

# Influence of Micro-Structure on the Fatigue Crack Propagation in Bridge Steel <sup>†</sup>

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**Abstract:** The use of high strength steels (HSS) allows designing lighter, slenderer and simpler structures with high structural performance. In general, the use of HSS leads to weight reduction of the whole structure, which compensates the higher cost of such a material comparing to the conventional construction steels. Knowledge of the fatigue resistance of material plays the key role during design and maintenance of the bridge structures. This contribution brings a comparison of the fatigue crack growth resistance of S355 J0 steel. Differences in microstructure and the texture of material structure could generally play a role in the fatigue crack growth. This study shows that in the case of studied steel texture of material structure has an influence on material fatigue behavior in Paris' law regime.

**Keywords:** bridge steel; Paris' law; microstructure; stress intensity factor; fatigue properties; S355 J0

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## 1. Introduction

Fatigue crack propagation, an important part of the fatigue life of a bridge component, is controlled by the local properties at the crack tip. Therefore, in the case of bridge steels with a heterogeneous microstructure, it is difficult to characterize their fatigue properties since the crack propagates through different microstructural regions with different mechanical properties. By finding the effect of each phase on fatigue crack propagation, fatigue properties of bridge steels can be well understood and consequently optimized.

The aim of this contribution is to compare experimentally obtained fatigue growth rates of long cracks in two different microstructures of S355 J0 steel grade and to determine the influence of chemical composition and structure texture on the behaviour of long fatigue cracks in this steel grade. The fatigue crack propagation is characterized by means of crack growth curves experimentally determined on compact tension (CT) specimens by using ASTM E647 [1] standard. The experimentally obtained results are discussed with results already published in [2,3]. This contribution aims to extend the experimental study done by Seitl et al. published in [4].

## 2. Theoretical Background

The fracture mechanics-based approach is used for the lifetime prediction of structures with existing cracks. The material characteristic, i.e., the crack growth curve, is experimentally determined on notched specimens, most often on CT specimens. There exist two basic testing approaches for crack growth rate determination which are based either on K-decreasing or on K-increasing

procedure ( $K$  is the stress intensity factor) [1]. The fatigue crack growth rate for applied loading is defined by the crack length increment for given number of loading cycles. Paris' law [5], expressed by the Equation (1)

$$\frac{da}{dN} = C(\Delta K)^m \tag{1}$$

is often used for the description of the fatigue crack growth.  $C$  and  $m$  are material constants,  $da/dN$  is the fatigue crack growth rate ( $da$ —crack length increment,  $dN$ —corresponding number of cycles) and  $\Delta K$  is the stress intensity factor range. The number of load cycles to failure ( $N_f$ ) can be calculated by integrating the crack propagation between an initial crack length ( $a_i$ ) and critical crack length ( $a_c$ ), Equation (2).

$$N_f = \int_{a_i}^{a_c} \frac{da}{C(\Delta K)^m}, \tag{2}$$

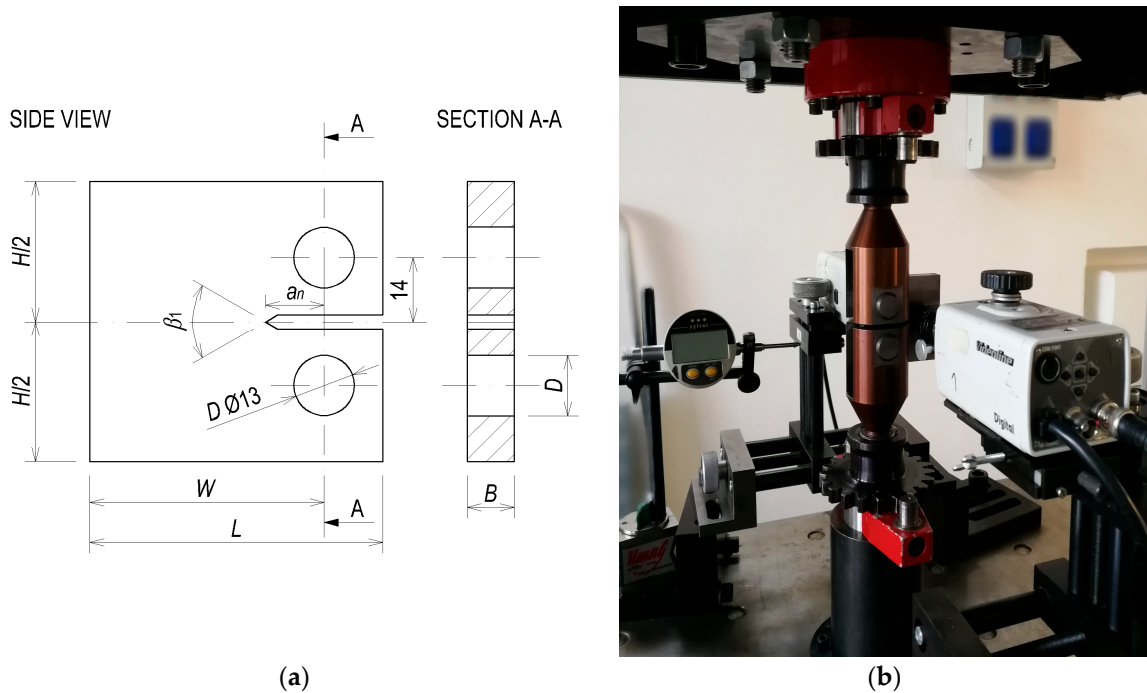
The stress intensity factor ranges for cracks in CT specimens can be computed using Equation (3) according to ASTM E647 Standard [1].

$$\Delta K = \frac{\Delta F}{B\sqrt{W}} \frac{(2+\alpha)}{(1-\alpha)^{\frac{3}{2}}} (0.886 + 4.64\alpha - 13.32\alpha^2 + 14.72\alpha^3 - 5.6\alpha^4), \tag{3}$$

where  $\alpha = a/W$ ,  $a$  is the crack length,  $W$  is the width of the specimen,  $B$  is the thickness of the specimen and  $\Delta F$  is the applied load range. The crack length increment is calculated as an average value of two crack length measurements performed on both sides of the CT specimen during the experiment. In this study the  $K$ -decreasing method according to the ASTM E647 [1] was used to obtain the values near the threshold values. From this point, the constant load value was used for the estimation of the crack growth rate of the tested specimens in the Paris' region.

*CT Specimen*

The tested CT specimens (see Figure 1-left) had dimensions:  $L = 62.5$  mm,  $W = 50$  mm,  $B = 10$  mm,  $D = 13$  mm,  $a_i = 12.5$  mm,  $H/2 = 30$  mm and the angle  $\beta_1 = 60^\circ$ .



**Figure 1.** CT test specimen— geometry (a) and actual test setup (b).

The fatigue crack growth experiments were carried out at a computer-controlled testing machine (Amsler—20 kN, see Figure 1 right). Tests were conducted under load control. The stress ratio  $R =$

$F_{min}/F_{max} = 0.1$ , where  $F_{min}$  and  $F_{max}$  refer to the minimum and maximum load of a sinusoidal wave in each cycle. The load frequency used for the tests varied from 96 (for the shortest cracks) to 42 Hz (for the longest cracks). The controlled values for temperature and relative humidity were  $23 \pm 2$  °C and 50%, respectively.

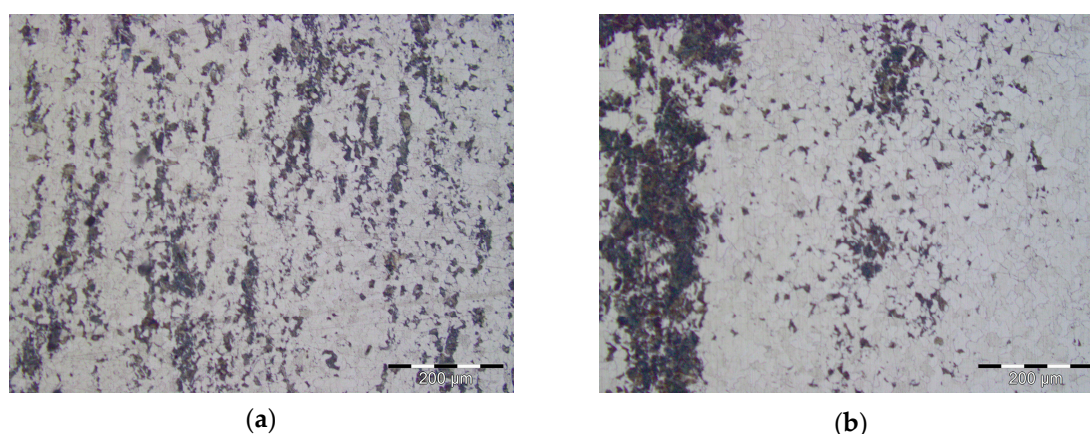
### 3. Material

The chemical composition of the investigated steel grade is specified in EN 10025-2:2004 standard [6] and is presented in Table 1. Chemical composition of the experimental material was verified by producer and it is in agreement with the standard presented in this paper.

**Table 1.** Chemical composition in percentage by weight (max. wt.%) of the used steel grades according to EN 10025-2:2004 standard [6].

Steel Grade	C	Mn	Si	P	S	N	Cu	CEV
S355 J0	0.2	1.6	0.55	0.035	0.035	0.012	0.55	0.47

The microstructure of the tested material was created with polyhedral grains of ferrite and pearlite, while the pearlitic colonies are elongated in the rolling direction, Figure 2. The average grain size estimated with linear intersection method was  $13.6 \pm 1.9$  μm for the S355 J0 steel. Small particles present in the grains and on the grain boundaries were observed in investigated specimen.



**Figure 2.** Structure of specimen S355\_J0\_FM (a) and S355\_J0\_RM (b) made from the HS S355 J0 grade, the crack was propagating in the horizontal direction; etched with 2% Nital, light optical microscope

### 4. Results and Discussion

The experimental measurements were done on two different specimens made from S355 J0 material. These specimens had different surface structure. One specimen showed fine surface structure and the other one showed rough surface structure, therefore specimens were marked as “S355\_J0\_FM”, “S355\_J0\_RM” respectively.

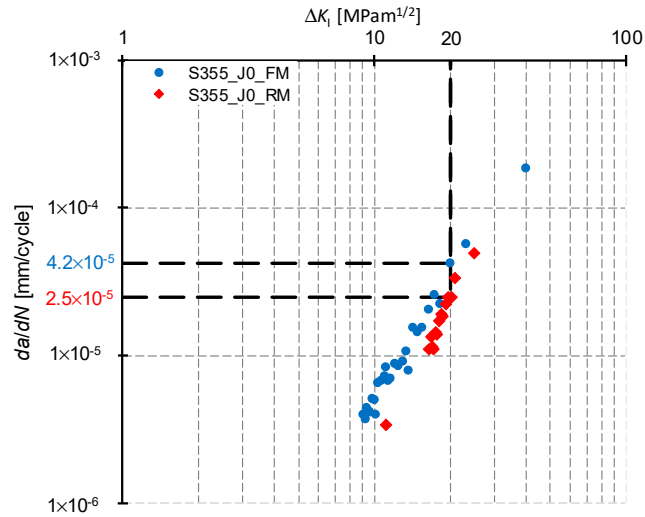
The material constants  $C$  and  $m$  measured from the Paris’ law region of these two specimens were compared to the data published by de Jesus [2]. Values of the exponent  $m$  can range from 2.0 to 7.0 with most values being between 3.0 and 4.0 [7]. The measured material constant  $m$  shows good agreement with data from literature S355 J0 steel grade, comparison of the experimental data are stated in the Table 2.

To quantify the influence of the microstructure on the fatigue crack grow rate of the S355 J0 steel a crack grow rates under the constant value of  $\Delta K_I$  were measured. The same value of  $\Delta K_I$  was chosen as  $20 \text{ MPam}^{1/2}$  then crack growth rates  $da/dN$  were measured. The evaluated crack growth rate  $da/dN$  for S355\_J0\_FM is  $4.2 \times 10^{-5} \text{ mm/cycle}$  and for S355\_J0\_RM specimen is  $2.5 \times 10^{-5} \text{ mm/cycle}$ . This difference can be seen (marked by dash line) in Figure 3.

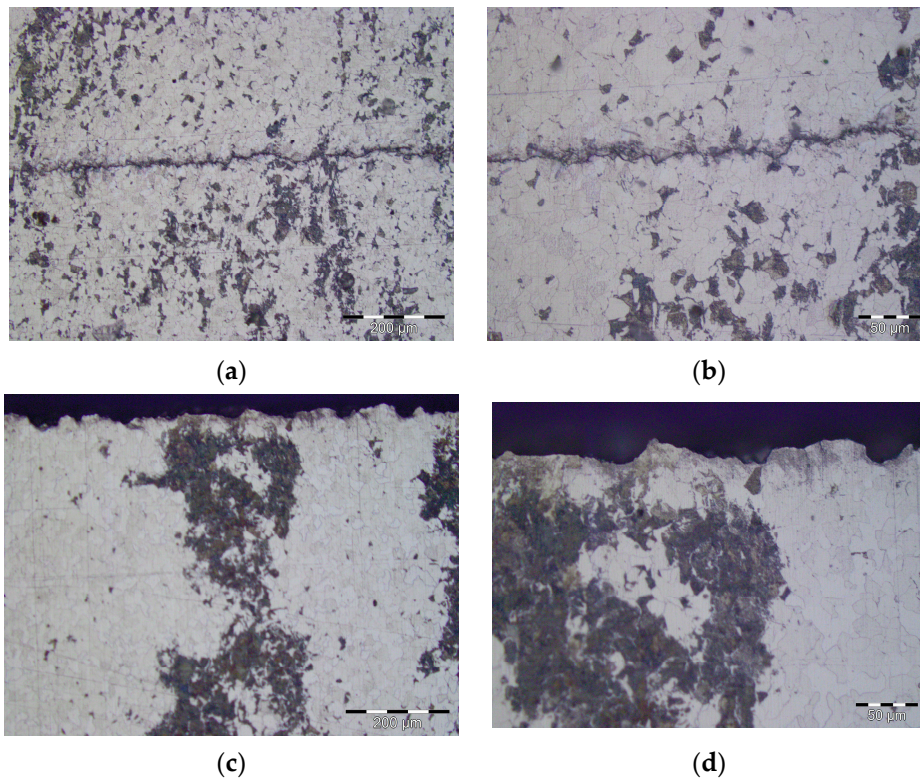


**Table 2.** The Comparison of material constants  $C$  and  $m$ , from experimental measurement and literature de Jesus et al. [2].

Material	$C$ [mm/(cycle·MPa·m <sup>0.5</sup> )]	$m$ [-]
de Jesus	$6.0 \times 10^{-10}$	3.561
S355_J0_FM	$1.0 \times 10^{-8}$	2.697
S355_J0_RM	$6.0 \times 10^{-10}$	3.553



**Figure 3.** Comparison of the fatigue crack growth rates for S355\_J0\_FM and S355\_J0\_RM specimens and estimation of the crack grow rates.



**Figure 4.** The comparison of the microstructures with two magnifications 200 μm and 50 μm of the S355\_J0\_FM (a) 200 μm (b) 50 μm and S355\_J0\_RM (c) 200 μm and (d) 50 μm at same value of stress intensity range  $\Delta K_I = 20$  MPam<sup>1/2</sup> with various crack propagation rate.

For this constant value of  $\Delta K_I = 20 \text{ MPam}^{1/2}$  from given position from measurement a microstructure of the specimen was investigated. The difference of microstructure is show in Figure 4.

## 5. Conclusions

In this paper the influence of the material microstructure on the fatigue crack rates of the S355 J0 steel grade was investigated. This effect can be expressed by means of changes of the material constants in the Paris' law. These changes of material constants due to microstructure can lead to the significant reduction of fatigue life time of the structure made from steel with fine microstructure. These fatigue properties of S355 J0 will be used in further probabilistic evaluation e.g., [8–10] and fatigue modelling for calculation as presented in e.g., [11–16].

**Author Contributions:** S.S., P.M. and J.K. have prepared analyzes and have analyzed the measured experimental data, P.P. has performed experimental measurements and S.F. has captured microstructure of the specimens.

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**Conflicts of Interest:** The authors declare no conflict of interest.

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