

Article

Region-Searching of Multiple Autonomous Underwater Vehicles: A Distributed Cooperative Path-Maneuvering Control Approach

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Abstract: In this article, a distributed cooperative path-maneuvering control approach is developed for the region-searching of multiple autonomous underwater vehicles under both dynamic uncertainties and ocean currents. Salient contributions are as follows: (1) by virtue of boustrophedon motions and trigonometric functions, the coverage path-planning design is first proposed to generate multiple parameterized paths, which can guarantee that the region-searching is successfully completed by one trial; (2) combining with sliding mode and adaptive technique, distributed maneuvering control laws for surge and yaw motions are employed to drive vehicles to track the assigned paths, thereby contributing to the cooperative maneuvering performance with high accuracy; (3) by the aid of graph theory, the distributed signal observer-based consensus protocols are developed for path parameter synchronization, and successfully apply to maintain the desired formation configuration. The globally asymptotical stability of the closed-loop signals is analyzed via the direct Lyapunov approach, and simulation studies on WL-II are conducted to illustrate the remarkable performance of the proposed path-maneuvering control approach.

Keywords: autonomous underwater vehicles; region-searching; distributed path-maneuvering control; adaptive sliding mode; consensus protocols

1. Introduction

Autonomous underwater vehicles (AUVs), characterized by small size, autonomy and intelligence [\[1](#page-15-0)[,2\]](#page-15-1), have become an indispensable platform in various engineering applications, such as demining, searching sunken ships and resources [\[3](#page-15-2)[,4\]](#page-15-3). In particular, the path-maneuvering control of AUVs has received widespread attention in both marine engineering and control community [\[5](#page-15-4)[–7\]](#page-15-5). However, AUVs in practice inevitably suffer from dynamic uncertainties and/or ocean currents. As an effective remedy of this problem, the robust control is extensively investigated such as adaptive dynamic surface control [\[8](#page-15-6)[,9\]](#page-15-7), sliding mode control [\[10–](#page-15-8)[12\]](#page-15-9), and finite-time control [\[13,](#page-15-10)[14\]](#page-16-0). Besides, to reconstruct and compensate dynamic uncertainties, neural network- [\[15](#page-16-1)[,16\]](#page-16-2) and fuzzy logic system- [\[17,](#page-16-3)[18\]](#page-16-4) based state observers have also received increasing attention.

The individual vehicle system inevitably poses a threat to large-range and long-time engineering applications due to its restricted payloads and communications. In this context, the cooperative path-maneuvering control (CPMC) approaches of multiple vehicles have been significantly developed, such as the leader follower [\[19](#page-16-5)[,20\]](#page-16-6), the consensus [\[21,](#page-16-7)[22\]](#page-16-8), and others [\[23](#page-16-9)[,24\]](#page-16-10). Obviously, the cooperative system stimulates agents to collaborate with their neighbors and thus completes the assigned mission, which can provide higher efficiency and effectiveness than the individual system [\[25–](#page-16-11)[27\]](#page-16-12). By virtue of information interactions, the cooperative path-maneuvering is expected to drive multi-vehicles along

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predesigned paths with a predefined formation configuration [\[28](#page-16-13)[,29\]](#page-16-14). In [\[30\]](#page-16-15), a decentralized path-maneuvering control approach using Lyapunov technique and graph theory was proposed for multiple vehicles in the presence of discrete-time periodic communications. In [\[31\]](#page-16-16), the proper interaction laws consisting of collision avoidance, velocity matching and flocking centering are employed using potential functions, where multiple vehicles are recognized as an integral whole. Taking environment disturbances into consideration, a robust cooperative path-maneuvering controller is employed by combining with adaptive neural network and dynamic surface control [\[32\]](#page-16-17). In [\[33\]](#page-16-18), a decentralized speed adaptive technique for cooperative path-maneuvering of AUVs was developed, where minimized communication variables are designed. It should be noted that this approach depends on the assumption that the desired speed signal is available for each AUV and thus enhances system communication burden. Fortunately, a consensus-based adaptive maneuvering control law within a bidirectional communication topology is developed to identify the orbital speed signal [\[34\]](#page-16-19), thereby removing the restriction in [\[33\]](#page-16-18). However, the abovementioned paths must be constructed by a set of convex and closed orbits, thereby partly limiting the maneuverability and flexibility of AUVs.

Another problem related to the CPMC is the path-planning design. By virtue of artificial potential functions [\[35\]](#page-16-20), genetic algorithms [\[36\]](#page-16-21), and particle swarm optimization [\[37\]](#page-16-22), fruitful path-planning approaches have been developed whereby vehicles can autonomously perform various missions. In [\[38\]](#page-16-23), a multiple sub-target artificial potential function field is presented where the global optimal path is generated using a heuristic A* algorithm. In [\[39\]](#page-16-24), a smooth Bezier path is generated in the presence of the end point constraints. In [\[40\]](#page-16-25), a multilayer path planner with collision avoidance and path correction is proposed by utilizing the B-Spline and Poisson distribution. However, the path-planning problem associated with cooperative region-searching of multiple AUVs is still largely open.

In this article, we focus on the unresolved problem of cooperative region-searching for multiple under-actuated AUVs. A distributed cooperative path-maneuvering control (DCPMC) approach including coverage path-planning (CPP) design, individual pathmaneuvering control (IPMC) design, and path parameter synchronization (PPS) design is developed for AUVs under dynamic uncertainties and ocean currents. To ensure complete coverage for the special region, a novel CPP method is proposed where multiple smooth paths are exclusively generated by combining with boustrophedon motions and trigonometric functions. Within the IPMC design, distributed adaptive sliding control laws are developed for AUVs to track the assigned paths despite the existence of dynamic uncertainties and ocean currents. Within the PPS design, the neighbor states-based consensus protocol is developed to synchronize path parameters. Besides, a distributed signal observer is designed using graph theory and thus relaxes the condition requiring the known speed in [\[34\]](#page-16-19). Moreover, via theoretical derivation and the Lyapunov direct approach-based stability analysis, we show that the resulting closed-loop signals are globally asymptotically stable (GAS).

The remainder of this article is organized in sections. Section [2](#page-1-0) describes the AUV model and the problem of cooperative path-maneuvering. Section [3](#page-3-0) presents the DCPMC approach and provides the stability analysis. Section [4](#page-10-0) gives simulation studies for illustrations. Finally, in Section [5,](#page-15-11) the conclusions of this article are reviewed.

2. Problem Statement

2.1. AUV Model

As shown in Figure [1,](#page-2-0) two types of reference frames, namely, the body-fixed frame and the earth-fixed frame, are widely used to describe AUV motions. Consider a cooperative system consisting of *N* under-actuated AUVs. Following the reference [\[41\]](#page-16-26), the 3 degree-offreedom (DOF) kinematics of the *i*th AUV is written as

$$
\dot{\eta}_i = R(\psi_i)v_i,\tag{1}
$$

in which $\eta_i \in R^3$ is a position state vector of coordinate (x_i, y_i) and heading angle ψ_i of the *i*th AUV in the earth-fixed frame; $v_i \in R^3$ is a velocity vector of surge velocity u_i , sway velocity v_i , and yaw angular velocity r_i in the body-fixed frame; $R(\psi_i)$ is a 3-DOF rotation matrix defined as

$$
R(\psi_i) = \begin{bmatrix} \cos \psi_i & -\sin \psi_i & 0\\ \sin \psi_i & \cos \psi_i & 0\\ 0 & 0 & 1 \end{bmatrix}
$$
 (2)

with properties as follows: $R^T(\psi_i)R(\psi_i) = I$, $||R(\psi_i)|| = 1$, and $R(\psi_i) = R(\psi_i)S(r_i)$, where

$$
S(r_i) = \begin{bmatrix} 0 & -r_i & 0 \\ r_i & 0 & 0 \\ 0 & 0 & 0 \end{bmatrix}.
$$
 (3)

Under complex uncertainties including dynamic uncertainties and time-varying ocean currents, the dynamics is given by

$$
(\hat{M}_i + \tilde{M}_i)\dot{v}_i = -(\hat{C}_i(v_i) + \tilde{C}_i(v_i))v_i - (\hat{D}_i(v_i) + \tilde{D}_i(v_i))v_i + \tau_i + d_i,
$$
\n(4)

in which $\tau_i = [\tau_{ui}, 0, \tau_{ri}]^T \in R^3$ is the control input vector, and $d_i = [d_{ui}, d_{vi}, d_{ri}]^T \in R^3$ denotes ocean currents. The inertia matrix $\hat{M}_i \in R^{3 \times 3}$, the coriolis matrix $\hat{C}_i(v_i) \in R^{3 \times 3}$, and the damping matrix $\hat{D}_i(v_i) \in R^{3 \times 3}$ are given by

$$
\hat{M}_{i} = \begin{bmatrix} \hat{m}_{11} & 0 & 0 \\ 0 & \hat{m}_{22} & 0 \\ 0 & 0 & \hat{m}_{33} \end{bmatrix}, \hat{C}_{i}(v_{i}) = \begin{bmatrix} 0 & 0 & -\hat{m}_{22}v_{i} \\ 0 & 0 & \hat{m}_{11}u_{i} \\ \hat{m}_{22}v_{i} & -\hat{m}_{11}u_{i} & 0 \end{bmatrix},
$$
\n(5a)
\n
$$
\hat{D}_{i}(v_{i}) = \begin{bmatrix} \hat{d}_{u} + \hat{d}_{uu}|u_{i}| & 0 & 0 \\ 0 & \hat{d}_{v} + \hat{d}_{vv}|v_{i}| & 0 \\ 0 & 0 & \hat{d}_{r} + \hat{d}_{rr}|r_{i}| \end{bmatrix},
$$
\n(5b)

in which \hat{M}_i , \hat{C}_i and \hat{D}_i are the nominal values, and the corresponding parameter perturbations are defined as $\widetilde{M}_i = M_i - \hat{M}_i$, $\widetilde{C}_i = C_i - \hat{C}_i$, $\widetilde{D}_i = D_i - \hat{D}_i$.

Figure 1. Cooperative path-maneuvering geometry of autonomous underwater vehicles (AUVs). **Figure 1.** Cooperative path-maneuvering geometry of autonomous underwater vehicles (AUVs).

2.2. Problem Formualtion

Given a parameterized curved path without temporal constraints, a "virtual vehicle" $(x_{di}(\varpi_i), y_{di}(\varpi_i))$ is regarded as the tracking target for the *i*th AUV. As depicted in Figure [1,](#page-2-0) a path reference frame is established where the axes are parallel to the surge velocity and sway velocity of the AUV. For an AUV located at the point (x_i, y_i) , the maneuvering errors between (x_i, y_i) and $(x_{di}(\omega_i), y_{di}(\omega_i))$ in the path reference frame are

$$
\begin{bmatrix} x_{ei} \\ y_{ei} \end{bmatrix} = \begin{bmatrix} \cos \psi_i & -\sin \psi_i \\ \sin \psi_i & \cos \psi_i \end{bmatrix}^T \begin{bmatrix} x_{di}(\omega_i) - x_i \\ y_{di}(\omega_i) - y_i \end{bmatrix},
$$
 (6)

in which *xei* and *yei* represent the along-maneuvering error and cross-maneuvering error, respectively. Taking time derivatives of *xei* and *yei* along (1), we get

$$
\begin{bmatrix} \dot{x}_{ei} \\ \dot{y}_{ei} \end{bmatrix} = \begin{bmatrix} \cos(\psi_{di} - \psi_i) & 0 \\ 0 & \sin(\psi_{di} - \psi_i) \end{bmatrix} \begin{bmatrix} u_{di} \\ u_{di} \end{bmatrix} - \begin{bmatrix} u_i \\ v_i \end{bmatrix} + \begin{bmatrix} 0 & \dot{\psi}_i \\ -\dot{\psi}_i & 0 \end{bmatrix} \begin{bmatrix} x_{ei} \\ y_{ei} \end{bmatrix}, (7)
$$

in which $\psi_{di} = \text{atan2}(\dot{y}_{di}, \dot{x}_{di})$; And $u_{di} > 0$ is the virtual vehicle's speed along the desired path, which can be expressed by

$$
u_{di} = \dot{\omega}_i \sqrt{\dot{x}_{di}^2 + \dot{y}_{di}^2}.
$$
\n(8)

By combining with (4) and (7), the path-maneuvering error dynamics is rewritten as

$$
\begin{cases}\n\dot{x}_{ei} = y_{ei}\psi_i - u_i + \overline{u}_{di}\omega_i \cos(\psi_{di} - \psi_i), \\
\dot{y}_{ei} = -x_{ei}\psi_i - v_i + \overline{u}_{di}\omega_i \sin(\psi_{di} - \psi_i), \\
\dot{\psi}_i = r_i, \\
m_{11}\dot{u}_i = m_{22}v_ir_i - (d_u + d_{uu}|u_i|)u_i + d_{ui} + \tau_{ui}, \\
m_{22}\dot{v}_i = -m_{11}u_ir_i - (d_v + d_{vv}|v_i|)v_i + d_{vi} + 0, \\
m_{33}\dot{r}_i = (m_{11} - m_{22})u_iv_i - (d_r + d_{rr}|r_i|)r_i + d_{ri} + \tau_{ri},\n\end{cases}
$$
\n(9)

in which $\overline{u}_{di} = \sqrt{\dot{x}_{di}^2 + \dot{y}_{di}^2}$.

The maneuvering control objective is to propose a DCPMC approach for multiple AUVs with the dynamics (4) such that the assigned multiple parameterized paths can be synchronously tracked. Specifically, the objective is written as

$$
\begin{cases}\n\lim_{t \to \infty} |x_{ei}| \to 0, \\
\lim_{t \to \infty} |y_{ei}| \to 0,\n\end{cases}
$$
\n(10)

and

$$
\lim_{t \to \infty} |\varpi_i - \varpi_j| \to 0,
$$
\n(11)

in which $i = 1, \ldots, N$, $j = 1, \ldots, N$, and $i \neq j$.

3. Distributed Cooperative Path-Maneuvering Control

3.1. Coverage Path-Planning Design

To ensure the complete coverage for one region, a boustrophedon motion-based CPP is proposed and generates multiple desired paths. As shown in Figure [2,](#page-4-0) consider a search region $a \times b$, together with the sensed area 2*R* of an AUV, where *a* and *b* denote the area length and the area width, respectively. Our goal is to construct continuous curve paths and achieve complete coverage for this region.

Boustrophedon paths are firstly proposed for the CPP, where Boustrophedon means "the trajectory of the ox". As an ox drags a plow in one region, it crosses the full length of the region in a straight line, turns around, and then traces a new straight line path adjacent to previous one. By repeating the aforementioned procedure, the ox achieves complete coverage for this region [\[42,](#page-16-27)[43\]](#page-16-28).

Figure 2. Boustrophedon paths. **Figure 2.** Boustrophedon paths.

coverage for this region \mathcal{A}

Remark 1. Due to the vehicle maneuverability and minimum gyration radius constraints, boustro*strophedon paths mentioned above cannot be tracked directly. To this end, smooth boustrophedon phedon paths mentioned above cannot be tracked directly. To this end, smooth boustrophedon paths paths are designed using trigonometric functions and deployed in Figure 3. are designed using trigonometric functions and deployed in Figure [3.](#page-4-1)*

Figure 3. Illustration of the coverage path-planning (CPP) design.

 ω_1 , ω_2 , and ω_3 . Besides, each continuous curve path is composed of sub-path and has periodicity, where path variable starts at "the circle point" and ends at "the star point".
By virtue of boustrophedon paths and trigonometric functions, the desired multip In what follows, the smooth paths are parameterized by time-dependent variables

By virtue of boustrophedon paths and trigonometric functions, the desired multiple paths for region-searching are expressed as "the starts at "the starte point".

$$
\begin{cases}\nP_{d_1}(\varpi_1) = P_{d_{11}}(\varpi_1) \cup P_{d_{12}}(\varpi_1) \dots \cup P_{d_{18}}(\varpi_1), \\
P_{d_2}(\varpi_2) = P_{d_{21}}(\varpi_2) \cup P_{d_{22}}(\varpi_2) \dots \cup P_{d_{29}}(\varpi_2), \\
P_{d_3}(\varpi_3) = P_{d_{31}}(\varpi_3) \cup P_{d_{32}}(\varpi_3) \dots \cup P_{d_{39}}(\varpi_3),\n\end{cases}
$$
\n(12)

 $2 \left(\frac{\omega_3}{2}, \frac{\omega_3}{2}, \frac{\omega_4}{2}, \dots, \frac{\omega_4}{2}, \frac{\omega_5}{2}, \dots \right)$
meterized curved path and composed of P_i $\frac{1}{2}$ 3 31 32 39 α as α 3 α $d_3 = \frac{d_3}{d_3}(\omega_3) \cup \frac{d_4}{d_3}(\omega_3) \dots \cup \frac{d_3}{d_3}(\omega_3)$
anneterized curved path and composed of $P_{d_{ii}}$
scribed as described as **described** as eterized curved path and composed of P_{d_i}
ibed as $(w_3) = r_{d_{31}}(w_3) \cup r_{d_{32}}(w_3) \dots \cup r_{d_{39}}(w_3)$
arameterized curved path and compo described a in which P_{d_i} denotes the parameterized curved path and composed of $P_{d_{ii}}$. To be specific, the complete paths can be described as the complete paths can be described as

$$
\begin{cases}\nP_{d_{11}} = (-2R, \omega_1), \omega_1 \in (-\infty, \omega_{11}]; \\
P_{d_{12}} = (-R - R \cos((\omega_1 - \omega_{11})R^{-1}), \omega_{11} + R \sin((\omega_1 - \omega_{11})R^{-1})), \omega_1 \in (\omega_{11}, \omega_{12}]; \\
P_{d_{13}} = (\omega_1 - \omega_{12} - R, \alpha_{11}b), \omega_1 \in (\omega_{12}, \omega_{13}]; \\
P_{d_{14}} = (7R + R \sin((\omega_1 - \omega_{13})R^{-1}), \alpha_{12}b + R \cos((\omega_1 - \omega_{13})R^{-1})), \omega_1 \in (\omega_{13}, \omega_{14}]; \\
P_{d_{15}} = (8R, \omega_{14} - \omega_1 + \alpha_{13}b), \omega_1 \in (\omega_{14}, \omega_{15}]; \\
P_{d_{16}} = (8.5R - 0.5R \cos((\omega_1 - \omega_{15})\pi\omega_1^{-1}), -0.5R \sin((\omega_1 - \omega_{15})\pi\omega_1^{-1})), \omega_1 \in (\omega_{15}, \omega_{16}]; \\
P_{d_{17}} = (\omega_1 - \omega_{16} + 8.5R, -\alpha_{14}b), \omega_1 \in (\omega_{16}, \omega_{17}]; \\
P_{d_{18}} = (9.5R + 0.5R \sin((\omega_1 - \omega_{17})\pi\omega_1^{-1}), -0.5R \cos((\omega_1 - \omega_{17})\pi\omega_1^{-1})), \omega_1 \in (\omega_{17}, \omega_{18}];\n\end{cases}
$$
\n(13)

$$
\begin{cases}\nP_{d_{21}} = (0, \omega_2), \omega_2 \in (-\infty, \omega_{21}]; \\
P_{d_{22}} = (R - R \cos((\omega_2 - \omega_{21})\pi\omega_2^{-1}), \omega_{21} + R \sin((\omega_2 - \omega_{21})\pi\omega_2^{-1})), \omega_2 \in (\omega_{21}, \omega_{22}]; \\
P_{d_{23}} = (\omega_2 - \omega_{22} + R, \alpha_{21}b), \omega_2 \in (\omega_{22}, \omega_{23}]; \\
P_{d_{24}} = (5R + R \sin((\omega_2 - \omega_{23})\pi\omega_2^{-1}), \alpha_{22}b - R \cos((\omega_2 - \omega_{23})\pi\omega_2^{-1})), \omega_2 \in (\omega_{23}, \omega_{24}]; \\
P_{d_{25}} = (6R, \omega_{24} - \omega_2 + \alpha_{23}b), \omega_2 \in (\omega_{24}, \omega_{25}]; \\
P_{d_{26}} = (7R - R \cos((\omega_2 - \omega_{25})\pi\omega_2^{-1}), -\alpha_{24}b - R \sin((\omega_2 - \omega_{25})\pi\omega_2^{-1})), \omega_2 \in (\omega_{25}, \omega_{26}]; \\
P_{d_{27}} = (\omega_2 - \omega_{26} + 7R, -\alpha_{25}b), \omega_2 \in (\omega_{26}, \omega_{27}]; \\
P_{d_{28}} = (11R + R \sin((\omega_2 - \omega_{27})\pi\omega_1^{-1}), -\alpha_{26}b - R \cos((\omega_2 - \omega_{27})\pi\omega_1^{-1})), \omega_2 \in (\omega_{27}, \omega_{28}]; \\
P_{d_{29}} = (12R, \omega_2 - \omega_{28} - \alpha_{27}b), \omega_2 \in (\omega_{28}, \omega_{29}];\n\end{cases}
$$
\n(14)

and

 *Pd*³¹ = (2*R*, *v*3), *v*³ ∈ (−∞, *v*31]; *Pd*³² = 2.5*^R* − 0.5*^R* cos (*v*³ − *^v*31)*πω*−¹ 3 , *^v*³¹ ⁺ 0.5*^R* sin (*v*³ − *^v*31)*πω*−¹ 3 , *^v*³ [∈] (*v*31, *^v*32]; *Pd*³³ = (*v*³ − *v*³² + 2.5*R*, *α*31*b*), *v*³ ∈ (*v*32, *v*33]; *Pd*³⁴ = 3.5*^R* + 0.5*^R* sin (*v*³ − *v*33)*πω*−¹ 3 , *^α*32*^b* [−] 0.5*^R* cos (*v*³ − *v*33)*πω*−¹ 3 , *^v*³ [∈] (*v*33, *^v*34]; *Pd*³⁵ = (4*R*, *v*³⁴ − *v*³ + *α*33*b*), *v*³ ∈ (*v*34, *v*35]; *Pd*³⁶ = 5*R* − *R* cos (*v*³ − *v*35)*R* −1 , −*α*34*b* − *R* sin (*v*³ − *v*35)*R* −1 , *v*³ ∈ (*v*35, *v*36]; *Pd*³⁷ = (*v*³ − *v*³⁶ + 5*R*, −*α*35*b*), *v*³ ∈ (*v*36, *v*37]; *Pd*³⁸ = 13*R* − *R* sin (*v*³ − *v*37)*R* −1 , −*α*36*b* − *R* cos (*v*³ − *v*37)*R* −1 , *v*³ ∈ (*v*37, *v*38]; *Pd*³⁹ = (14*R*, *v*³ − *v*³⁸ − *α*37*b*), *v*³ ∈ (*v*38, *v*39]; (15)

in which ω_{1i} , ω_{2i} , and ω_{3i} are sub-path variables. α_{1i} , α_{2i} , and α_{3i} are design parameters with respect to the special region. ω_1 , ω_2 , and ω_3 are path parameters without temporal constraints.

Remark 2. *Based on the practical considerations, three AUVs are widely applied to varieties of missions, and because of this, three boustrophedon paths are constructed and tracked in this article.*

3.2. Individual Path-Maneuvering Control Design

In practice, dynamic uncertainties and/or ocean currents have an adverse impact on path-maneuvering performance. It is highly desirable to develop robust control laws for under-actuated AUVs. To this end, the dynamics (4) can be rewritten as

$$
\begin{cases}\n\hat{m}_{11}\dot{u}_i = \hat{m}_{22}v_i r_i - \varphi_{1i}^T \zeta_{1i} + \delta_{1i} + \tau_{ui}, \\
\hat{m}_{22}\dot{v}_i = -\hat{m}_{11}u_i r_i - \varphi_{2i}^T \zeta_{2i} + \delta_{2i} + 0, \\
\hat{m}_{33}\dot{r}_i = (\hat{m}_{11} - \hat{m}_{22})u_i v_i - \varphi_{3i}^T \zeta_{3i} + \delta_{3i} + \tau_{ri},\n\end{cases}
$$
\n(16)

in which δ_{1i} , δ_{2i} , and δ_{3i} denote nonlinear dynamics determined by

$$
\begin{cases}\n\delta_{1i} = d_{ui} - \tilde{m}_{11}\dot{u}_i + \tilde{m}_{22}v_i r_i, \\
\delta_{2i} = d_{vi} - \tilde{m}_{22}\dot{v}_i - \tilde{m}_{11}u_i r_i, \\
\delta_{3i} = d_{ri} - \tilde{m}_{33}\dot{r}_i + (\tilde{m}_{11} - \tilde{m}_{22})u_i v_i,\n\end{cases}
$$
\n(17)

with $\varphi_{1i} = [u_i|u_i|$, $u_i]^T$, $\varphi_{2i} = [v_i|v_i|, v_i]^T$, $\varphi_{3i} = [r_i|r_i|, r_i]^T$, $\zeta_{1i} = [d_{uu}, d_u]^T$, $\zeta_{2i} = [d_{vv}, d_v]^T$, and $\zeta_{3i} = [d_{rr}, d_r]^T$.

In order to achieve (10), virtual control laws are designed at the kinematic hierarchy, and then adaptive sliding controllers are designed at the dynamic hierarchy. In the following, we define $u_{ei} = u_i - \alpha_{ui}$, $r_{ei} = r_i - \alpha_{ri}$, $\psi_{ei} = \psi_{di} - \psi_i$, $\omega_i = v_{si}(\omega_i) - v_i$, $\tilde{\zeta}_{1i} = \hat{\zeta}_{1i} - \zeta_{1i}$ and $\tilde{\zeta}_{3i} = \tilde{\zeta}_{3i} - \zeta_{3i}$, in which α_{ui} and α_{ri} are virtual control variables; v_{si} is a distributed speed assignment and *υⁱ* is a consensus protocol variable to accomplish synchronized motions; $\hat{\zeta}_{1i}$ and $\hat{\zeta}_{3i}$ are the estimations for uncertainties.

In this context, the path-maneuvering error dynamics for kinematics is reformulated by

$$
\begin{cases}\n\dot{x}_{ei} = y_{ei}\psi_i - u_i + \overline{u}_{di}(v_{si} - v_i)\cos\psi_{ei},\\ \n\dot{y}_{ei} = -x_{ei}\psi_i - v_i + \overline{u}_{di}(v_{si} - v_i)\sin\psi_{ei},\\ \n\dot{\psi}_{ei} = \psi_{di} - r_i.\n\end{cases}
$$
\n(18)

Consider the following Lyapunov candidate function

$$
V_1 = k_1(1 - \cos \psi_{ei}) + \frac{1}{2} \left(x_{ei}^2 + y_{ei}^2 \right), \tag{19}
$$

in which $k_1 > 0$.

Differentiating V_1 along the error dynamics (18) yields

$$
\dot{V}_1 = x_{ei} \Big(y_{ei} \psi_i - u_i + \overline{u}_{di} (v_{si} - v_i) \cos \psi_{ei} \Big) + y_{ei} \Big(-x_{ei} \dot{\psi}_i \n- v_i + \overline{u}_{di} (v_{si} - v_i) \sin \psi_{ei} \Big) - k_1 \sin \psi_{ei} \Big(\dot{\psi}_{di} - r_i \Big) \n= x_{ei} (\overline{u}_{di} (v_{si} - v_i) \cos \psi_{ei} - u_i) - y_{ei} v_i + \sin \psi_{ei} (\overline{u}_{di} \n(v_{si} - v_i) y_{ei} - k_1 \Big(\dot{\psi}_{di} - r_i \Big) \Big).
$$
\n(20)

Then, choosing the virtual control laws *αui* and *αri* as

$$
\begin{cases}\n\alpha_{ui} = \overline{u}_{di}\omega_i \cos \psi_{ei} + k_2 x_{ei} + \overline{u}_{di}\nu_i, \\
\alpha_{ri} = \psi_{di} + \frac{1}{k_2} \overline{u}_{di}\omega_i y_{ei} + k_1 \overline{u}_{di}\omega_i \sin \psi_{ei},\n\end{cases}
$$
\n(21)

in which $k_2 > 0$.

Submitting (21) into (20), there is

$$
\dot{V}_1 = -k_1^2 \overline{u}_{di} \dot{\omega}_i \sin^2 \psi_{ei} - k_2 x_{ei}^2 + \overline{u}_{di} v_i x_{ei} - u_{ei} x_{ei} - k_1 r_{ei} \sin \psi_{ei} - v_i y_{ei}.
$$
 (22)

Note that α_{ui} and α_{ri} could not be directly applied in practice. In this context, consider the Lyapunov candidate function

$$
V_2 = V_1 + \frac{1}{2} \left(u_{ei}^2 + r_{ei}^2 \right). \tag{23}
$$

Differentiating V_2 along u_{ei} and r_{ei} , we have

$$
\dot{V}_2 = -k_1^2 \overline{u}_{di} \dot{\omega}_i \sin^2 \psi_{ei} - k_2 x_{ei}^2 + \overline{u}_{di} v_i x_{ei} - k_1 r_{ei} \sin \psi_{ei} \n- u_{ei} x_{ei} - v_i y_{ei} + u_{ei} (\dot{u}_i - \dot{u}_{di}) + r_{ei} (\dot{r}_i - \dot{r}_{di}).
$$
\n(24)

By combining with sliding mode and adaptive technique, we design the maneuvering control laws for surge and yaw motions as follows

$$
\begin{cases}\n\tau_{ui} = \hat{m}_{11}(\dot{\alpha}_{ui} - k_3 u_{ei} + x_{ei} - \eta_1 \text{sgn}(u_{ei})) + \varphi_{1i}^T \hat{\zeta}_{1i} - \hat{m}_{22} v_i r_i, \\
\tau_{ri} = \hat{m}_{33}(\dot{\alpha}_{ri} - k_4 r_{ei} + k_1 \sin \psi_{ei} - x_{ei} - \eta_2 \text{sgn}(r_{ei})) + \varphi_{3i}^T \hat{\zeta}_{3i} - (\hat{m}_{11} - \hat{m}_{22}) u_i v_i,\n\end{cases}
$$
\n(25)

with update laws

$$
\begin{cases}\n\dot{\hat{\zeta}}_{1i} = -\lambda_1 u_{ei} \varphi_{1i}, \\
\dot{\hat{\zeta}}_{3i} = -\lambda_3 r_{ei} \varphi_{3i},\n\end{cases}
$$
\n(26)

in which k_3 , k_4 , η_1 , η_2 , λ_1 , and λ_3 are positive constants, and sgn(u_{ei}) is the switching function given by

$$
sgn(u_{ei}) = \begin{cases} 1, u_{ei} > 0; \\ 0, u_{ei} = 0; \\ -1, u_{ei} < 0. \end{cases}
$$
 (27)

In addition, $sgn(r_{ei})$ has the similar structure with (27).

Remark 3. *To overcome chattering of the discontinuous switch function, a saturation function is employed by*

$$
sat(u_{ei}; \mu, \sigma) = \begin{cases} \operatorname{sgn}(u_{ei}), & |u_{ei}| > \mu; \\ \frac{\operatorname{sig}^{\sigma}(u_{ei})}{\mu^{\sigma}}, & |u_{ei}| \leq \mu; \end{cases}
$$
(28)

with $\mu > 0$ *and* $0 < \sigma < 1$.

Substituting the control laws (25) into (24), and using dynamics (16), we have

$$
V_2 = -k_1^2 \overline{u}_{di} \omega_i \sin^2 \psi_{ei} - k_2 x_{ei}^2 + \overline{u}_{di} v_i x_{ei} - u_{ei} x_{ei} - k_1 r_{ei} \sin \psi_{ei} - v_i y_{ei} + u_{ei} (\delta_{1i} / \hat{m}_{11} + x_{ei} - \eta_1 \text{sgn}(u_{ei}) - \varphi_{1i}^T \zeta_{1i} / \hat{m}_{11} + \varphi_{1i}^T \zeta_{1i} / \hat{m}_{11} - k_4 u_{ei}) + r_{ei} (\delta_{3i} / \hat{m}_{33} - k_5 r_{ei} + k_1 \sin \psi_{ei} - \eta_2 \text{sgn}(r_{ei}) - \varphi_{3i}^T \zeta_{3i} / \hat{m}_{33} + \varphi_{3i}^T \zeta_{3i} / \hat{m}_{33}) = -k_1^2 \overline{u}_{di} \omega_i \sin^2 \psi_{ei} - k_2 x_{ei}^2 + \overline{u}_{di} v_i x_{ei} - k_3 u_{ei}^2 - k_4 r_{ei}^2 - v_i y_{ei} + u_{ei} (\delta_{1i} / \hat{m}_{11} - \eta_1 \text{sgn}(u_{ei}) + \varphi_{1i}^T \zeta_{1i} / \hat{m}_{11}) + r_{ei} (\delta_{3i} / \hat{m}_{33} - \eta_2 \text{sgn}(r_{ei}) + \varphi_{3i}^T \zeta_{3i} / \hat{m}_{33}).
$$
\n(29)

The key result on individual path-following control is summarized as follows.

Theorem 1. *Consider the dynamics (9) and the control laws (25) with (26). Then, error signals uei, rei, ψei, xei, and yei asymptotically converge to zero.*

Proof of Theorem 1. Consider the Lyapunov candidate function

$$
V_3 = V_2 + \frac{1}{2\lambda_1 \hat{m}_{11}} \tilde{\zeta}_{1i}^T \tilde{\zeta}_{1i} + \frac{1}{2\lambda_3 \hat{m}_{33}} \tilde{\zeta}_{3i}^T \tilde{\zeta}_{3i},\tag{30}
$$

whose time derivative is

$$
\dot{V}_{3} = -k_{1}^{2} \overline{u}_{di} \dot{\omega}_{i} \sin^{2} \psi_{ei} - k_{2} x_{ei}^{2} + \overline{u}_{di} v_{i} x_{ei} - k_{3} u_{ei}^{2} - k_{4} r_{ei}^{2} + u_{ei} (\delta_{1i} / \hat{m}_{11} - \eta_{1} \text{sgn}(u_{ei})) - v_{i} y_{ei}
$$
\n
$$
+ r_{ei} (\delta_{3i} / \hat{m}_{33} - \eta_{2} \text{sgn}(r_{ei})) + u_{ei} \varphi_{1i}^{T} \tilde{\zeta}_{1i} / \hat{m}_{11} + r_{ei} \varphi_{3i}^{T} \tilde{\zeta}_{3i} / \hat{m}_{33} + \tilde{\zeta}_{1i} \tilde{\zeta}_{1i} / \lambda_{1} \hat{m}_{11} + \tilde{\zeta}_{3i} \tilde{\zeta}_{3i} / \lambda_{3} \hat{m}_{33}.
$$
\n(31)

Using the fact that $v_i y_{ei} = \varepsilon v_i y_{ei} + (1 - \varepsilon) v_i y_{ei}$ and the update laws (26), (31) becomes

$$
\dot{V}_{3} = -k_{1}^{2} \overline{u}_{di} \dot{\omega}_{i} \sin^{2} \psi_{ei} - k_{2} x_{ei}^{2} + \overline{u}_{di} v_{i} x_{ei} - k_{3} u_{ei}^{2} - k_{4} r_{ei}^{2} + u_{ei} (\delta_{1i} / \hat{m}_{11} - \varepsilon v_{i} y_{ei} / u_{ei} - \eta_{1} \sin(u_{ei})) + r_{ei} (\delta_{3i} / \hat{m}_{33} - \eta_{2} \sin(r_{ei}) - (1 - \varepsilon) v_{i} y_{ei} / r_{ei}),
$$
\n(32)

in which $0 < \varepsilon < 1$.

By the aid of the boundedness of the sway velocity v_i , i.e., $|v_i(t)| \leq 2\vartheta/d_v + |v_i(t_0)|$ *e*^{−0.5*dv*^(*t*−*t*₀) mentioned in [\[17\]](#page-16-3), we have the following inequality}

$$
\begin{cases} \left| \delta_{1i} / \hat{m}_{11} - \varepsilon v_i y_{ei} / u_{ei} \right| \leq \eta_1, \\ \left| \delta_{3i} / \hat{m}_{33} - (1 - \varepsilon) v_i y_{ei} / r_{ei} \right| \leq \eta_2. \end{cases}
$$
\n(33)

Note that synchronization variable *υⁱ* is zero since path parameter synchronization is not considered in the IPMC design. Then, substituting (33) into (32), there is

$$
\dot{V}_3 < 0, \ \forall \|[u_{ei}, r_{ei}, \psi_{ei}, \widetilde{\zeta}_{1i}, \widetilde{\zeta}_{3i}, x_{ei}, y_{ei}]^T\| \neq 0, \tag{34}
$$

which implies that error signals *uei*, *rei*, *ψei*, *xei*, and *yei* asymptotically converge to zero. Thus, the proof is completed. \square

3.3. Path Parameter Synchronization Design

Consider the communication topology described by a graph $G = (0, \varepsilon)$ with the set of nodes = $\{n_1, \ldots, n_N\}$ and the set of edges $\varepsilon = \{(n_i, n_j) \in \times\}$. In this graph, a node represents an AUV and the element (n_i, n_j) describes information interactions from one node to another. $A_{(G)} = [a_{ij}] \in R^{N \times N}$ denotes an adjacency matrix, where $a_{ij} = 1$ $f(n_i, n_j) \in \varepsilon$; Otherwise, $a_{ij} = 0$, and the Laplacian matrix $L_{(G)} = [l_{ij}] \in R^{N \times N}$ is introduced with

$$
l_{ij} = \begin{cases} -a_{ij}, & \text{if } i \neq j \text{ and } j \in N_i; \\ \sum_{j \in N_i} a_{ij}, & \text{if } i = j; \\ 0, & \text{otherwise} \end{cases}
$$
(35)

It is worth noting that *L*(*G*) in the connected graph is symmetric and positive semidefinite. In addition, the corresponding eigenvector with minimum eigenvalue of zero is $I = [1, \dots, 1]^T$, thus resulting in $L_{(G)}I = 0_N$.

Consider that parts of the nodes are able to obtain the speed assignment. The accessorial matrix $D_{(G)} = \text{diag}\{d_1, \ldots, d_N\}$ is proposed, where $d_i = 1$ represents that the node can get information from the speed assignment and $d_i = 0$ represents that it cannot. In addition, the symmetric and positive definite matrix $H_{(G)} = L_{(G)} + D_{(G)}$ is satisfied for the undirected connected graph.

The key result on the distributed signal observer is collected as follows.

Theorem 2. *Consider that only the parts of AUVs receive the speed assignment and the distributed signal observer given by*

$$
\dot{\hat{\sigma}}_{si} = -\sum_{j=0}^{N} a_{ij} (\hat{\sigma}_{si} - \hat{\sigma}_{sj}), \qquad (36)
$$

 i *n which* $\hat{v}_{s0} = v_{s0}$ *is a desired speed assignment and satisfies* $\dot{v}_{s0} = 0$. Then, $\lim_{t\to\infty} |\hat{v}_{si}| \to v_{s0}$, *can be guaranteed.*

Proof of Theorem 2. Define observation error as

$$
\widetilde{v}_{si} = \widehat{v}_{si} - v_{s0}.\tag{37}
$$

Taking time derivatives along (36) yields

$$
\dot{\tilde{v}}_{si} = \dot{\tilde{v}}_{si} - \dot{v}_{s0} \n= -\sum_{j=0}^{N} a_{ij} [(\hat{v}_{si} - \hat{v}_{s}) - (\hat{v}_{sj} - \hat{v}_{s})] \n= -\sum_{j=0}^{N} a_{ij} (\tilde{v}_{si} - \tilde{v}_{sj}).
$$
\n(38)

Defining $\widetilde{v}_s = [\widetilde{v}_{si}] \in R^{N \times 1}$, we have

$$
\dot{\widetilde{v}}_s = -\left(H_{(G)} \otimes I\right) \widetilde{v}_s. \tag{39}
$$

Consider the Lyapunov candidate function $V_s = 1/2 \tilde{v}_s^T \tilde{v}_s$ and differentiating V_s
 \tilde{v}_s (20) vialde along (39) yields

$$
\dot{V}_s = -\tilde{v}_s^T \left(H_{(G)} \otimes I \right) \tilde{v}_s \le 0, \tag{40}
$$

which implies that observation error \tilde{v}_{si} can converge to zero. This proof is complete. \Box

Furthermore, based on neighbors' parameter information, we define the following coordination errors

$$
e_i = \sum_{j=1}^{N_i} a_{ij} (\varpi_i - \varpi_j), \tag{41}
$$

and design the consensus protocol as

$$
v_i = \xi_i e_i - \xi_i \overline{u}_{di} x_{ei}, \qquad (42)
$$

in which $\zeta_i > 0$.

In vector form, (42) can be presented as

$$
\begin{cases} \n\dot{\omega} = \hat{v}_s(\omega) - v, \\ \neq L_{(G)}\omega, \n\end{cases} \n\tag{43}
$$

here $\varpi = [\varpi_i] \in R^{N \times 1}$, $\hat{v}_s = [\hat{v}_{si}] \in R^{N \times 1}$, $v = [v_i] \in R^{N \times 1}$, and $e = [e_i] \in R^{N \times 1}$. Computing the time derivative of *e* and using $L_{(G)}I = 0_N$, we obtain

$$
\dot{e} = \dot{L}_{(G)}\omega + L_{(G)}(\hat{v}_s(\omega) - v)
$$

= $-L_{(G)}v.$ (44)

Eventually, the path-maneuvering error dynamics (9) is rewritten as

$$
\begin{cases}\n\dot{x}_{ei} = y_{ei}\psi_i - u_i + \overline{u}_{di}(v_s - v_i)\cos\psi_{ei}, \\
\dot{y}_{ei} = -x_{ei}\psi_i - v_i + \overline{u}_{di}(v_s - v_i)\sin\psi_{ei}, \\
\dot{\psi}_{ei} = \psi_{di} - r_i, \\
m_{11}\dot{u}_i = m_{22}\overline{v}_i r_i - (d_u + d_{uu}|u_i|)u_i + d_{ui} + \tau_{ui}, \\
m_{22}\dot{v}_i = -m_{11}\overline{u}_i r_i - (d_v + d_{vv}|v_i|)v_i + d_{vi} + 0, \\
m_{33}\dot{r}_i = (m_{11} - m_{22})\overline{u}_i \overline{v}_i - (d_r + d_{rr}|r_i|)r_i + d_{ri} + \tau_{ri}, \\
\dot{e} = -L_{(G)}\overline{v}.\n\end{cases}
$$
\n(45)

3.4. Stability Analysis

The following theorem exhibits the stability result of the proposed DCPMC approach.

Theorem 3. *Consider a cluster of AUVs with dynamics (4), the control laws (25), and the consensus protocol (42). Then, all error signals in the closed-loop system are GAS.*

Proof of Theorem 3. Consider the following Lyapunov candidate function

$$
V = \sum_{i=1}^{N} \left\{ \frac{1}{2} \left(x_{ei}^2 + y_{ei}^2 + u_{ei}^2 + r_{ei}^2 \right) + k_2 (1 - \cos \psi_{ei}) + \frac{1}{2\lambda_1 \hat{m}_{11}} \tilde{\zeta}_1^T \tilde{\zeta}_1 + \frac{1}{2\lambda_3 \hat{m}_{33}} \tilde{\zeta}_3^T \tilde{\zeta}_3 \right\} + \frac{1}{2} e^T P e. \tag{46}
$$

Using Lemma 1 in [\[39\]](#page-16-24) yields that $\omega^T L_{(G)} \omega = e^T P e$, where P is a positive definite matrix. In this context, (46) becomes

$$
V = \sum_{i=1}^{N} \left\{ \frac{1}{2} \left(x_{ei}^2 + y_{ei}^2 + u_{ei}^2 + r_{ei}^2 \right) + k_2 (1 - \cos \psi_{ei}) + \frac{1}{2\lambda_1 \hat{m}_{11}} \tilde{\zeta}_1^T \tilde{\zeta}_1 + \frac{1}{2\lambda_3 \hat{m}_{33}} \tilde{\zeta}_3^T \tilde{\zeta}_3 \right\} + \frac{1}{2} \omega^T L_{(G)} \omega, \tag{47}
$$

whose derivative along (45) is

$$
\dot{V} = \sum_{i=1}^{N} \left\{ -k_1^2 \overline{u}_{di} \dot{\omega}_i \sin^2 \psi_{ei} - k_2 x_{ei}^2 - k_3 u_{ei}^2 - k_4 r_{ei}^2 + u_{ei} (\delta_{1i} / \hat{m}_{11} - \varepsilon v_i y_{ei} / u_{ei} + r_{ei} (\delta_{3i} / \hat{m}_{33} - \eta_2 \text{sgn}(r_{ei}) - (1 - \varepsilon) v_i y_{ei} / r_{ei}) \right\} + \sum_{i=1}^{N} \left\{ \overline{u}_{di} v_i x_{ei} - \omega_i^T L_{(G)} v_i \right\} \tag{48}
$$
\n
$$
\leq \sum_{i=1}^{N} \left\{ -k_1^2 \overline{u}_{di} \dot{\omega}_i \sin^2 \psi_{ei} - k_2 x_{ei}^2 - k_3 u_{ei}^2 - k_4 r_{ei}^2 \right\} - \lambda_{\min}(\xi) ||\gamma||^2,
$$

in which $\xi = \text{diag}\{\xi_1, \ldots, \xi_N\}$, $\gamma = -e + \overline{u}_d x_e$, $x_e = [x_{ei}] \in R^{N \times 1}$, and $\overline{u}_d = \text{diag}$ ${\{\overline{u}_{d1}, \ldots, \overline{u}_{dN}\}}$. By straightforward derivation [\[44\]](#page-17-0), one can conclude that all error signals are GAS. This concludes the proof. \square

Remark 4. By virtue of the inequality $x^T Sym(L(G))x \ge \zeta_s(Sym(L(G)))||x-1_N1_N^Tx/N||^2$ and Sym $(L_{(G)}) = (L_{(G)} + L_{(G)}^T)/2$ in [\[45\]](#page-17-1), one can conclude that $e^T Pe \geq \xi_s \|\varpi - \sum_{i=1}^N \varpi_i / N\|^2$ and $\lim_{t\to\infty} |\varpi_i - \varpi_j| \to 0$.

4. Simulation Studies

To evaluate the proposed DCPMC approach, simulation studies are given in this
22. The weight of the AUV called WL-II is 45 kg and the length is 1.46 m. Bosides section. The weight of the AUV called WL-II is 45 kg and the length is 1.46 m. Besides, model parameters are illustrated in Table [1](#page-10-1) [\[46\]](#page-17-2).

Table 1. Model parameters of WL-II.

Let the information topology among the three AUVs be described by Figure 4, where only AUV 2 can receive the speed signal. Consider the special region with $a = 180m$, only AUV 2 can receive the speed signal. Consider the special region with $a = 180m$, $b = 300m$, and the sensed area of an AUV with $R = 10m$. The desired coverage paths are generated by (13), (14) and (15), where path parameters are illustrated in Table [2.](#page-11-0) *r* e

Figure 4. Communication topology of three AUVs.

Figure 4. Communication topology of three AUVs. $\sigma_{vi} = \sigma_2 \kappa_2$, and $\sigma_{ri} = \sigma_3 \kappa_3$, where $\kappa_i > 0$ are constants and σ_i are zero-mean Gaussian white noise processes [\[47](#page-17-3)[,48\]](#page-17-4). The initial conditions of three AUVs are set as $v_1(0) = [0, 0, 0]^T$, $v_2(0) = [0, 0, 0]^T$, and $v_3(0) = [0, 0, 0]^T$. \int_{0}^{T} , $\eta_{2}(0) = [-5, -50, \pi/3]^{T}$, $\eta_{3}(0) = [25, -48, \pi/3]^{T}$. , $[2]^T$, For the sake of simulation setup, dynamic uncertainties and ocean currents are assumed as follows: $\tilde{m}_{11} = -0.1\hat{m}_{11}$, $\tilde{m}_{22} = -0.1\hat{m}_{22}$, $\tilde{m}_{33} = -0.15\hat{m}_{33}$, $d_{ui} = \vartheta_1 \kappa_1$, $d_{vi} = \vartheta_2 \kappa_2$, and $d_{ri} = \vartheta_3 \kappa_3$, where $\kappa_i > 0$ are constants and ϑ_i are zero-mean Gaussian follows: $\eta_1(0) = [-30, -55, \pi/6]^T$, $\eta_2(0) = [-5, -50, \pi/3]^T$, $\eta_3(0) = [25, -48, \pi/2]^T$,

As shown in Figure 5, three AUVs convertion of the sign and keep a synchronized formation si As shown in Figure 5, three AU vs converge to the desired paths generated by the CFF
design and keep a synchronized formation simultaneously. Besides, taking readability into are zero-mean Gaussian white noise pro-
Service pro-As shown in Figure [5,](#page-11-1) three AUVs converge to the desired paths generated by the CPP

consideration, Figure [6](#page-12-0) shows the snapshots of three AUVs in different time periods. From Figures [7](#page-12-1) and [8,](#page-12-2) it is observed that along-maneuvering and cross-maneuvering errors can converge to the origin in a short time. The small chattering existing in tracking errors is caused by the switching of the straight line and the curve. The path parameter variables of three AUVs are given in Figure [9](#page-13-0) and almost synchronized. Figures [10](#page-13-1) and [11](#page-13-2) show the AUV heading angles, surge and yaw angular velocities, respectively. Note that the desired paths are designed using piecewise function, thereby leading to slight chattering for control inputs in Figure [12.](#page-14-0) The observation results on speed signals are given in Figure [13,](#page-14-1) which ¹ can exhibit excellent observation performance. Furthermore, comparisons on maneuvering errors and velocity signals of individual AUV are given in Figure [14.](#page-14-2) It is apparent that the proposed approach has better maneuvering performance than the backsteeping-based

2nnroach (*n*, - 0, *i* - 1, 2), thus contributing to loss abrasion of actuators approach ($\eta_i = 0$, $i = 1, 2$), thus contributing to less abrasion of actuators.

> **Table 2.** Path parameters and control parameters. , leters and control parameters.

Figure 5. Region-searching performance of three AUVs under the distributed cooperative path-**Figure 5.** Region-searching performance of three AUVs under the distributed cooperative pathmaneuvering control (DCPMC) approach. maneuvering control (DCPMC) approach.

Figure 6. Snapshots of three AUVs in different time periods.

Figure 7. Along-maneuvering errors of three AUVs. **Figure 7.** Along-maneuvering errors of three AUVs. **Figure 7.** Along-maneuvering errors of three AUVs.

Figure 8. Cross-maneuvering errors of three AUVs. **Figure 8.** Cross-maneuvering errors of three AUVs.

Figure 9. Figure 9. Path parameter variables of three AUVs. Path parameter variables of three AUVs.

Figure 8. Cross-maneuvering errors of three AUVs.

Figure 10. Heading angles of three AUVs. **Figure 10.** Heading angles of three AUVs. **Figure 10.** Heading angles of three AUVs.

Figure 11. Surge and yaw angular velocities of three AUVs. **Figure 11.** Surge and yaw angular velocities of three AUVs. **Figure 11.** Surge and yaw angular velocities of three AUVs.

Figure 12. Figure 12. Control inputs of three AUVs. Control inputs of three AUVs.

Figure 13. Observations for the speed signal. **Figure 13.** Observations for the speed signal. **Figure 13.** Observations for the speed signal.

Figure 14. Comparisons on maneuvering errors and velocity signals of individual AUV. Fi**gure 14.** Comparisons on maneuvering errors and velocity signals of individual AUV.
 Figure 14. Comparisons on maneuvering errors and velocity signals of individual AUV.

Figure 14. Comparisons on maneuvering errors and velocity signals of individual AUV.

5. Conclusions

In this article, a DCPMC approach was developed for a cluster of under-actuated AUVs under undirected communication topology. By the aid of smooth boustrophedon paths, the CPP is designed for the region-searching, where multiple parameterized paths are generated and used to be tracked. Within the dynamics controllers, adaptive sliding mode techniques are used to design maneuvering control laws, which can drive vehicles to track the assigned paths despite the existence of dynamic uncertainties and ocean currents. Besides, the consensus protocol with distributed signal observer is developed for path parameter synchronization, thus contributing to the desired formation configuration. Rigorous theoretical derivation and proof have exhibited that the resulting closed-loop signals are GAS, and simulation results have illustrated the effectiveness of the DCPMC approach.

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