

Article

Research on Equivalent Static Load of High-Rise/Towering Structures Based on Wind-Induced Responses

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Abstract: A method of assessing equivalent static wind loads that can represent all the real ultimate states of a high-rise building and towering structure has still not been fully determined in wind engineering. Based on random vibration theory, the wind-induced response and equivalent static wind loading of high-rise buildings and towering structures are investigated using the vibration decomposition method. Firstly, the structural wind-induced mean response, background response, resonant response and background and resonant coupled response are studied in the time and frequency domains. Secondly, a new gust load factor (*GLF*) assessment method suitable for wind-induced displacement, bending moment and shear force response at any height of the structure is proposed, and a typical high-rise building is used as an example for comparison with the previous research results, in order to verify the effectiveness of the method in this paper. The results show the following: for high-rise buildings and towering structures, the percentage of the coupled components in the total pulsation response is less than 2%, and the influence can be ignored; the *GLF* based on bending moment (*MGLF*) and the *GLF* based on shear force (*QGLF*) increase significantly with height, and the traditional *GLF* methods underestimate the maximum wind effects.



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Keywords: time domain method; frequency domain method; background and resonance coupled components; wind induced dynamic responses; equivalent static wind load

1. Introduction

Wind load is one of the loads that must be considered in engineering design. For highly flexible, low-damping and light-mass structures, such as high-rise buildings [1,2], bridges [3,4], circular cylinder structures [5,6], wind turbines [7–9], railway catenary [10], cables [11,12] and transmission towers [13], the wind vibration responses are very sensitive and highly susceptible to wind vibration disasters, and determining their wind vibration response and equivalent wind load is one of the core problems of structural wind engineering. Furthermore, the structural wind-induced response is the basis for the study of equivalent static wind loads. In the past, the wind loads of structures were mainly determined directly by anemometers. In the 1960s, Davenport AG [14,15] first introduced random vibration theory into the analysis of the along-wind vibration response of high-rise buildings, divided the structure's total pulsation response into background and resonance components, and proposed the gust load factor method (*GLF*) based on the first-order displacement response, in order to calculate the equivalent static wind load of the structure. Zhang Xiangting [16] argued that the equivalent pulsating wind load on the structure could instead be assessed by the inertial force of the first-order pulsating displacement response, and then proposed the inertial wind load method (*IWL*), which had been adopted into the Chinese code. Kasperski et al. [17,18] proposed an accurate assessment method for the structural background response of low-rise buildings with high stiffness, named the load–response correlation (*LRC*) method. Zhou Yin et al. [19,20] proposed the basal

moment array wind load factor (MGLF) method, which is characteristic of the advantages of LRC and IWL, and has been adopted by the American code. Subsequently, equivalent static wind load assessment methods were proposed by Holmes, Gu Ming, Xie Zhuangning, Lin Yongjun, Ke Shitang and others [21–29]. The above-mentioned methods all take the responses at key locations, such as peak displacement and peak internal force, as the equivalent value, without considering other locations. However, with the development of building shape and height, considerations of only the first-order vibration mode cannot guarantee the security of the structure. On the other hand, with progress in the theory of the wind vibration response calculation, not only the higher-order mode of the wind vibration but also the background and resonant coupling components can be taken into consideration, so there is still a need to investigate the equivalent static wind loads.

The main research methods of structural wind-induced response studies include theoretical analysis, field measurements, wind tunnel experiments and numerical simulations [1,2,5,7,30], and the theoretical analysis can be classified into two types: the time domain method and the frequency domain method. The advantage of the time domain method is that it has a wide range of applications, given its ability to obtain the dynamic response of the displacement, bending moment and shear force, and it can take into account the influence of material's nonlinearity, but the disadvantage is its time-consuming method of calculation. The advantage of the frequency domain method is that the concept is clear and the computational cost is small; the disadvantage is that it cannot consider nonlinear problems. In fact, with the development of computer hardware and frequency-domain computational theory, the disadvantages of the time domain method are gradually being overcome, and the results of the frequency domain method are becoming more and more accurate. Gu Ming et al. [31] proposed an along-wind vibration response analysis method that takes the higher-order modes and inter-modal coupling terms into consideration, and it is based on non-constant load tests performed in the frequency domain. Li Shouke et al. [32] derived a refined assessment method for wind-induced response, considering background, resonance, and background–resonance coupling components, which is based on the combination of the modal acceleration method and the principle of stochastic dynamics two-input single-output system. Zou Lianghao et al. [33] proposed a simplified assessment method for the wind vibration response of high-rise buildings, considering the second-order mode in conjunction with the Chinese code. Zhang Junfeng et al. [34] analyzed the division of the background and resonant wind responses in the time domain, and established a method to calculate the background and resonant coupling components in the time domain.

This paper presents a method for calculating the background component, the resonance component, and the background and resonance coupled component of the wind-driven vibration response of a structure in the time and frequency domains, and presents an equivalent static wind load assessment method for any location on the structure. The method combines the advantages of the traditional MGLF method and the IWL method. The equivalent static wind load at each position of the structure is obtained directly from the wind-induced displacement, the bending moment, and the shear force response at that position, instead of considering the response at a specific position, so that the equivalent static wind load at any position of the structure can be obtained accurately and the result can be more reasonable. The accuracy of the method is also verified using a high-rise building as an example. The work of this paper provides a reference for the study of equivalent static wind load.

2. Refinement of Structural Wind-Induced Response Assessment Method

For a multi-degree-of-freedom structure, the basic equation of motion under wind load is

$$[M]\ddot{Y}(z, t) + [C]\dot{Y}(z, t) + [K]Y(z, t) = P(z, t) \quad (1)$$

where $[M]$, $[C]$ and $[K]$ are the mass matrix, damping matrix and stiffness matrix of the structure, respectively, $P(z, t)$ is the pulsating wind load time course at node z , and $\ddot{Y}(z, t)$, $\dot{Y}(z, t)$ and $Y(z, t)$ are the acceleration time course, velocity time course and total pulsating displacement time course, respectively.

Decoupling the above structural dynamic equilibrium equations yields N mutually independent equations expressed by generalized coordinates. The normalization of the vibration mode in relation to mass gives the following equations:

$$\ddot{q}_j(t) + 2\zeta_j\omega_j\dot{q}_j(t) + \omega_j^2q_j(t) = F_j(t) \tag{2}$$

$$Y(z, t) = \sum_{j=1}^N \phi_j(z)q_j(t) \tag{3}$$

$$F_j(t) = \sum_{i=1}^N P(z_i, t)\phi_{ji} = \sum_{i=1}^N \rho C_D(z) A_i \bar{U}(z_i) u(z, t) \phi_{ji} \tag{4}$$

where $q_j(t)$ is the j th modal generalized coordinate time course, ζ_j is the j th modal damping ratio, ω_j is the j th modal natural vibration angular frequency, $F_j(t)$ is the j th modal generalized load, ϕ_{ji} is the vibration displacement component of the i th node of the j th vibration mode, ρ is the air density, generally taken as 1.25 kg/m^3 , C_D is the quasi-constant wind pressure coefficient, A_i is the windward area of the i th node, $\bar{U}(z_i)$ is the average wind speed of incoming flow at height z_i , and $u(z_i, t)$ is the time course of incoming pulsating wind speed at height z_i .

2.1. The Full Three-Component Expressions in the Frequency Domain

The average response of a high-rise building under wind loads is

$$\bar{r}(z) = \int_0^H \bar{P}(z') i(z, z') dz' \tag{5}$$

where $\bar{P}(z')$ is the average wind load at height z' , and $i(z, z')$ is the influence function (including the influence of displacement, bending moment and shear force), which represents the response generated at position z when a unit load is applied at position z' [23].

The frequency domain method is mostly used for the analysis of the structural wind vibration response, given its advantages of clear physical concepts and speed. When the wind speed at one spatial point reaches a maximum, it usually does not reach a maximum at another point in a certain range at the same time. Therefore there is a certain spatial correlation between the wind speeds at two points in the space [35]. From Equation (4), the structure's generalized wind load mutual spectrum $S_{FiFj}(f)$ can be expressed as:

$$S_{FiFj}(f) = \sum_{k=1}^N \sum_{l=1}^N \phi_{ik}\phi_{jl}\rho^2 C_D^2 A_k \bar{U}_k A_l \bar{U}_l R_X R_Z \sqrt{S_{u_k}(f) S_{u_l}(f)} \tag{6}$$

where R_X, R_Z are the horizontal and vertical coherence function, \bar{U}_i is the average wind speed at the i th node, and $S_{u_i}(f)$ is the pulsating wind speed spectrum.

According to Equation (2), if the generalized wind load mutual spectrum $S_{FiFj}(f)$ is known, then the total pulsation response spectrum $S_{q_j}(f)$ of the modal generalized coordinates q_j can be expressed as Equation (7), and then using the modal superposition principle (Equation (3)), the response spectrum $S_{y_i}(f)$ at node i of the structure can be obtained, as in Equation (8).

$$S_{q_j}(f) = \sum_{i=1}^N S_{FiFj}(f) H_i^*(f) H_j(f) \tag{7}$$

$$S_{yi}(f) = \sum_{j=1}^N \sum_{k=1}^N \phi_{ji} \phi_{ki} H_j^*(f) H_k(f) S_{FjFk}(f) \tag{8}$$

$$H_k(f) = \frac{1}{K_k^*} \cdot \frac{1}{1 - (f/f_k)^2 + i2\zeta_k(f/f_k)} \tag{9}$$

where $H_k(f)$ is the k th modal complex frequency response function, and $H_j^*(f)$ is the conjugate of $H_j(f)$. The response spectra of the bending moment and shear force can be obtained by replacing the vibration displacement ϕ in Equation (8) with bending moment or shear force.

$$\sigma_{Ti} = \left(\int_0^\infty |S_{yi}(f)| df \right)^{1/2} \tag{10}$$

The total pulsation response σ_{Ti} can be calculated according to Equation (10), and it contains three components: the background component σ_B , the resonance component σ_R , and the background–resonance coupling component σ_{BR} . Because of the complicated nature of the calculation for σ_T , researchers often choose to determine each component, and then calculate σ_T according to Equation (11). The expressions for each component are as follows.

$$\sigma_{Ti} = (\sigma_{Bi}^2 + \sigma_{Ri}^2 + \sigma_{BRi}^2)^{1/2} \tag{11}$$

$$\hat{\sigma}_{Bi} = g_u \left(\int_0^\infty \int_0^H \int_0^H \int_0^W \int_0^W \frac{p(x_1, z_1, f) p(x_2, z_2, f) i(z_i, z_1) i(z_i, z_2) dx_1 dx_2 dz_1 dz_2 df}{p(x_1, z_1, f) p(x_2, z_2, f)} \right)^{1/2} \tag{12}$$

$$\hat{\sigma}_{Ri} = g_R \left(\int_0^\infty \sum_{j=1}^N \sum_{k=1}^N \phi_{ji} \phi_{ki} S_{FjFk}(f) [H_j^*(f) - \frac{1}{k_j^*}] [H_k^*(f) - \frac{1}{k_k^*}] df \right)^{1/2} \tag{13}$$

$$\hat{\sigma}_{BRi} = 2\rho_{BR} \left(\int_0^\infty \sum_{j=1}^N \sum_{k=1}^N \phi_{ji} \phi_{ki} S_{FjFk}(f) \left[\frac{1}{k_j^*} H_k^*(f) + \frac{1}{k_k^*} H_j^*(f) - \frac{2}{k_j^* k_k^*} \right] df \right)^{1/2} \tag{14}$$

$$\overline{p(x_1, z_1, f) p(x_2, z_2, f)} = (\rho C_D \bar{U}_H)^2 \left(\frac{z_1}{H}\right)^\alpha \left(\frac{z_2}{H}\right)^\alpha R_Z R_X S_u(f) \tag{15}$$

$$g_R = \sqrt{2 \ln(f_1 T)} + 0.5772 / \sqrt{2 \ln(f_1 T)} \tag{16}$$

$$\rho_{BR} = \sigma_{BR}^2 / (\sigma_B \sigma_R) \tag{17}$$

where $\overline{p(x_1, z_1, f) p(x_2, z_2, f)}$ is the pulsating wind pressure mutual spectrum at the structure’s surface points (x_1, z_1) and (x_2, z_2) , g_u is the peak wind speed factor, generally taken as 3.5, g_R is the peak factor of the resonant response, and ρ_{BR} is the background resonant coupled response mutual relationship number [32].

The above equation is the full three-component expression of the structural wind response in the frequency domain, and it can be seen that the expressions for σ_R and σ_{BR} are quite complicated, and are not convenient for engineering applications. As high and flexible structures such as high-rise buildings and towering structures have a sparse natural frequency, the wind vibration response of such structures is mainly related to the first few modes, so the simplification of the calculation of such buildings can be carried out by ignoring the contribution of the coupling term between each structural vibration mode and the coupling terms of background and resonance. At this point, the total pulsation response of the structure can be expressed via the following equation.

$$\sigma_T = \left(\sigma_B^2 + \sum_j^n \sigma_{Rj}^2 \right)^{1/2} \tag{18}$$

$$\sigma_{Rj} = g_R \left(S_{Fj}(f_j) \int_0^\infty |H_j(f)|_2 df \right)^{1/2} = g_R \frac{1}{K_j^*} \left(S_{Fj}(f_j) \frac{\pi f_j}{4\zeta_j} \right)^{1/2} \varphi_j \quad (19)$$

2.2. The Full Three-Component Expressions in Time Domain

$Y(z, t)$ can be obtained by the direct solution of the above dynamic equilibrium equation. From the concept and development of the background and resonant responses, it is known that the background and resonant responses can be calculated by the following equation:

$$Y_B(z, t) = P(z, t) / [K] \quad (20)$$

$$Y_R(z, t) = Y(z, t) - Y_B(z, t) \quad (21)$$

After obtaining the $Y(z, t)$, $Y_B(z, t)$ and $Y_R(z, t)$, the variances of the respective responses σ_Y^2, σ_B^2 and σ_R^2 can be calculated. It is important to note that there must also be coupling effects in the vicinity of the structural natural frequency, since both $Y_B(z, t)$ and $Y_R(z, t)$ contain components of this frequency [34].

$$\sigma_{BR} = \sigma_T^2 - \sigma_B^2 - \sigma_R^2 \quad (22)$$

So far, the assessment methods for the structural background component, resonance component and background–resonance coupling component in the time domain wind response have been elaborated. The time domain calculation can directly give a dynamic response, such as displacement, bending moment and shear force.

2.3. Comparison of Time–Frequency Domain Results for the Full Three Components

In this section, the wind-induced vibration response of a single-pole lightning rod structure will be studied to compare the frequency domain method and time domain method. The study will include displacement, bending moment and shear force. A lightning rod is a typical towering structure with a small natural frequency and significant wind-induced vibration effect, which often causes engineering accidents [36]. The physical parameters of the lightning rod investigated in this section are: total height 60 m, tower tip height 2 m, tower height 58 m. It is composed of seven sections of variable-thickness thin-walled steel pipe, with a diameter from 1.5 m tapered to 0.15 m, a wall thickness from 16 mm tapered to 5 mm; the material is Q345 steel, and the steel pipes are connected by plug-in adder flange. The steel pipe size is shown in Figure 1b. The simulation was carried out in the ANSYS finite element software, using Beam188 unit, and each of the lower six sections of steel pipe were divided into five units; each upper section was equivalent to 1 unit, and a total of 31 units were used in the model. In order to simplify the calculation, it does not include the tip of the tower, and does not consider the overlap areas between the steel pipes; the steel cylinder is only the interface connection and rigid connection. The five orders of vibration pattern for structure X-direction are shown in Figure 1c, with frequencies of 0.75, 2.09, 4.31, 7.46 and 11.66 Hz. Only the along-wind wind-induced response is analyzed in the simulation.

The lightning rod operates in a Class B site with a basic wind speed $U_{10} = 35.8$ m/s, wind profile power index $\alpha = 0.15$, turbulence $I_{10} = 0.14$, and air density $\rho = 1.25$ kg/m³. The section drag coefficient C_D takes the value of 0.6 according to the load code [37], the Davenport wind spectrum was considered in this case, and the Davenport coherence function exponential decay coefficient is $C_Z = 7$, $C_X = 8$. The harmonic superposition method is used to simulate the pulsating wind load time course, and the other parameters are selected according to the specifications. Rayleigh damping is used for the simulation and the modal damping ratio is taken as $\zeta = 0.01$.

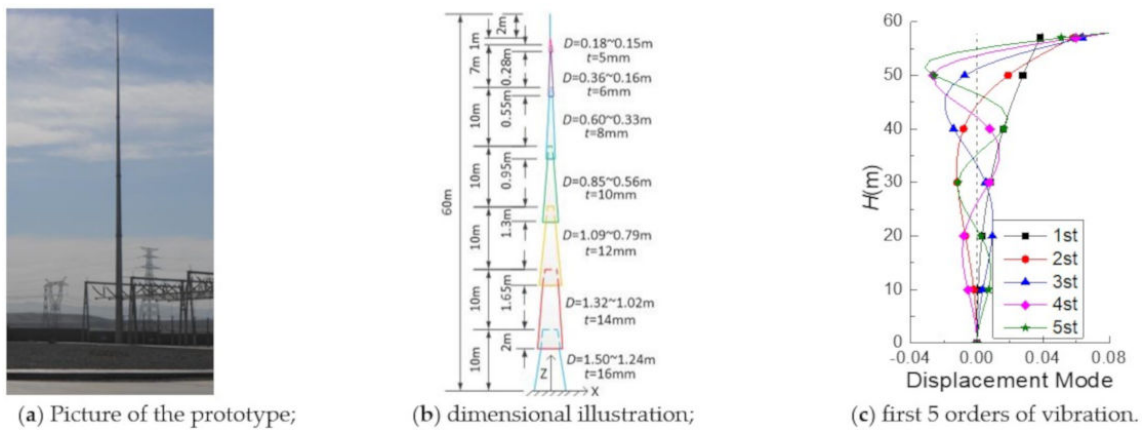


Figure 1. Geometry parameters and the FE model of the lightning rod.

Figure 2 shows the curve of the displacements, bending moments and shear forces (σ_T , σ_B and σ_R , respectively), obtained by the frequency domain method and the time domain method, as well as the percentage of the coupled components in the total pulsation response obtained by the time domain method. As shown in the picture, the σ_T values obtained from the frequency domain and time domain methods are in good agreement, and the deviations in the displacement of the tower tip, the base bending moment and the base shear force are all within 1.5%. As regards the displacement and bending moment responses, the background response obtained by the frequency domain method is smaller, and the resonance response is larger, than those obtained by the time domain method. The percentage of σ_{BR} in the total pulsation response for each variable is less than 2%, so the effect of neglecting σ_{BR} on the total pulsation response is small.

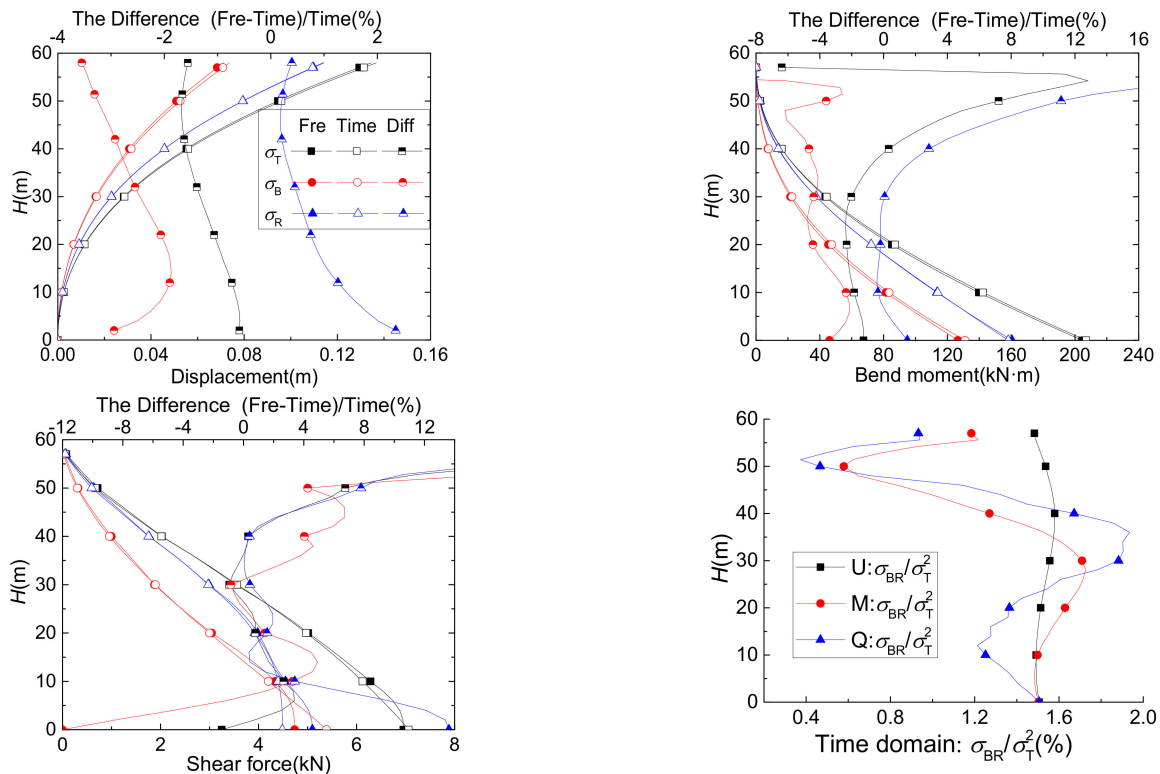


Figure 2. Comparison of time/frequency domain results and the percentage of coupling components.

3. The Assessment Method for the Equivalent Static Wind Load of Arbitrary Structural Response

Theoretical Analysis

As in previous studies, for high-rise buildings and towering structures, this paper still assumes that the equivalent static wind load on the structure is equal to the product of the mean wind load and the gust load factor, which can be expressed as follows:

$$\hat{P}(z) = GLF(z) \cdot \bar{P}(z) \tag{23}$$

$$GLF(z) = 1 + \frac{\sigma_T(z)}{\bar{r}(z)} = 1 + \sqrt{\frac{\sigma_B(z)^2}{\bar{r}(z)^2} + \sum_i^n \frac{\sigma_{Ri}(z)^2}{\bar{r}(z)^2}} = 1 + \sqrt{G_B(z)^2 + \sum_i^n G_{Ri}(z)^2} \tag{24}$$

$$\sigma_B(z) = g_u \left(\int_0^\infty \int_0^H \int_0^H \int_0^W \int_0^W \overline{p(x_1, z_1, f)p(x_2, z_2, f)i(z, z_1)i(z, z_2)} dx_1 dx_2 dz_1 dz_2 df \right)^{1/2} \tag{25}$$

$$\sigma_{Rj}(z) = \int_0^H \hat{P}_{Rj}(z') i(z, z') dz' \tag{26}$$

$$\begin{aligned} \hat{P}_{Rj}(z) &= m(z)(2\pi f_j)^2 \sigma_{YRj} \varphi_j(z) \\ &= g_R \left(S_{Fj}(f_j) \frac{\pi f_j}{4\zeta_j} \right)^{1/2} \varphi_j(z) \\ &= g_R \left(\sum_{k=1}^N \sum_{l=1}^N \phi_{ik} \phi_{jl} \rho^2 C_D^2 A_k \bar{U}_k A_l \bar{U}_l R_X R_Z \sqrt{S_u(f_j) S_u(f_j) \frac{\pi f_j}{4\zeta_j}} \right)^{1/2} \varphi_j(z) \end{aligned} \tag{27}$$

where $\hat{P}_{Rj}(z)$ is the j th order modal resonant equivalent wind load at height z of the structure, and its value is equal to the inertia force caused by the j th order modal resonant displacement [38].

The background response $\sigma_B(z)$ and resonant response $\sigma_R(z)$ can be obtained according to Equations (25) and (26), respectively, and then bringing them into the GLF expression (Equation (24)) can give the expressions of the $G_B(z)$ and $G_R(z)$ at any position on the structure.

The background gust load factor $G_B(z)$ of the structure's response when under wind load at any location is

$$\begin{aligned} G_B(z) &= \frac{\sigma_B(z)}{\bar{r}(z)} \\ &= \frac{g_B \sqrt{\int_0^\infty \int_0^H \int_0^H \int_0^W \int_0^W \overline{p(x_1, z_1, f)p(x_2, z_2, f)i(z, z_1)i(z, z_2)} dx_1 dx_2 dz_1 dz_2 df}}{\int_0^H \bar{P}(z') i(z, z') dz'} \\ &= \frac{2g_B I_H H^\alpha}{\int_0^h z'^{2\alpha} i(z, z') dz' + \int_h^H z'^{2\alpha} i(z, z') dz'} \sqrt{\int_0^\infty S_u^*(f) |K_Z(\alpha, z, f)|^2 |J_X(f)|^2 df} \end{aligned} \tag{28}$$

$$\begin{aligned} |K_Z(\alpha, z, f)|^2 &= \int_0^z \int_0^z z_1^\alpha z_2^\alpha R_Z(z_1, z_2, f) i(z, z_1) i(z, z_2) dz_1 dz_2 \\ &\quad + \int_0^z \int_z^H z_1^\alpha z_2^\alpha R_Z(z_1, z_2, f) i(z, z_1) i(z, z_2) dz_1 dz_2 \\ &\quad + \int_z^H \int_0^z z_1^\alpha z_2^\alpha R_Z(z_1, z_2, f) i(z, z_1) i(z, z_2) dz_1 dz_2 \\ &\quad + \int_z^H \int_z^H z_1^\alpha z_2^\alpha R_Z(z_1, z_2, f) i(z, z_1) i(z, z_2) dz_1 dz_2 \end{aligned} \tag{29}$$

Following the method of GLF based on displacement (DGLF), Davenport [15] defined the integral term in the structural force spectrum with respect to height z as a vertical joint receiver function. To make the above expression formally identical to the DGLF method,

this paper also draws on this idea, and defines the integral term with respect to height z as a new function (Equation (29)), but the difference is that we use four integral terms.

The j th order modal resonant gust wind load factor $G_{Rj}(z)$ of the structure, in terms of all responses at any location under wind load, is

$$G_{Rj}(z) = \frac{\sigma_{Rj}(z)}{\bar{r}(z)} = \frac{\int_0^H \hat{P}_{Rj}(z')i(z, z')dz'}{\int_0^H \bar{P}(z')i(z, z')dz'} \tag{30}$$

$$= \frac{g_R \int_0^H \left(\sum_{k=1}^N \sum_{l=1}^N \phi_{ik}\phi_{jl}\rho^2 C_D^2 A_k \bar{U}_k A_l \bar{U}_l R_X R_Z \sqrt{S_u(f_j) S_u(f_j) \frac{\pi f_j}{4\zeta_j}} \right)^{1/2} \varphi_j(z)i(z, z')dz'}{\frac{1}{2}\rho \bar{U}_H^2 C_D W \int_0^H \left(\frac{z'}{H} \right)^{2\alpha} i(z, z')dz'}$$

According to Equations (28) and (30), $G_B(z)$ and $G_R(z)$ can be determined by the height z and the influence function $i(z, z')$.

4. Example and Analysis

Yin Zhou [20] used a high-rise building as an example to illustrate the correctness of their proposed method of basal bending moment gust wind load factor (MGLF). In the following, the *GLF*, G_B and G_{Rj} of the same structure are calculated according to the method given in this paper. The physical parameters of the structure are: dimensions $H \times W \times D = 200 \times 50 \times 40$ m, natural frequency $f_1 = 0.22$ Hz; damping ratio $\zeta = 0.01$; first order vibration mode $\varphi_1(z) = (z/H)^\beta$; mass distribution $m(z) = m_0(1 - \lambda(z/H))$; $m_0 = 5.5 \times 10^5$ kg/m; section drag coefficient $C_D = 1.3$. The wind environment parameters are: fundamental wind speed $\bar{U}_{10} = 30$ m/s; wind profile power index $\alpha = 0.15$; and turbulence $I_{10} = 0.2$, while the wind spectrum type is Davenport wind spectrum, and the coherence function parameter $C_X = C_Z = 11.5$. The following four operating conditions are obtained by adjusting the values of the vibration index β and the mass discount factor λ , respectively (Table 1).

Table 1. The values of the parameters for the four working conditions [20].

Condition	Vibration Index β	Mass Discount Factor λ
1	1.0	0.0
2	1.6	0.0
3	1.0	0.2
4	1.6	0.2

Zhou Yin (2001) calculated the values of *GLF*, G_B and G_{R1} via the DGLF method proposed by Davenport, as well as the QGLF method, and the MGLF method, which was proposed by him. The DGLF method directly takes the first-order vibration mode as linear, i.e., the vibration index $\beta = 1$ (of course, the DGLF method can also obtain results when $\beta \neq 1$), but the DGLF method itself cannot consider the mass discount of the structure along with the height. Therefore, Zhou could only derive the result of one of the above four conditions using the DGLF method. The MGLF method also divides the structure σ_T into two parts: the σ_B is calculated directly based on the wind load and influence function, and the σ_R is calculated by the indirect method, which distinguishes the influence of β and λ , so the results of the four conditions are different. The results are presented in reference [20].

According to Equations (24), (28) and (29), the *GLF*, G_B and G_{Rj} values at arbitrary positions are calculated, and when the height is taken as 0 or H , the *GLF*, G_B and G_{R1} values at these positions can also be obtained—the results are shown in Table 2. The values in parentheses are the results relative to the results of Zhou Yin [20]. It can be seen that the *GLF* values of the base bending moments obtained by the method proposed in this paper are identical to those obtained by Zhou Yin [20]. The values of base shear force *GLF* and top displacement *GLF* deviate slightly, but the maximum deviation is only 0.6%, meaning they are consistent with each other. The expressions of the basal bending

moment GLF and top displacement GLF obtained according to the method of this paper are identical to those of Zhou Yin [20]. The deviations in the result may be caused by numerical integration. Due to the different values derived for the respective integration step and the upper integration limit when numerical integration is performed, the integration results will be somewhat deviant. The specific values of these parameters are not given by Zhou Yin [20]. The first-order frequency of this high-rise building is $f_1 = 0.22$ Hz; the trapezoidal integration formula is used in this paper for the numerical integration calculation, and the frequency integration step is $df = 1/1000$ Hz, while the upper integration limit is $f_{MAX} = 6$ Hz. Other values are also taken ($f_{MAX} = 1$ Hz, 3 Hz and 9 Hz; $df = 1/500$ Hz and $1/2000$ Hz) for verification to ensure the calculation's accuracy. The results show that the resulting G_B decreases slightly when $df = 1/500$ Hz compared to $df = 1/1000$ Hz, but it remains unchanged when $df = 1/2000$ Hz, and the resulting G_B decreases slightly when $f_{MAX} = 1$ Hz and 3 Hz, compared to $f_{MAX} = 6$ Hz, while it remains unchanged when $f_{MAX} = 9$ Hz, which indicates that the df and f_{MAX} used in the numerical integration are reasonable.

Table 2. GLF of feature locations obtained in this paper.

Condition	Top Displacement			Basal Bend Moment			Basal Shear Force		
	G_{YB}	G_{YR}	$DGLF$	G_{MB}	G_{MR}	$MGLF$	G_{QB}	G_{QR}	$QGLF$
1	0.6520 (1.000)	0.9761 (0.998)	2.1738 (0.999)	0.6520 (1.000)	0.9761 (1.000)	2.1738 (1.000)	0.6560 (0.994)	0.8275 (1.002)	2.0560 (1.000)
2	0.6591	1.0302	2.2230	0.6520 (1.000)	0.9532 (1.000)	2.1549 (1.000)	0.6560 (0.994)	0.7460 (1.003)	1.9934 (0.999)
3	0.6520	0.9761	2.1738	0.6520 (1.000)	0.9761 (1.000)	2.1738 (1.000)	0.6560 (0.994)	0.8438 (1.001)	2.0688 (0.999)
4	0.6591	1.0302	2.2230	0.6520 (1.000)	0.9589 (1.000)	2.1596 (1.000)	0.6560 (0.994)	0.7612 (1.002)	2.0049 (1.000)

The values of the GLF of the tip displacement and its components G_{YB} and G_{YR} , obtained under the four working conditions, are the results for condition 1. Since the effects of β and λ can all be considered when calculating the displacement response using the method in this paper, different results can be obtained under each of the four conditions, and because of that, the results under conditions 2, 3 and 4 will not be compared.

From Table 2, it can be seen that under each working condition, the top displacement GLF is not less than the base bending moment GLF , with a maximum deviation of about 3.1%, and the base bending moment GLF is greater than the base shear force GLF , with a maximum deviation of 7.5%. The σ_B values of the base bending moment and shear force are obtained directly from the load and influence function (Equation (25)), which does not contain parameters β and λ ; therefore, the bending moment and shear force G_B are not affected by β and λ , and G_B is constant under all four conditions. Further, the base bending moment G_B is slightly smaller than the base shear force G_B , since the displacement influence function is obtained according to the assumption of the first-order vibration mode, and G_{YB} increases with the increase of β . Since the σ_R values of the base bending moment and shear force are obtained according to the resonant displacement inertia force $\hat{P}_R(z')$ and the influence function (Equation (26)), and $\hat{P}_R(z')$ contains the effects of β and λ , the σ_R response of each condition is not same, and both G_R values decrease with the increase in β . For the displacement of σ_R , although both $\hat{P}_R(z')$ and the influence function contain the parameters β and λ , the parameter λ can be removed, so G_{YR} is not affected by parameter λ , and only increases with the increase in β . The G_{MR} decreases with the increase in β , but remains unchanged or increases when λ increases from 0 to 0.2, which is mainly because λ can be approximately removed when $\beta = 1$. When only the first-order vibration mode is considered, the G_{QR} of base shear force decreases with the increase in β , and increases with the increase in λ .

Taking the above high-rise building as an example, the *GLF* values of each response at any position of the structure under different working conditions are obtained, and the different *MGLF* values and *QGLF* values relative to *DGLF* for each working condition are given in terms of *DGLF* values. The results are shown in Figure 3.

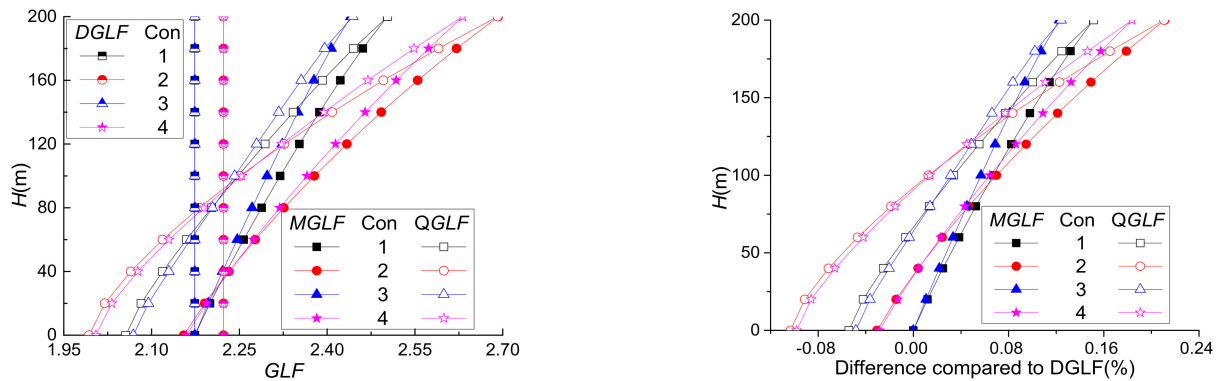


Figure 3. *GLF* values obtained by the method in this paper and their difference rates.

From Figure 3, it can be seen that with the increase in building height, the *MGLF* and *QGLF* corresponding to the four working conditions increase significantly, and the *DGLF* remains unchanged along the height. This is mainly because this paper uses the influence function based on the first-order vibration mode to calculate the *DGLF*, so the total pulsation response and the average response are in the form of the first-order vibration mode, and for *MGLF* and *QGLF*, the increase rate increases with the increase in β and decreases with the increase in λ . Under the same working conditions, the *MGLF* is larger than both the *QGLF* and the *DGLF*, and only the value of *DGLF* at the basal level is larger (slightly) than the *MGLF*, but the deviation is within 3%. Therefore, if *DGLF* and *QGLF* are used as the equivalent wind load in the structural design, the moment response of the structure may be underestimated; however, the use of the *MGLF* value for the equivalent wind load calculation is more important, because the design of the structure is biased towards safety, which is corroborated by the popular idea of using the bending moment as the control parameter for towering structures. Under the same working conditions, the basal *QGLF* value is the smallest, but as the building's height increases, the *QGLF* value gradually increases, and when the height is above 100 m, the *QGLF* under each working condition is greater than the *DGLF*. As the building's height increases, the difference between the *MGLF* and *QGLF* gradually decreases, until the tops tend to be equal, and the *MGLF* is greater than the *DGLF*. Until the top is reached, the difference between the *MGLF* value and the *DGLF* value gradually increases, and the maximum deviation between the *MGLF* and *QGLF* is about 22%, compared with *DGLF*. The position of maximum deviation is the top of the structure, so the wind effect of the structure will be seriously underestimated when the wind load design is based only on the base moment *GLF*.

5. Conclusions

This paper presents an assessment method for the equivalent static wind load based on random vibration theory and quasi-steady aerodynamic theory, and investigated a method for analyzing the background response, the resonant response, and the coupling components between these two responses, to wind vibration in the time domain and the frequency domain. The main conclusions are as follows:

(1) The displacement and bending moment responses obtained by the time domain and frequency domain methods are in good agreement, and the deviation in the base shear force response of the structure is slightly increased. However, the deviation in the total pulsation response is within 1.5%, and the percentage of σ_{BR} in the total pulsation response of each response is within 2%, so the influence of neglecting σ_{BR} on the total pulsation response is small;

(2) The *MGLF* and *QGLF* obtained by the new assessment method in this paper increase significantly with height, while the traditional *GLF* methods underestimate the maximum wind effects. The *DGLF* is constant as the height increases, and the *MGLF* at the top of the structure is about 22% larger than the *DGLF*. At this time, if only the *DGLF* or the base location *MGLF* values are considered, the wind effect of the structure will be seriously underestimated. The *DGLF*, *MGLF* and *QGLF* values increase as the β increases, and the *MGLF* and *QGLF* values decrease as the λ increases;

(3) The differences between the *GLF* obtained in this paper and the traditional gust load factor method are compared through the case of a high-rise building, and the validity of this paper's method is verified. The *GLF* assessment method of this paper considers the higher-order vibration pattern of the structure and the whole wind-induced response, including displacement, bending moments and shear forces. It has a wide range of applications.

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