

Article

The Effect of Geometrical, Operational, Mixing Methods, and Rheological Parameters on Discharge Coefficients of Internal-Mixing Twin-Fluid Atomizers

Farid A. Hammad ^{1,2,*} , Kai Sun ¹, Jan Jedelsky ³  and Tianyou Wang ^{1,*}¹ State Key Laboratory of Engines, Tianjin University, Tianjin 300072, China; sunkai@tju.edu.cn² Mechanical Power Engineering Department, Faculty of Engineering, Tanta University, Tanta 31521, Egypt³ Faculty of Mechanical Engineering, Brno University of Technology, Technicka 2896/2, 616 69 Brno, Czech Republic; jedelsky@fme.vutbr.cz

* Correspondence: faridhammad87@tju.edu.cn (F.A.H.); wangtianyou@tju.edu.cn (T.W.); Tel.: +86-022-85356295 (T.W.)

Received: 2 April 2020; Accepted: 5 May 2020; Published: 11 May 2020



Abstract: Accurate prediction of the discharge coefficient (C_D) for internal-mixing twin-fluid (IMTF) atomizers is challenging, the effect of control factors remains inadequately understood, and comparative data on the C_D of IMTF atomizers are unavailable. This work presents an experimental study on C_D for different IMTF atomizers with a wide range of factors, including the gas-to-liquid ratio (GLR), the inlet-overpressure ratio ($\Delta p_{\text{mix}}/p_{\text{amb}}$), the orifice length-to-diameter ratio (L_o/d_o), and the liquid viscosity (μ_L). Five atomizers with different internal-mixing principles were probed on a cold test rig, including the frequently studied outside-in-gas (OIG) and inside-out-gas (IOG) effervescent types, the recently-introduced outside-in-liquid (OIL) and air-core-liquid-ring (ACLR) atomizers, and our new design named the swirling-air-core-liquid-ring (SACLR) atomizer. The results demonstrate that C_D is governed mainly by GLR, and reduces if GLR, L_o/d_o , or μ_L is increased. An increase in $\Delta p_{\text{mix}}/p_{\text{amb}}$ causes a C_D reduction up to $\Delta p_{\text{mix}}/p_{\text{amb}} = 0.98$, and C_D increases for a higher $\Delta p_{\text{mix}}/p_{\text{amb}}$. Surprisingly, differences in C_D amid examined atomizers were found negligible, although the flow visualization inside the orifice showed a significantly different flow character for each one of the atomizers. Finally, a general C_D correlation fitting with an $R^2 \geq 0.99$ for all the tested nozzles was proposed. The results amend the present knowledge, allow design optimization, and provide flow rate prediction for a variety of IMTF atomizers.

Keywords: twin-fluid nozzles; internal-mixing methods; discharge coefficient; atomization; two-phase flow

1. Introduction

Twin-fluid (TF) atomizers were found to be more advantageous over other liquid atomization devices in numerous applications including gas turbines [1–3], scramjet [4], internal combustion engines [5,6], furnaces, and boilers [7–9]. They have been used for spray drying in the food industry [10–15], humidification, dust control, gas cooling, spray coating [16], pharmaceutical or consumer products [17], and water mist fire suppression systems [18].

TF atomizers use different approaches to produce two-phase mixtures for the intensification of liquid break-up. The TF atomizers in the present study significantly differ, for example, from the flash-boiling atomizers, even though both these categories work with a two-phase flow at the exit orifice. Flash-boiling atomizers use a single supply line with the bulk liquid phase only, and the two-phase flow inside the atomizer is a product of a phase-change process (i.e., an intense gas phase

nucleation or flashing boiling process). This effect can be achieved by exposing the bulk liquid to a sudden and sharp pressure drop (during an isothermal expansion process) or by isobaric heating. As a result, the liquid changes from a sub-cooled state to vapor or a superheated liquid state [19]. The typical flashing boiling atomizer contains two sharp-edged orifices (0.1 and 5 mm in diameter, respectively) with an expansion chamber that separates them. The flashing of the bulk liquid, which is usually pre-heated, takes place through the first small orifice. Due to the specific design, liquid physical properties, and the working conditions, a small portion from the liquid vaporizes to create a two-phase flow inside this atomizer. On the other hand, it is difficult for the phase change phenomena inside the internal-mixing twin-fluid (IMTF) atomizers to occur for reasons that are discussed in Section 2.2. In the TF atomizers used here, both the gas and liquid phases are separately introduced into the atomizer and are mixed into a two-phase fluid. The gas–liquid mixing can be carried out internally or externally. Internal-mixing twin-fluid (IMTF) nozzles feature lower atomizing gas consumption and higher energy transfer compared to their external-mixing twin-fluid (EMTF) counterparts (e.g., airblast atomizers [20]) [5,21–23], which reduce operating costs.

The gas and liquid enter the IMTF nozzle through separate ports and interact inside a mixing chamber forming a two-phase mixture with a character varying depending on the mixing conditions (mixing chamber geometry, fluid pressures, and rheological properties). The mixture discharges from the exit orifice forming a spray. The discharge coefficient (C_D) measures the extent to which the liquid flow uses the available exit orifice area [22].

The purpose of an atomizer is to deliver a specified amount of a liquid into a target place (defined with spray cone angle, droplet concentration, and liquid mass flux) and disperse it with the required quality (the spray characteristics such as droplet size and velocity, and droplet size distribution, are considered here). The amount of liquid discharged per time is represented with the flow rate or more conveniently with the C_D of the atomizer. The importance of each factor category depends on the application. Precise liquid dosing and metering in a whole range of operating conditions are crucial in power generation units, such as combustion engines, turbines, burners, and furnaces; the throughput of the liquid governs mixing ratios of fuel with air in reacting sprays and directly affects combustion efficiency as well as pollutant formation. The amount of discharged liquid plays an essential role in process engineering and medical applications, particularly in mixing and dosing devices, chemical reactors, or columns for pollutant removal and capture. It is also considered in agriculture, food industry, or spray coating devices.

The atomized liquid is delivered to the target application in a specified amount within a range of operating conditions. Therefore, the discharge coefficient (C_D) knowledge allows for predicting the actual liquid flow rate for given operating conditions; thus, playing an important role in the nozzle design process [22,24–26]. It is advantageous to have a prediction tool for discharge characteristics of the atomizer before its realization. Iterative or trial–error techniques of fabrication and testing of flow characteristics in the whole range of operating conditions are costly and time-consuming. Such a calculation procedure giving correct data simplifies the atomizer design, and the final experimental verification at only several operation points is required. The discharge coefficient is also a suitable parameter for the validation of numerical simulations [27] that are challenging due to the two-phase flow nature as pointed out below. The C_D is defined as a ratio of the actual liquid mass flow rate to the theoretical flow rate as can be described by Equation (1) [28,29]:

$$C_D = \frac{m_L}{A_o \sqrt{2\rho_L \Delta p_{\text{mix}}}} \quad (1)$$

where m_L , A_o , ρ_L , and Δp_{mix} are the actual liquid mass flow rate, the exit orifice area, the liquid density, and the inlet-overpressure, respectively.

The C_D determination of the IMTF nozzles is not such an easy task because both phases flow through the same exit orifice area [26]. Additionally, the gas–liquid flow is complex, random, and challenging in the study since this two-phase flow combines the characteristics of a deformable interface and the compressibility of one of the phases. A notable effort was paid to study the factors that mark the C_D of the IMTF nozzles in order to obtain the C_D predictive correlations found in the literature. Previous works are summarized in Table 1. All previous investigations are consistent in the effect of gas-to-liquid ratio (GLR) on C_D (i.e., an increase in GLR leads to a non-linear decrease in the C_D , while the decrement rate is rapid at low GLRs). However, there is some contradiction about the effect of liquid properties, the inlet-overpressure ratio, and the exit orifice structure on the C_D . As a result, a comparison between the previously proposed models with the same conditions and geometries showed a notable difference in predictions of about 25%. Ochowiak [24] concluded that the full model for the C_D calculation is unavailable yet, since there are complex relations between dependent and independent factors that affect the nozzle discharge.

Applications of IMTF nozzles cover a wide range of operating conditions and liquid rheological properties. The literature review (i.e., Table 1) suggests that the investigated factor ranges are in many cases very limited and narrow. Some factors were not included in the investigations at all, and a general and precise C_D model has not yet been presented. Each of the studies was carried out on a single IMTF nozzle type, and only two old types (outside-in-gas (OIG) and inside-out-gas (IOG)) were examined. Therefore, each of the presented correlations is valid only for the type used, and no discharge characteristics are available for the newly introduced and promising IMTF nozzles. Additionally, the different IMTF nozzles possess a different way and intensity of mixing. This results in forming different internal flow regimes and characteristics [30,31]. Models based on the physical analysis of the two-phase discharge (i.e., [28]) suggest that, at the same operating conditions, the internal flow form affects the discharge characteristics; homogeneous gas-liquid flow (e.g., the homogenous bubbly flow) results in a C_D decrease, and it is vice versa for separated flow (e.g., slug flow). Therefore, the internal-mixing method is expected to affect C_D values, and the literature shows that this factor has not yet been investigated.

Consequently, in the present work, we tried to address these gaps by conducting a detailed experimental study on the discharge characteristics (i.e., C_D) of different types of IMTF nozzles. Several factors were considered, including operation conditions, liquid types, and exit orifice geometry. The covered ranges of these factors were widened in comparison with the available literature. To study the effect of the internal-mixing method factor, for the first attempt, five types of IMTF nozzles with widely different internal-mixing methods were tested under the same conditions, liquid types, and orifice geometry. Two of the selected nozzles (outside-in-gas (OIG) (and inside-out-gas (IOG)) are well-known and frequently used [1,5,10,13,24,28,30–36]. The following two designs (outside-in-liquid (OIL) and air-core-liquid-ring (ACLR)) were introduced recently; they are promising due to a better atomization performance [10,30,31,34], [37,38]. The last (swirling-air-core-liquid-ring (SACLR)) nozzle represents our new design, inspired by the nozzles described by García et al. [7] Stähle and colleagues [34]. Finally, we proposed a precise model for C_D prediction as a function of all the investigated factors.

Table 1. Summary of the published discharge coefficient (C_D) equations.

Ref.	Varied Factors						
	Nozzle Geometry			Working Conditions		Physical Properties	
	Type	do (mm)	do/D _{mix} (-)	L _o /d _o (-)	GLR (%)	Δp _{mix} /p _{amb} (-)	μ _L (kg/m·s)
[26]	OIG	1.2–2	0.19–0.3	0.5	1–12	1.36–2.04	0.001–0.1
[39]	OIG	2.5–4	0.31–0.5	0.5	0.5–20	1.48–2.96	0.001
[28]	OIG	2.5	0.45–0.18	0.28	2–10	0.99–4.9	0.0185
[35]	IOG	1.7–4.9	0.09–0.25	0.3–0.88	2–46	N/C ¹	0.001
[24]	IOG	2.7	0.14	N/M ²	1.4–57	N/M	0.001–0.182
Conclusions							
Ref.	Proposed Model						Notes
[26]	$C_D = c \left(1 - \frac{Q_G}{Q_G + Q_L} \right)^{0.3} \left(1 + \frac{1}{GLR} \right)^{0.15}$						- c is a constant, depends on the liquid properties, and is equal to 0.385 for water. C_D decreases with increasing gas-to-liquid ratio (GLR), while its value increases when the inlet-overpressure increases. The effect of viscosity is negligible.
[39]	$C_D = 0.3 - 0.0002 [GLR \times Re^{0.5}]$						- For horizontal flow. This model is valid for $2000 < Re < 20,000$, with a coefficient of determination (R^2) of 0.982. The inlet-overpressure has a small effect on C_D ; increasing the gas pressure leads to decrease the C_D .
[28]	$C_D = 0.62 \left(\frac{\mu_L}{\mu_w} \right)^{0.04} \left(\frac{\sigma_L}{\sigma_w} \right)^{0.02} \times \left(\frac{L_o}{d_o} \times \sin(2\beta) \right)^{0.5} \times \frac{G}{\sqrt{(2\rho_L \Delta p)}} \left(\frac{1}{1 + GLR} \right)^{-0.11}$						- The terms that reflect the effect of physical properties were adopted from Chin and Lefebvre [29]. G can be obtained by combining two models (the homogeneous flow model (HFM) and the separated flow model (SFM)) of two-phase discharge, which can be obtained via the online multiphase flow calculator [40]. C_D increases when liquid viscosity, surface tension, or GLR increases, while it is vice versa for the orifice length-to-diameter ratio.
[35]	$C_D = 0.0088 \left(GLR \times \frac{d_o}{D_{mix}} \right)^{-0.75} \pm 14\%$						- The atomizer construction shows a strong effect on C_D , in which C_D increases when the orifice diameter d_o decreases.
[24]	$C_D = 0.0361 \left(\frac{\mu_L}{\mu_w} \right)^{-0.0475} \times GLR^{-0.8}$						- C_D decreases with increasing liquid viscosity or GLR. In addition, C_D is affected by the flow character and the atomization mechanism.

¹ N/C: not constant; the mixing chamber pressure changes with gas–liquid ratio. ² N/M: not mentioned. Abbreviations: OIG: outside-in-gas; IOG: inside-out-gas.

2. Materials and Methods

Experiments for measurement of C_D and other relevant data were conducted by a cold atmospheric test rig at room temperature of 20 °C. A schematic drawing for the test rig is shown in Figure 1. The air was used as the gas phase, while water and aqueous glycerol solutions simulated the liquid phase. A wide range of operation factors was covered during the nozzle testing (i.e., GLR ranged between 1% and 65% and $\Delta p_{\text{mix}}/p_{\text{amb}}$ varied between 0.29 and 5.9).

An illustration of the operating conditions used for all the nozzles testing such as mass flow rate of the gas and the liquid, and the corresponding GLRs with their uncertainty values are documented in Table S1. Note that the $\Delta p_{\text{mix}}/p_{\text{amb}}$ was kept constant, while the GLR varied. Increasing the liquid flow rate resulted in a gradual decrease in the gas flow rate. The μL ranged between 0.001 and 0.2132 Pa·s (see Table 2), and the L_o/d_o ratio of the exit orifice varied between 0.5 and 4.2, while the convergence angle upstream the orifice (2β) was kept constant at 90° (see Table 3). The rheological properties of the liquids were found as follows. The density was calculated by dividing the known liquid volume by its weight, while the viscosity was measured by a controlled stress-rotary rheometer (DHR-2, TA Instruments, New Castle, DE, USA). Viscosity measurements were conducted at a constant shear rate of 10 s^{-1} .

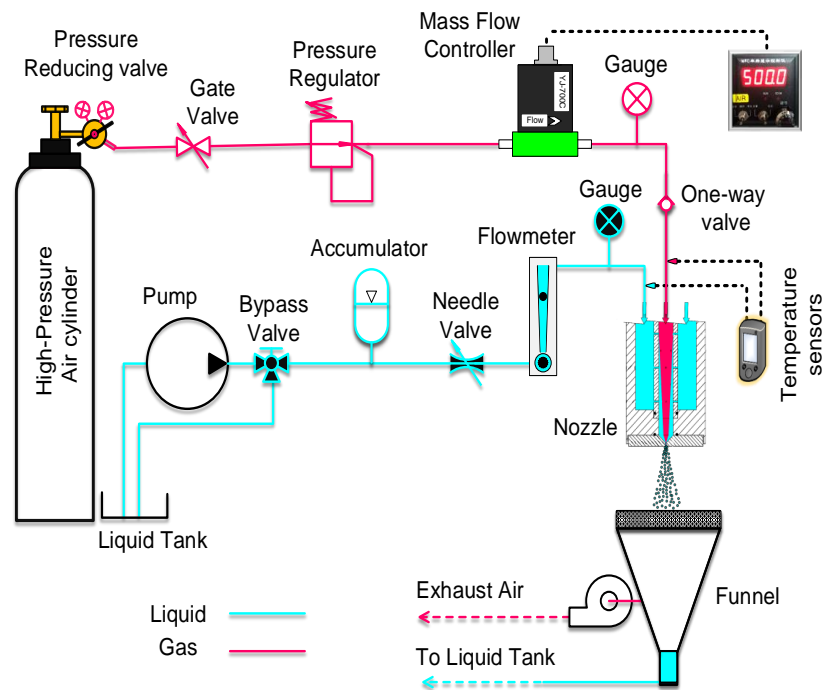


Figure 1. Schematic diagram of the fluid supply setup.

Table 2. Rheological properties: density (ρ), viscosity (μ), surface tension (σ), and the corresponding uncertainty values u_r (γ) of the tested aqueous glycerol solutions at 20 °C.

Mixture	Glycerol (wt%)	ρ (kg/m^3)	u_r (ρ) (%)	μ (Pa·s)	u_r (μ) (%)	σ (kg/s^2)	u_r (σ) (%)
M1	0	998.2	-a	0.001	-a	0.0731	-a
M2	70	1190	1	0.0241	4.68	0.0679	-a
M3	80	1216	1	0.0697	8.91	0.0672	-a
M4	90	1241	1	0.2132	7.73	0.0665	-a
Properties of the air at 20 °C.							
Air		1.23		1.81×10^{-5}			

-a: values were taken from the literature [7,41–43].

Table 3. The tested exit orifices geometry: exit orifice diameter (d_o) and length (L_o).

No.	d_o (mm)	L_o (mm)	β (°)	L_o/d_o (–)
1	0.85	0.42	45	≈0.5
2	0.85	0.77	45	0.9
3	0.85	1.46	45	≈1.7
4	0.85	2.2	45	≈2.6
5	0.85	2.89	45	3.4
6	0.85	3.57	45	4.2

2.1. Fluid Supply System

The liquid was driven by a commercial plunger pump (AR-RC 14.16 N, Modena, Italy) through a cylindrical accumulator used for damping the flow fluctuations. Sets of devices such as a needle valve for volume flow adjustment and a rotameter (DK800, KROHNE, Duisburg, Germany) were used. Rotameters with measuring ranges of 0.25–2.5 L/h, 1.2–12 L/h, 4–40 L/h, and 16–160 L/h were used for measuring the liquid volumetric flow rate with an accuracy of $\pm 2.5\%$ of the measuring value. The pressure gauge was fixed to measure the liquid pressure, as shown in Figure 1. The flow directed to the nozzle was controlled by a bypass valve. The pressurized air was delivered from a high-pressure air cylinder, passed through a gate valve, a pressure regulator used to control the $\Delta p_{\text{mix}}/p_{\text{amb}}$. A Bourdon pressure gauge (ZYIA, Zhejiang, China, accuracy $\pm 1.6\%$ of the full scale) was fixed very close to the nozzle in order to measure the gas pressure. A non-return valve was used for the prevention of the water backflow to the gas line. A thermal mass flow controller (YJ-700CF—Air—10–500 SCCM, and YJ-700CV—Air—0.5–25 SLPM, Nanning, KONXIN, China, accuracy $\pm 1\%$ of FS) measured the gas mass flow rate. The funnel collected the sprayed liquid. Two K-type temperature sensors were located after the pressure gauges and used for measuring the gas and liquid temperatures.

When conducting the experiments of high viscosity mixtures, the plunger pump was replaced with a gear one. Additionally, before these experiments, the liquid flow meters were calibrated for metering the mixtures with different viscosities. The uncertainties for liquid density, liquid flow rate, and gas flow rate were used in the error analysis calculations for both the GLR and C_D according to Holman [44]. The application of the mass flow controller for gas precluded the issue of gas density change when varying the pressure that would affect the true flow rate value. It also allowed us to achieve a large number of flow rate measurements (see Table S1) required for accurate and precise regression analysis.

The inlet conditions such as the flow fluctuations of the supplied fluids, eventual presence of dissolved gas in the liquid or inlet turbulence intensity can affect the performance of the atomizers and harm their comparison. The inlet fluid fluctuations were prevented using several precautions. The liquid supply line was equipped with a cylindrical accumulator, which provided a damping of liquid fluctuations eventually caused by the plunger pump. Additionally, to avoid the pulsations in the gas supply line, a high-pressure air cylinder was used instead of a reciprocating compressor to supply the pressurized air to the atomizer. Moreover, to avoid the effect of the possible presence of dissolved gas in the liquid, a treated water was used instead of the tap water. The treated water also prevents the metal components of the test rig to get oxidized and rusted.

2.2. Tested Nozzles

All tested IMTF nozzles were designed to work nearly under the same operation conditions; they feature comparable dimensions, as displayed in Figure 2, but their internal-mixing method for the two phases differs.

The OIG nozzle (Figure 2a) was investigated many times [1,5,10,13,28,30–34,36]. The mixing method is usually as follows: The liquid flows inside the main perforated chamber (aerator), while the gas is injected into the flowing liquid stream from outside through the aerator holes, perpendicularly

to the main axis, resulting in a formation of a two-phase flow inside the mixing chamber that is located in the space between the last row of injection holes and the top of the exit orifice. The aerator contains 20 injection holes, each 1 mm in diameter, arranged in five rows. These rows are separated by 10 mm and turned by 45° to the neighboring one. The internal flow pattern is mainly influenced by the GLR and varies among bubbly, slug, intermittent, and annular flow [13,36].

The OIL nozzle (Figure 2b) was first introduced by Mlkvik et al. [30] in 2015 and studied later [31,42]. Its design is similar to the OIG nozzle, but the mixing method is totally different; the gas flows through the main perforated chamber, while the liquid enters into the flowing gas stream from outside through the holes at the chamber wall. This mixing method is most similar to the Y-Jet nozzle mixing. The internal flow pattern for the OIL nozzle was theoretically predicted to be annular at any GLR [30,31,42], but this hypothesis has not yet been experimentally proven.

The IOG nozzle (Figure 2c) has been frequently investigated as well [5,24,28,35,45,46]. In this configuration, the liquid flows inside the main chamber, while the gas enters into the aerator tube and bubbles outward to mix with the liquid. The aerator, as in the previous designs, contains 20 injection holes, each 1 mm in diameter, arranged in five rows. These rows are separated by 10 mm and turned by 45° to the neighboring one. The aerator internal diameter is 4 mm, and its wall is 1 mm thick. The internal flow pattern is heavily influenced by the GLR and thus similar to the OIG configuration.

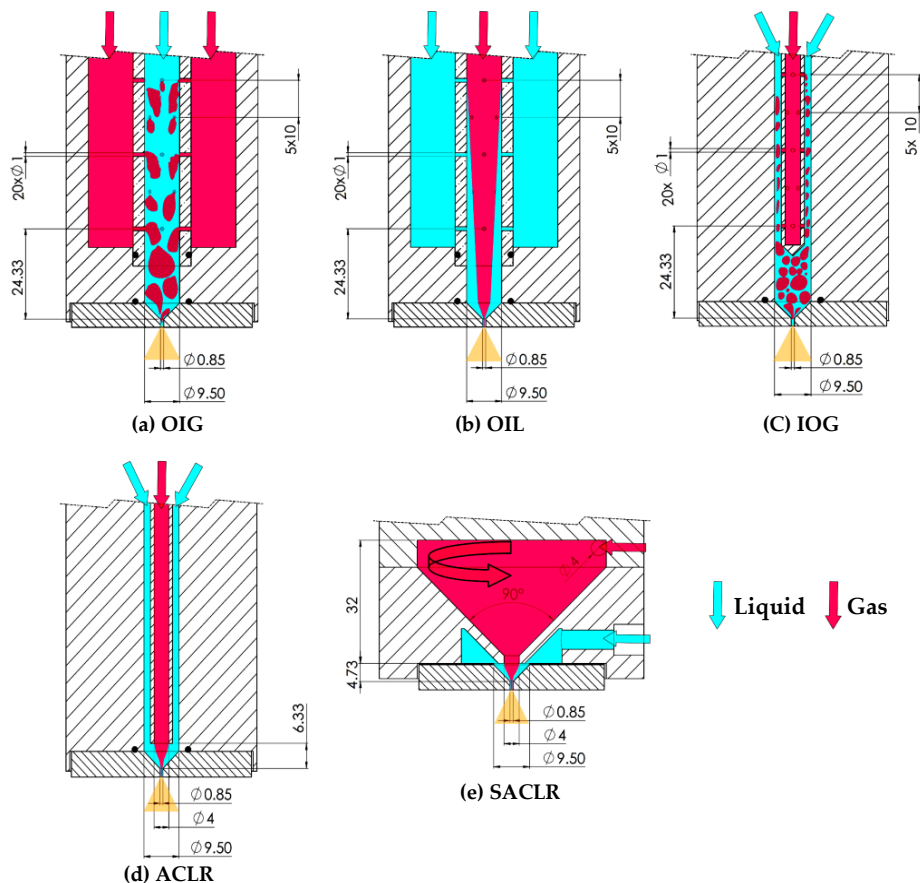


Figure 2. The geometry of the tested internal-mixing twin-fluid (IMTF) nozzles with their main dimensions in mm: (a) outside-in-gas (OIG); (b) outside-in-liquid (OIL); (c) inside-out-gas (IOG); (d) air-core-liquid-ring (ACLR); (e) swirling-air-core-liquid-ring (SACLR).

Correspondingly, the ACLR nozzle (Figure 2d) was designed and introduced by Stähle et al. [34] and studied later by other researchers [10–12]. In contrast to the OIG and IOG nozzles, the air in the ACLR nozzle is not injected into the liquid through multiple holes perpendicularly to the main axis, but through a single capillary positioned coaxially to the main axis and it exits very close to the

discharge orifice. This arrangement produces a high-velocity air core surrounded by a liquid annulus in the mixing position, which provides an annular flow pattern inside the orifice at any GLR [34].

Finally, the new SACLRL nozzle was designed to introduce an additional type of internal mixing which involves a swirling motion. Figure 2e shows that, in this case, the air enters through a tangential inlet at the top and swirls inside the conical conduct towards the exit hole at the base. This hole is coaxial to the main atomizer axis and ends very close to the top of the discharge orifice. The air exits this hole to enter into the mixing chamber in the form of a swirling air core, while the liquid is introduced parallel to this core, which forms a liquid ring around the swirling core directly upstream the exit orifice.

Considering the phase change phenomena, it is difficult for the liquid inside the IMTF atomizers to be exposed to an intense nucleation at any region. Neither the internal design nor the operating conditions help to initiate such a process. As shown in the literature review, and in the present study, these atomizers usually work under very low inlet-overpressure ($\Delta p_{\text{mix}}/p_{\text{amb}} < 6$) and the mixing of the two phases carries out at a constant pressure. This means that the liquid pressure outside the main perforated chamber (i.e., in the OIL design case) is slightly higher than that inside the chamber. In addition, the number and the diameter of the holes in the chamber are large (i.e., 1 mm, see Figure 2) in comparison with the first, small orifice of the flash-boiling atomizer (i.e., 0.1 mm in diameter). In addition, the liquid temperature is low and is usually equal to the room temperature. Consequently, the liquid flows through these holes with a negligible pressure drop, and therefore without any phase transition. After the mixing process, the two-phase (water–air) mixture flows through the convergent/exit orifice region (the convergence angle upstream the orifice is 90°). During this, the average mixture pressure gradually drops down until it equalizes with the atmospheric pressure. Furthermore, it was observed in other studies [47–49] that the nucleation phenomena considerably reduces and delays if a small amount of dissolved gas is present in the pure liquid. In other words, the gas bubble presence in the liquid inhibits the phase change process if compared to the pure liquid case. Therefore, the cold water–air mixture flowing through the convergent/exit orifice region in the IMTF atomizer is not likely to be exposed to the phase change process. It is expected that this phenomenon does not affect the discharge characteristics of the IMTF atomizers.

2.3. Internal Flow Visualization

For the flow visualization inside the exit orifice, the segment containing the contraction zone and exit orifice was machined (with $d_o = 0.85$ mm, and $L_o/d_o \approx 1.1$) in a square cross-section acrylic (Plexiglas) block to reduce the optical distortion caused by the curved shapes. This part was backlit illuminated by a diffused 500 W light. A high-speed camera (Fastcam SA 1.1, Photron, Tokyo, Japan) with a long distance microscope (Questar-QM1) was used to capture the flow images with 2000 frames/s rate and 1024×1024 pixel image resolution.

2.4. Methodology of Data Analysis

The literature review revealed that C_D is affected by working conditions [26], the exit office size and geometry [28,35], and the physical properties of the phases [24,28]. Here a general function applies:

$$f(C_D, Q_G, Q_L, \Delta p_{\text{mix}}, p_{\text{amb}}, L_o, d_o, \mu_L, \mu_G, \rho_L, \rho_G) = 0 \quad (2)$$

After performing the dimensional analysis, C_D can be expressed by Equation (3):

$$C_D = f\left(\text{GLR}, \frac{L_o}{d_o}, \frac{\Delta p_{\text{mix}}}{p_{\text{amb}}}, \frac{\mu_L}{\mu_G}\right) \quad (3)$$

The form of C_D correlation can be obtained by applying a non-linear regression analysis with the least-squares method to the experimental data. Note that the gas-to-liquid mass flow ratio is defined as shown by Equation (4):

$$\text{GLR} = \frac{m'_G}{m'_L} \quad (4)$$

The effect of the investigated factors on C_D was evaluated via an analysis of variance (ANOVA) test. ANOVA is a form of statistics hypothesis testing that determines the effect of one or more factors on an outcome variable by comparing the means of different groups. ANOVA is widely used in the analysis of experimental data [50]. The test result, which is usually presented by the probability (p -value) that is calculated from the null hypothesis and the experimental data sample, is considered statistically significant when the p -value is less than a pre-specified threshold. A predictor that keeps a low p -value has a significant effect on the model accuracy as changing that predictor value causes a considerable change in the model response. On the other hand, high p -value predictors do not have a significant effect on the model response. For the identification of significant differences, the alpha value was set equal to 0.05.

$$\alpha_i = \frac{A_G}{A_o} \quad (5)$$

$$\text{CV}_{\alpha_i} = \frac{\sigma_{\alpha}}{\alpha_{\text{avg}}} \quad (6)$$

$$\sigma_{\alpha} = \sqrt{\frac{1}{N} \sum_1^N (\alpha_i - \alpha_{\text{avg}})^2} \quad (7)$$

$$S = \frac{U_G(1 - \alpha_{\text{avg}})}{U_L \alpha_{\text{avg}}} \quad (8)$$

For each case, 250 flow visualization images were captured and processed, and this number was found sufficient for the minimization of bias errors in the measurement. Thereafter, MATLAB software was used for determining the gas phase boundaries at the entrance of the exit orifice and subsequently determining the diameter of the gas core (see Figure 3). The gas core diameter was used to measure the time variation of the gas phase inside the exit orifice, which corresponds with the internal flow regime [51,52]. Additionally, this diameter can be used to indicate some parameters for the gas–liquid flow (i.e., the gas void fraction and the corresponding slip ratio), as represented by Equations (5)–(8).

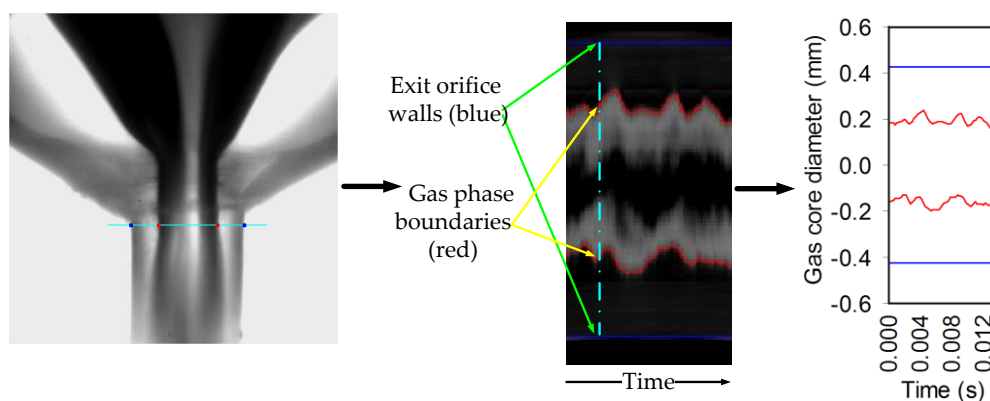


Figure 3. Processing procedure for determining the gas core diameter at the entrance of the exit orifice.

α_i is the time-varying void fraction, A_G is the cross-sectional area occupied by the gas phase, A_o is the orifice cross-sectional area, CV_{α_i} is the coefficient of variation of α_i , which reflects its fluctuation, σ_{α} is the standard deviation of α_i , α_{avg} is the time-averaged void fraction, and N is the image number. The CV_{α_i} increase indicates a wide distribution range of α_i and its high fluctuations. S is the slip ratio (also called the velocity ratio), which represents the actual velocities ratio of both gas and liquid phases, respectively. U_G and U_L are phase superficial velocities of the gas and liquid, respectively.

2.5. Analysis of the Experimental Errors

The uncertainties in the experimental results are caused by the errors in the primary measurements. Experimental error analysis was provided according to Holman [44]. Let R be a function given as

$R = f(X_1, X_2, X_3 \dots X_n)$, where $X_1, X_2, X_3, \dots, X_n$ are independent variables with uncertainties of $W_{X_1}, W_{X_2}, W_{X_3}, \dots, W_{X_n}$, respectively. Therefore, the uncertainty in the R value (W_R) can be calculated as follows:

$$W_R = \pm \sqrt{\left(\frac{\partial R}{\partial X_1} \times W_{X_1}\right)^2 + \left(\frac{\partial R}{\partial X_2} \times W_{X_2}\right)^2 + \dots + \left(\frac{\partial R}{\partial X_n} \times W_{X_n}\right)^2} \quad (9)$$

and the relative error is

$$E_R = \frac{W_R}{R} \times 100\% \quad (10)$$

The relative error of the dependent variables is determined as follows:

Density:

$$W_\rho = \pm \sqrt{\left(\frac{1}{v} \times W_m\right)^2 + \left(\frac{-m}{v^2} \times W_v\right)^2} \quad (11)$$

Mass flow rate of liquid and gas:

$$W_m = \pm \sqrt{(\rho \times W_Q)^2 + (Q \times W_\rho)^2} \quad (12)$$

Gas-to-liquid ratio:

$$W_{GLR} = \pm \sqrt{\left(\frac{1}{m_l} \times W_{m_g}\right)^2 + \left(\frac{-m_g}{m_l^2} \times W_{m_l}\right)^2} \quad (13)$$

Discharge coefficient:

$$W_{C_D} = \pm \sqrt{\left(\frac{1}{A_o \sqrt{2\rho_1 \Delta p}} \times W_{m_l}\right)^2 + \left(\frac{-m_l}{A_o^2 \sqrt{2\rho_1 \Delta p}} \times W_{A_o}\right)^2 + \left(\frac{-m_l}{2A_o \rho_1^{(3/2)} \sqrt{2 \Delta p}} \times W_{\rho_1}\right)^2 + \left(\frac{-m_l}{2A_o \Delta p^{(3/2)} \sqrt{2 \rho_1}} \times W_{\Delta p}\right)^2} \quad (14)$$

The relative errors for the C_D are displayed as error bars at each measuring point in the following result graphs. For instance, the uncertainties for mass flow and GLR are documented in Table S1.

3. Results and Discussion

In this section, the results of the OIL nozzle are presented, while the results of the other nozzles are presented in Figures S1–S4. The comparisons between the obtained results of the tested nozzles will be illustrated in Section 3.4.

3.1. The Effect of GLR and Orifice Structure on C_D

The effect of both GLR and L_o/d_o on the C_D of the OIL nozzle is shown in Figure 4. It is clear that C_D features a sharp decrease when increasing the GLR to $\approx 30\%$, while a further GLR increase has little effect on C_D (see Figure 4a where the absolute values for the C_D are documented). This behavior results from the link between the GLR and the flow area occupied by the liquid inside the exit orifice [26]. The actual area available for the liquid flow decreases with any increase in the gas flow rate (i.e., with increasing GLR). It is apparent that the discharge coefficients of the twin-fluid atomizers are, due to the gas blockage, lower than that reported for plain-orifice atomizers and single-fluid nozzles [53], simplex [54], and even spill-return pressure-swirl atomizers [55]. It moreover drops down below 0.1

for GLRs increasing above 10%. Figure 4b documents that, with a constant GLR, C_D decreases with an increasing L_o/d_o ratio. Here the friction force on the liquid flow plays a role, as it is proportional to the exit orifice length.

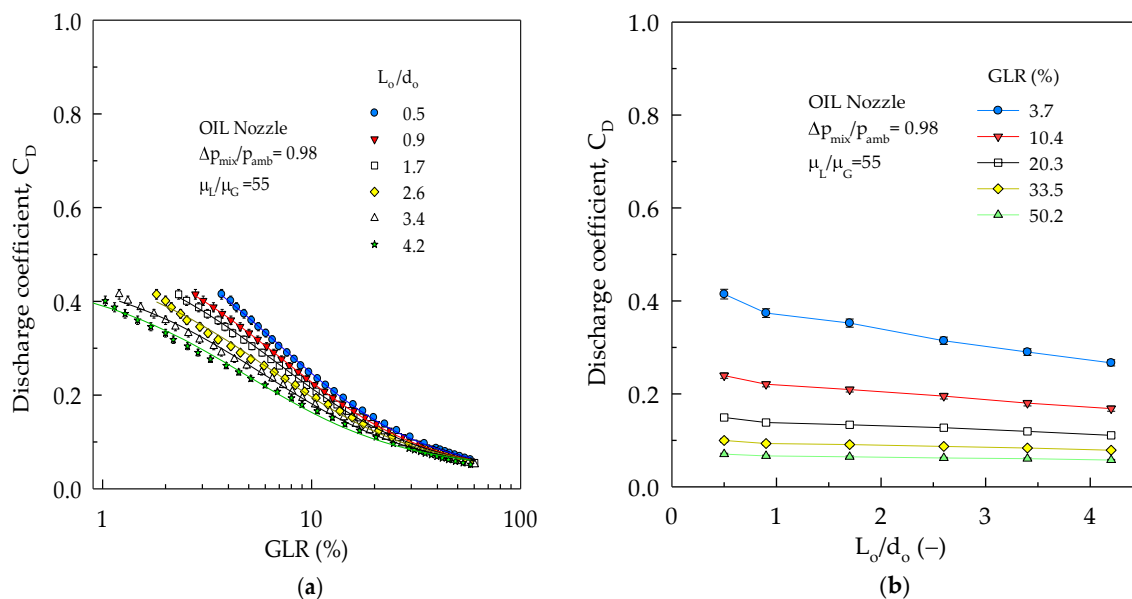


Figure 4. The effect of (a) GLR on C_D at different orifice configurations and of (b) L_o/d_o on C_D at different GLRs.

Additionally, the sensitivity of C_D to the L_o/d_o variation is high at low GLRs, while it gradually decreases with the GLR increase. The obtained results show that, when L_o/d_o increases from 0.5 to 4.2, there is a pronounced drop of C_D by 35.6% and 18% at GLRs of 3.7% and 50.2%, respectively. The reason is that, at low GLRs, the liquid occupies the largest portion of the exit orifice and is exposed to the largest friction force. Therefore, a small increase in the exit orifice length results in significant resistance to the liquid flow, and a notable C_D reduction. At high GLRs, the amount of liquid passing through the orifice is very small relative to the gas amount, and the friction force acting on this liquid is also small.

It was found that an exponential decay function $C_D = A \times \text{EXP}(B/(GLR + C))$ fits the experimental data accurately as can be seen in Figure 4. The results of other nozzles (OIG, IOG, ACLR, and SACLR) show similar trends as the OIL nozzle (see Figure S1). Regarding the results of the IOG nozzle, it is consistent with Jedelsky and colleagues [28] and Chin and Lefebvre [29] and inconsistent with Ochowiak [45]. C_D was found to increase to a maximum value as L_o/d_o increases to 2.04 and then to decrease with a further increase in L_o/d_o . Finally, results from the ANOVA show a statistically significant difference in C_D with varying GLRs and L_o/d_o (i.e., a p -value equal to 6.94×10^{-27} and 5.49×10^{-20} , respectively), so the effect of both GLRs and L_o/d_o factors on C_D cannot be neglected.

3.2. The Effect of the Inlet-Overpressure Ratio on C_D

The effect of $\Delta P_{mix}/P_{amb}$ (P_{amb} was constant in all the experiments, i.e., atmospheric pressure) on the C_D of the OIL nozzle is shown in Figure 5. Figure 5a documents the notable effect of GLRs on C_D at different $\Delta P_{mix}/P_{amb}$, while the effect of the $\Delta P_{mix}/P_{amb}$ is shown in Figure 5b. Figure 5b shows that an increase in $\Delta P_{mix}/P_{amb}$ at constant GLRs results in the C_D reduction to its minimum value, which occurred at $\Delta P_{mix}/P_{amb} = 0.98$. Then C_D increased for higher $\Delta P_{mix}/P_{amb}$ (i.e., > 0.98). Figure 6 illustrates the reasons for this non-monotonic phenomenon. Increasing the ΔP_{mix} at a constant GLR causes a decrease in the gas volume flux at the exit orifice due to the high gas compressibility compared to liquids. As a result, the liquid mass flux (G) increases. G features a significant change at $\Delta P_{mix} \geq 0.1$ MPa in comparison with its change at lower ΔP_{mix} . According to the known relation $C_D \sim$

$G/(\Delta p_{mix})^{0.5}$ [39], at $\Delta p_{mix} < 0.1$ MPa, the effect of the pressure increase is predominant and overcomes the effect of the G increase, so C_D decreases; it is vice versa for $\Delta p_{mix} > 0.1$ MPa (see Figures 5b and 6). Additionally, as aforementioned, Figure 5b shows that the increase in GLR suppresses the effect of $\Delta p_{mix}/p_{amb}$ on C_D because of a decrease in the liquid amount. This phenomenon is of significant importance for IMTF atomizers; firstly, it is related to C_D , which is important and an influential factor for atomizers. Secondly, this phenomenon occurs at regimes of low pressures and low GLRs (see Figure 5b), where the nozzles feature the highest atomization efficiency [56].

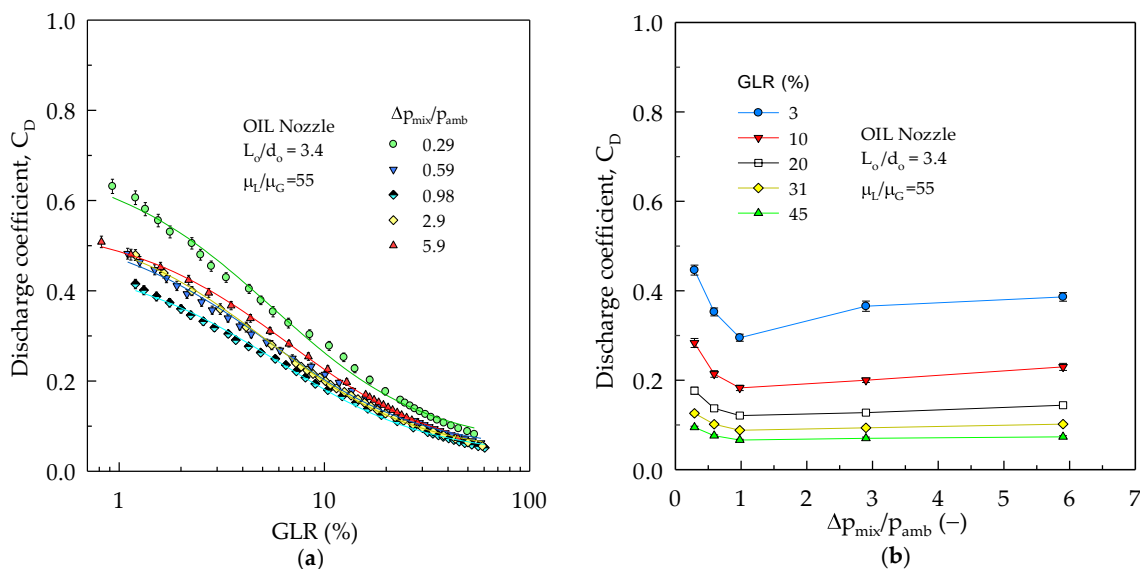


Figure 5. The effect of (a) GLR on C_D at different $\Delta p_{mix}/p_{amb}$ values and of (b) $\Delta p_{mix}/p_{amb}$ on C_D at different GLRs.

Additionally, we found that the other IMTF nozzles (OIG, IOG, ACLR, and SACLR) perform similarly with respect to the OIL nozzle (see Figure S2). It is worth mentioning that we observed this non-monotonic phenomenon in a wider range for the $\Delta p_{mix}/p_{amb}$ in this study than those in the literature (see Table 1). This finally explains and unifies the previous contradicting results by Chen and Lefebvre [26] and Ramamurthi et al. [39], who found that C_D monotonically increased and decreased with the $\Delta p_{mix}/p_{amb}$ in narrower operation ranges, respectively. Lastly, results from ANOVA show that the influence of $\Delta p_{mix}/p_{amb}$ on C_D is statistically significant and cannot be neglected (i.e., a p -value is equal to 1.88×10^{-10}).

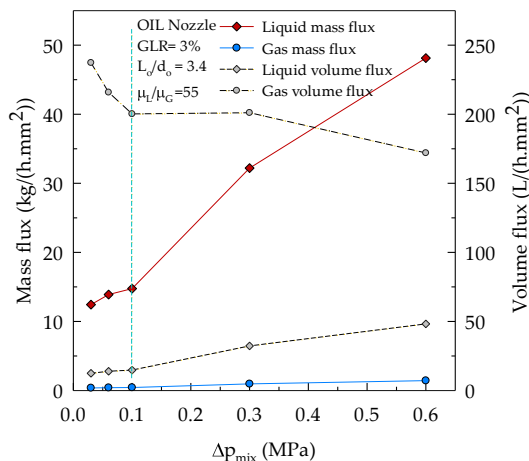


Figure 6. The effect of Δp_{mix} on the mass and volume fluxes at a constant GLR for the OIG nozzle.

3.3. The Effect of Liquid Viscosity on C_D

The effect of viscosity on the C_D of the OIL nozzle is shown in Figure 7. C_D decreases as the μ_L/μ_G increases at constant GLRs, as shown in the figures. This effect can be interpreted via the relation between C_D and the Reynolds number for the liquid flow (Re_L) (Equation (15)):

$$Re_L = \frac{\rho_L U_L d_o}{\mu_L} \quad (15)$$

where U_L is the cross-sectional average velocity of the liquid phase. C_D depends on Re_L in a linear manner [24,35,57], as confirmed in Figure 8a. The value of Re_L drops with rising liquid viscosity at constant GLRs (Figure 8b). By increasing the liquid viscosity, the internal friction of the liquid increases, which leads to a decrease in the discharge velocity of the liquid as well as the suppression of the flow turbulence [58,59], and C_D consequently decreases. The same effect on Re_L was found with the increase in GLR (Figure 8b). The sensitivity of C_D to the variation in μ_L/μ_G is high at low GLRs and it gradually decreases with increasing GLR; the increase of μ_L/μ_G from 55 to 11,779 results in a reduction in C_D by 53.7% at GLR = 3.7%, while it is 44% at GLR = 45.8%.

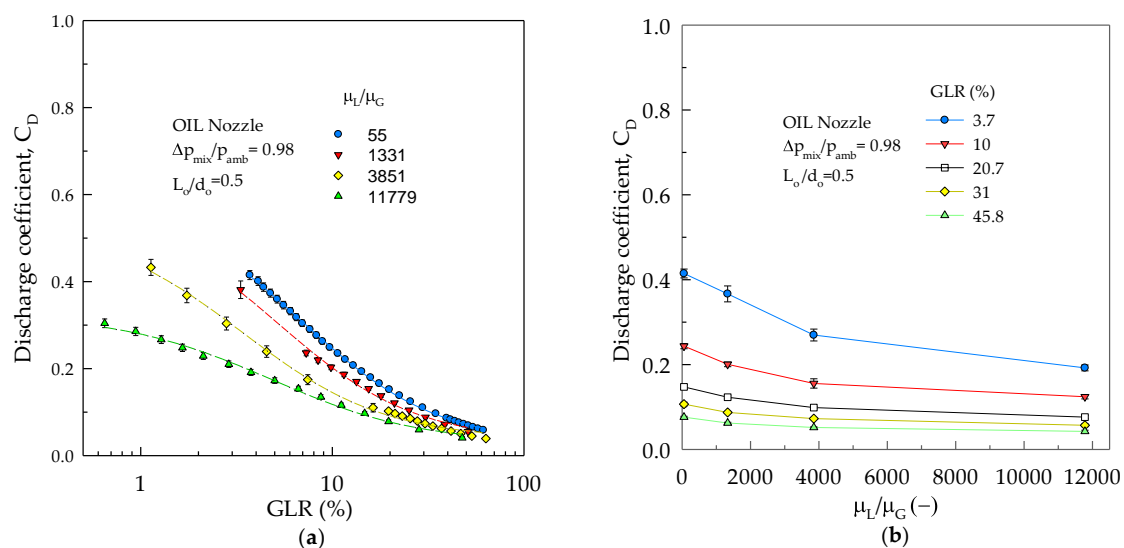


Figure 7. The effect of (a) GLR on C_D at different μ_L/μ_G and of (b) μ_L/μ_G on C_D at different GLRs.

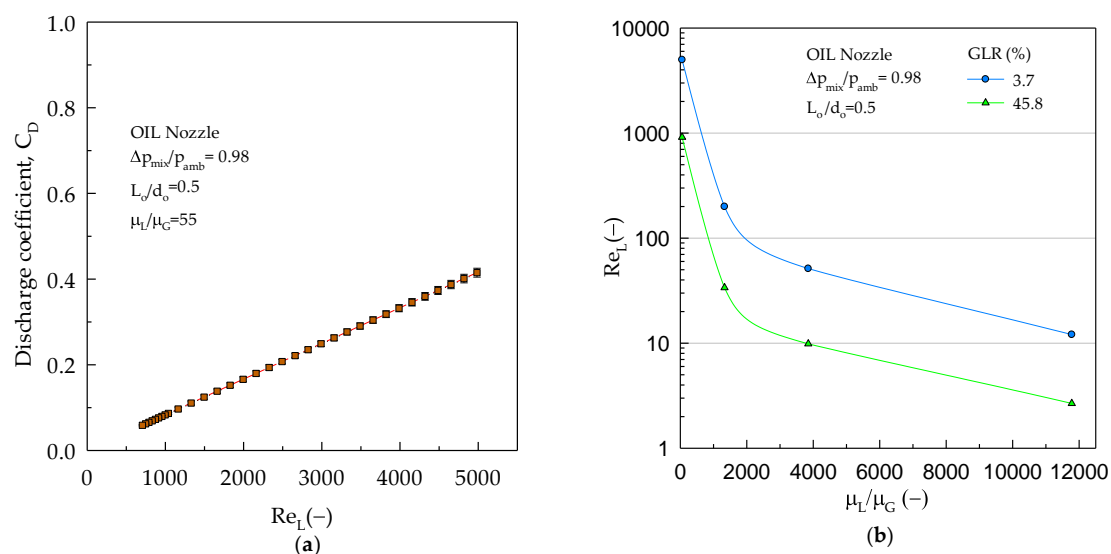


Figure 8. (a) The relation between Re_L and C_D ; (b) the relation between μ_L/μ_G and Re_L at different GLRs.

The results of other IMTF nozzles show similar tendencies as the OIL nozzle (see Figure S3). Our findings on the viscosity effect are in good agreement with the results reported by Ochowiak [24] for the IOG nozzles, while they are inconsistent with those reported by Chen and Lefebvre [26] for the IOG type. Once more, ANOVA results show that the p -value of the effect of μ_L/μ_G on the C_D is equal to 4.60×10^{-13} , so this effect cannot be neglected.

3.4. The Effect of the Mixing Method on C_D

As mentioned, the nozzle type outlines the mixing process. The influence of the mixing on C_D is shown in Figure 9 at different conditions. It is clear in Figure 9a–d that the differences in the C_D values are very small and may even approach zero. This finding was confirmed by ANOVA results that showed insignificant differences in C_D values among different nozzles (i.e., p -value is greater than 0.15). Overall, the effect of the mixing method factor is negligible.

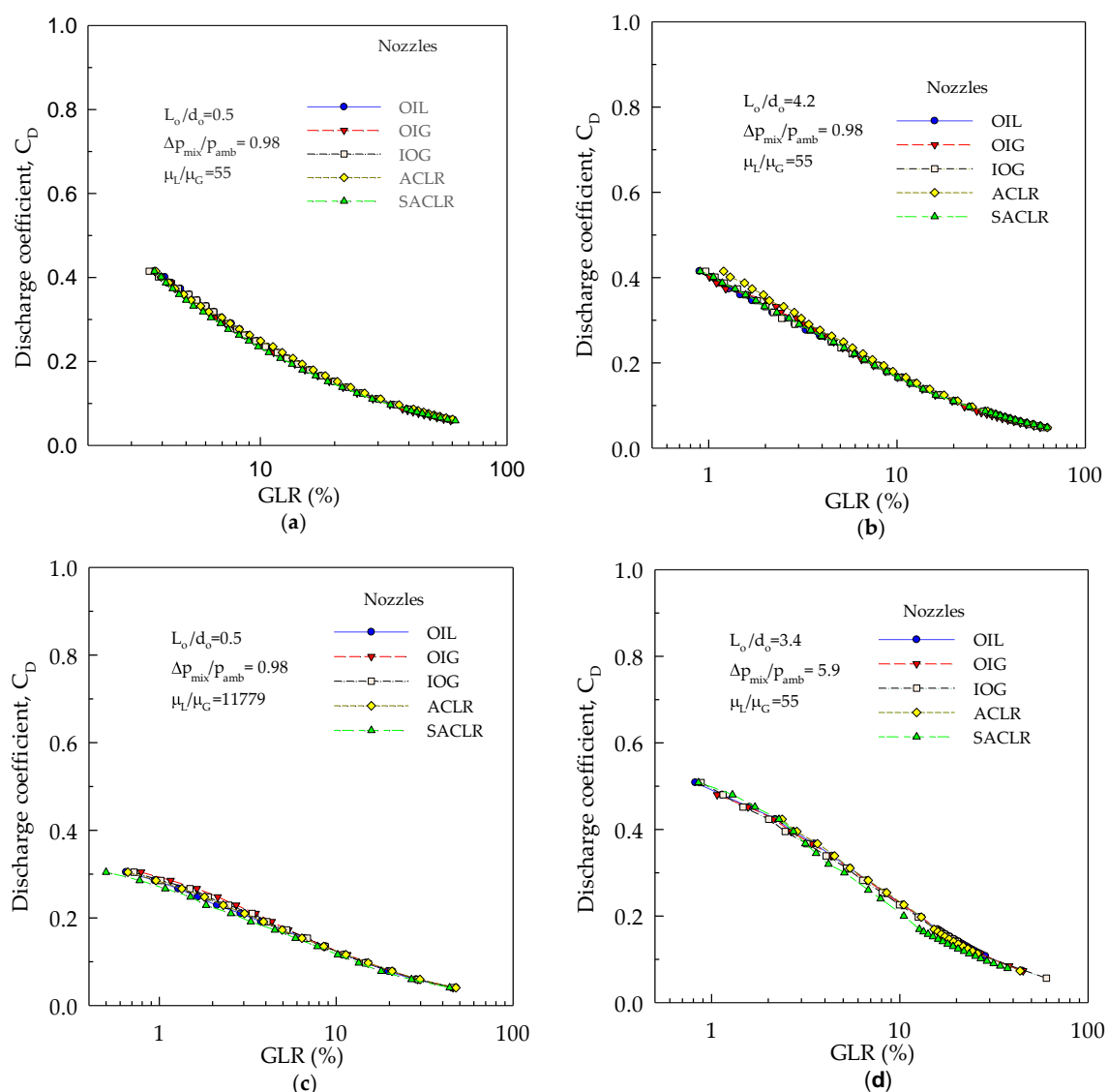


Figure 9. The effect of the mixing method on C_D at different conditions: (a) low values of all parameters; (b) high L_o/d_o value; (c) high μ_L/μ_G value; (d) high values of $\Delta p_{mix}/p_{amb}$ and L_o/d_o .

It is interesting that every nozzle type possesses its own internal flow pattern and characteristics, because the mixture formation and its character are strongly influenced by the mixing process [30,31], especially at low GLRs. The literature sources (see Table 4) show that the internal flow regime varies

from one nozzle to another between bubbly, slug, intermittent, and annular. Shepard [60] found that the flow patterns can change during the residence time of the mixture inside the atomizer, so the regimes at the exit orifice can differ from those at the gas injection area. Our visualization results for the actual flow inside the exit orifice for the tested nozzles showed significant differences in the flow nature between nozzles.

Table 4. Summary of the internal flow pattern of the tested IMTF nozzles from the literature.

Author/s	Nozzle Type	Investigation		Results of Internal Flow Pattern
		Type	Region	
Stähle et al. [13]	OIG	Visualization	Mixing chamber	Bubbly flow (GLR <2%) Intermittent (2% < GLR < 10%) Annular flow (GLR ≥10%)
Zaremba et al. [31]	OIL	Theoretically	Mixing chamber	Annular flow at any operation condition
Jobehdar et al. [61]	IOG	Visualization	Mixing chamber	Bubbly flow (GLR ≤0.1%) Slug flow (GLR = 0.7%) Annular (GLR >2.2%)
Stähle et al. [34]	ACLR	Visualization	Contraction zone and exit orifice.	Annular flow at any operation condition

Figure 10 shows the character of α_i at the entrance of the exit orifice for the tested nozzles at the given condition (i.e., GLR = 1.2%, $\Delta p_{\text{mix}}/p_{\text{amb}} = 0.98$, and $\mu_L/\mu_G = 55$). The variation in the character of α_i between the tested atomizers is certainly related to the nature of the internal flow regime inside the mixing chamber. It is clear in Figure 10a–e that α_i is not stable and features a different character for each nozzle, which means the internal flow regime inside each nozzle is quite different from the others. This also indicates that the internal flow of these nozzles is inherently unsteady, but with varying levels. Only the exit orifice for the OIL nozzle continuously possesses an annular flow, while the other nozzles possess different flow regimes.

The corresponding CV_{α_i} values for α_i in Figure 10, while C_D values and slip ratios for each nozzle are shown in Figure 11. Figure 11a shows that the CV_{α_i} features substantial differences between nozzles. In contrast, the differences in C_D between nozzles were not significant (i.e., p -value is equal to 0.24). These results reveal that the C_D of the IMTF nozzles is independent of the form of the internal flow regime.

It is worth mentioning that the C_D prediction model based on the analytical physical models of two-phase discharge (i.e., homogeneous flow model (HFM) and separated flow model (SFM), which correspond to the two-phase flow discharge without slip between phases and the discharge with maximum slip, respectively) predict that the C_D for the SFM should be higher than that for the HFM by three times [62]. The present experimental data show that there is no difference (≈ 0) in C_D values for the nozzles that feature the highest and lowest slip ratio (i.e., SACL and OIL, respectively) (see Figure 11a,b). Therefore, C_D prediction models should not rely solely on the analytical models of two-phase discharge.

Additionally, a comparison between the inlet turbulence intensity (Tu_1) values of the continuous single-phase flow entering the mixing chamber of all the atomizers is displayed in Figure 12. The gas represents continuous single phase for the OIL atomizer, while it is the liquid in the other configurations (OIG, IOG, ACLR, and SACL), as shown in Figure 2. Tu_1 can be calculated as follows [63,64]:

$$Tu = 0.16 \cdot (Re)^{-(1/8)} \quad (16)$$

Figure 12 compares the differences in Tu_1 among the tested nozzles. It suggests that the mixing process of the two phases, and consequently the nature of the produced two-phase mixture were possibly affected by the inlet conditions. The maximum difference in the Tu_1 is about 7.4% and applies

between the OIL atomizer (i.e., the highest Tu_1) and the SACLRL atomizer (i.e., the lowest Tu_1). While the difference in the C_D between the same atomizers (under the same operating condition) is only about 0.4%. It is implied that the C_D values were negligibly affected by the differences in Tu_1 .

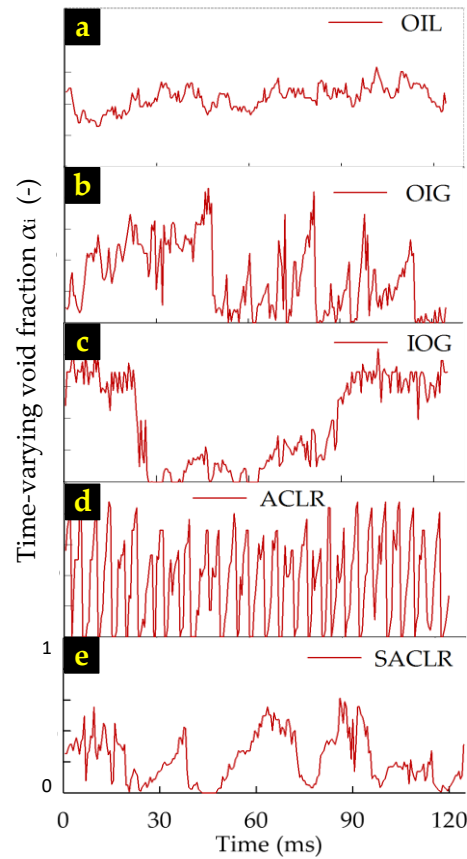


Figure 10. Effect of the nozzle type (i.e. (a) OIL; (b) OIG; (c) IOG; (d) ACLR; and (e) SACLRL) on the time-varying void fraction (α_i) inside the exit orifice of each nozzle under the operating condition of $GLR = 1.2\%$, $\Delta p_{mix}/p_{amb} = 0.98$, and $\mu_L/\mu_G = 55$.

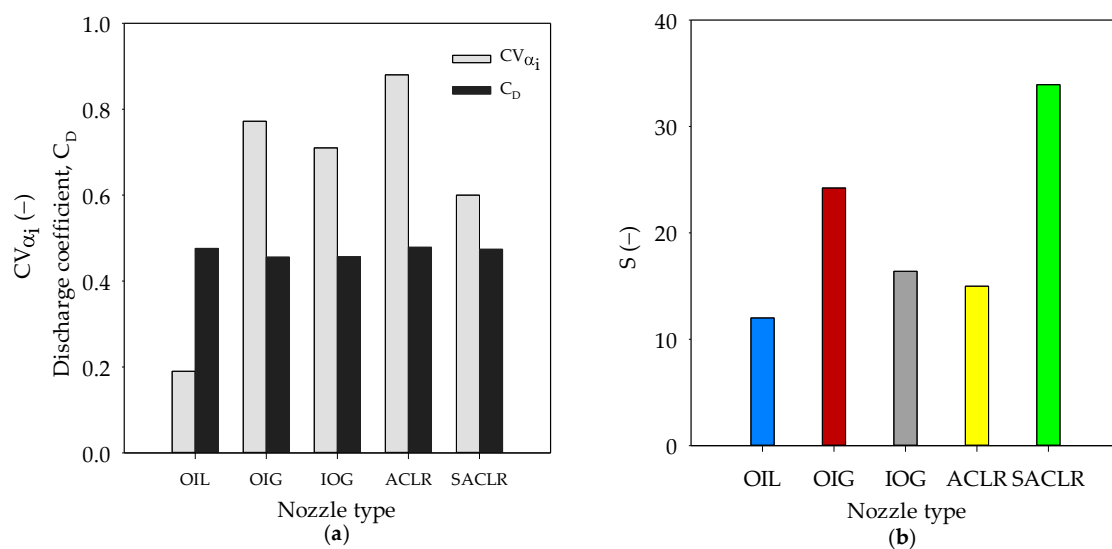


Figure 11. (a) Variation coefficient of gas void fraction and C_D ; (b) slip ratio of the flow inside the exit orifice for the tested nozzles: at $GLR = 1.2\%$, $\Delta p_{mix}/p_{amb} = 0.98$, $\mu_L/\mu_G = 55$.

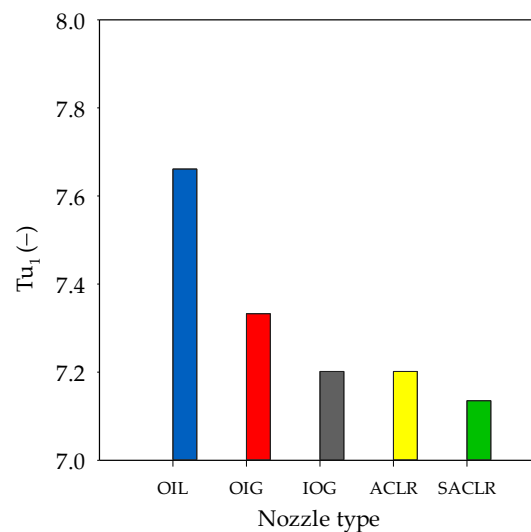


Figure 12. The inlet turbulence intensity (Tu_1) for the continuous single phase entering the mixing chamber for the tested atomizers at the condition of $GLR = 1.2\%$, $\Delta p_{mix}/p_{amb} = 0.98$, and $\mu_L/\mu_G = 55$.

Finally, Table 4 indicates that the internal flow inside the ACLR was annular flow, while our visualization showed a different flow character. This may be due to the change in the ratio between exit orifice diameter and gas capillary diameter; it was about 0.2 and 1 in the present and previous study, respectively. We set these dimensions so as to test all the nozzles under the same operating conditions and exit orifice geometry. Additionally, it is interesting that the flow nature inside the exit orifice of the OIL nozzle was more stable than the other nozzles. This behavior gives an advantage to this type, given that this stability will be reflected in the quality of the spray. Thus, the OIL nozzle can be useful in applications that require a more stable spray with high quality (combustion, surface coating, and powder generation).

3.5. The Proposed C_D Model and a Comparison with Published Models

Since the effect of the mixing method is negligible, a general C_D prediction model can be developed for different IMTF nozzles. The non-linear least-squares regression technique was applied to the experimental data of C_D for each nozzle, and it was found that the C_D can be correlated as follows:

$$C_D = 814 \cdot e^{(-0.086 \frac{L_e}{d_o})} \cdot \left[e^{(-2.89 \frac{\Delta p_{mix}}{P_{amb}})} + 0.496 \left(\frac{\Delta p_{mix}}{P_{amb}} \right)^{0.161} \right] \cdot \left(\frac{\mu_L}{\mu_G} + 11505 \right)^{-1} \cdot e^{(\frac{0.51}{GLR+0.179})} \quad (17)$$

This general equation is valid for all the IMTF nozzles (i.e., internal-mixing types). Table 5 summarizes the values of the coefficient of determination (R^2) and the sum of the squared residuals (SS). A comparison of the experimentally found C_D with the C_D predicted by Equation (17) for the OIL nozzle is shown in Figure 13, and the others (OIG, IOG, ACLR, and SACLR) are documented in Figure S4. The figures and the table confirm that the proposed C_D model fits the experimental data for each mixing type very closely with $R^2 \geq 0.99$.

A comparison between the present model with the other models published for the OIG and IOG nozzles in Figure 14 shows that the C_D predicted by the published models for both nozzles features a significant dispersion along the expected trend. This could be because of the neglect of some important factors in these equations, or because the ranges of the investigated factors are too narrow and inconsistent with present experiments (see Table 1), and/or because the effect of some factors is inconsistent with the findings of the present work. Note that the SFM ($a = 0$, $b = 1$) with polytropic expansion was selected during the calculations of Jedelsky's model in Figure 14a.

Table 5. Values of R^2 and sum of the squared residuals (SS) for the tested IMTF nozzles in Equation (17).

Nozzle	R^2 (-)	SS (-)
OIL	0.992	0.064
OIG	0.994	0.049
IOG	0.990	0.069
ACLR	0.991	0.090
SACLR	0.991	0.070

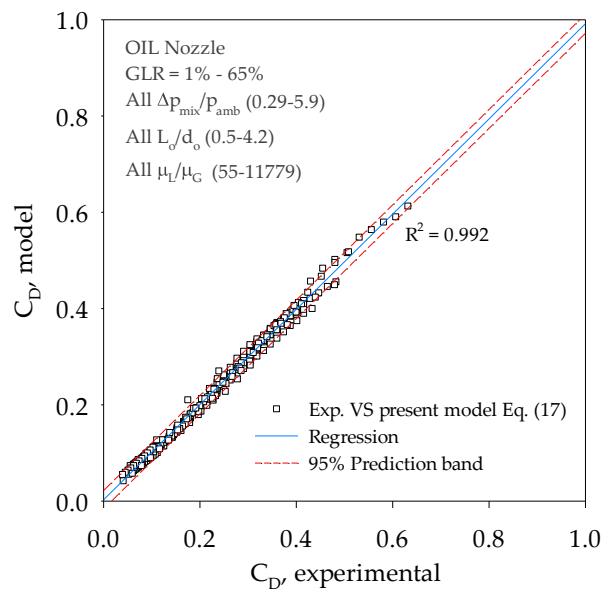


Figure 13. Comparison of the experimental and predicted C_D from the present model in Equation (17) for the OIL nozzle.

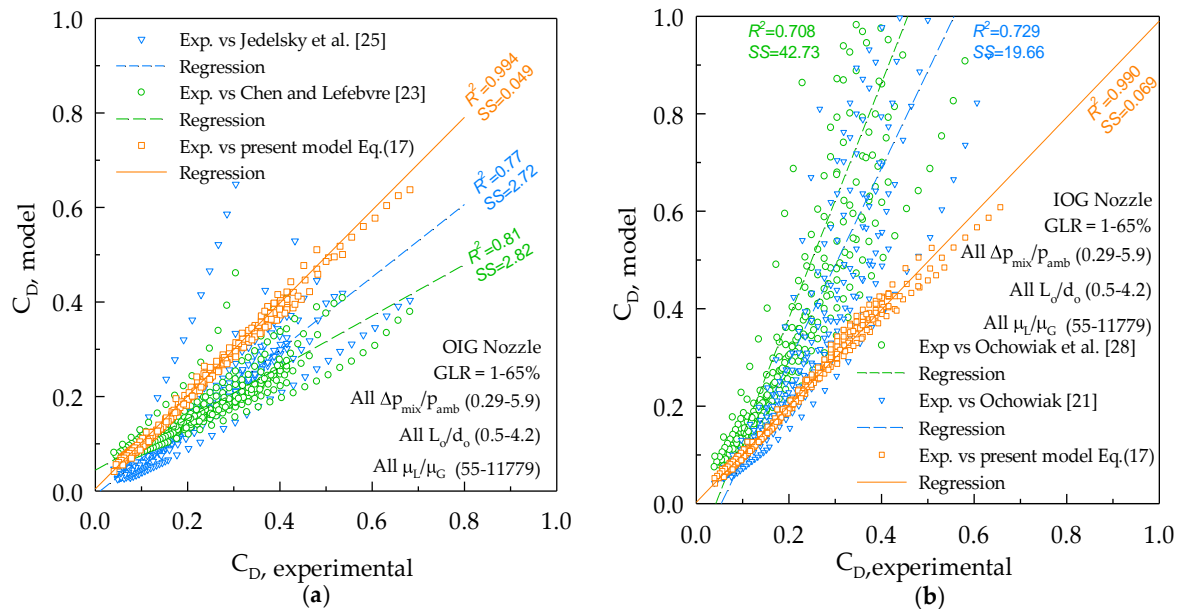


Figure 14. Comparison of the experiments and published models for (a) the OIG nozzle and (b) the IOG nozzle.

4. Conclusions

In this paper, a detailed experimental study was carried out to investigate and analyze the effect of many factors upon the discharge coefficient of different types of internal-mixing twin-fluid atomizers. These factors mainly are the gas-to-liquid mass flow ratio, the inlet-overpressure ratio, the orifice length-to-diameter ratio, the internal mixing method of the two phases, and the liquid viscosity. Five different types of IMTF nozzles—OIL, OIG, IOG, ACLR, and SACL—were tested under the same working conditions, liquid types, and exit orifice geometry. Additionally, the internal flow at the entrance of the exit orifice was visualized using a high-speed camera at a certain operating condition for each nozzle. Based on the results, the main conclusions are as follows:

1. The GLR, L_o/d_o , μ_L/μ_G , and $\Delta p_{mix}/p_{amb}$ have a significant effect on the C_D of IMTF nozzles, while the effect of the mixing-method factor is negligible.
2. C_D decreases with the increase of GLR, L_o/d_o , or μ_L/μ_G , and the same trend was observed with the increase of $\Delta p_{mix}/p_{amb}$ up to a value of 0.98, but C_D increases with any further increase in $\Delta p_{mix}/p_{amb}$.
3. The sensitivity of C_D to changes in L_o/d_o , $\Delta p_{mix}/p_{amb}$, or μ_L/μ_G is high at a low GLR but declines at a high GLR.
4. Flow visualization inside the exit orifice showed a diverse internal flow nature for each nozzle, which indicates that the C_D of the IMTF nozzles is independent of the form of the internal flow regime at any given condition.
5. The internal flow inside the exit orifice of the tested IMTF nozzles is inherently unsteady, but with varying levels; the OIL nozzle features a more stable internal flow behavior than the other nozzles.
6. Finally, a general well-fitting C_D prediction model (Equation (17)) with an $R^2 \geq 0.99$ was proposed, which is valid for any type of IMTF nozzle and with wider validity ranges for the included factors.

Supplementary Materials: The following are available online at <http://www.mdpi.com/2227-9717/8/5/563/s1>, Figure S1: Effect of GLR on C_D at different L_o/d_o for different nozzles: (a) OIG; (b) IOG; (c) ACLR; and (d) SACL, Figure S2: Effect of GLR on C_D at different $\Delta p_{mix}/p_{amb}$ for different nozzles: (a) OIG; (b) IOG; (c) ACLR; and (d) SACL, Figure S3: Effect of GLR on C_D at varying μ_L/μ_G for different nozzles: (a) OIG; (b) IOG; (c) ACLR; and (d) SACL, Figure S4: Comparison of the experimental and predicted C_D from the present model Eq. (17) for different nozzles: (a) OIG, (b) IOG, (c) ACLR, and (d) SACL, Table S1: A sample of operational conditions.

Author Contributions: Conceptualization, F.A.H., J.J., and T.W.; methodology, F.A.H., J.J., and K.S.; validation, F.A.H., K.S., J.J., and T.W.; investigation, F.A.H. and K.S.; writing—original draft preparation, F.A.H.; writing—review and editing, K.S., J.J., and T.W.; visualization, F.A.H.; supervision, T.W. All authors have read and agreed to the published version of the manuscript.

Funding: This research was supported by the National Natural Science Funds for Distinguished Young Scholar (No. 51525603), the National Natural Science Foundation of China (No. 51606134), and the project LTAIN19044 funded from the program INTER-EXCELLENCE by the Ministry of Education, Youth and Sports of the Czech Republic.

Conflicts of Interest: The authors declare no conflicts of interest.

References

1. Lefebvre, A.H. A novel method of atomization with potential gas turbine applications. *Def. Sci. J.* **1988**, *38*, 353–361. [[CrossRef](#)]
2. Wang, X.F.; Chin, J.S.; Lefebvre, A.H. Influence of gas-injector geometry on atomization performance of aerated-liquid nozzles. *Int. J. Turbo Jet Engines* **1989**, *6*, 271–280. [[CrossRef](#)]
3. Roesler, T.C.; Lefebvre, A.H. Studies on aerated-liquid atomization. *Int. J. Turbo Jet Engines* **1989**, *6*, 221–230. [[CrossRef](#)]
4. Gadgil, H.P.; Raghunandan, B.N. Some features of spray breakup in effervescent atomizers. *Exp. Fluids* **2011**, *50*, 329–338. [[CrossRef](#)]
5. Sovani, S.D.; Sojka, P.E.; Lefebvre, A.H. Effervescent atomization. *Prog. Energy Combust. Sci.* **2001**, *27*, 483–521. [[CrossRef](#)]

6. Sovani, S.D.; Crofts, J.D.; Sojka, P.E.; Gore, J.P.; Eckerle, W.A. Structure and steady-state spray performance of an effervescent diesel injector. *Fuel* **2005**, *84*, 1503–1514. [[CrossRef](#)]
7. García, J.A.; Lozano, A.; Alconchel, J.; Calvo, E.; Barreras, F.; Santolaya, J.L. Atomization of glycerin with a twin-fluid swirl nozzle. *Int. J. Multiph. Flow* **2017**, *92*, 150–160. [[CrossRef](#)]
8. Hájek, J.; Dohnal, M.; Vondál, J.; Broukal, J. Analysis of effervescent spray quality for oil-fired furnace application. *Clean Technol. Environ. Policy* **2015**, *17*, 1195–1205. [[CrossRef](#)]
9. Zhou, Y.; Zhang, M.; Yu, J.; Zhu, X.; Peng, J. Experimental investigation and model improvement on the atomization performance of single-hole Y-jet nozzle with high liquid flow rate. *Powder Technol.* **2010**, *199*, 248–255. [[CrossRef](#)]
10. Wittner, M.O.; Karbstein, H.P.; Gaukel, V. Spray performance and steadiness of an effervescent atomizer and an air-core-liquid-ring atomizer for application in spray drying processes of highly concentrated feeds. *Chem. Eng. Process. Process. Intensif.* **2018**, *128*, 96–102. [[CrossRef](#)]
11. Kleinhans, A.; Georgieva, K.; Wagner, M.; Gaukel, V.; Schuchmann, H.P. On the characterization of spray unsteadiness and its influence on oil drop breakup during effervescent atomization. *Chem. Eng. Process. Process. Intensif.* **2016**, *104*, 212–218. [[CrossRef](#)]
12. Stähle, P.; Schuchmann, H.P.; Gaukel, V. Performance and Efficiency of Pressure-Swirl and Twin-Fluid Nozzles Spraying Food Liquids with Varying Viscosity. *J. Food Process. Eng.* **2015**, *40*, 1–12. [[CrossRef](#)]
13. Stähle, P.; Gaukel, V.; Schuchmann, H.P. Investigation on the Applicability of the Effervescent Atomizer in Spray Drying of Foods: Influence of Liquid Viscosity on Nozzle Internal Two-Phase Flow and Spray Characteristics. *J. Food Process. Eng.* **2014**, *38*, 474–487. [[CrossRef](#)]
14. Schröder, J.; Günther, A.; Wirth, K.E.; Schuchmann, H.P.; Gaukel, V. Effervescent Atomization of Polyvinylpyrrolidone Solutions: Influence of Liquid Properties and Atomizer Geometry on Liquid Breakup and Spray Characteristics. *At. Sprays* **2013**, *23*, 1–23. [[CrossRef](#)]
15. Schröder, J.; Kraus, S.; Rocha, B.B.; Gaukel, V.; Schuchmann, H.P. Characterization of gelatinized corn starch suspensions and resulting drop size distributions after effervescent atomization. *J. Food Eng.* **2011**, *105*, 656–662. [[CrossRef](#)]
16. Ashgriz, N. *Handbook of Atomization and Sprays: Theory and Applications*; Springer: Berlin/Heidelberg, Germany, 2011; ISBN 9781441972637.
17. Broniarz-Press, L.; Ochowiak, M.; Woziwodzki, S. Atomization of PEO aqueous solutions in effervescent atomizers. *Int. J. Heat Fluid Flow* **2010**, *31*, 651–658. [[CrossRef](#)]
18. Li, S.; Li, W.; Liu, Y.; Ji, C.; Zhang, J. Experimental investigation of the performance and spray characteristics of a supersonic two-phase flow ejector with different structures. *Energies* **2020**, *13*, 1166. [[CrossRef](#)]
19. Bar-kohany, T.; Rashkovan, A. Flash-boiling atomization. *Prog. Energy Combust. Sci.* **2008**, *34*, 417–439. [[CrossRef](#)]
20. Holz, S.; Braun, S.; Chaussonnet, G.; Koch, R.; Bauer, H.J. Close nozzle spray characteristics of a prefilming airblast atomizer. *Energies* **2019**, *12*, 2835. [[CrossRef](#)]
21. Hede, P.D.; Bach, P.; Jensen, A.D. Two-fluid spray atomisation and pneumatic nozzles for fluid bed coating/agglomeration purposes: A review. *Chem. Eng. Sci.* **2008**, *63*, 3821–3842. [[CrossRef](#)]
22. Lefebvre, A.H.; McDonell, V.G. *Atomization and Sprays*; CRC Press, Taylor & Francis Group: Boca Raton, FL, USA, 2017; Volume 2, ISBN 0891166033.
23. Mujumdar, A.S. *Handbook of Industrial Drying*; CRC Press: Boca Raton, FL, USA, 2006; ISBN 1574446681.
24. Ochowiak, M. The experimental study on the viscosity effect on the discharge coefficient for effervescent atomizers. *Exp. Therm. Fluid Sci.* **2013**, *50*, 187–192. [[CrossRef](#)]
25. Gong, J.S.; Fu, W.B. The experimental study on the flow characteristics for a swirling gas-liquid spray atomizer. *Appl. Therm. Eng.* **2007**, *27*, 2886–2892. [[CrossRef](#)]
26. Chen, S.K.; Lefebvre, A.H. Discharge coefficients for plain-orifice effervescent atomizers. *At. Sprays* **1994**, *4*, 275–290.
27. Maly, M.; Jedelsky, J.; Slama, J.; Janackova, L.; Sapik, M.; Wigley, G. Internal flow and air core dynamics in Simplex and Spill-return pressure-swirl atomizers. *Int. J. Heat Mass Transf.* **2018**, *123*, 805–814. [[CrossRef](#)]
28. Jedelsky, J.; Jicha, M.; Slama, J.; Otahal, J. Development of an effervescent atomizer for industrial burners. *Energy Fuels* **2009**, *23*, 6121–6130. [[CrossRef](#)]
29. Chin, J.S.; Lefebvre, A.H. A Design Procedure for Effervescent Atomizers. *ASME J. Eng. Gas. Turbines Power* **1995**, *117*, 226–271. [[CrossRef](#)]

30. Mlkvik, M.; Stähle, P.; Schuchmann, H.P.; Gaukel, V.; Jedelsky, J.; Jicha, M. Twin-fluid atomization of viscous liquids: The effect of atomizer construction on breakup process, spray stability and droplet size. *Int. J. Multiph. Flow* **2015**, *77*, 19–31. [CrossRef]
31. Zaremba, M.; Weiß, L.; Malý, M.; Wensing, M.; Jedelský, J.; Jícha, M. Low-pressure twin-fluid atomization: Effect of mixing process on spray formation. *Int. J. Multiph. Flow* **2017**, *89*, 277–289. [CrossRef]
32. Liu, M.; Duan, Y.; Zhang, T. Evaluation of effervescent atomizer internal design on the spray unsteadiness using a phase/Doppler particle analyzer. *Exp. Therm. Fluid Sci.* **2010**, *34*, 657–665. [CrossRef]
33. Liu, M.; Duan, Y.; Zhang, T.; Xu, Y. Evaluation of unsteadiness in effervescent sprays by analysis of droplet arrival statistics - The influence of fluids properties and atomizer internal design. *Exp. Therm. Fluid Sci.* **2011**, *35*, 190–198. [CrossRef]
34. Stähle, P.; Gaukel, V.; Schuchmann, H.P. Comparison of an Effervescent Nozzle and a Proposed Air-Core-Liquid-Ring (ACLR) Nozzle for Atomization of Viscous Food Liquids at Low Air Consumption. *J. Food Process. Eng.* **2015**, *40*, e12268. [CrossRef]
35. Ochowiak, M.; Broniarz-Press, L.; Rozanski, J. The discharge coefficient of effervescent atomizers. *Exp. Therm. Fluid Sci.* **2010**, *34*, 1316–1323. [CrossRef]
36. Huang, X.; Wang, X.; Liao, G. Visualization of two phase flow inside an effervescent atomizer. *J. Vis.* **2008**, *11*, 299–308. [CrossRef]
37. Wittner, M.O.; Ballesteros, M.A.; Link, F.J.; Karbstein, H.P.; Gaukel, V. Air-core-liquid-ring (ACLR) atomization part II: Influence of process parameters on the stability of internal liquid film thickness and resulting spray droplet sizes. *Processes* **2019**, *7*, 616. [CrossRef]
38. Wittner, M.O.; Karbstein, H.P.; Gaukel, V. Air-core-liquid-ring (ACLR) atomization: Influences of gas pressure and atomizer scale up on atomization efficiency. *Processes* **2019**, *7*, 139. [CrossRef]
39. Ramamurthi, K.; Sarkar, U.K.; Raghunandan, B.N. Performance characteristics of effervescent atomizer in different flow regimes. *At. Sprays* **2009**, *19*, 41–56. [CrossRef]
40. Jedelsky, J.; Michalica, K. Calculation of Gas-Liquid Two-Phase Flow Discharge from Orificess. Available online: <http://www.energetickeforum.cz/ext/2pf/discharge/> (accessed on 9 July 2015).
41. Forsythe, W.E. *Smithsonian Physical Tables*, 9th ed.; Smithsonian Institution: Washington, DC, USA, 1956.
42. Zaremba, M.; Kozák, J.; Malý, M.; Weiß, L.; Rudolf, P.; Jedelský, J.; Jícha, M. An experimental analysis of the spraying processes in improved design of effervescent atomizer. *Int. J. Multiph. Flow* **2018**, *103*, 1–15. [CrossRef]
43. Takamura, K.; Fischer, H.; Morrow, N.R. Physical properties of aqueous glycerol solutions. *J. Pet. Sci. Eng.* **2012**, *98–99*, 50–60. [CrossRef]
44. Holman, J.P. *Experimental Methods for Engineers*, 8th ed.; McGraw-Hill: New York, NY, USA, 2012.
45. Ochowiak, M. Discharge coefficient of effervescent atomizers with the swirl motion phenomenon. *Exp. Therm. Fluid Sci.* **2016**, *79*, 44–51. [CrossRef]
46. Konstantinov, D.; Marsh, R.; Bowen, P.J.; Crayford, A. Effervescent Atomization for Industrial Energy-Technology Review. *At. Sprays* **2010**, *20*, 525–552. [CrossRef]
47. Mori, Y.; Hijikata, K.; Nagatani, T. Effect of dissolved gas on bubble nucleation. *Int. J. Heat Mass Transf.* **1976**, *19*, 1153–1159. [CrossRef]
48. Forest, T.W.; Ward, C.A. Effect of a dissolved gas on the homogeneous nucleation pressure of a liquid. *J. Chem. Phys.* **1977**, *66*, 2322–2330. [CrossRef]
49. Forest, T.W.; Ward, C.A. Homogeneous nucleation of bubbles in solutions at pressures above the vapor pressure of the pure liquid. *J. Chem. Phys.* **1978**, *69*, 2221–2230. [CrossRef]
50. Montgomery, D.C. *Design and Analysis of Experiments*, 8th ed.; Wiley: New York, NY, USA, 2013; ISBN 978-1118-14692-7.
51. Sun, C.; Ning, Z.; Qiao, X.; Lv, M.; Li, Y.; Zhao, J.; Wang, X. Study on effervescent spray morphology based on internal gas-liquid two-phase flow patterns. *Eur. J. Mech. B/Fluids* **2019**, *74*, 123–138. [CrossRef]
52. Stähle, P.; Gaukel, V.; Schuchmann, H.P. Influence of feed viscosity on the two-phase flow inside the exit orifice of an effervescent atomizer and on resulting spray characteristics. *Food Res. Int.* **2015**, *77*, 55–62. [CrossRef]
53. Hong, M.; Jeon, J.; Lee, S.Y. Discharge Coefficient of Pressure-Swirl Atomizers with Low Nozzle Opening Coefficients. *J. Propuls. Power* **2012**, *28*, 1–5. [CrossRef]

54. Ohrn, T.R.; Senger, D.W.; Lefebvre, A.H. Geometrical Effects on Discharge Coefficients for Plain-Orifice Atomizers. *At. Sprays* **1991**, *1*, 137–153. [[CrossRef](#)]
55. Maly, M.; Sapik, M.; Cejpek, O.; Wigley, G.; Katolicky, J.; Jedelsky, J. Effect of spill orifice geometry on spray and control characteristics of spill-return pressure-swirl atomizers. *Exp. Therm. Fluid Sci.* **2019**, *106*, 159–170. [[CrossRef](#)]
56. Jedelsky, J.; Jicha, M. Energy conversion during effervescent atomization. *Fuel* **2013**, *111*, 836–844. [[CrossRef](#)]
57. Catlin, C.A.; Swithenbank, J. Physical processes influencing effervescent atomizer performance in the slug and annular flow regimes. *At. Sprays* **2001**, *11*, 575–595.
58. Essien, S.; Archibong-Eso, A.; Lao, L. Discharge coefficient of high viscosity liquids through nozzles. *Exp. Therm. Fluid Sci.* **2019**, *103*, 1–8. [[CrossRef](#)]
59. Wozniwodzki, S.; Broniarz-Press, L.; Ochowiak, M. Transitional Mixing of Shear-Thinning Fluids in Vessels with Multiple Impellers. *Chem. Eng. Technol.* **2010**, *33*, 1099–1106. [[CrossRef](#)]
60. Shepard, T.G. Bubble Size Effect on Effervescent Atomization. Ph.D. Thesis. 2011. Available online: <https://conservancy.umn.edu/handle/11299/113573> (accessed on 9 May 2020).
61. Jobehdar, M.H.; Siddiqui, K.; Gadallah, A.H.; Chishty, W.A. Effect of single-and multi-hole bubble breakers on the effervescent atomization process. *At. Sprays* **2016**, *26*, 135–162. [[CrossRef](#)]
62. Jedelsky, J.; Jicha, M. Prediction of Discharge Coefficient of Internally-Mixed Twin-Fluid Atomizers. In Proceedings of the 24th European Conference on Liquid Atomization and Spray Systems ILASS-Europe, Lisbon/Estoril, Portugal, 5 September 2011; pp. 1–4.
63. Rouina, S.; Ravelli, S.; Barigozzi, G. Combined experimental and CFD investigation of flat plate film cooling through fan shaped holes. *Int. J. Turbomach. Propuls. Power* **2019**, *4*, 7. [[CrossRef](#)]
64. Barigozzi, G.; Mosconi, S.; Perdichizzi, A.; Ravelli, S. The effect of hot streaks on a high pressure turbine vane cascade with showerhead film cooling. *Int. J. Turbomach. Propuls. Power* **2017**, *2*, 15. [[CrossRef](#)]



© 2020 by the authors. Licensee MDPI, Basel, Switzerland. This article is an open access article distributed under the terms and conditions of the Creative Commons Attribution (CC BY) license (<http://creativecommons.org/licenses/by/4.0/>).