b](#page-11-0),c, the fibrous silicon (Si) network surrounding the Al-matrix grains in the as-built microstructure was decomposed, followed by dynamic precipitation of spherical Si particles. Following LSP, the nanoscale Si particles were precipitated in a size range of 100–500 nm and homogeneously disseminated in the affected area, as demonstrated in Figure [12c](#page-11-0). The microstructure of the machined surface utilizing high-intensity LSP (machined surface + high-LSP) is shown in Figure [12d](#page-11-0)–f. The area near the surface was influenced by the circular stress waves that started at the shot position along the surface and extended to a depth of 10 μ m, as shown in Figure [12d](#page-11-0). As compiled in Figure [12e](#page-11-0), microcracks and the layers beneath the sample surface were also discovered. The microcracks vanished at a depth $>10 \mu m$ from the sample surface. It is worth noting that the microcracks did not emerge inside the microstructure of the as-built + LSP sample, implying that the sample's original surface texture influences the beginning of these cracks after LSP. The absence of the stress waves pattern and the use of a high surface covering factor value increased the microstructure homogeneity of the nano-recrystallized grains to depths of more than $10 \mu m$ (200%). The area affected inside the machined surface + high-LSP sample was extended to a depth of $130-150 \mu m$, as shown in SEM pictures (Figure [12f](#page-11-0)). The microstructure of the machined surface + LSP sample is depicted in Figure [12g](#page-11-0); no stress wave patterns were seen due to the use of low-intensity LSP, which reduced the plastic deformation strength. The machined surface + LSP sample depth was roughly $110-120 \mu m$, which was smaller than the machined surface + high-LSP sample. Due to the low-intensity LSP, Figure [12h](#page-11-0), i show incomplete dynamic precipitation of Si particles. The Si particles precipitated in the machined surface + LSP sample were more significant than those in the machined surface + LSP sample. These results were in relation to Cho et al.'s findings [\[70\]](#page-29-6).

LSP was employed by Chen et al. [\[71\]](#page-29-7) to modify the surface properties of nano-TiC particle-reinforced Inconel 625 nanocomposites. The effect of LSP was examined on surface morphology, residual stress, microhardness, microstructure and high-temperature oxidation behavior of AM parts. It was discovered that the strong plastic deformation caused by LSP could eliminate pores in the as-built sample. With a $460 \mu m$ hardened layer, the maximum hardness was 462 HV, and the surface stress state was switched from tensile to compressive. The (111) and (200) diffraction FWHM values expanded, attributed to grain refinement and a rise in lattice strain of the samples. It was also discovered that the LSP induced the transformation of a substantial number of columnar dendritic structures in the as-built sample into cellular dendritic structures. The walls with a high dislocation density were generated in the LSP sample. Jiang et al. [\[72\]](#page-29-8) performed systematic analyses on 3D printed Ti6Al4V alloy specimens. They determined that LSP can refine microstructure,

suppress residual stresses, and delay crack propagation. Still, it cannot eliminate the inherent defects in an AM part, such as un-melted powders, lack of fusion and clusters of α-phase, which significantly reduces the fatigue performance. Chi et al. [\[73\]](#page-29-9) applied a combination of heat treatment and LSP to change the microstructure and mechanical characteristics of Ti17 titanium alloy. The results showed that severe plastic deformation was induced in the surface layer, which, in turn, led to a high-level surface compressive residual stress (~763 MPa). Meanwhile, high-density dislocations and mechanical twins were analyzed in coarse α-phases after treatment, which gradually evolved into refined *α*-phases. The samples' elongation was significantly enhanced by 15% while ensuring *α*-phases. The samples' elongation was significantly enhanced by 15% while ensuring original ultimate tensile strength (1153 MPa).

Figure 12. Microstructure development due to LSP of AlSi10Mg samples under various resolution: **Figure 12.** Microstructure development due to LSP of AlSi10Mg samples under various resolution: (a-c) as-built + LSP; (d-f) machined su[rfa](#page-29-5)ce + High-LSP, and (g-i) machined surface + LSP [69]; published under MDPI open-access license. published under MDPI open-access license.

Lu et al. [\[74\]](#page-29-10) performed the LSP on Ti6Al4V samples to alter the mechanical properties
consine mismeturetures. Mechanical properties of metallis components are influenced by effective-reflection and procedure of LSP was examined on sur-position are marketiced.
By microstructure parameters such as grain size, dislocation density and distribution, and density. Figure 13 depicts typical cross-sectional views of all specimens. In all cross-sections, there are few pores but no apparent cracks or incomplete dissolution. As demonstrated in Figure 13a, some long columnar grains may be detected inside the horizontal AM specimen, and these long columnar grains are parallel to the building direction. The temperature of the filen poor generated by the faser steadily fowers from the bottom plane to the top surface throughout AM processing, while the solidification speed gradually increases. The melt pool solidifies from the base plane, and such solidification circumstances cause columnar grains formation. The latter layer's laser beam will cause the previous columnar by changing microstructures. Mechanical properties of metallic components are influenced of the melt pool generated by the laser steadily lowers from the bottom plane to the top

grains to re-melt, acting as a nucleus for the epitaxial development of the heavily textured grains. A substantial amount of fine acicular martensite can be seen inside the preceding grains, which is linked with the improved micro-hardness of AM specimens. Figure 13b depicts the cross-sectional microstructure of the horizontal AM-LSP specimen. There are no visible long columnar previous grains that can be refined by LSP and transformed into equiaxed grains. As presented in Figure [13c](#page-12-0), the vertical AM specimen contains many action of variations in the variation of variations. The variation of variations in the variation of variations in the variation of variat acicular martensites of varying lengths. Typical previous grains in the shape of irregular polygons can also be seen in the vertical AM specimen, which differs from the best cantel polygons can also be seen in the vertical AM specimen, which differs from the horizontal β but allowed the section in the vertical time operator, which differently from the horizontal one. Ti6Al4V is $\alpha + \beta$ dual-phase alloy, but no phase was seen in the AM specimens. It can be explained by the ultra-fast cooling rate (10^3-10^6 K/s) that immediately converts phases into supersaturated solid solutions. However, these phases have different crystal structures than the parent phase, namely α' martensite structures. A significant number of acicular α' martensite is parallel to 45°. At the cross-section of the vertical AM specimen, several overlapping α' martensite emerges in various areas. The vertical AM-LSP specimen's typical cross-sectional microst[ruc](#page-12-0)ture is shown in Figure 13d. Similar microstructures, such as acicular α' martensite, may be found in cross-sections of both AM-LSP specimens (Figure $10b$,d). There is modest acicular martensite in A[M-L](#page-8-1)SP specimens compared to both AM specimens. Still, the density of acicular martensite in the surface layer significantly increases due to LSP.

stances cause columnar grains formation. The latter layer's laser beam will cause the pre-

Figure 13. Optical images of Ti6Al4V samples' cross-sections: horizontal (a) AM and (b) AM-LSP specimens, and vertical (**c**) AM and (**d**) AM-LSP specimens [74]; with permission from Elsevier. specimens, and vertical (**c**) AM and (**d**) AM-LSP specimens [\[74\]](#page-29-10); with permission from Elsevier.

Sidhu et al. [75] investigated the effects of LSP on AM Inconel 718 specimens. With Sidhu et al. [\[75\]](#page-29-11) investigated the effects of LSP on AM Inconel 718 specimens. With the increase in laser energy density, it was found that both compressive residual stress the increase in laser energy density, it was found that both compressive residual stress and hardness increased after LSP treatment. After high-energy LSP treatment, the as-built samples presented the compressive residual stress of 875 MPa, and the hardness increased samples presented the compressive residual stress of 875 MPa, and the hardness increased from 468 to 853 HV. from 468 to 853 HV.

4.2. Laser Polishing (LP) 4.2. Laser Polishing (LP)

LP is a technique to improve the surface roughness of AM-ed parts. When the laser energy irradiates the material surface during LP, morphological apexes quickly attain the melting temperature. Due to gravity and surface tension, the liquified material reorganizes melting temperature. Due to gravity and surface tension, the inpairiod material reorganizes to the same level after the melt pool is generated. The heat-affected zone (HAZ) temperature lowers rapidly once the laser beam stops scanning the surface, resulting in melt-pool solidification and reduction in surface roughness [\[76–](#page-29-12)[78\]](#page-29-13). LP is a technique that re-melts to energy irradiates the material surface during LP, morphological apexes quickly attain the modify the surface morphology without affecting the bulk characteristics [\[79\]](#page-29-14).

Mai and Lim [\[80\]](#page-29-15) applied laser irradiation to polish AISI 304 stainless steel surface. The melting depth was in the sub-microscopic region, while the polishing rate was in the range of $5-15 \text{ cm}^2/\text{min}$. Due to LP, surface roughness decreased from 195 to 75 nm. LP induced an increase in specular surface reflectance of 14% and a decrease in diffusive reflectance of 70%. The heterogeneous microhardness distribution was converted to a homogeneous distribution. LP could improve pitting corrosion resistance because of the microstructural changes generated by laser rapid melting and re-solidification. The melting action is beneficial in sealing the micro-pores and -cracks, hence reducing surface scratches. Guo et al. [\[81\]](#page-29-16) used
surface in an authority of Atsimila and No determine the laser energies laser meanstrated an experimental orthogonal design (OD) to determine the laser operational parameters that the experimental duration in attaining a superior surface finishing of AISI 01 to reduce the experimental duration in attaining a superior surface finishing of AISI 01 tool steel by pulsed Nd:YAG laser. The results demonstrated that the orthogonal design made it possible to swiftly and effectively obtain the optimum parameters. Based on OD analysis and experimental data, the optimum LP parameters are pulse energy = 1 J, feed rate = $300-400$ mm/min, pulse length = 3 ms and pulse frequency = $20-25$ Hz. These conditions reduced the surface roughness from 0.40 to $0.12 \mu m$ by applying these condi-tions. Lamikiz et al. [\[82\]](#page-29-17) employed LP for selective laser sintered metallic components. The results indicated clear roughness reductions for AISI 420 stainless steel + bronze. Surface
calculated that roughness was reduced to 80% (7.5 to 1.2 μ m). The metallurgical studies indicated that the HAZ had no cracks or porosity. It can be deduced that laser-affected areas had a more uniform composition than untreated parts. On the other hand, the resultant surfaces were tougher and more homogeneous than [the](#page-29-14) initial components. Ma et al. [79] demonstrated the ability of a fiber laser to polish the rough surface of Ti6Al4V and $\alpha + \beta$ Ti-alloy alloys produced by AM. Figure 14a,e show macro-scale photos of Ti-alloy surfaces after [LP.](#page-13-0) As exhibited in Figure 14b,f, the rough samples were polished, and laser me[ltin](#page-13-0)g trails were visible on the surface of Ti-alloys. Figure [14c](#page-13-0),d,g,h show that after LP, the surface roughness of Ti-alloys decreased from 90 to 4.5 of Ti and $\alpha + \beta$ Ti-alloys decreased from 90 to 4 μm and 80 to 4.5 μm, respectively.

Figure 14. (a–d) Ti6Al4V alloy (laser-polished region, SEM, Topographies image of as-received and LP-ed surfaces), and (**e–h**) α + β Ti-alloy (laser-polished region, SEM, Topographies image of received and LP-ed surfaces) [79[\]; w](#page-29-14)ith permission from Elsevier. as-received and LP-ed surfaces) [79]; with permission from Elsevier.

Lee et al. [\[83\]](#page-29-18) investigated the AM of an $\alpha + \beta$ Ti-alloy. A continuous-wave fiber laser was used to treat the surface of the manufactured products. The powder particles were re-melted using LP, and new surface morphology was attained. The results are shown in Figure [15a](#page-14-0),b. It can be seen that the as-fabricated samples' surface yielded an enormously random surface with high peaks-valleys values. The samples after LP are presented in Figure 15c,d. One can observe that LP generated smooth [sur](#page-14-0)faces with minor peak-valleys was used to treat the surface of the manufactured products. The powder particles were values. One of the benefits of LP is that the surface conditions are usually improved without utilizing any extra material.

Figure 15. (a,b) Optical image and color map of $\alpha + \beta$ Ti-alloy before polishing, and (c,d) optical image and color map of $\alpha + \beta$ Ti-alloy after polishing [\[83](#page-29-18)]; with permission from Elsevier.

values. One of the benefits of L is that the surface conditions are usually improved with-surface conditions are usually improved with-surface conditions are usually improved with-surface α

Zhou et al. [\[84](#page-29-19)] developed a transient numerical model to identify the surface roughness evolution during the LP process. AlSi10Mg samples were polished using various laser hatch distances and laser beam scanning directions to understand surface characteristics. It was found that after LP, the surface roughness was reduced from 12.5 to 3.7 µm due to the re-melting of the material. The hardness value was improved from 112.3 to 176.9 HV with grain refinement and surface defects eli[min](#page-14-1)ation. Figure 16a,b show that the LP process eliminated the part's pores. After LP, the peaks-valleys variation was reduced significantly, as shown in Figure 16c,d.

Figure 16. (a,b) SEM images of AlSi10Mg alloy before and after laser polishing, and (c,d) mor- $\frac{1}{2}$ and $\frac{1}{2}$ are $\frac{1}{2}$; published under $\frac{1}{2}$; published under MDPI open-access and after $\frac{1}{2}$; published under MDPI open-access and $\frac{1}{2}$; published under MDPI open-phologies of AlSi10Mg alloy before and after laser polishing [\[84\]](#page-29-19); published under MDPI openlicense. access license.

Zhou et al. [\[85\]](#page-29-20) investigated Ti-alloys LP experimentally to identify an optimum laser energy density to reduce the surface roughness. A combination of operating conditions, in-

cluding laser = 150 W, overlapping percentage = 95% and laser scanning speed = 20 mm/s, reduces the average surface roughness from 3.09 to 0.56 μ m. Martensitic structures appeared in the material due to repeated heating and fast cooling thermal cycles, resulting in appeared in the material due to repeated heating and fast cooling thermal cycles, resulting a 25% increment in hardness value, as shown in Figure [17.](#page-15-0) in a 25% increment in hardness value, as shown in Figure 17. energy density to reduce the surface roughness. A combination of $\frac{1}{20}$ mm $\frac{1}{20}$ mm $\frac{1}{20}$ $\frac{1}{100}$ which is the 150 W $\frac{1}{100}$ W $\frac{1}{100}$ W $\frac{1}{100}$ and last $\frac{1}{100}$ and last $\frac{1}{100}$ in $\frac{1}{100}$ $\frac{1}{100}$ $\frac{1}{100}$ $\frac{1}{100}$ $\frac{1}{100}$ $\frac{1}{100}$ $\frac{1}{100}$ $\frac{1}{100}$ $\frac{1}{100$ meared in the material due to repeated begins and fest cooling thermal evalue reguling in $\frac{25\%}{9}$ increment in bardness value as shown in Figure 17 $\sum_{i=1}^n$ in Figure 15. In Figure 17. In Figure 17.

Figure 17. Evolution of hardness across the material [\[85\]](#page-29-20); with permission from Elsevier.

Aviles et al. [\[86](#page-29-21)] used the LP process to improve the surface roughness of medium carbon AISI 1045 steel, resulting in enhanced fatigue life. It has been demonstrated that LP improved the fatigue life from 10^5 to 10^6 cycles by decreasing the surface roughness from 15 to 5 μ m. Chen et al. [\[87\]](#page-29-22) studied the surface morphology and microstructure of AISI 316 L stainless steel parts before and after the LP process. They found that the average surface roughness decreased from 4.84 to 0.65 μ m. After LP, the average grain dimension was also reduced, while the angle of grain boundaries increased from $2°$ to $5°$. A maximum hardness equal to 262 HV was achieved. Rosa et al. [\[88\]](#page-29-23) investigated the effect of LP on AISI 316L stainless steel surface roughness. They found that the surface roughness was reduced to up to 96% after 5 laser passes. After LP, the surface did not collapse, while the surface to pography became smooth, as shown in Figure [18.](#page-15-1) It can be seen that the final surface still contained several cracks and defects. These defects significantly reduce the fatigue strength. Furthermore, LP yielded a "form" distortion due to the heat transfer from the laser to the top surface. $\frac{1}{\sqrt{2}}$ factor strength. Furthermore, LP yielded a $\frac{1}{\sqrt{2}}$ yielded a $\frac{1}{\sqrt{2}}$ transfer to the heat tra

Figure 18. Surface morphology before and after laser polishing (LP) [88]; with permission from **Figure 18.** Surface morphology before and after laser polishing (LP) [\[88\]](#page-29-23); with permission from Journal of laser applications.

LP does not influence the microstructure phase type significantly. Obeidi et al. [\[89\]](#page-29-24) applied LP on AM AISI 316L stainless steel samples. A $CO₂$ laser beam operating in continuous wave mode was used for polishing. The operating conditions, including laser tinuous wave mode was used for polishing. The operating conditions, including laser power, the sample's rotating speed, the number of scanning passes, the laser beam's focal power, the sample's rotating speed, the number of scanning passes, the laser beam's focal position, and the overlap (%) of the laser scans between adjacent passes, were considered. position, and the overlap (%) of the laser scans between adjacent passes, were considered. The samples' roughness was decreased from 10.4 to 2.7 µm, while no significant change The samples' roughness was decreased from 10.4 to 2.7 μm, while no significant change in microstructure phase-type was analyzed. Tian et al. [90] applied LP on electron beam in microstructure phase-type was analyzed. Tian et al. [\[90](#page-29-25)] applied LP on electron beam melted-Ti6Al4V components to improve surface quality. It was found that the surface melted-Ti6Al4V components to improve surface quality. It was found that the surface roughness was reduced by 75% (0.51 μm), which is analogous to a computerized numerical control machined surface. However, the texture of the re-melted layer changed, resulting in a modest increase in sub-surface hardness (15%). The technique generated a high amount of near-surface tensile residual stresses. LP increased the tensile residual stresses in the component's surface, up to 580 MPa. Chen et al. [\[91\]](#page-29-26) studied the influence of LP on surface modification and corrosion behavior of AM AISI 316L. Laser scanning speed and number of passes were varied to evaluate their effect on the surface quality and corrosion resistance. The results indicated that LP reduced the surface roughness from 4.75 to 0.49 μ m while incorporating partially melted powders originally on the as-printed surface layer. X-ray diffraction (XRD) results indicated no considerable phase change after LP. The sub-surface microhardness increased from 1.82 to 2.89 GPa.

Let $\mathcal{L}_\mathcal{D}$ does not influence the microstructure phase type significantly. Obeiding et al. $\mathcal{R}_\mathcal{D}$

Although the open circuit potential (OCP) cannot measure corrosion resistance, it can Although the open circuit potential (OCP) cannot measure corrosion resistance, it can reflect the corrosion tendency of the sample in the electrolyte. Metals with a lower OCP are more sensitive to electron loss and oxidation processes. The OCP vs. time curves for the as-printed and LP-ed samples in a 0.4 mol/L HCl solution are shown in Figure [19.](#page-16-0) The anode changes of the OCP indicate the production of the passivation film. In contrast, the cathode shifts of the OCP indicate the dissolution of the film or the absence of film formation. As shown in Figure [19,](#page-16-0) all curves include anode shifts at the start of the corrosion test due to $\frac{1}{2}$ the formation of the passivation coating. However, the OCP varies around a fixed value
the formation of the passivation coating. However, the OCP varies around a fixed value due to the passivation film's stability as time passes. After 1200 s of immersion in 0.4 mol/L
Help due to the passivation film's stability as time passes. After 1200 s of immersion in 0.4 mol/L HCl solution, the final stable OCP for the as-printed sample is around -0.483 V. When only 1 laser scanning pass was used, and the OCP stabilized at −0.469 V and −0.471 V.
The OCP stabilized scann d. 0.461 V and the OCP stabilized at −0.469 V and −0.471 V. The OCP stabilized around −0.461 V and −0.463 V for 3 laser scanning passes. The LP FIRE OCP Stabilized around −0.481 V and −0.463 V for 3 laser scanning passes. The *Et* samples exhibited anode shifts of OCP compared to the as-printed sample, which show stamples exhibited anode shifts of OCP compared to the as-printed sample, which show
that LP can minimize the sample's corrosion tendency. Furthermore, the time required from the LP sample to establish a stable OCP is reduced when compared to the as-printed for the LP sample to establish a stable OCP is reduced when compared to the as-printed $\frac{1}{2}$ and $\frac{1}{2}$ samples to establish a stable $\frac{1}{2}$ of the reduced when computed to the as-printed sample, indicating that the LP samples produced the passivation layer faster than the as-printed sample. p_{initial} and the LP samples produced the passivation layer faster that the passivation layer faster than the passivation layer faster than α

Figure 19. Corrosion test for as-printed and LP AISI 316 L samples [91]; with permission from Else-**Figure 19.** Corrosion test for as-printed and LP AISI 316 L samples [\[91\]](#page-29-26); with permission from Elsevier.

Lee et al. [\[83\]](#page-29-18) applied the LP to improve the fatigue performance of AM Ti6Al4V parts. Lee et al. [83] applied the LP to improve the fatigue performance of AM Ti6Al4V The strain-life fatigue plot is depicted in Figure [20.](#page-17-0) A different symbol represents each surface and stress relief condition: (a) green triangles for specimens with as-built surface condition (AB) , (b) blue diamonds for specimens with LP, (c) red circles for specimens with laser polished surface condition and secondary stress relief (LPSR), and (d) black squares for specimens with machined/polished surface condition as the baseline (M/P) . As indicated in Figure [20,](#page-17-0) the fatigue strength of LP specimens was improved at lower strain levels (i.e., intermediate cycle fatigue (ICF) and high cycle fatigue (HCF)) compared lower strain levels (i.e., intermediate cycle fatigue (ICF) and high cycle fatigue (HCF)) to AB specimens. In the HCF regime, the fatigue lifetimes of LP specimens were at least one order of magnitude higher than those of AB specimens.

Figure 20. Fatigue data of LP-ed Ti6Al4V samples [83]; with permission from Elsevier. **Figure 20.** Fatigue data of LP-ed Ti6Al4V samples [\[83\]](#page-29-18); with permission from Elsevier.

Nonetheless, LP specimens displayed lower fatigue strengths at greater strain levels Nonetheless, LP specimens displayed lower fatigue strengths at greater strain levels than AB specimens (i.e., low cycle fatigue (LCF)) due to the residual stresses generated on the LP specimens' surface due to the LP process. Although LP caused unavoidable residual stresses, the influence of surface roughness on fatigue strength in the HCF regime be more substantial than the residual stresses. As a result, even when additional stress can be more substantial than the residual stresses. As a result, even when additional stress release was not used, LP specimens had longer fatigue lifetimes than AB specimens in the release was not used, LP specimens had longer fatigue lifetimes than AB specimens in the ICF and HCF regimes. On the other hand, surface roughness has less influence on LCF ICF and HCF regimes. On the other hand, surface roughness has less influence on LCF due to the material's high plastic deformation. As a result, high plastic strain amplitudes due to the material's high plastic deformation. As a result, high plastic strain amplitudes may interact with tensile residual stresses to cause even more negative effects. In the LCF may interact with tensile residual stresses to cause even more negative effects. In the LCF regime, the contribution of residual stress may be more significant than the influence of regime, the contribution of residual stress may be more significant than the influence of surface roughness. LPSR specimens outperformed both AB and LP specimens in terms of surface roughness. LPSR specimens outperformed both AB and LP specimens in terms of fatigue strength. fatigue strength.

Furthermore, under 0.004 mm/mm strain amplitude, one LPSR specimen reached 10⁷ Furthermore, under 0.004 mm/mm strain amplitude, one LPSR specimen reached $10⁷$ reversals, equivalent to M/P specimens. Because it was the sole variation between the LP and LPSR specimens, the fatigue testing findings implied that residual stresses were satisfactorily reduced by a stress relief method (one hour at 700 °C under an argon atmosphere). However, several tests at 0.004 mm/mm strain amplitude failed before reaching 10^7 reversals, with one test failing before 10^5 cycles. At this strain amplitude, the difference between the shortest and longest measured fatigue life is two orders of the difference between the shortest and longest measured fatigue life is two orders of magnitude; hence, fatigue strength improvement is significant compared to AB and LP specimens. In general, the improvement in fatigue resistance in ICF and HCF regimes f_n follows the AB \lt LP \lt LPSR \lt M/P hierarchy, demonstrating that LP can be a viable technique of improving the following of AM parts be lessoning configurations. technique of improving the fatigue resistance of AM parts by lowering surface roughness.

4.3. Conventional Machining Process (CMP): Milling, Rolling, Chemical Machining and 4.3. Conventional Machining Process (CMP): Milling, Rolling, Chemical Machining and Abrasive Machining Abrasive Machining

CMP is the traditional manufacturing process employed to enhance the manufactured

CMP is the traditional manufacturing process employed to enhance the manufactured parts' dimensional accuracy and surface quality. Owing to the popularity and acceptability
at a suide layel, they are also arralayed in AM for the used we assessing at manufactured at a wide level, they are also employed in AM for the post-processing of manufactured at a wide level, they are also employed in AM for the post-processing of manufactured
parts. Bai et al. [\[92\]](#page-30-0) studied the machining of the ASTM A131 steel AM components. They parts. Bai et al. [92] studied the materialing of the ASTM A191 steel AM components. And y
used computerized numerical control (CNC) milling machine. After post-processing, the samples' surface roughness was smoothened from 22.7 to 0.6μ m, as shown in Figure [21.](#page-18-0) Figure consider the graces was smoothed from 22.7 to 0.6 μm, as shown and gas 22.7 However, the milling process did not affect the hardness value significantly.

Figure 21. Surface morphology of AM samples (**a**) top surface before milling, (**b**) side surface before milling, (**c**) top surface after milling and (**d**) side surface after milling [\[92\]](#page-30-0); with permission from Elsevier.

Lopes et al. [93] applied milling process on high strength low alloy steel parts pro-Lopes et al. [\[93\]](#page-30-1) applied milling process on high strength low alloy steel parts produced by AM process. It was found that the surface roughness upgraded with an increase in the cutting speed and a reduction in tool feeding rate. The maximum surface roughness = 0.641 µm was attained when milling was carried out using cutting speed = 30 m/min and feeding rate = 0.0345 mm/tooth. On the other hand, the least surface roughness = 0.206 µm was obtained when cutting speed = 65 m/min and feeding rate = 0.0115 mm/tooth. Honnige et al. [\[94\]](#page-30-2) investigated the impact of vertical inter-pass and post-deposition side
Honnige et al. [94] investigated the impact of vertical inter-pass and post-deposition side rolling on the aluminum (2319) walls manufactured by AM process. They found that resid-
rolling on the aluminum (2319) walls manufactured by AM process. They found that residual stresses and part defects decreased considerably. Furthermore, the hardness increased
using the ultrarity on increased in applied load due to work hardness of take as above in significantly with an increment in applied load due to work hardening effects, as shown in
Eisywe 22 Figure [22.](#page-19-0)

In another study by Scherillo [\[95\]](#page-30-3), surface finishing of AM-ed AlSi10Mg was carried out by samples' immersion in a chemical solution. A mixture of $HNO₃$ and HF was used at

85 ℃ for 75 min. The samples were rinsed in ultrasonic water + acetic acid bath to remove insoluble products. Figure [23](#page-19-1) shows a considerable decrease in surface roughness (Sa), ten-point height (Sz), kurtosis (Sku), skewness (Ssk), and symmetrical distribution of peaks and valleys (Sdq) with respect to the etching time. The chemical solution mainly acted upon the peaks, resulting in flat surfaces.

Figure 22. Evolution of hardness with an increment in applied load [\[94\]](#page-30-2); with permission from Elsevier. p_{square} and value to the etching term in the etching to the chemical solution p_{old} solution matrices with an increment in applied foad p_{old}

Figure 23. Effect of chemical machining on (a) surface roughness, (b) ten-point height, (c) kurtosis, (d) skewness, (e) peaks and valleys symmetrical distribution, and (f) etching rate [\[95\]](#page-30-3); with permission sion from Elsevier. from Elsevier.

There are various abrasive machining methods. Zhang et al. [\[96\]](#page-30-4) used magnetic abrasive finishing to polish the AISI 316L stainless steel at multiple angles. The surface finish was improved by up to 76%, and all the defects were eliminated. The impacts of ultrasonic abrasive polishing (UAP) on the surface quality of AM parts were examined by Wang et al. [\[97\]](#page-30-5). The Smoothed Particle Hydrodynamics methodology was used to simulate the impact action of abrasive particles. The results showed that UAP could remove partially melted particles, thus reducing the surface roughness from 5.02 to 2.93 μ m. Teng et al. [\[98\]](#page-30-6) investigated the grinding and abrasion processes for AM parts and found that the surface roughness was reduced from 7.0 to $0.15 \mu m$. In the beginning, there are a lot of burrs and pores on the sample's surface that result in high peaks and valleys variations in a provided sample, thus presenting an increased surface roughness value. The grinding and abrasion processes help in removing the burrs and pores within the provided specimen, which ultimately decreases the surface roughness. Guo et al. [\[99\]](#page-30-7) performed an experimental and mathematical investigation on abrasive flow machining to enhance the interior surface quality of Inconel 718. The surface quality was improved up to 56% compared to the as-built surface. Han et al. [\[100\]](#page-30-8) identified the effects of abrasive flow machining on fatigue resistance. It was found that the fatigue resistance was improved up to 26% by using an abrasive flow machining process. Yamaguchi et al. [\[101\]](#page-30-9) observed a significant reduction in surface roughness (100 to 0.1 μ m) and the compressive residual stress formation after performing a sanding, polishing, and ball burnishing on AM AISI 316L stainless steel.

Furthermore, Table [1](#page-21-0) compiles the advantages and disadvantages of conventional machining processes.

Table 1. Advantages and disadvantages of conventional machining processes.

Table 1. *Cont.*

4.4. Heating Processes (HP)

HP can significantly alleviate the residual stresses, lessen cracks and homogenize the microstructure [\[110–](#page-30-18)[113\]](#page-30-19). There are several HPs; however, the commonly used processes are (a) solution heat treatment (SHT), hot isostatic pressing (HIP), and T6-heat treatment (T6-HT). Various studies have been carried out on the effect of HPs on the microstructure and mechanical characteristics of the AM parts [\[34,](#page-27-26)[114](#page-30-20)[–118\]](#page-30-21). The HIP is a typical thermomechanical process that combines high temperature and pressure to eliminate the pores and increase the produced parts' density. The average temperature for this process is between 1000–2000 ◦C. In a closed container with a working pressure of 200 MPa, high-pressure inert gas is used as the pressure medium. The parts are pressed equally in all directions with high temperature and pressure. As a result, the parts produced have a high density, outstanding homogeneity, and exceptional performance [\[119\]](#page-30-22). Short production cycles, low energy usage, maximizing material utilization by increasing material qualities, and allowing for smaller, lighter-weight, high-strength parts are all characteristics of the HIP. The HIP can cure or eradicate intrinsic flaws and porosity in the additive manufactured parts [\[120](#page-30-23)[,121\]](#page-31-0). The HIP has been found to significantly improve the fatigue strength of Ti6Al4V produced by electron beam melting [\[120,](#page-30-23)[121\]](#page-31-0). In these studies, HIP optimized the mechanical qualities of EBM parts. Due to the reduction in porosity and un-melted material, as well as the coarsening of microstructure, the HIP treatment can yield parts having excellent mechanical qualities [\[116\]](#page-30-24).

Goel et al. [\[122\]](#page-31-1) investigated the effects of two different post-processing treatments: (a) HIP and (b) HIP + heat treatment on Inconel 718 alloy manufactured by electron beam melting (EBM). Gas and shrinkage porosity, as well as lack-of-fusion flaws, were identified in the manufactured parts. Figure [24a](#page-22-0) depicts the total defect measured in both as-built and post-treated specimens. Figure [24b](#page-22-0),c show a graphic representation of flaws in the manufactured parts. During printing, the gas was entrapped in the powder, causing the "circular" shaped defects [\[123\]](#page-31-2). During solidification, the shrinkage porosity (SP)

resulted from inter-dendritic shrinking [\[124\]](#page-31-3) and consequently appeared along with the build direction, as shown in Figure [24a](#page-22-0). It is essential to distinguish SP from liquation build direction, as shown in Figure 24a. It is essential to distinguish SF from liquation cracking, as it is usually linked to the secondary phases $[125,126]$ $[125,126]$. SP was identified as an effective mechanism that contributed to the defect formation. Incomplete fusion between effective mechanism that contributed to the defect formation. Incomplete fusion between the melted layers causes lack-of-fusion (LoF). Both the HIP and HIP + heat treatment post-treatments resulted in a significant defects reduction, as seen in Figure [24a](#page-22-0),c. Creep, post deduction becaused in a significant defects reduction, as seen in Figure 24a,c. Creep, and diffusion mechanisms are mainly responsible for defects elimination during HIP [\[127\]](#page-31-6). and diffusion includibility are mainly responsible for defects elimination during HIP [127].
The post-treatment products had only 0.01% defects (Figure [24c](#page-22-0)). $\frac{1}{2}$ is the matrix sultant from $\frac{1}{2}$ and consequently appeared along with the $\frac{1}{2}$ cracking, as it is usually linked to the secondary phases $\frac{1}{2}$, $\frac{1}{2}$, the method layers causes layers cause in \mathbb{R}^n , both the HIP and HIP an The post-treatment products had only 0.01% defects (Figure 24c).

Figure 24. (**a**) Defect content in all the investigated specimens as determined by image analysis, (**b**) **Figure 24.** (a) Defect content in all the investigated specimens as determined by image analysis, (**b**) optical micrograph showing defects in the as-built material, and (**c**) optical micrograph showing defects elimination in the HIP product [\[122\]](#page-31-1); with permission from Elsevier.

Leon et al. [\[128\]](#page-31-7) provided a thorough investigation on the stress-corrosion vulnerabilbility of EBM Ti6Al4V for (a) as-build and (b) after HIP. All the related tests were con-ity of EBM Ti6Al4V for (a) as-build and (b) after HIP. All the related tests were conducted ducted at 3.5 wt.% NaCl solution at room temperature. Figure 25a–f show the macro- and at 3.5 wt.% NaCl solution at room temperature. Figure [25a](#page-23-0)–f show the macro- and micromicro-structure of longitudinal and cross-sections of as-built and HIP-EBM samples. Due structure of longitudinal and cross-sections of as-built and HIP-EBM samples. Due to the epitaxial growth of the parent grains, both specimens had typical columnar microstrucstructures [32]. As indicated in Figure 25a,b, the macro-structure in as-built conditions tures [\[32\]](#page-27-24). As indicated in Figure [25a](#page-23-0),b, the macro-structure in as-built conditions was finer and less uniform than the HIP sample macrostructure. Furthermore, the scanning pass of the electron beam was nearly indistinguishable after the HIP heat treatment, despite the columnar macro-structure being primarily retained. Due to the relatively rapid cool-ing conditions in the EBM process, the microstructure of the as-built sample (Figure [25c](#page-23-0)) 25c) reveals the presence of three main phases: (a) discontinuous grain boundary, (b) fine reveals the presence of three main phases: (a) discontinuous grain boundary, (b) fine αα-Widmanstätten, and (c) α-primary that nucleates ideally. The microstructure of the HIP Widmanstätten, and (c) α-primary that nucleates ideally. The microstructure of the HIP sample (Figure [25e](#page-23-0)) displayed a more extensive and more continuous $\alpha_{g,b}$ -phase, but the primary α-phase and the coarser Widmanstätten α and β structure were retained [14,31]. primary α-phase and the coarser Widmanstätten α and β structure were retained [\[14,](#page-27-6)[31\]](#page-27-23). It is evident from the expanded magnifications of the β-phase in as-built (Figure [25d](#page-23-0)) and HIP samples (Figure [25f](#page-23-0)) that the HIP procedure has resulted in accelerated β-phase growth. According to Dai et al. [\[35\]](#page-27-27), it should improve corrosion resistance and the continuity of the nuity of the β-phase inside the microstructural region. In the HIP sample, agglomeration β-phase inside the microstructural region. In the HIP sample, agglomeration of β-phase significantly reduced the overall area of the interfaces between the α - and β-phases.

Karami et al. [\[129\]](#page-31-8) performed HIP on AM-ed Ti6Al4V lattice structures. During HIP, 920 °C temperature was attained that is just below the β-transus temperature of 996 ◦C, that transformed the acicular-*α*´-martensitic structure into β- and α-phases. Initially, during the HIP process, the α-phase is formed along the *α*´-boundaries. By increasing the α-plates, vanadium withdrew to a nearby regime and β-phase nucleated due to vanadium enrichment. After HIP, the final microstructure consisted of α -platelets embedded in α-/β-grain boundaries [\[130\]](#page-31-9), as shown in Figure [26.](#page-23-1)

Figure 25. Ti6Al4V EBM printing (**a**) as-built macrostructure, (**b**) macrostructure after HIP, (**c**,**d**) as-Figure 25. Ti6Al4V EBM printing (a) as-built macrostructure, (b) macrostructure after HIP, (c,d) as-built microstructure, and (e,f) HIP microstructure [\[128\]](#page-31-7); with permission from Elsevier. $\mathcal{L}_{\text{diff}}$ interconducture, and (\mathcal{L}_{μ}) in Figure

Figure 26. Microstructure and EBSD maps of samples manufactured by (**a**) continuous laser mode **Figure 26.** Microstructure and EBSD maps of samples manufactured by (**a**) continuous laser mode and (**b**) pulses laser mode [\[129\]](#page-31-8); published under open-access license of Elsevier.

Figure 26. Air and EBSD manufactures manufactures continuously continuously continuously continuously continuous solution treatment + ageing, the delta-phases accumulated within the Laves-phases, the solution treatment + ageing, the delta-phases accumulated within the Laves-phases, the micro-segregation decreases. After homogenization + solution treatment + ageing, Laves-
micro-segregation decreases. After homogenization + solution treatment + ageing, Lavespriase annost vanished. The nacture toughness results indicated that as built samples showed the minimum fracture toughness owing to the most negligible elasticity modulus $t_{\rm crit}$ as a built sample samples contained $\mu_{\rm crit}$ and $\mu_{\rm crit}$ and a minor $\mu_{\rm crit}$ and a minor $\mu_{\rm crit}$ and yielding strength. The sample showed the elasticity modulus and yielding strength Yu et al. [\[131\]](#page-31-10) investigated the effect of the post-processing technique on the microstrucstructure and fracture toughness of Inconel 718 produced by AM process. It was found ture and fracture toughness of Inconel 718 produced by AM process. It was found that that as-built samples contained γ-columnar dendrites and a minor quantity of gamma + as-built samples contained γ-columnar dendrites and a minor quantity of gamma + Laves Laves eutectics within inter-dendritic regimes. After using direct ageing, a heat treatment eutectics within inter-dendritic regimes. After using direct ageing, a heat treatment prophase almost vanished. The fracture toughness results indicated that as-built samples

close to the as-built samples using direct gaining. However, the fracture toughness of solution treatment + aging and homogenization + solution treatment + aging samples increased by 56% and 91% compared to as-built samples. Careri et al. [\[132\]](#page-31-11) investigated the effects of two post-processing techniques: (a) thermal post-processing and (b) machining, on stresses formation, microstructure evolution, hardness and surface characteristics of AM parts. Due to the material's higher ductility than as-built parts, the results demonstrated that the strategy of combining heat treatment with machining presented the best machining conditions. In addition, an increment in hardness and a reduction in surface roughness were found.

Brandl et al. [\[133\]](#page-31-12) executed T6-HT at 525 °C for 6 h, water quenching at room temperature, ageing at 165 \degree C for 7 h, and machining the AlSi10Mg parts. The results showed that heat treatment increased the fatigue resistance up to 50%, while a combination of heat treatment at (300 $^{\circ}$ C) and T6-HT enhanced the fatigue resistance by 120% compared to the asbuilt parts. The primary reason for such an extraordinary result was the transformation of dendritic shapes from Si-spheroids that decreased cracks formation. Maamoun et al. [\[113\]](#page-30-19) conducted solution heat treatment for 1 h at 530 ◦C and 5 h at 530 ◦C, and T6-HT for 5 h at 530 ◦C. These processes were performed after annealing (200 ◦C and 300 ◦C). It was found that solution heat treatment and T6-HT elevated the microstructure homogenization. Furthermore, it was noticed that solution heat treatment reduced the hardness by nearly 50% compared to as-built components. However, T6-HT enhanced the sample hardness by 70% compared to as-built components. Gu et al. [\[134\]](#page-31-13) investigated the effects of T6-peakhardening on Al 2219 alloy. They found that this treatment assisted the as-built part in enhancing the yield strength and ultimate strength from 305 to 450 MPa that was much higher than wrought Al 2219 alloy parts. Zhuo et al. [\[135\]](#page-31-14) examined the influence of two annealing temperatures on AlSi10Mg components: (a) 300 °C for 2 h and (b) 535 °C for 1 h. It was detected that the first set of heat treatments was better in reducing the tensile residual stresses from 111 to 13 MPa. Fousova et al. [\[136\]](#page-31-15) described that T6-HT could decrease the strength of AlSi10Mg parts at the verge of material stability improvement. It was suggested that such components could not be used for high-strength applications.

Table [2](#page-24-1) collects the commonly used post-processing techniques identified based on the literature review.

Material Removal Processes Mechanical treatments Machining Polishing **Grinding** • Milling • Conventional • Magnetically driven • Hydrodynamic cavitation Ultrasonic cavitation Chemical treatments • Chemical etching • Chemical polishing No material removal processes Mechanical treatments • Rolling Shot peening Laser-based treatment • Laser shock peening Laser polishing

Table 2. Various post-processing techniques applied in AM.

5. Conclusions and Future Outlook

5.1. Conclusions

This paper has summarized numerous post-processing technologies and their applications in additive manufacturing (AM) processes. In AM, there are various types of defects, including pores, cracks, anisotropy, residual stresses, thermal stresses, laser spattering,

and poor surface roughness. Different post-processing techniques, including laser shock peening, laser polishing, conventional machining, and heating processes, have been explored to resolve these issues. These processes have proved their capability to improve the mechanical properties and reduce the residual stresses formation and surface finish of AM products. Based on this study, the following significant conclusions have been deduced:

- The laser shock peening (LSP) has been applied to the bulk and thinner parts. However, this process leads to severe strain generation in the case of more delicate parts. The percentage of overlap between the two laser scans plays an essential role in controlling the final surface roughness value. LSP is inherited with the local grain refinement phenomenon, resulting in elevated hardness value. Another significant factor is the laser beam intensity and the laser wavelength type that control the surface regularities for a given specimen.
- Besides LSP, laser polishing (LP) plays an essential role in controlling surface roughness and hardness. LP usually acts on the peaks of a given surface, thus increasing the surface reflectivity. Moreover, the number of laser passes also determines a specimen's surface quality. It has been identified that laser polishing can decrease the surface roughness up to 95% concerning the as-built specimen. The hardness value is usually maximum at the top of an LP-ed specimen and declines while travelling from the top to the bottom surface.
- Conventional machining processes have proved their viability for the post-processing of AM parts. They are commonly applied to amplify surface characteristics such as surface roughness and skewness. However, their effect on hardness improvement has not been proved yet. These findings are valid in the case of the milling and turning process. The rolling process has also been used as a post-processing technique. It has been identified that the rolling process significantly improves surface roughness and hardness. Besides the mechanical conventional machining processes, there are various chemical-conventional machining processes. The chemical reagent is usually utilized to improve a given surface's roughness in this category.
- Thermal post-processing techniques have commonly been used to eliminate pores, enhance corrosion resistance, and improve mechanical properties. There are various thermal post-processing techniques: (a) solution heat treatment, (b) hot isostatic pressing, and (c) T6-heat treatment. These techniques involve grain refinement and deposited layers' compactness at elevated temperatures, reducing porosity up to 99.99%.

5.2. Future Outlook

In this section, a few points for the future outlook of post-processing techniques are proposed:

- Various researchers and scientists have carried out efforts to facilitate AM postprocessing commercialization. Each part produced via AM process contains multiple defects. These defects decrease the life expectancy of a produced part and limit its utilization. To prescribe a particular post-processing technique for specific defects, there is a need to establish standardization. These efforts will guarantee a manufactured part's surface integrity and adequate mechanical characteristics, thus avoiding failure during application.
- These post-processing techniques have not been explored for ceramic-reinforced metal matrix composites (CMMCs). In CMMCs, the ceramic particulates are mixed with metallic powder materials and can be used in applications where the properties of ceramics and metal materials are required. CMMCs have elevated hardness and high melting-point and require specific tooling compared to metal materials. CMMCs have gained attention worldwide; hence, their post-processing techniques require special attention from scientists and researchers.
- There is an urgent need for process automation of post-processing techniques in this modern era. These automated solutions can advance production effectiveness. It

can be done via machine or deep learning techniques, useful in process automation, process control, and optimal solutions.

- Process simulations play an essential role in identifying the effect of operating conditions on the final part characteristics. Various simulation techniques for laser shock peening, laser polishing, and conventional machining processes are available. However, the heat treatment simulation techniques have not been well-developed so far. Heat treatment techniques are commonly used in AM process to reduce or eliminate part's porosity that can increase the operational life of the printed components. Developing the simulation models for heat treatment can assist in understanding the multi-physical processes involved while porosity elimination.
- In AM, improving the parts' is a priority; however, reducing the cycle time is also one of the significant challenges. During AM manufacturing, the post-processing techniques consume almost 43% of the total time [\[137\]](#page-31-16). During manufacturing, the in situ control of part's quality can reduce or even eliminate the post-processing technique. It can be done by controlling and optimizing the process at the layer level. Simulation models and in situ monitoring techniques can be developed and applied to understand the AM process at the layer level. It can, in turn, reduce the post-processing technique requirements, thus reducing cycle time and production cost.

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